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NASA Contractor Report 166418

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AIRCRAFT SEAT CUSHION FIRE-BLOCKING LAYERS.
FULL SCALE: TEST DESCRIPTION AND RESULTS
Final Report, Sep. 1981 - Sep. 1982 (Douglas
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Final Report

**Study for the Optimization of Aircraft
Seat Cushion Fire-Blocking Layers — Full Scale —
Test Description and Results**

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Kenneth J. Schutter and Fred E. Duskin

**McDonnell Douglas Corporation
Douglas Aircraft Company
Long Beach, California 90846**

**CONTRACT NASA2-11095
MAY 1982**



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**Ames Research Center
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16. Abstract <p>This report describes the work done by Douglas Aircraft Company under contract to the National Aeronautic and Space Agency, Ames Research Center (NASA ARC) to determine the burn characteristics of presently used and proposed seat cushion materials and types of constructions. These tests were conducted in the Douglas Cabin Fire Simulator (CFS) at the Space Simulation Laboratory, Huntington Beach, California. Thirteen different seat cushion configurations were subjected to full-scale burn tests. The fire source used was a quartz lamp radiant energy panel with a propane pilot flame. During each test, data were recorded for cushion temperatures, radiant heat flux, rate of weight loss of test specimens, and cabin temperatures. When compared to existing passenger aircraft seat cushions, the test specimens incorporating a fire barrier and those fabricated from advance materials, using improved construction methods, exhibited significantly greater fire resistance. Results of these tests were similar to those obtained from tests conducted by Douglas Aircraft Company under contract to NASA Johnson Space Center, Contract NAS9-16062.</p>					
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SECTION I

INTRODUCTION

Aircraft passenger seats represent a high percentage of the organic materials used in a passenger cabin. These organics can contribute to a cabin fire if subjected to a severe ignition source such as post-crash fuel fire. Since 1976, programs funded by NASA have been conducted at Douglas Aircraft Company to study and develop a more fire-resistant passenger seat. The first program dealt with laboratory screening of individual materials (Report No. NASA CR-152056, Contract No. NAS 2-9337). The second program continued laboratory screening of individual materials, conducted laboratory burn tests of multilayer materials, developed a full-scale standard fire source and prepared a preliminary fire-hardened passenger seat guideline (Report No. NASA CR-152184, Contract No. NAS 2-9337). The third program consisted of additional laboratory burn testing of multilayer materials, fabricating a fire-hardened three-abreast tourist class passenger seat, and a design guideline for fire-resistant seats (Contract No. NASA 2-9337, Report No. NASA CR-152408). The fourth program fabricated and burn tested full-scale seat cushions utilizing the fire blocking concept for protecting the inner cushion (Contract No. NASA 9-16026).

The tests documented in this report involve a continuation of full-scale burning of seat cushions utilizing the fire-blocking concept.

SECTION 2
SYMBOLS AND ABBREVIATIONS

Btu	British thermal unit
°C	Degrees Celsius (centigrade)
cm	Centimeter
cm ²	Square centimeter
DAC	Douglas Aircraft Company
°F	Degrees Fahrenheit
ft	Feet
hr	Hour
in.	Inch
kg	Kilogram
kg/m ²	Kilogram per square meter
kw	Kilowatt
lb	Pound
lb/ft ²	Pounds per square foot
lb/ft ³	Pounds per cubic foot
m	Meter
mm	Millimeter
min	Minutes
NASA	National Aeronautics and Space Administration
PCT, %	Percent
psi	Pounds per square inch
sec	Second
T	Thermocouple
W	Watt

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SECTION 3

TEST ARTICLES

3.1 Test Specimens

Thirteen different seat cushion constructions were tested (Table 1). Fire blocking, when incorporated, covered all sides of the cushion. All seams were sewn with nylon thread. The overall dimensions for the back cushions were 43 by 61 by 5 centimeters (17 by 24 by 2 inches). The bottom cushions dimensions were 46 by 50 by 8 centimeters (18 by 20 by 3 inches).

3.2 Materials

The 13 test specimens were fabricated using a combination of materials shown in Table 2. These materials were selected and supplied for use in this program by NASA-AMES Research Center.

All cushions were fabricated by Expanded Rubber and Plastics Corporation in Gardena, California.

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TABLE 1
SEAT CONSTRUCTIONS

Construction Number	Decorative Upholstery	Slip Cover	Fire Blocking	Foam
1	Wool-Nylon	None	None	F. R. Urethane*
2	Wool-Nylon	Cotton-Muslin	Vonar-3	F. R. Urethane
3	Wool-Nylon	Cotton-Muslin	Vonar-2	F. R. Urethane
4	Wool-Nylon	None	3/8 LS 200	F. R. Urethane
5	Wool-Nylon	None	Celiox 101	F. R. Urethane
6	Wool-Nylon	None	Norfab 11 HT-26-AL	F. R. Urethane
7	Wool-Nylon	Cotton-Muslin	Vonar-3	N. F. Urethane*
8	Wool-Nylon	None	Norfab 11 HT-26-AL	N. F. Urethane
9	Wool-Nylon	None	None	LS 200 Neoprene
10	Wool-Nylon	None	None	Polyimide
11	Polyester	None	None	Polyimide
12	Wool-Nylon	None	Norfab 11 HT-26	F. R. Urethane
13	Wool-Nylon	None	PBI	F. R. Urethane

*F. R. Urethane (Fire Retarded Urethane)
N. F. Urethane (Non-Fire Retarded Urethane)

TABLE 2
MATERIAL

Material	Source
#2043 urethane foam, fire-retardant (FR), 0.032 g/cm ³ (2.0 lb/ft ³) 43 ILD	North Carolina Foam Ind. Mount Airy, NC
Urethane foam, non-fire retardant (NF), 0.022 g/cm ³ (1.4 lb/ft ³) 24-35 ILD	CPR Division of Upjohn Torrance, Ca.
Vonar-3, 3/16-inch thick with Osnaburg cotton scrim (23.5 oz/yd ²) .079 g/cm ²	Chris Craft Industries Trenton, NJ
Norfab 11HT26-aluminized (12.9 oz/yd ²) .044 g/cm ² , aluminized one side only	Amatex Corporation Norristown, Pa
Gentex preox (celiox) (10.9 oz/yd ²) .037 g/cm ² , aluminized one side only	Gentex Corporation Carbondale, Pa
Wool nylon (0.0972 lb/ft ²) .0474 g/cm ² , 90% wool/100% nylon, R76423 sun eclipse, azure blue 78-3080 (ST7427-115, color 73/3252)	Collins and Aikem Albermarle, NC
Vonar 2, 2/16 inch thick, .068 g/cm ² , (19.9 oz/yd ²) osnaburg cotton scrim	Chris Craft Industries Trenton, NJ
LS-200 foam, 3/8" thick (33.7 oz/yd ²) .115 g/cm ² LS-200 foam, 3-4 inches thick (7.5 lb/ft ³) 0.12 g/cm ³	Toyad Corporation Latrobe, Pa
Polyimide Foam (1.05 lb/ft ³) .017 g/cm ³	Solar San Diego, Ca
100% polyester (10.8 oz/yd ²) .037 g/cm ² 4073/26	Langenthal Corporation Bellevue, Wa
Norfab 11HT26 Approximately (11.3 oz/yd ²) .038 g/cm ²	Gentex Corporation Carbondale, Pa
PBI Woven Cloth Approximately (10.8 oz/yd ²) .037 g/cm ²	Calanese Plastic Company Charlotte, NC

SECTION 4

TEST PROGRAM

4.1 Test Setup

All tests were conducted within the Cabin Fire Simulator (CFS). The CFS is a double-walled steel cylinder 12 feet in diameter and 40 feet long, with a double-door entry airlock at one end and a full-diameter door at the other. It is equipped with a simulated ventilation system and, for environmental reasons, all exhaust products are routed through an air scrubber and filter system. A view port in the airlock door allows the tests to be monitored visually. The radiant heat panels used in these tests were positioned as shown in Figures 1 and 2.

The radiant panels consisted of 46 quartz lamps producing a 10 watt/square centimeter heat flux at 6 inches from the surface of the panels. Prior to testing, the heat flux upon the cushion surface was mapped using calorimeters. Figure 3 shows the positions at which heat flux measurements were taken and their recorded values.

4.2 Instrumentation

The relative location of instrumentation for the tests is shown in Figure 4.

4.2.1 Post test still photographs were taken for each seat construction. These photographs are located in Appendix A. In addition, a video recording was made during each test.

4.2.2 Thermal Instrumentation

Temperatures were obtained using chromel-alumel thermocouples placed within the seat constructions. The number of thermocouples varied between 2 and 3 per cushion depending on whether or not a fire blocking layer was used (Figure 5). In the CFS, chromel-alumel thermocouples were located along the ceiling and at the cabin air exhaust outlet. Two heat flux sensors were installed facing the seat assembly. The upper calorimeter was used to monitor the heat flux given off by the radiant panels to insure consistency among tests. The thermocouple and calorimeter signals were fed through a Hewlett-Packard 3052A Automatic Data Acquisition System which provided a real-time printout of data (Figure 6).

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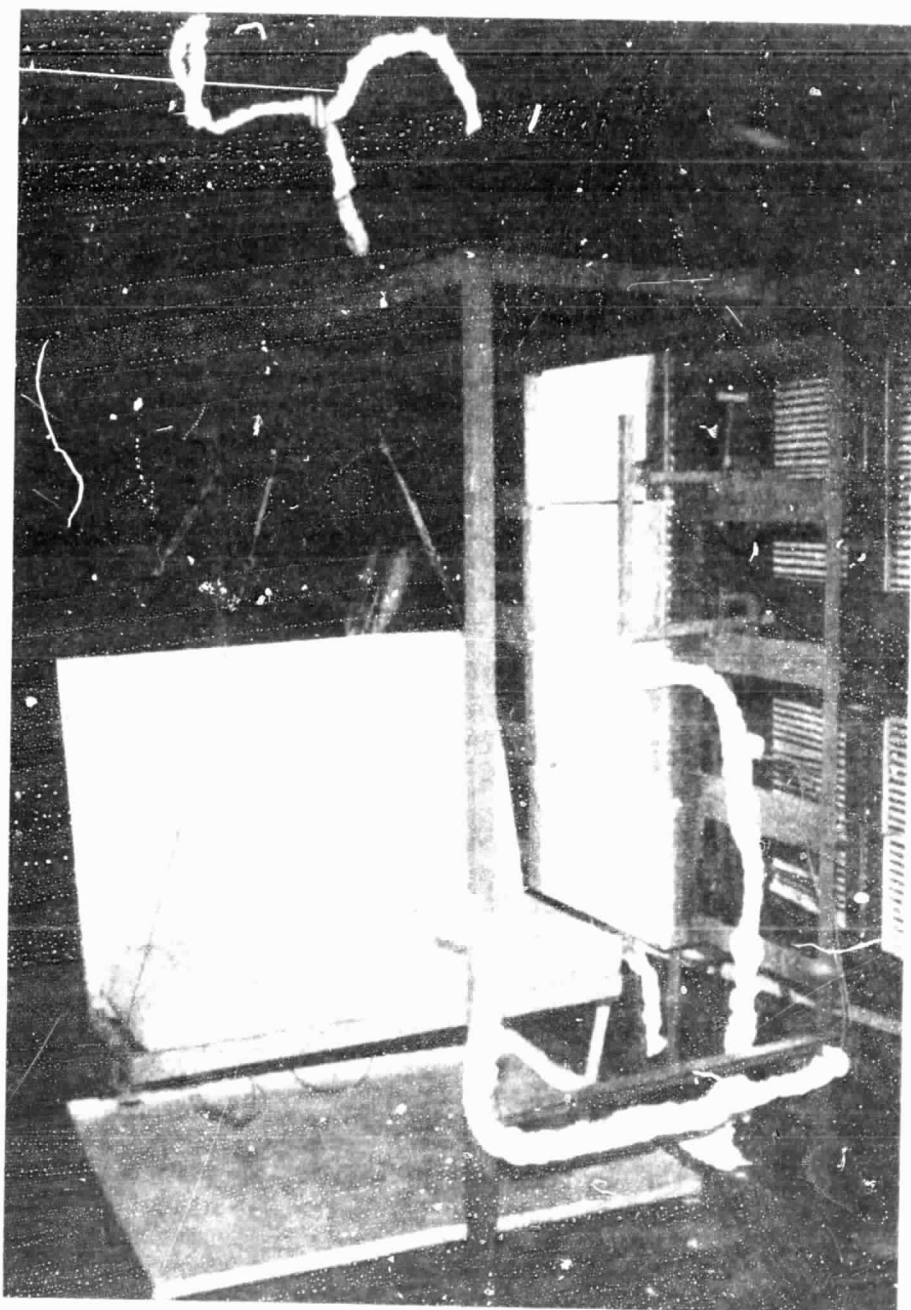


FIGURE 1. TEST STEUP

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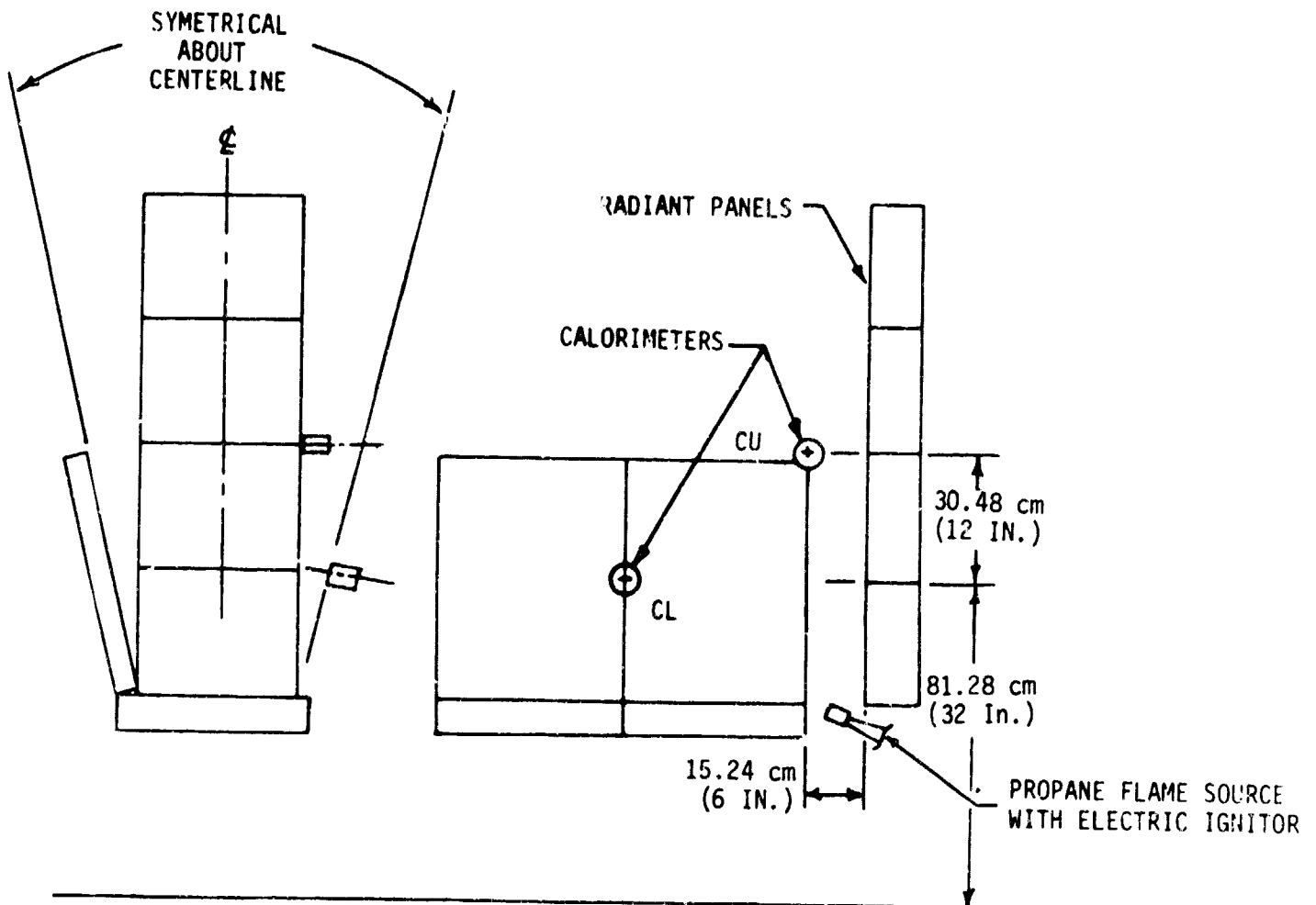


FIGURE 2. FUEL SOURCE AND CALORIMETER LOCATION

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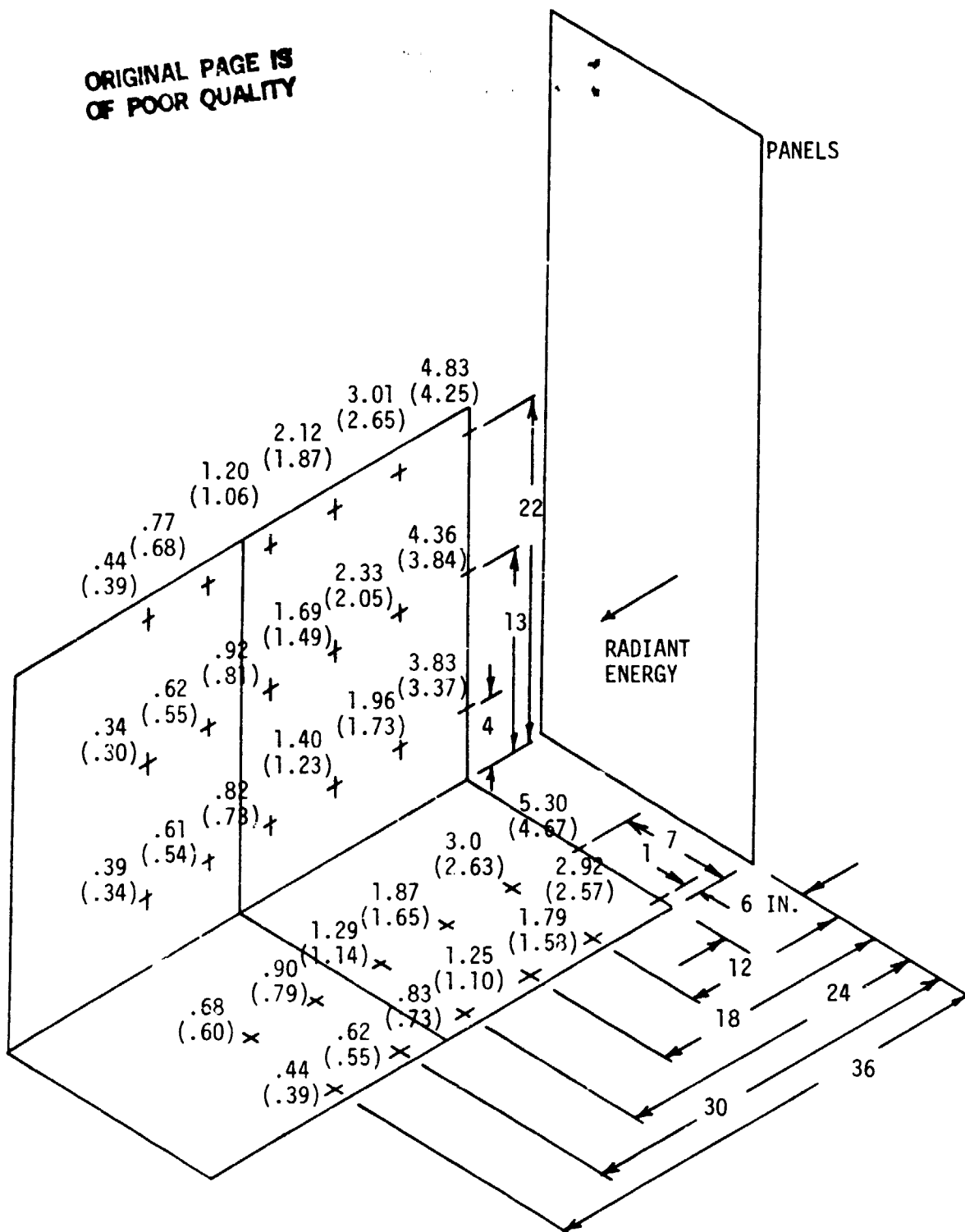
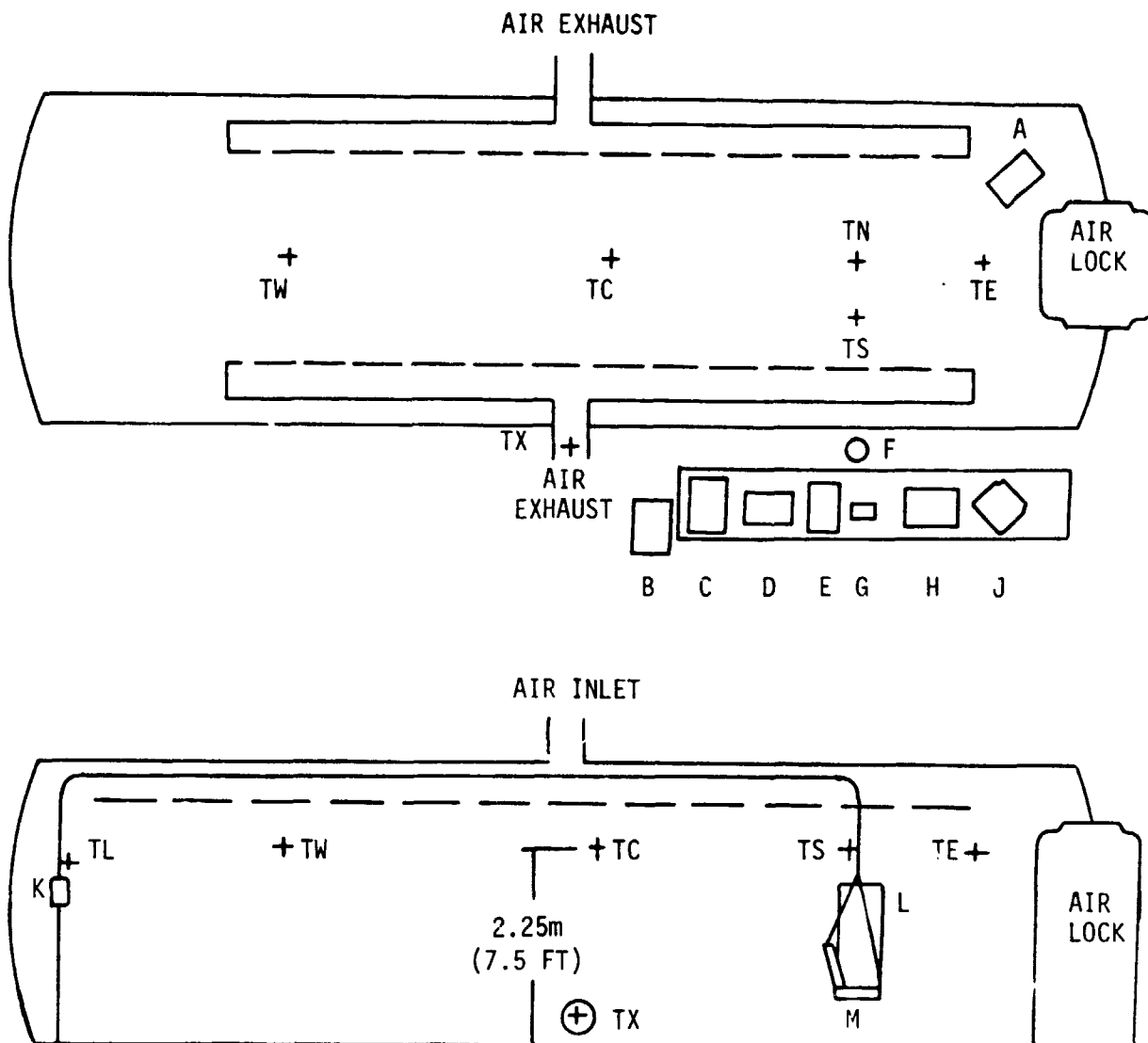


FIGURE 3. HEAT FLUX MAPPING w/cm² (Btu/FT²-SEC)



- LEGEND:
- A = VIDEO CAMERA
 - B = HEWLETT PACKARD SCANNER AND DIGITAL VOLTMETER
 - C = HEWLETT PACKARD CALCULATOR
 - D = HEWLETT PACKARD PLOTTER
 - E = HEWLETT PACKARD PRINTER
 - F = PROPANE ON/OFF VALVE
 - G = VARIABLE TRANSFORMER
 - H = VIDEO CASSETTE RECORDER
 - J = VIDEO MONITOR
 - K = 100 LB. LOAD CELL
 - L = RADIANT ENERGY PANELS
 - M = SEAT FRAME
 - T = THERMOCOUPLE

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FIGURE 4. CFS INSTRUMENTATION

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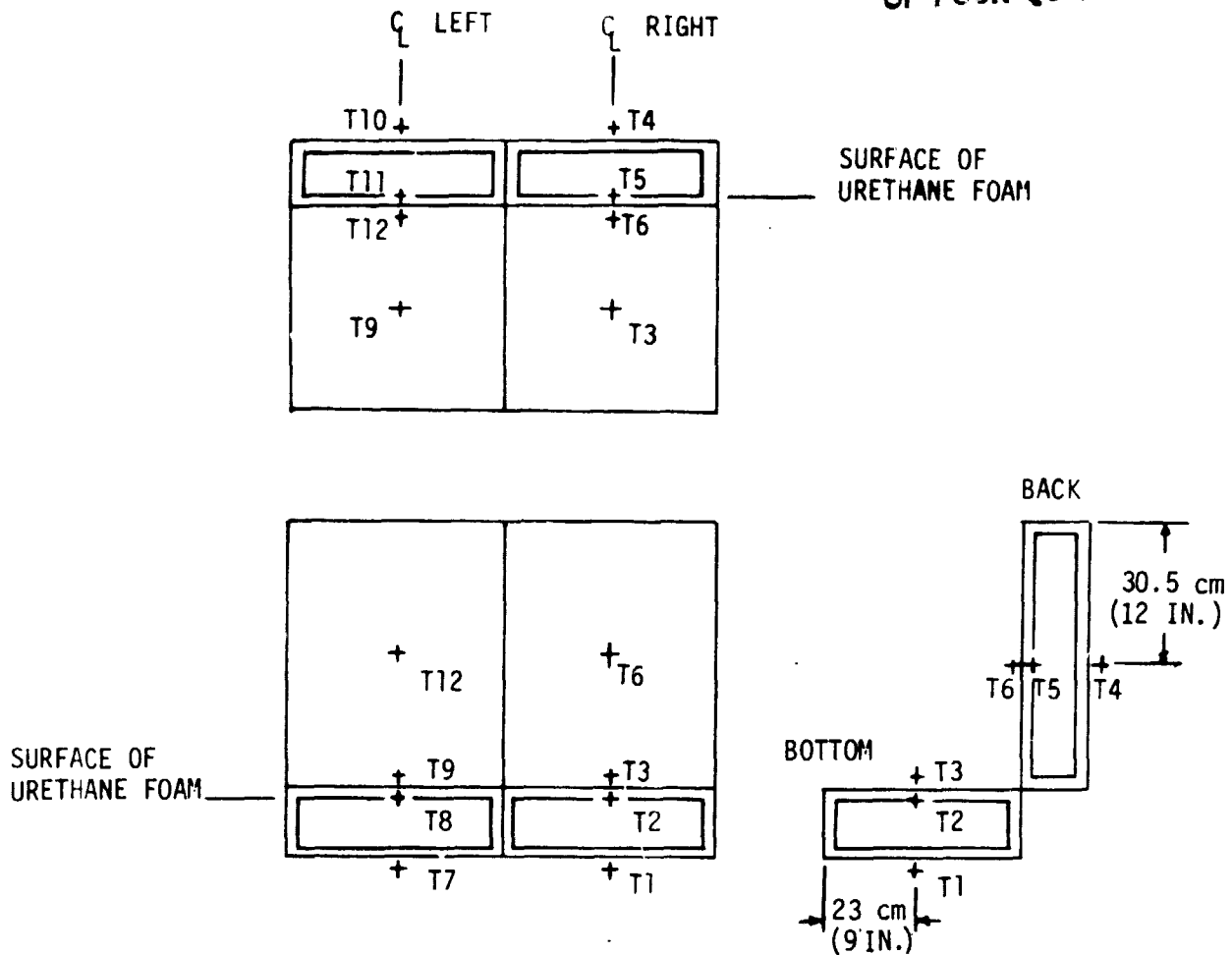


FIGURE 5. CUSHION THERMOCOUPLES (LOCATION AND IDENTIFICATION)

4.3 Test Procedures

Cushions instrumented with thermocouples were weighed, then positioned on the seat frame. The seat frame was rigged with suspension cables and hung from one end of a cable located in the ceiling of the CFS. The other end of the ceiling cable was attached to a load cell. Thermocouples, heat flux sensors, and load cells were checked for proper operation and calibration. The computer and video were started, the propane gas was ignited, and then the radiant panel was switched on. The radiant panels remained on for five minutes. After fifteen minutes, the tests were complete and post-test photos were taken of the cushion residue. The residue was removed from the seat frame and weighed.

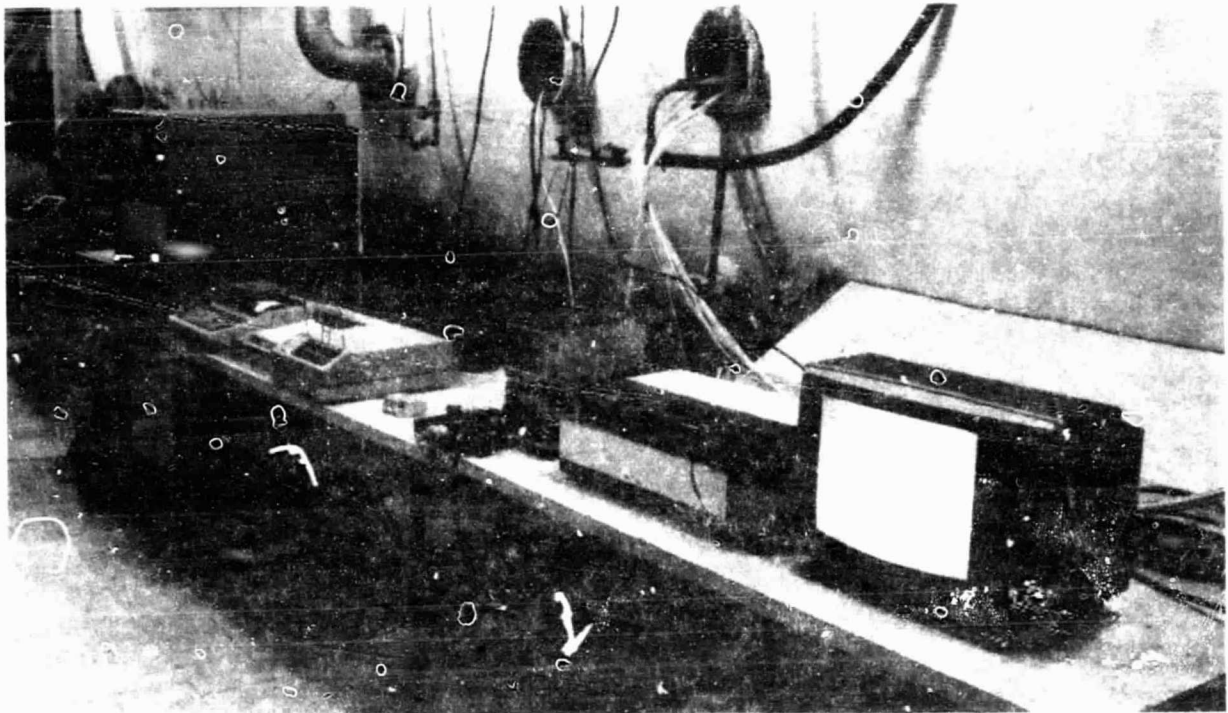


FIGURE 6. DATA ACQUISITION

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SECTION 5 TEST RESULTS

A total of 23 full-scale cushion burn tests were conducted. Each seat construction listed in Table 1 was tested twice with the exception of constructions 8, 11, 12 and 13. For these constructions, only enough material for one test was available. However, when two tests of the same construction were made, the results were identical and therefore a third test was considered unnecessary.

The purpose of these tests was to investigate the burning characteristics of cushion employing fire resistant designs. It was the peculiar designs and how the materials were used which were evaluated and not so much the individual materials themselves. To give an example, construction number 2 was designed to employ one layer of Vonar-3 as a fire blocking layer. The evaluation of the performance of this cushion was not so much decided on what material was used, Vonar-3, as the way in which it was used, one layer as fire blocking.

5.1 General

The constructions tested can be classified in four groups. These groups are standard cushion construction, standard cushion construction with a protective covering enveloping the urethane foam core, standard cushion construction with a protective covering enveloping non-fire retarded urethane foam core and standard cushion construction with the urethane foam core replaced by an advance fire resistant foam.

The test results of these constructions is graphically provided in plots presented in Appendix B. To aid in comparison of these constructions, the peak values for each test and the time at which they occurred were taken from the respective plots and are presented in Table 3. The weight loss results are in Table 4. Post-test photographs for each construction are located in Appendix B.

5.2 Standard Seat Construction

Construction number 1 is representative of the type of materials most commonly used in the construction of aircraft passenger seat cushions. These cushions were totally consumed by the fire in a matter of minutes.

Characteristically, the fire-retarded urethane foam thermally decomposes under the extreme heat into a fluid form and subsequently to a gas. In the fluid form, the urethane drips from the seat cushion onto the floor forming a puddle or pool. This pool of urethane fluid gives off gases which are ignited by burning debris falling from the seat. This results in a very hot pool fire engulfing the seat in a matter of minutes.

TABLE 3
TEST DATA PEAK VALUES

C = CALORIMETER T = THERMOCOUPLE S = SECONDS																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																										
CUSHION CONSTRUCTION	CL	BTU/FT ² -SEC										T ₁ °F	T ₂ °F	T ₃ °F	T ₄ °F	T ₅ °F	T ₆ °F	T ₇ °F	T ₈ °F	T ₉ °F	T ₁₀ °F	T ₁₁ °F	T ₁₂ °F	T _S °F	T _M °F	BEFORE LB	AFTER LB	DELTA LB	T _X °F	T _C °F	T _E °F	DP IN H ₂ O																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																										
1	TEST 1	4.0	1494	1666	1309	1218	1526	1690	1690	912	3225	2285	1341	747	750	7.4	0	7.4	293	308	388	11.11																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																				

TABLE 3
TEST DATA PEAK VALUES (CONTINUED)

		C = CALORIMETER										T = THERMOCOUPLE		S = SECONDS								
CUSHION CONSTRUCTION	CL	BTU/FT ² -SEC										TM	BEFORE LB	AFTER LB	DELTA LB	TX °F	TC °F	TE °F	OP IN. H ₂ O			
		T1 °F	T2 °F	T3 °F	T4 °F	T5 °F	T6 °F	T7 °F	T8 °F	T9 °F	T10 °F									T11 °F	T12 °F	
6 TEST 14	1.74 1005	705 3565	1002 3165	1012 1125	890 4765	807 1885	1352 1665	307 3805	120 3765	469 3185	127 6165	254 3785	490 3165	274 3125	473 2565	9.32	7.0	2.32	117 3165	185 3145	211 3185	6.64 585
7 TEST 15	1.39 965	793 9185	782 3085	1366 945	201 2965	1031 3365	1115 1205	135 6605	198 5565	412 3205	110 8925	176 6045	382 3225	277 2985	405 3105	11.25	8.45	2.8	115 3265	171 3125	204 3025	5.64 665
7 TEST 16	1.54 1065	950 8505	778 3165	1156 1265	983 6965	945 3145	1165 1365	148 8945	229 4645	400 3185	101 8825	175 5265	350 3185	269 3185	396 3145	11.01	8.10	2.93	108 3265	160 3185	200 3185	6.03 665
8 TEST 18	1.54 965	783 3665	1020 3085	1180 1045	1198 3425	880 2365	1133 1305	142 3825	334 3525	459 3185	146 4025	285 3705	393 3185	310 2865	507 2965	8.47	6.05	2.42	126 3305	206 3145	229 3185	6.08 645
9 TEST 8	1.51 1145	132 3225	1026 3185	168 3605	168 3605	968 3225	100 3945	280 3285	94 8485	280 3285	94 8485	323 3185	270 2245	333 2345	19.6			104 3105	143 2665	171 2925	5.34 625	
9 TEST 19	1.46 1185	145 3245	988 3185	161 4185	161 4185	516 3225	95 4645	308 3405	92 8825	308 3405	92 8825	318 3205	264 2245	329 2505	19.01	17.65	1.36	98 3065	137 3105	169 3105	5.43 645	
10 TEST 9	1.60 865	774 1765	1393 1085	1335 3345	1335 3345	1029 1245	411 3685	1061 2905	161 4585	1061 2905	161 4585	729 3325	290 2365	350 3065	5.05	3.60	1.45	113 3185	162 2305	185 2725	5.81 525	
10 TEST 6	1.67 1025	1153 2585	1425 985	1794 2425	1794 2425	1141 1345	556 4765	895 3265	152 5245	895 3265	152 5245	373 3885	302 2605	346 3105	6.48	3.70	2.78	111 3165	166 2745	194 3205	5.76 545	
11 TEST 20	1.46 1085	588 3685	1100 1105	886 2285	886 2285	1252 1545	134 3585	436 3245	176 4505	436 3245	176 4505	368 3225	261 2665	353 3025	4.20	3.67	0.53	104 3185	150 3245	184 3205	5.18 685	
12 TEST 21	1.63 1145	1504 3405	1231 3225	1414 1205	1091 3965	1204 3005	1450 2385	1268 3465	783 3945	1039 3805	1185 7925	628 4005	772 3205	324 2105	392 2105	9.1	3.66	5.54	137 3965	214 3325	238 3205	6.02 665
13 TEST 22	1.81 1005	1259 3225	1059 3205	1331 1125	698 3445	1052 1745	1462 1565	497 5065	405 3265	457 3185	131 6785	338 3425	452 3165	305 3005	432 2125	9.8	6.0	3.8	122 3125	182 2805	223 2525	6.53 745

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TABLE 4
WEIGHT DATA

Cushion Construction		Weight Before kg (LB)	Weight After kg (LB)	Weight Loss kg (LB)
1	Test 1	3.36 (7.4)	0 (0)	3.36 (7.4)
1	Test 17	3.40 (7.5)	0 (0)	3.40 (7.5)
2	Test 2	5.78 (12.75)	3.72 (8.20)	2.06 (4.55)
2	Test 4	5.43 (11.97)	3.76 (8.3)	1.67 (3.67)
3	Test 11	5.22 (11.5)	3.27 (7.2)	1.95 (4.3)
3	Test 12	5.22 (11.5)	3.27 (7.2)	1.95 (4.3)
4	Test 3	5.28 (11.65)	3.47 (7.65)	1.81 (4.0)
4	Test 10	5.42 (11.95)	3.54 (7.8)	1.88 (4.15)
5	Test 7	4.11 (9.05)	3.00 (6.62)	1.11 (2.23)
5	Test 13	4.17 (9.20)	2.95 (6.50)	1.22 (2.70)
6	Test 5	4.26 (9.40)	3.23 (7.13)	1.03 (2.27)
6	Test 14	4.23 (9.32)	3.18 (7.0)	1.05 (2.32)
7	Test 15	5.10 (11.25)	3.8 (8.45)	1.30 (2.80)
7	Test 16	5.00 (11.03)	3.67 (8.10)	1.33 (2.93)
8	Test 18	3.84 (8.47)	2.74 (6.05)	1.10 (2.42)
9	Test 8	8.89 (19.6)	N/A	--
9	Test 19	8.62 (19.01)	8.0 (17.65)	.62 (1.36)
10	Test 9	2.29 (5.05)	1.63 (3.60)	.66 (1.45)
10	Test 6	2.94 (6.48)	1.68 (3.70)	1.26 (2.78)
11	Test 20	1.91 (4.20)	1.66 (3.67)	.25 (.53)
12	Test 21	4.13 (9.10)	1.66 (3.66)	2.47 (5.54)
13	Test 22	4.45 (9.80)	2.72 (6.00)	1.73 (3.80)

5.3 Protected Fire-Blocked Standard Cushions

The purpose of the fire-blocking layer surrounding the urethane foam core is to thermally isolate the foam from the heat source by either conducting the heat laterally away and by providing an insulative char layer.

5.3.1 Aluminized Fabric

The celiox and norfab fire blocking constructions employed a reflective aluminum coating bonded to their outer surface.

All three constructions resulted in identical test results. These constructions were unable to protect the urethane foam in the cushions closest to the radiant heat source. They were able to slow down the burn rate of the urethane thus producing a less severe fire. This fire was unable to penetrate the adjacent cushions also protected by these materials.

Characteristically, in these constructions the urethane thermally decomposes within the fire-blocking layer and produces fluids and gases. The gas leaks through the cushion seams, ignites, burn and continues to open the seams. This results in a small controlled pool fire burning within the fire-blocking envelope with flames reaching through the seam areas. The radiant heat source in combination with the controlled pool fire, is adequate to thermally decompose the urethane foam on the closest side of the adjacent cushions. The heat source is not adequate to ignite these gases.

Reversing the edges at which the seams were located, i.e, placing the seams at the bottom edge instead of the top edge of the cushion, made no appreciable difference for the cushions adjacent to the fire source. Placing the seam on the bottom edge of the cushions farthest from the radiant panel helped to prevent the escaping gases from igniting, and the seam from opening. All cushions using this fire-blocking material were vented in the back to prevent ballooning of the cushions by the gas generated within them. However, the decomposed urethane tended to plug the vent and restrict the out-gasing. The overall final appearance of the cushion closest to the radiant panels showed a fragile, charred, empty fire-blocking envelope with its seams burned open.

The final appearance of the cushions farthest from the radiant panels showed a partially charred upholstery cover. The urethane cushion had some minor hollow spots. When the seams were placed on the bottom edge of the cushion, a fully intact fire-blocking envelope remained.

The percent weight loss between the fire and non-fire retarded urethane cushions was small, as shown by Figure 7.

5.3.2 Non-Aluminized Fire Blocking

Constructions 2, 3 and 7 used Vonar foam, construction 4 used LS-200 foam, construction 12 used non-aluminized norfab fabric and construction 13 used PBI fabric.

The constructions were unable to protect the urethane foams in the cushions closest to the radiant panels. However, they did slow down the burn rate of the urethane thus subjecting the adjacent cushion to a less intense fire.

The fire-blocking foams performed much like the aluminized fabric fire-blocking in that even though the heat was intense enough to thermally decompose the urethane into a fluid and gas, the fire blocking layer was able to contain and subdue the burning urethane. Flames exited where the fire-blocking char layer had fallen away.

The non-aluminized norfab fabrics were unable to contain the decomposed urethane. The urethane fluid dripped onto the floor where it pooled and ignited. The cushions were completely consumed when this floor fire engulfed it. The overall final appearance of the cushion remains closest to the radiant panels for foam fire blocking constructions 2, 3, 4 and 7 was thoroughly charred fire-blocking material void of all urethane foam.

The final appearance of the cushions farthest from the radiant panels were very similar. They varied in the amount of thermal decomposition of the urethane foam core, i.e., the size of the void or hollowing of the urethane. Construction number 2 using Vonar-3 material produced the smallest amount of urethane decomposition. It was followed by construction number 4, 3/8 LS 200 neoprene, and construction number 3, Vonar-2. Construction number 7 used a non-fire retarded urethane with Vonar-3. It did not fair as well as construction number 2 employing fire retarded urethane.

Typically, the foam fire-blocking layer adjacent to the urethane hollow spots were completely charred but intact.

5.4 Advanced Foam

Construction numbers 9, 10 and 11 used advanced foams in place of the urethane foam.

Construction number 9, LS 200 neoprene, produced a deep seated fire which did not produce a significant amount of heat or flames. It smoldered long after the test was completed and required total emersion in water to extinguish. This cushion had the lowest weight loss as shown by Figure 7. However, an all LS-200 neoprene seat cushion would result in a large aircraft weight impact because of its high density.

The foam in the seat cushion closest to the radiant panels was completely charred with the upholstery burned off of all surfaces except the bottom and back.

The foam in the seat cushions farthest from the radiant panels had a thick char on the edge closest to the heat source. This char gradually diminished halfway across the cushions. The upholstery on the back and bottom of these cushions was not burned.

Constructions 10 and 11, polyimide foam, had different upholstery materials. Construction 10, 90/10 wool-nylon upholstery, performed identically to a previous test program. The cushions closest to the radiant panels shrunk to one-half inch in thickness or less with a char of one-quarter inch or greater.

The cushion farthest from the radiant panels shrank to within one-half inch thickness with a char of one-quarter inch or less.

Characteristically, the polyimide foam thermally decomposes by giving off gases, and produces a char layer as it decreases in size.

The decomposing of the foam beneath the upholstery on the seat farthest from the radiant panel creates a pocket or void where the gases generated by the foam accumulates. When these trapped gases burn, the foam further thermally decomposes. Construction number 11, polyester upholstery, reacted differently from that characteristic of construction number 10. When the radiant panel was turned on, the polyester upholstery on the cushion farthest from the heat source rapidly decomposed into a liquid which dripped off the seat cushions.

With the upholstery gone, the majority of the gas from the decomposing polyimide foam escaped without igniting. These cushions decomposed less as exemplified by the small weight loss and a thinner char layer.

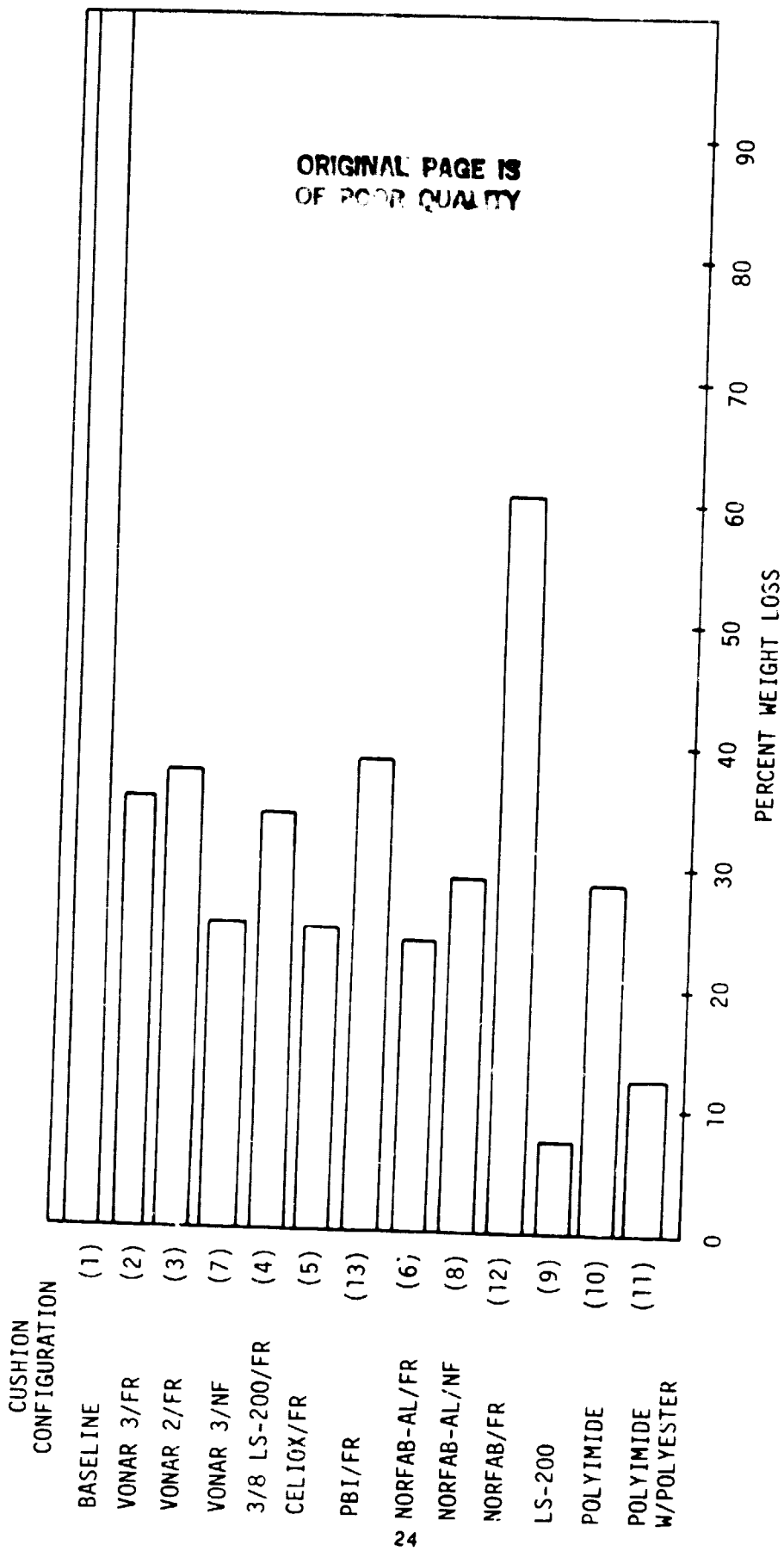


FIGURE 7. PERCENT WEIGHT LOSS

SECTION 6

CONCLUSIONS

Urethane foam decomposes into a volatile gas when exposed to a severe heat source. If this generated gas can be contained in such a manner as to prevent its igniting or to control the rate at which it burns, the severity of the fire will be reduced. This was clearly shown in the testing of standard cushion constructions with a protective covering, "fire-blocking", enveloping the urethane foam.

When the fire blocking was able to contain the decomposing urethane by-products, i.e., fluid and gas, the cushions closest to the heat source burned with less intensity, generated a minimum of heat and were unable to ignite the adjacent cushions. However, when the decomposing urethane fluid was able to escape from the fire-blocking envelope and pool on the floor, an uncontrolled fire erupted which resulted in total burning of all cushion materials.

Some of the Norfab and Celiox materials utilized aluminum coatings. It was not the aluminums reflecting properties which made the cushions perform well as it was its non-permeable properties. This coating helped contain the decomposed by-products and prevented propagation to the adjacent cushion.

Had the seams held and all the gases vented out the back of the cushions and away from the heat, the decomposing of the cushions may have been even less severe. Undoubtedly, the reflective properties had an effect in slowing down the decomposing of the urethane, but only by a few seconds. The reason being the emissivity and thermal conductivity of the aluminum coating was inadequate to resist the severe radiant energy being applied to the surfaces.

The charred foam fire-blocking layers did not act primarily as a heat barrier as they did a liquid and gas barrier. In the cushions farthest from the radiant source, the urethane foam still thermally decomposed. It formed a pocket of gas behind the intact charred envelope. This was verified in post test inspection. However, the gas escaped slowly and only created a small pilot flame. The flame extinguished itself when the radiant energy source was switched off.

The polyimide cushions are examples of a foam which thermally decomposes at high temperatures and generates gas and char but no noticeable liquids. The wool-nylon upholstery trapped gases between itself and the foam. When these gases ignited, the foam decomposed rapidly. The polyester upholstery decomposed from the cushions fast enough to prevent the trapping of these gases. Subsequently, the foam in the cushions decomposed at a slower rate. From these tests, it is concluded that no matter the foam used as a core for the cushion, if the gases generated by the foam can be expelled or contained in such a manner as to prevent their burning or reduce the rate at which they burn, a severe fire can be avoided or delayed. It is further concluded that if the thermal decomposition characteristics can be altered so as to slow down the generation of gas, the time before a fire becomes severe can be extended to the point where appropriate extinguishment of the fire may be possible.

SECTION 7

RECOMMENDATIONS

It is recommended that a study be made to incorporate cushion designs and fire-blocking materials which are thermally stable and nonpermeable to urethane fluids and gases to prevent or reduce the rate at which a seat cushion burns.

This study should include considerations for wearability of fire blocking layers, fatigue life of cushion foams and methods of venting decomposition gases from the cushion assembly. Test results from this program have shown that seam constructions significantly affect cushion burn performance. Therefore, seam constructions previously studied by the NASA seat program should be reconsidered in future cushion designs.

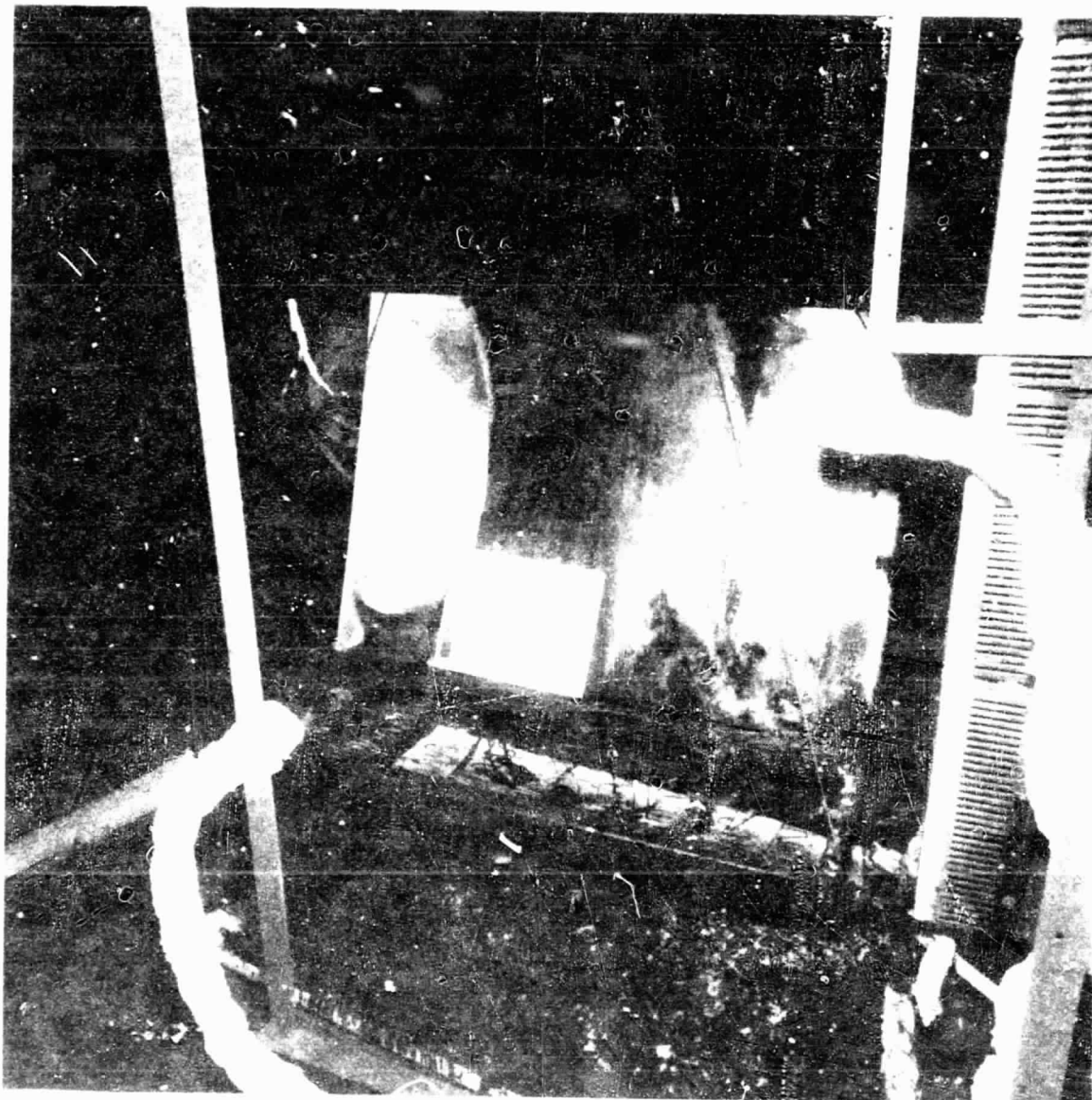
It is also recommended to use these studies as a basis to develop a design standard for a fire resistant passenger seat. This standard must be supported by inexpensive laboratory burn test methods that can verify these standards are being met.

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APPENDIX A
Posttest Photographs

Construction	Page
1 Test 1 17	30 31
2 Test 2 4	32 33
3 Test 11 12	34 35
4 Test 3 10	36 37
5 Test 7 13	38 39
6 Test 5 14	40 41
7 Test 15 16	42 43
8 Test 18	44
9 Test 8 19	45 46
10 Test 9 6	47 48
11 Test 20	49
12 Test 21	50
13 Test 22	51

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Construction
Number

1

Decorative
Upholstery

Wool-Nylon

Slip Cover

None

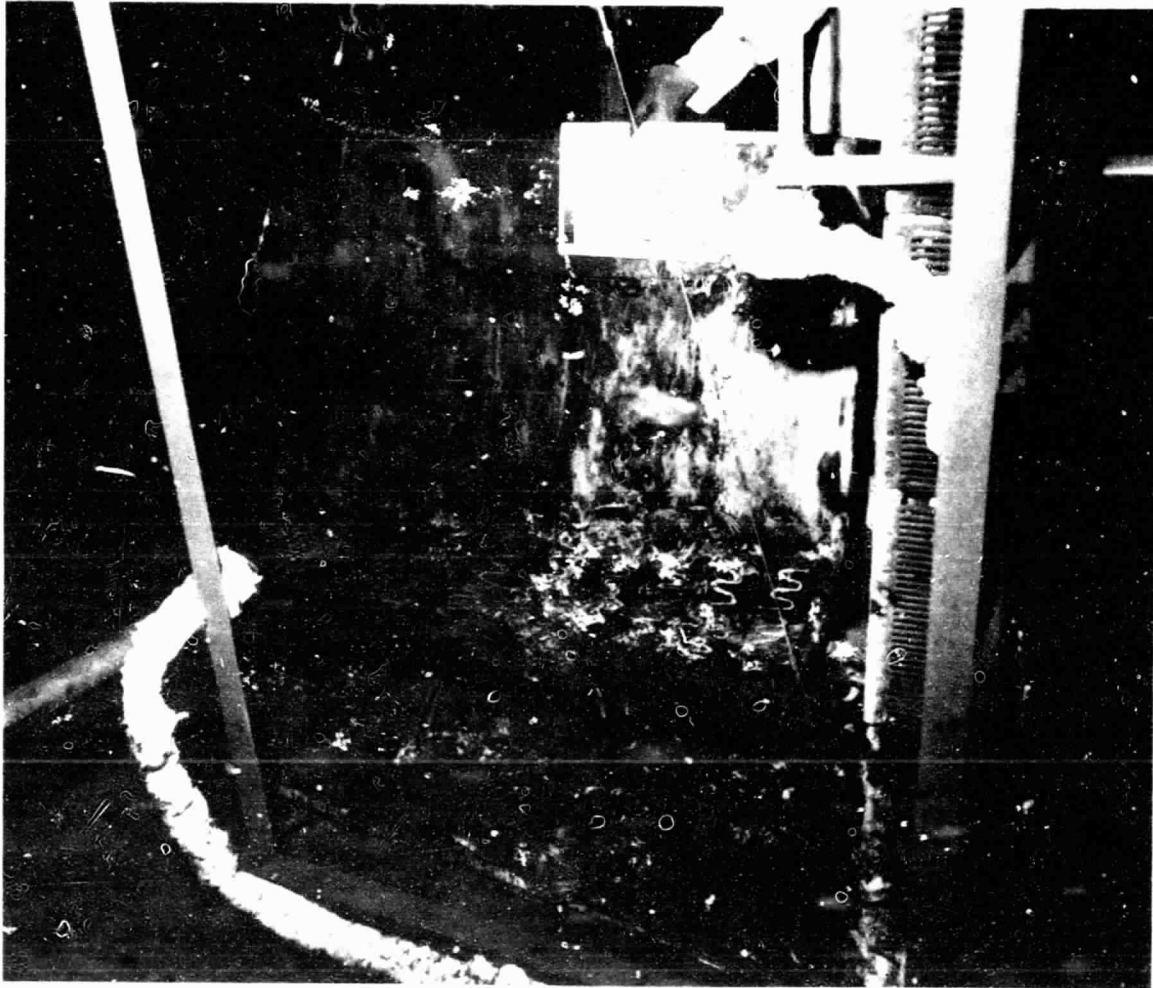
Fire Blocking

None

Foam

F. R. Urethane

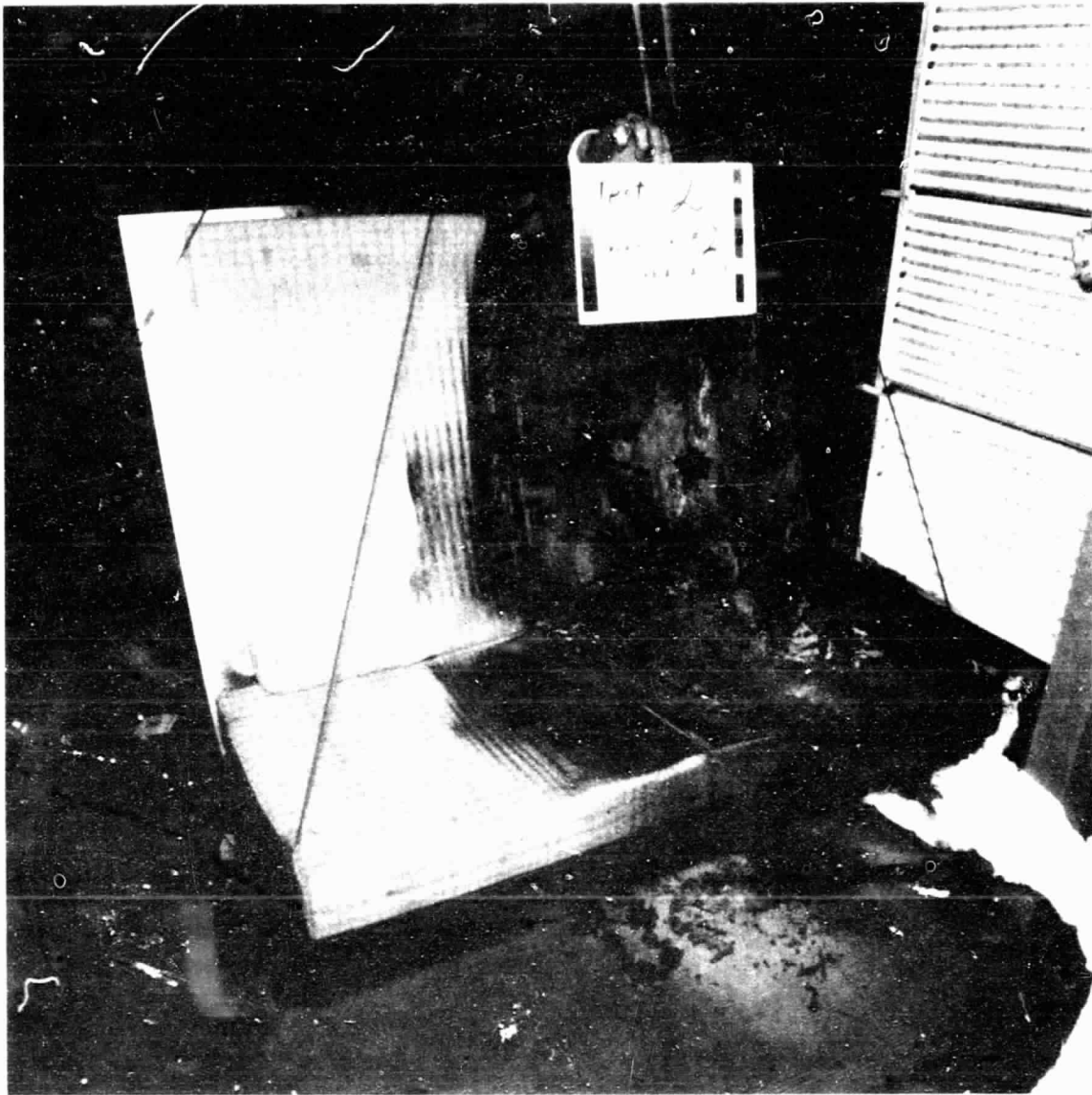
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Construction Number	Decorative Upholstery	Slip Cover	Fire Blocking	Foam
1	Wool-Nylon	None	None	F. R. Urethane

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BLACK AND WHITE PHOTOGRAPH



Construction
Number

2

Decorative
Upholstery

Wool-Nylon

Slip Cover

Cotton-Muslin

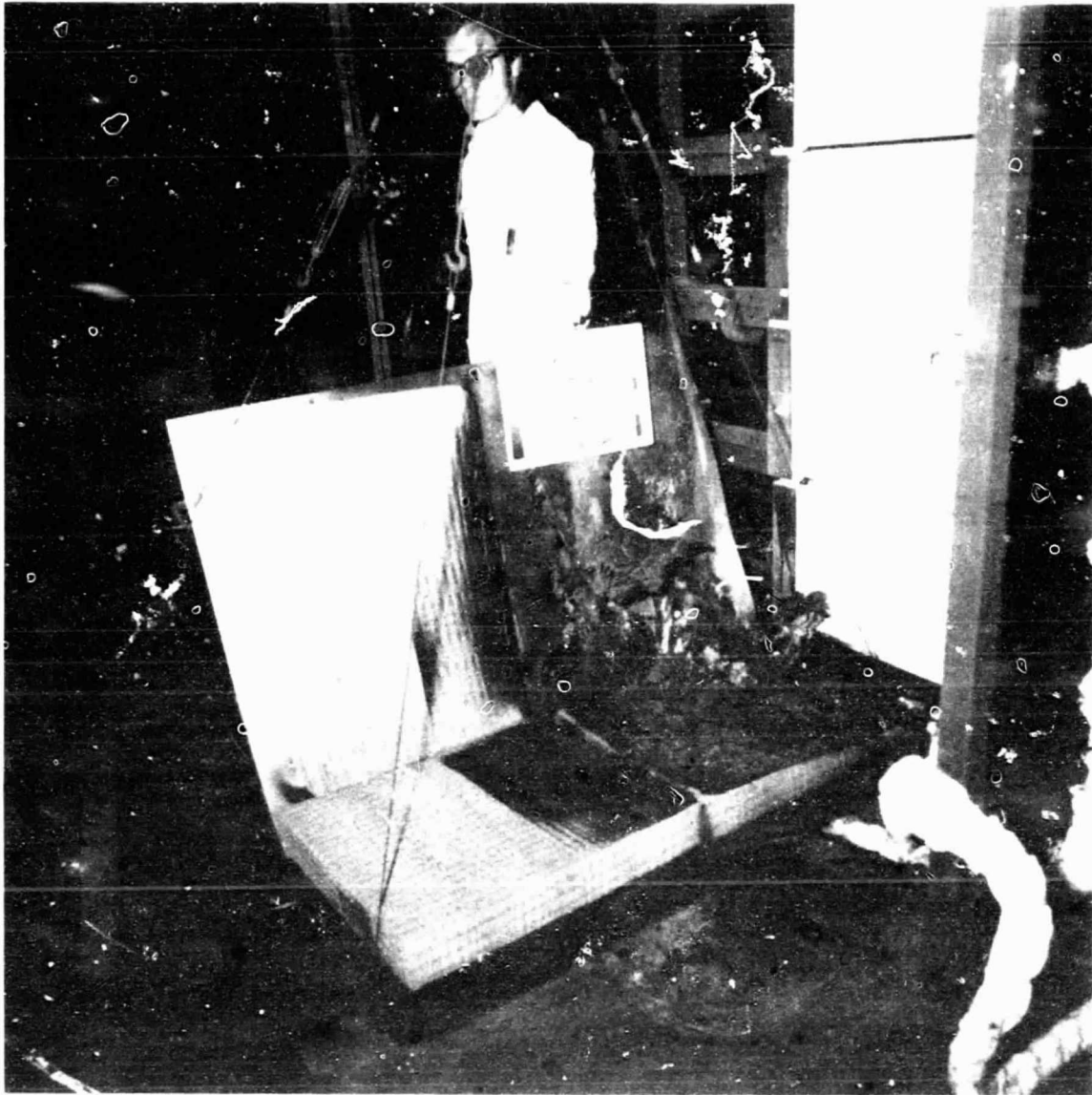
Fire Blocking

Vonar-3

Foam

F. R. Urethane

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Construction
Number

2

Decorative
Upholstery

Wool-Nylon

Slip Cover

Cotton-Muslin

Fire Blocking

Vonar-3

Foam

F. R. Urethane

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Construction
Number

3

Decorative
Upholstery

Wool-Nylon

Slip Cover

Cotton-Muslin

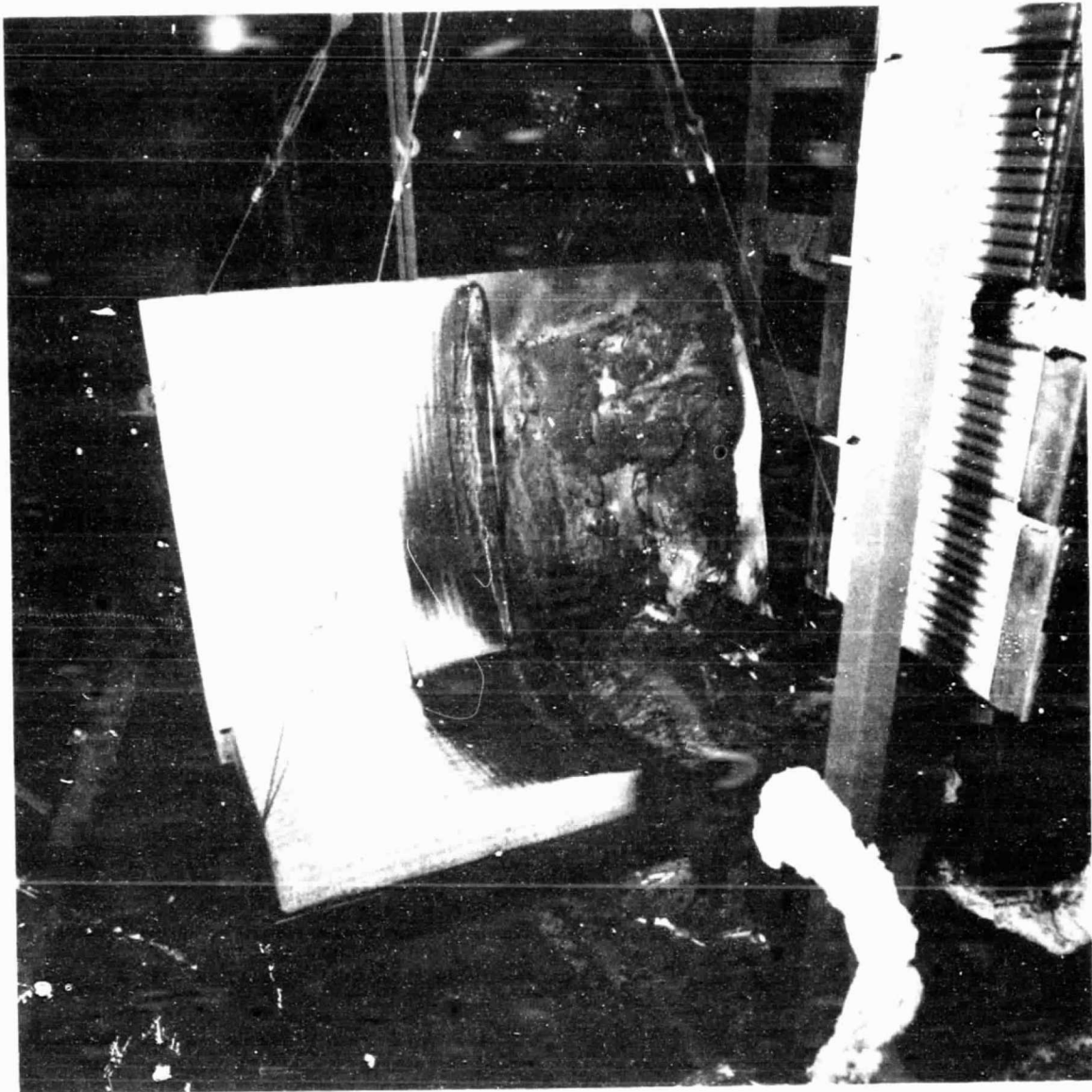
Fire Blocking

Vonar-2

Foam

F. R. Urethane

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Construction
Number

3

Decorative
Upholstery

Wool-Nylon

Slip Cover

Cotton-Muslin

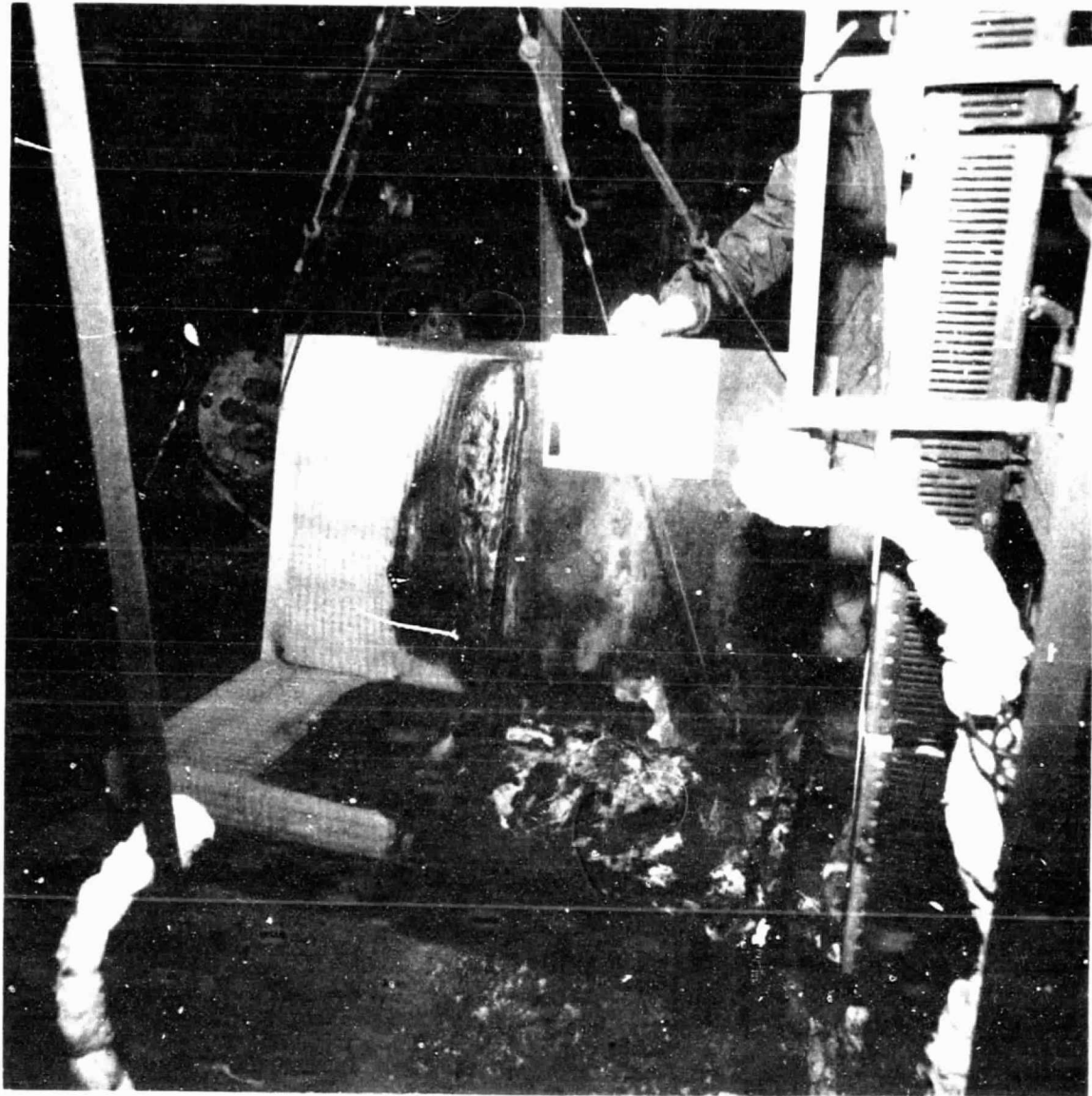
Fire Blocking

Vonar-2

Foam

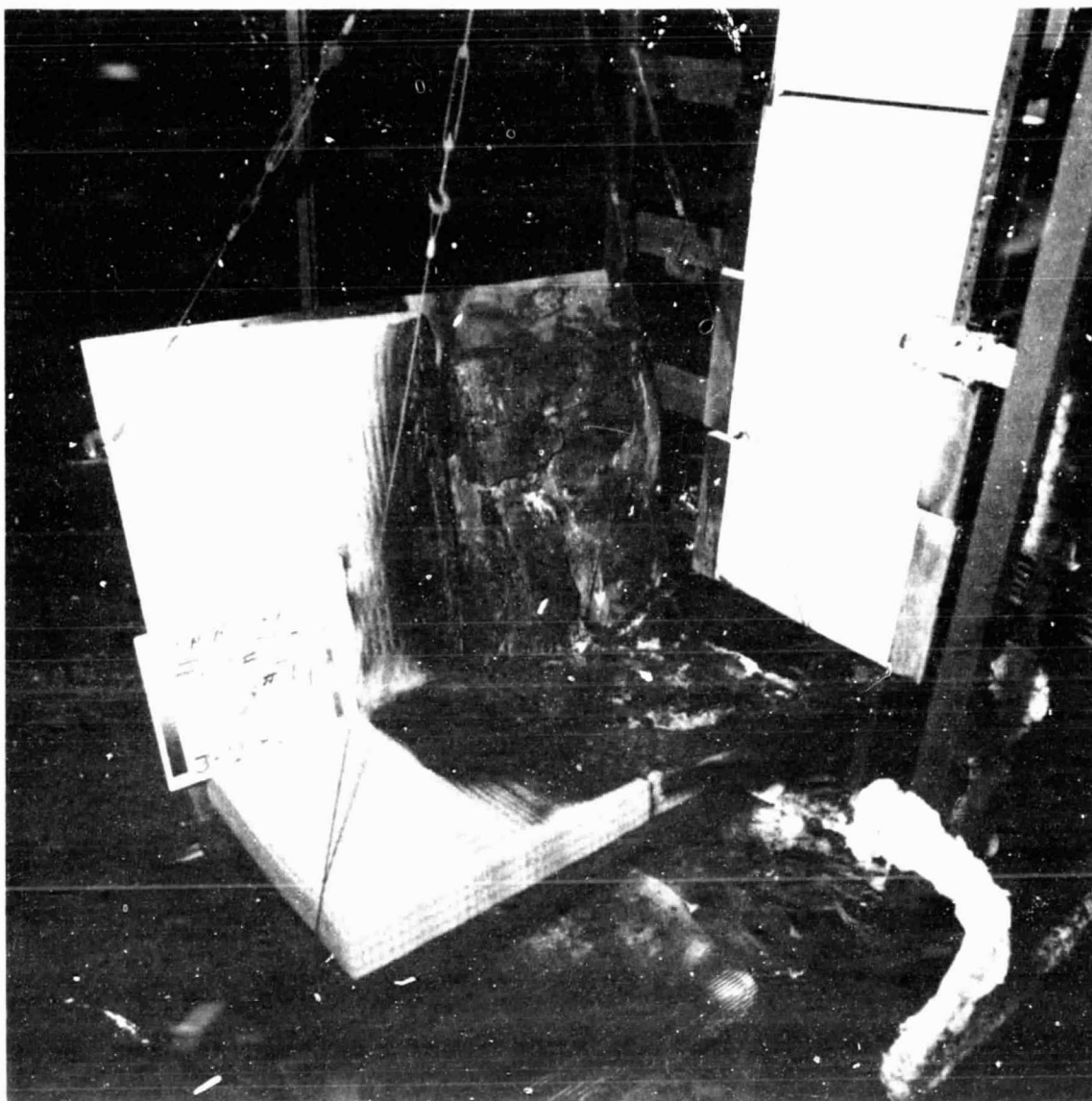
F. R. Urethane

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Construction Number	Decorative Upholstery	Slip Cover	Fire Blocking	Foam
4	Wool-Nylon	None	3/8 LS 200	F. R. Urethane

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Construction
Number
4

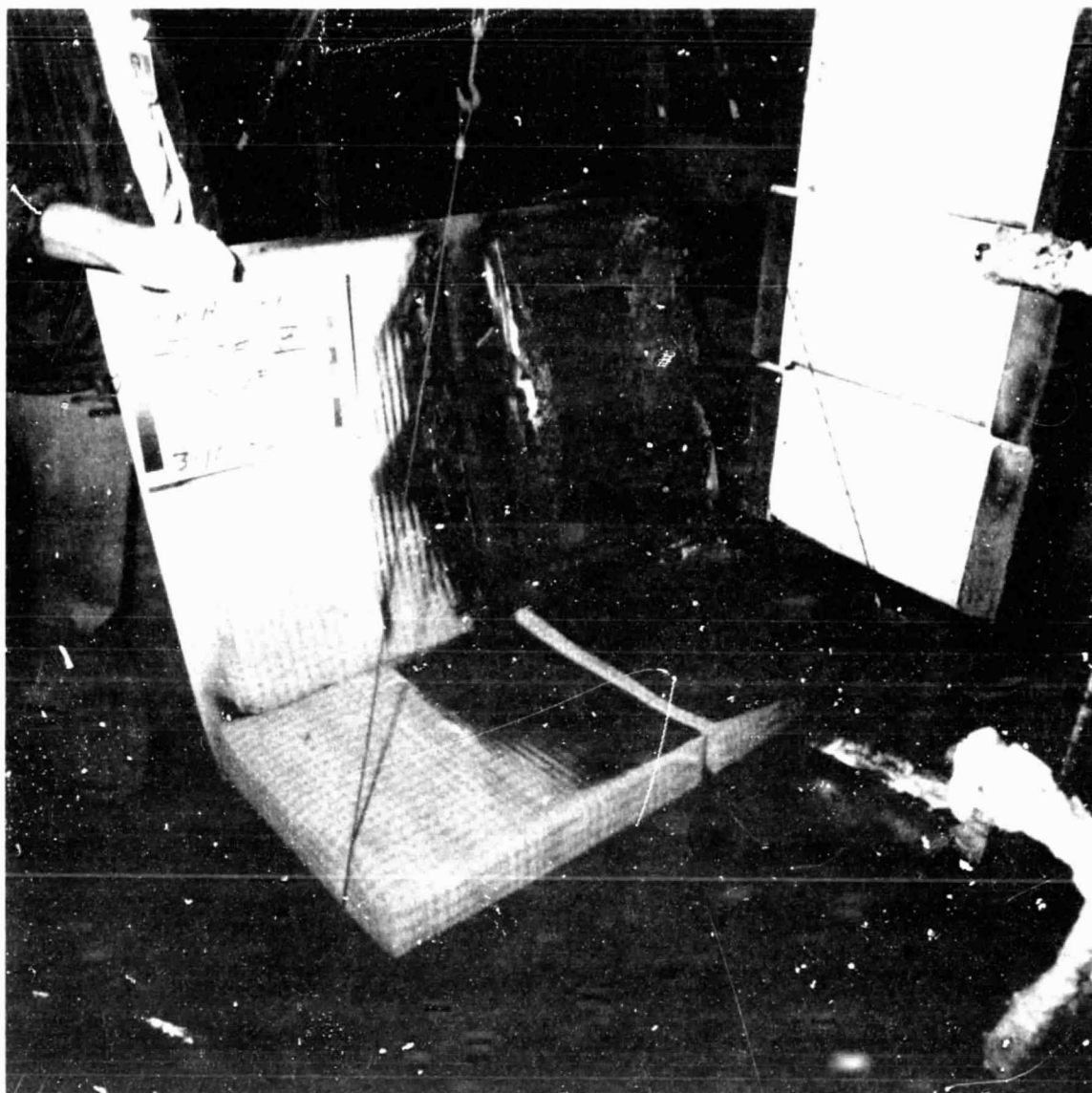
Decorative
Upholstery
Wool-Nylon

Slip Cover
None

Fire Blocking
3/8 LS 200

Foam
F. R. Urethane

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Construction
Number

5

Decorative
Upholstery

Wool-Nylon

Slip Cover

None

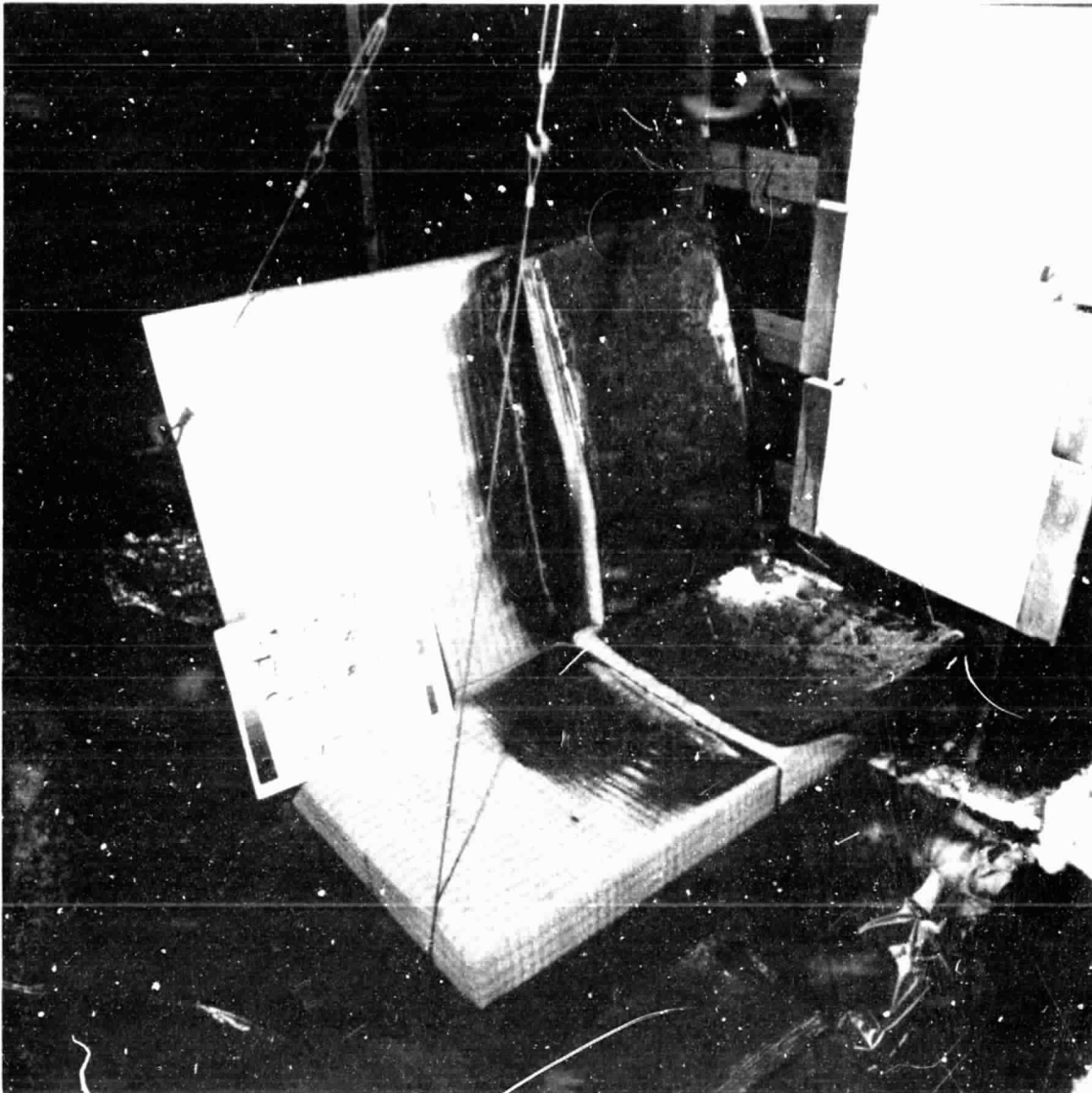
Fire Blocking

Celiox 101

Foam

F. R. Urethane

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Construction Number	Decorative Upholstery	Slip Cover	Fire Blocking	Foam
5	Wool-Nylon	None	Celiox 101	F. R. Urethane

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Construction
Number

6

Decorative
Upholstery

Wool-Nylon

Slip Cover

None

Fire Blocking

Norfab 11 HT-26-AL

Foam

F. R. Urethane

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Construction
Number

6

Decorative
Upholstery

Wool-Nylon

Slip Cover

None

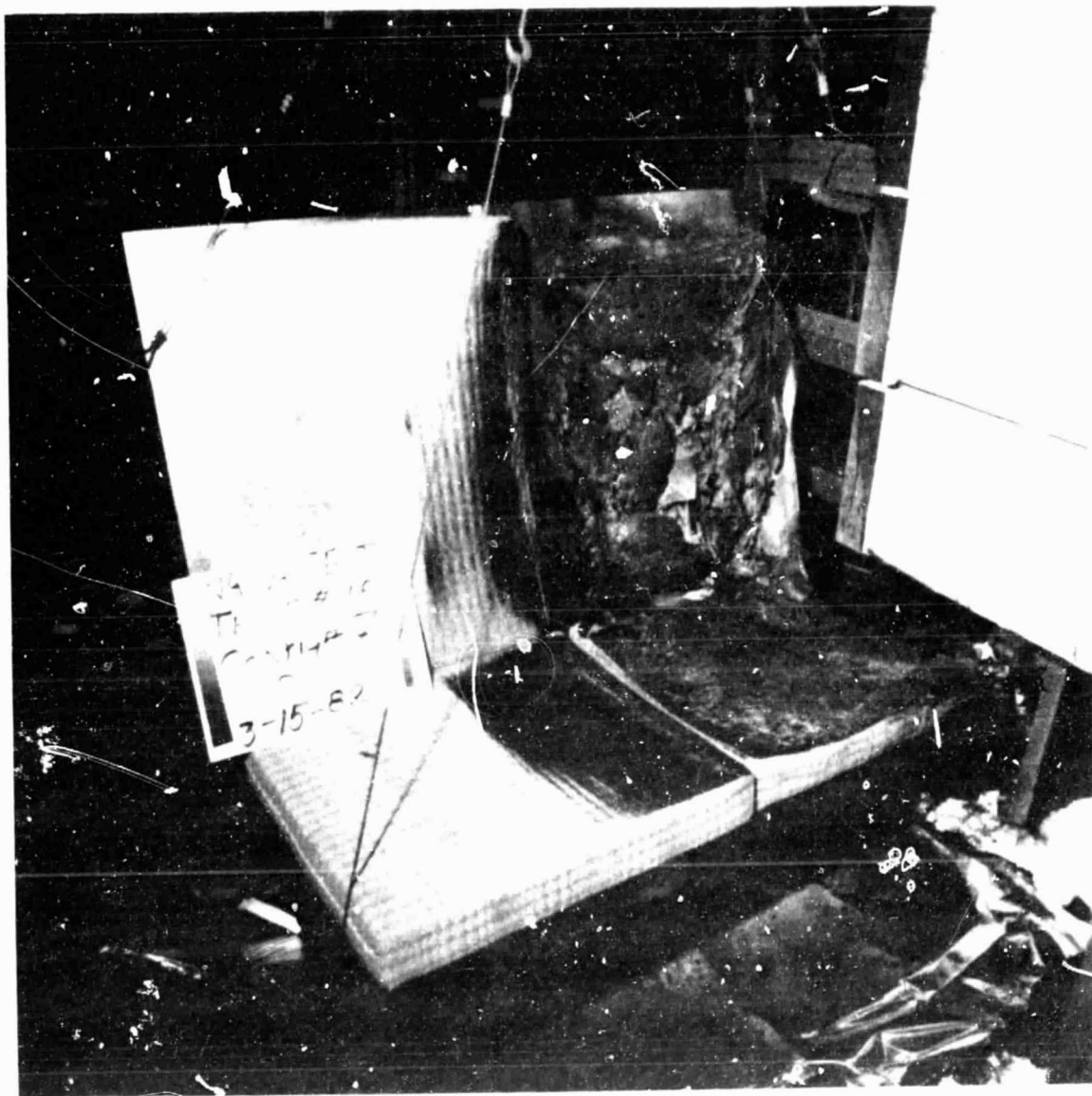
Fire Blocking

Morfab 11 HT-26-AL

Foam

F. R. Urethane

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Construction
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7

Decorative
Upholstery

Wool-Nylon

Slip Cover

Cotton-Muslin

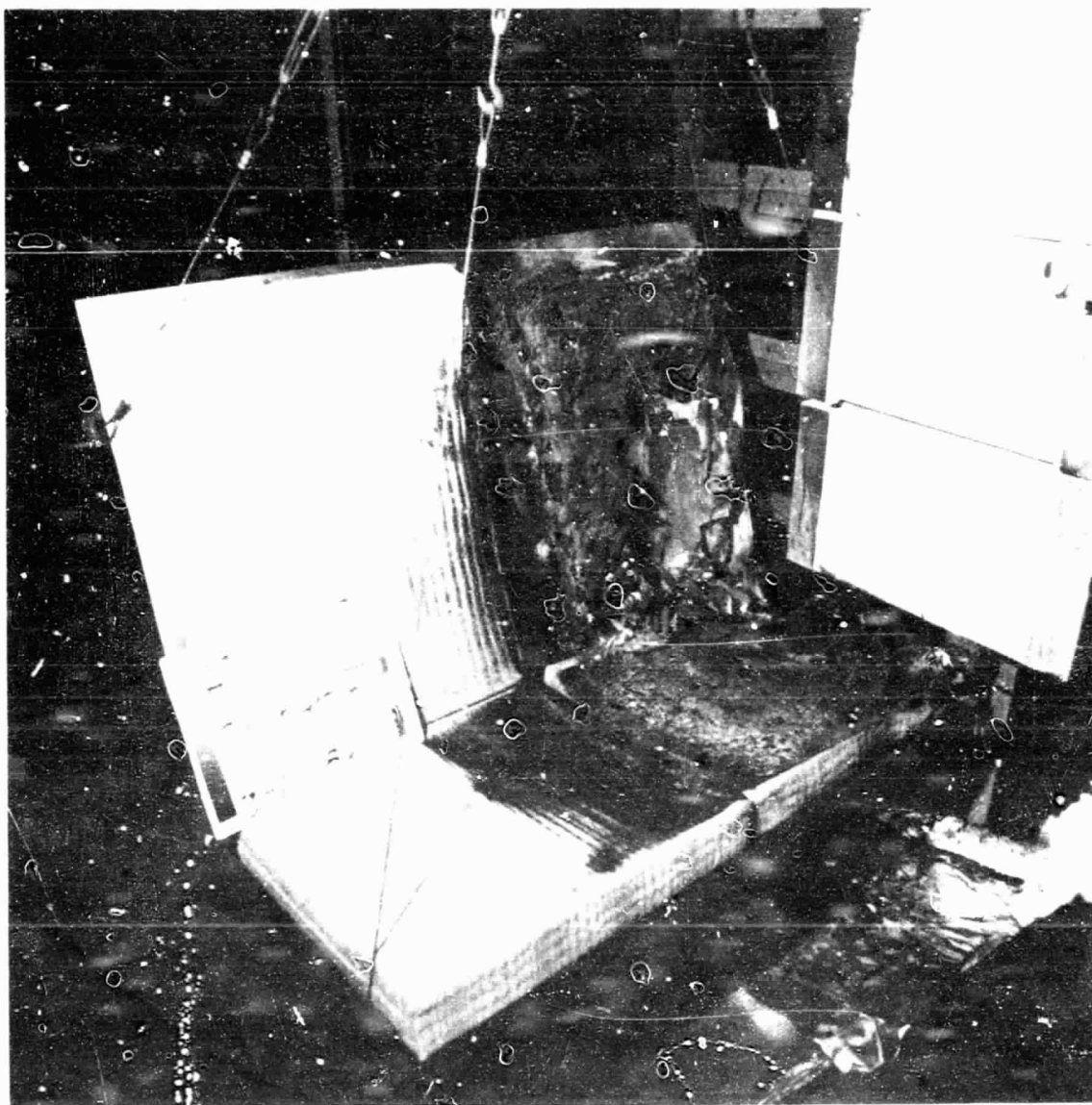
Fire Blocking

Vonar-3

Foam

N. F. Urethane

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Construction
Number

7

Decorative
Upholstery

Wool-Nylon

Slip Cover

Cotton-Muslin

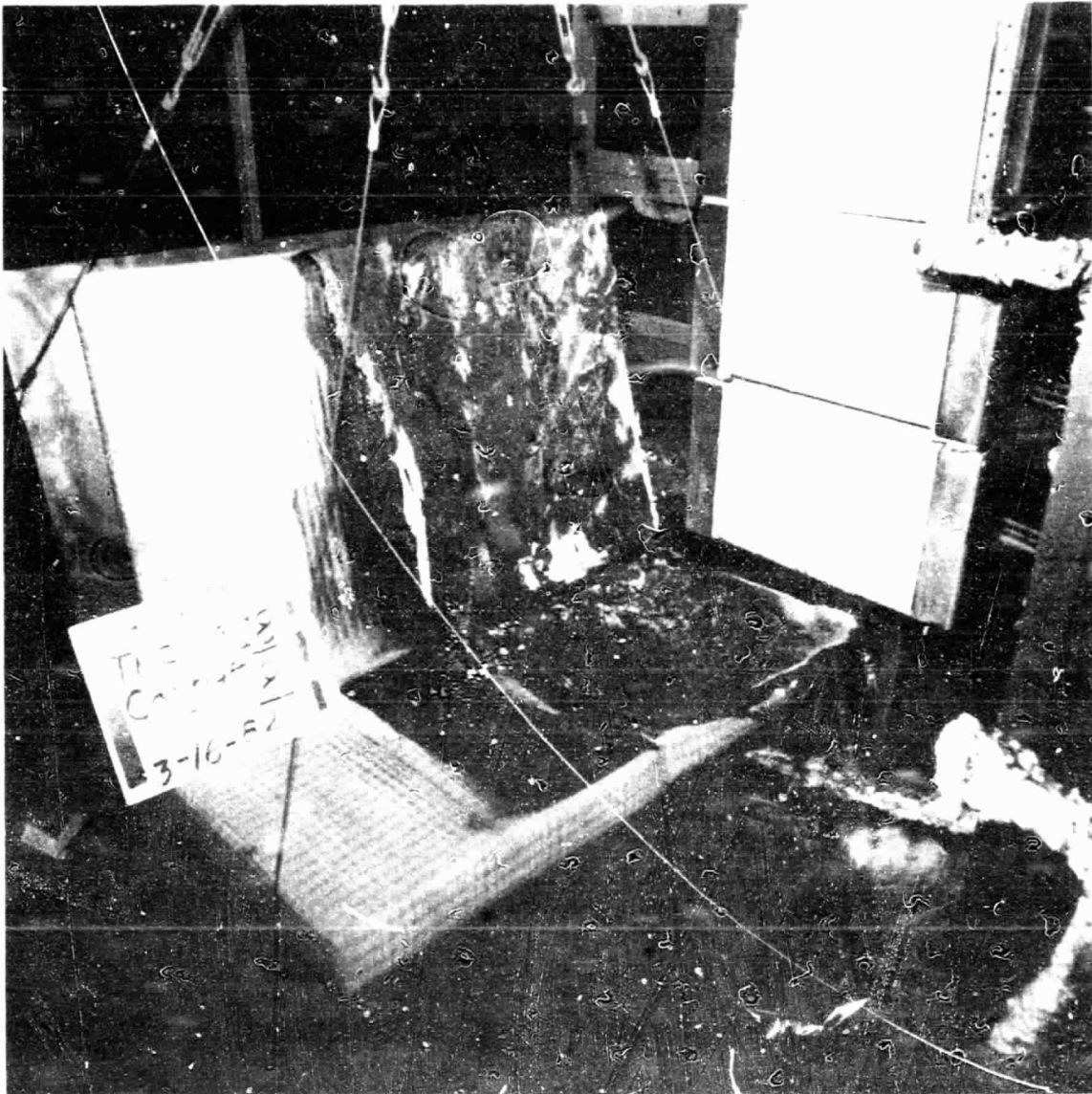
Fire Blocking

Vonar-3

Foam

N. F. Urethane

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Construction
Number
8

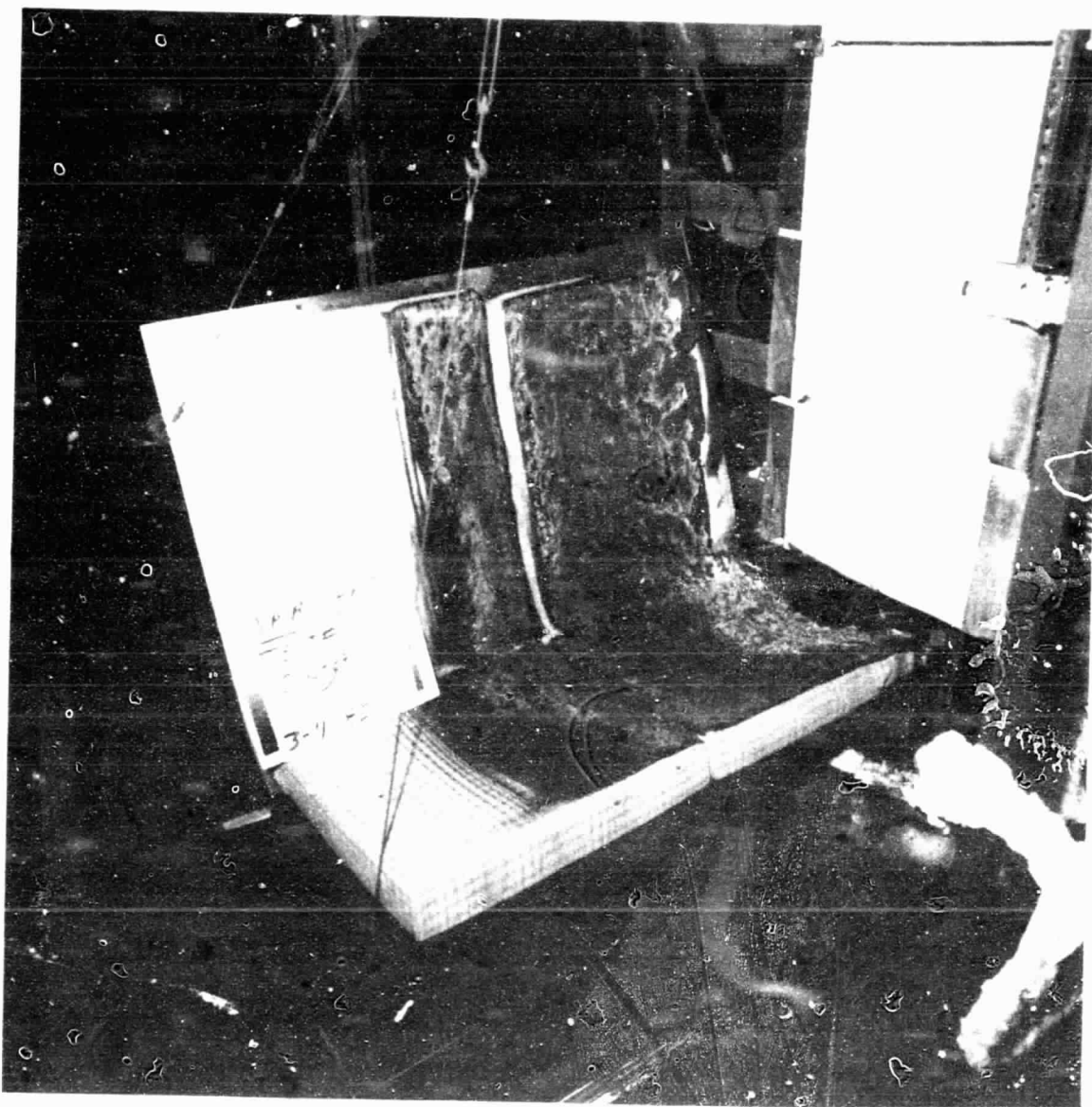
Decorative
Upholstery
Wool-Nylon

Slip Cover
None

Fire Blocking
Norfab 11 HT-26-AL

Foam
N. F. Urethane

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Construction
Number

9

Decorative
Upholstery
Wool-Nylon

Slip Cover
None

Fire Blocking
None

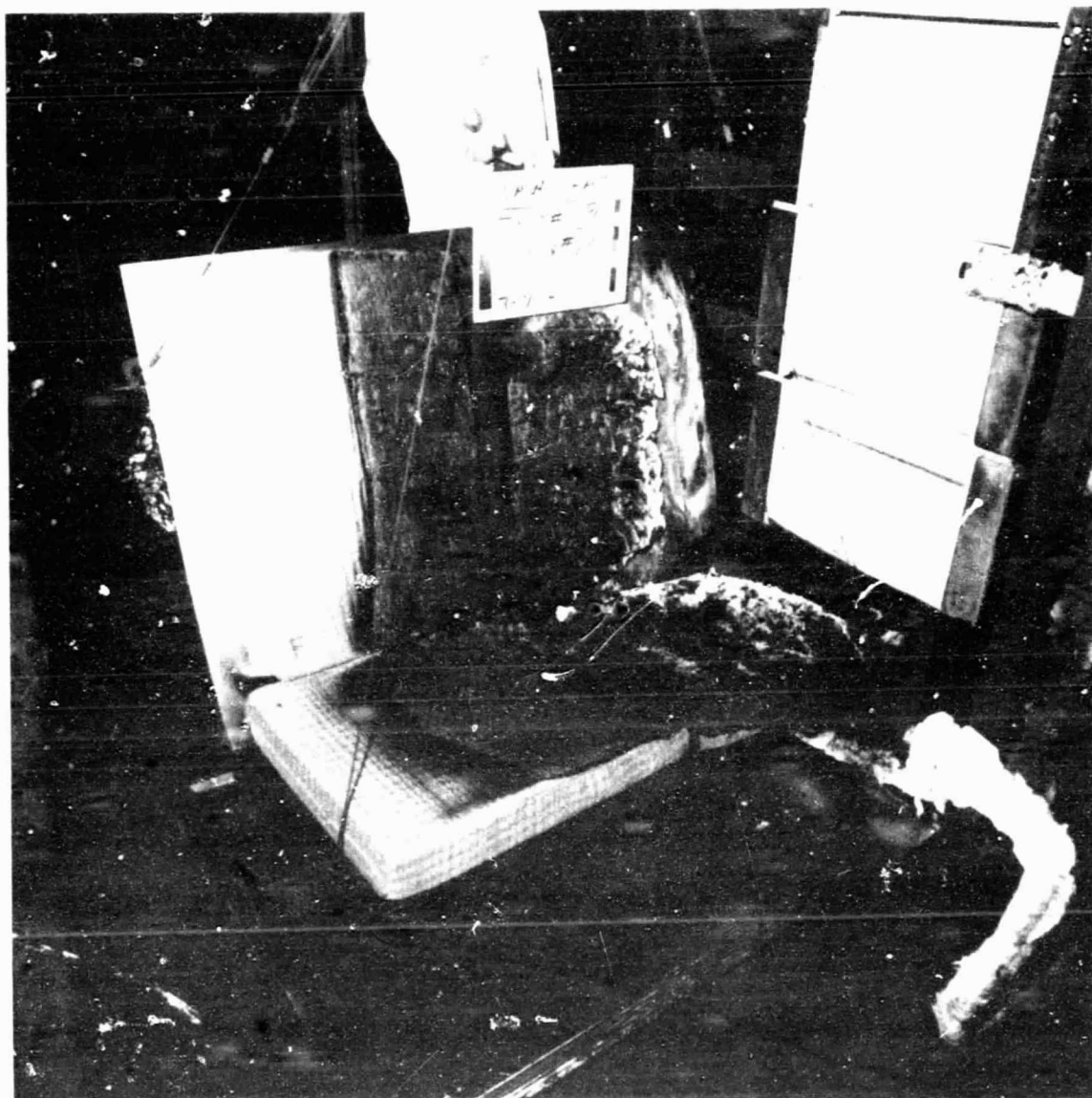
Foam
LS 200 Neoprene

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Construction Number	Decorative Upholstery	Slip Cover	Fire Blocking	Foam
8	Wool-Nylon	None	Norfab 11 HT-26-AL	N. F. Urethane



Construction
Number

10

Decorative
Upholstery

Wool-Nylon

Slip Cover

None

Fire Blocking

None

Foam

Polyimide

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Construction
Number
10

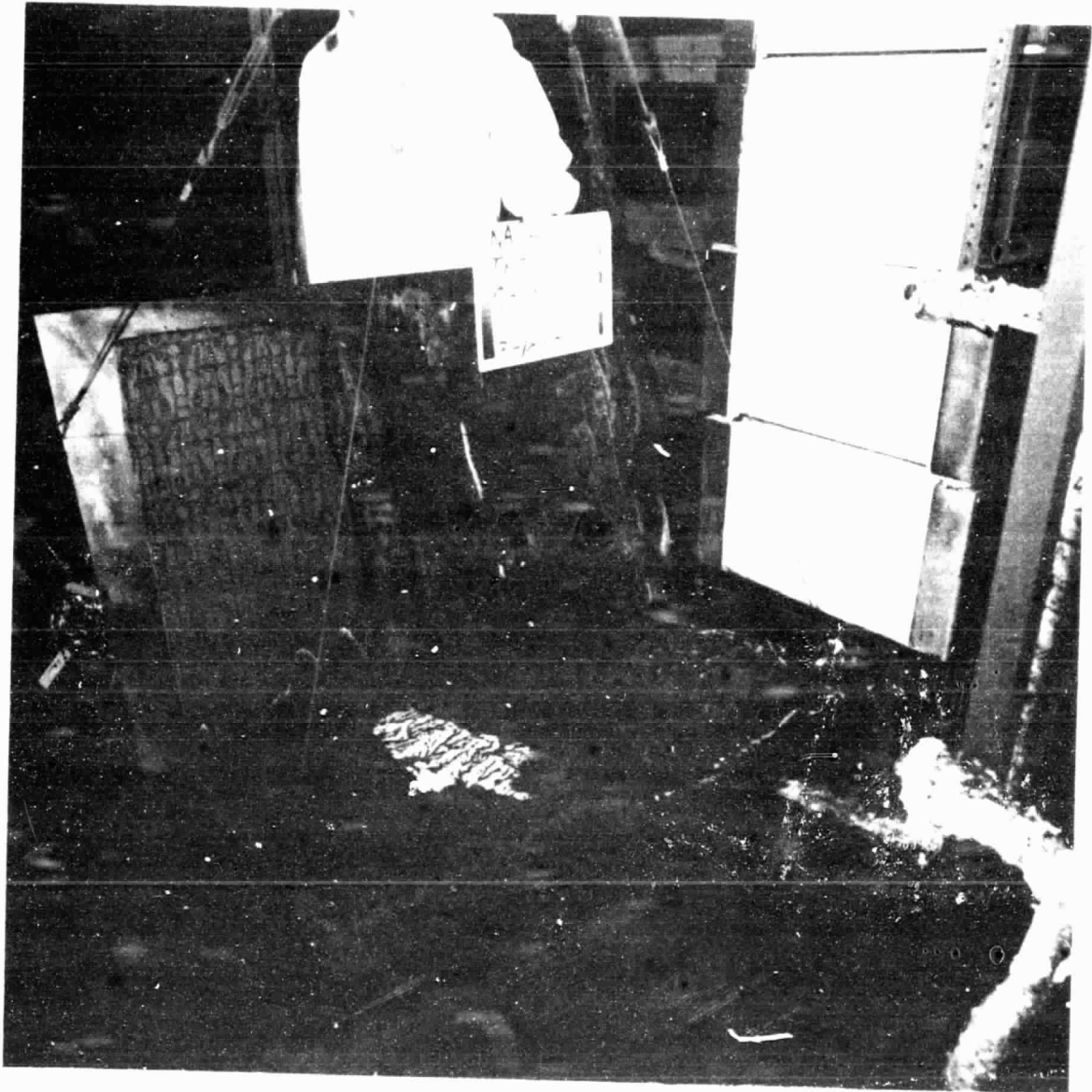
Decorative
Upholstery
Wool-Nylon

Slip Cover
None

Fire Blocking
None

Foam
Polyimide

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Construction
Number
11

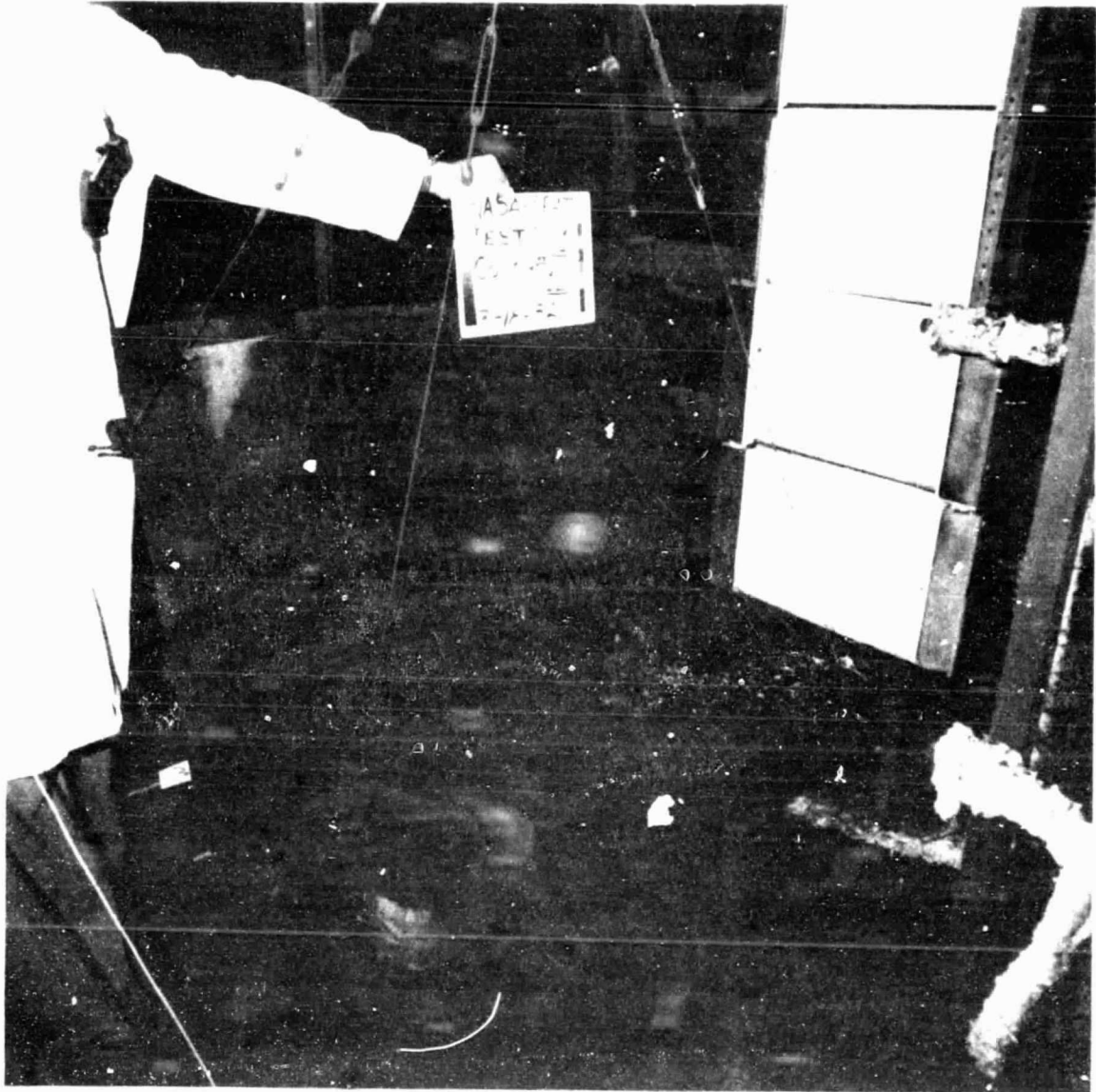
Decorative
Upholstery
Polyester

Slip Cover
None

Fire Blocking
None

Foam
Polyimide

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Construction
Number
12

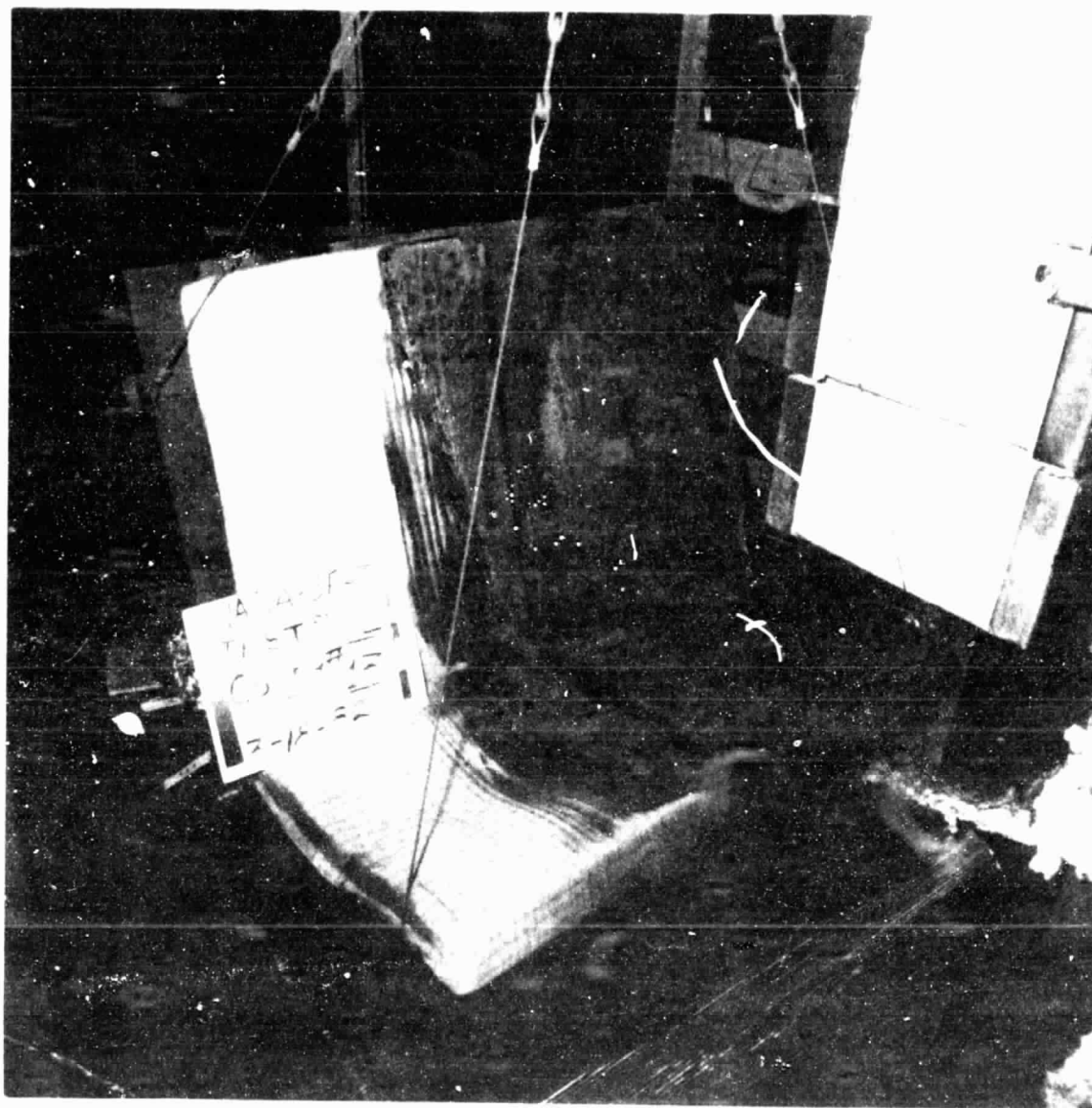
Decorative
Upholstery
Wool-Nylon

Slip Cover
None

Fire Blocking
Norfab 11 HT-26

Foam
F. R. Urethane

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Construction
Number

13

Decorative
Upholstery

Wool-Nylon

Slip Cover

None

Fire Blocking

PBI

Foam

F. R. Urethane

CO. 14-1-1 PART 1
SLA. 14-1-1-10-1-1-1

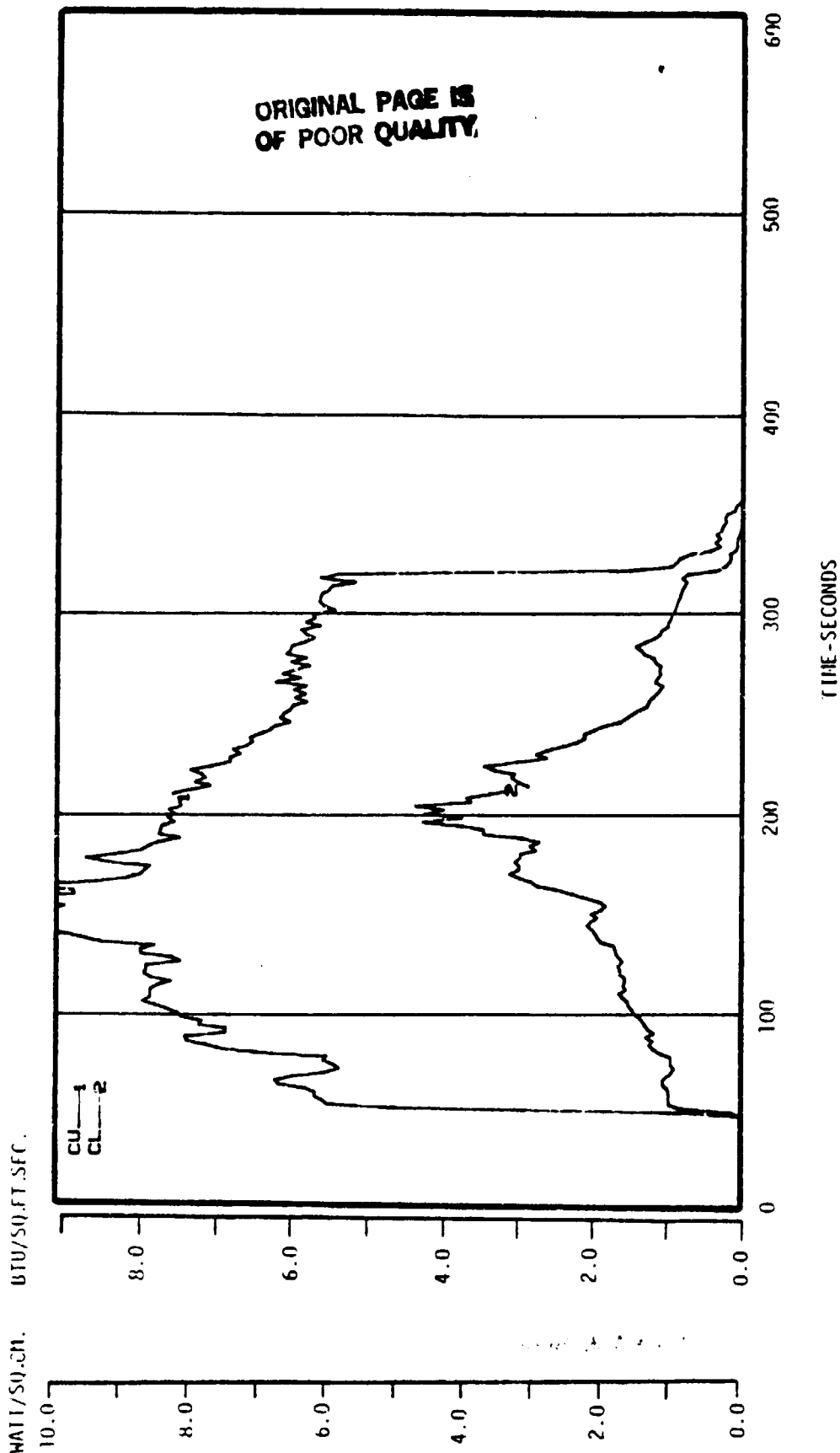
APPENDIX B

Test Plots

Construction	Page
1 Test 1 17	54 62
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3 Test 11 12	86 94
4 Test 3 10	102 110
5 Test 7 13	118 126
6 Test 6 14	134 142
7 Test 15 16	150 158
8 Test 18	166
9 Test 8 19	174 182
10 Test 9 6	190 198
11 Test 20	206
12 Test 21	214
13 Test 22	222

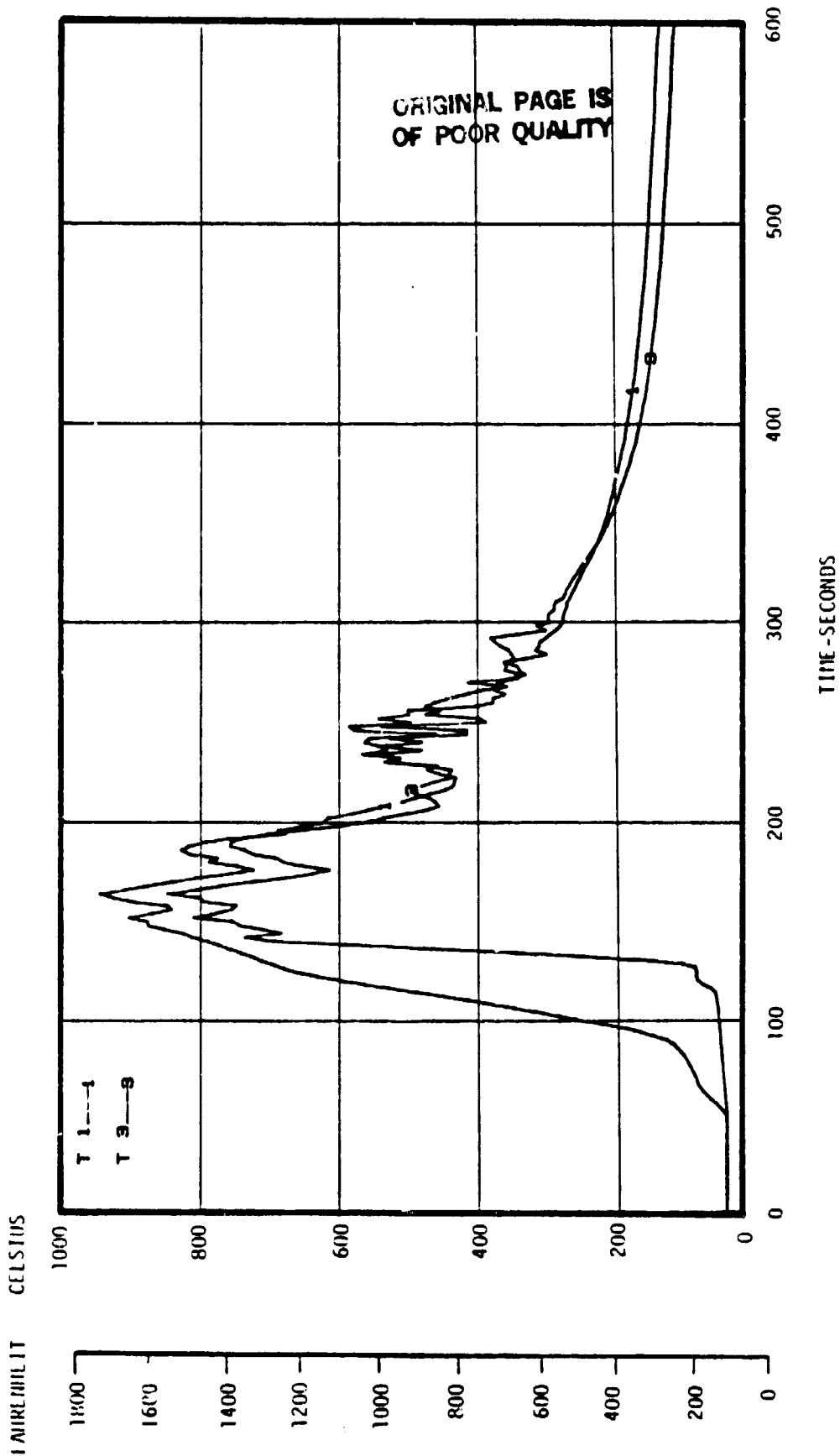
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HEAT FLUX



DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 55/05/02 08.38
NASA-MES FULL SCALE CUSHION BURN TEST NUMBER 1
CUSHION CONSTRUCTION NUMBER 1.0

SEAT CUSHION TEMPERATURES



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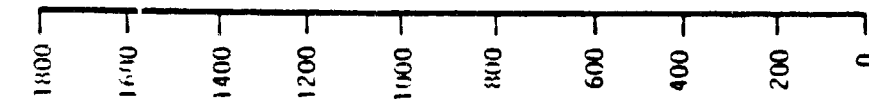
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CUSHION CONSTRUCTION NUMBER 1.0

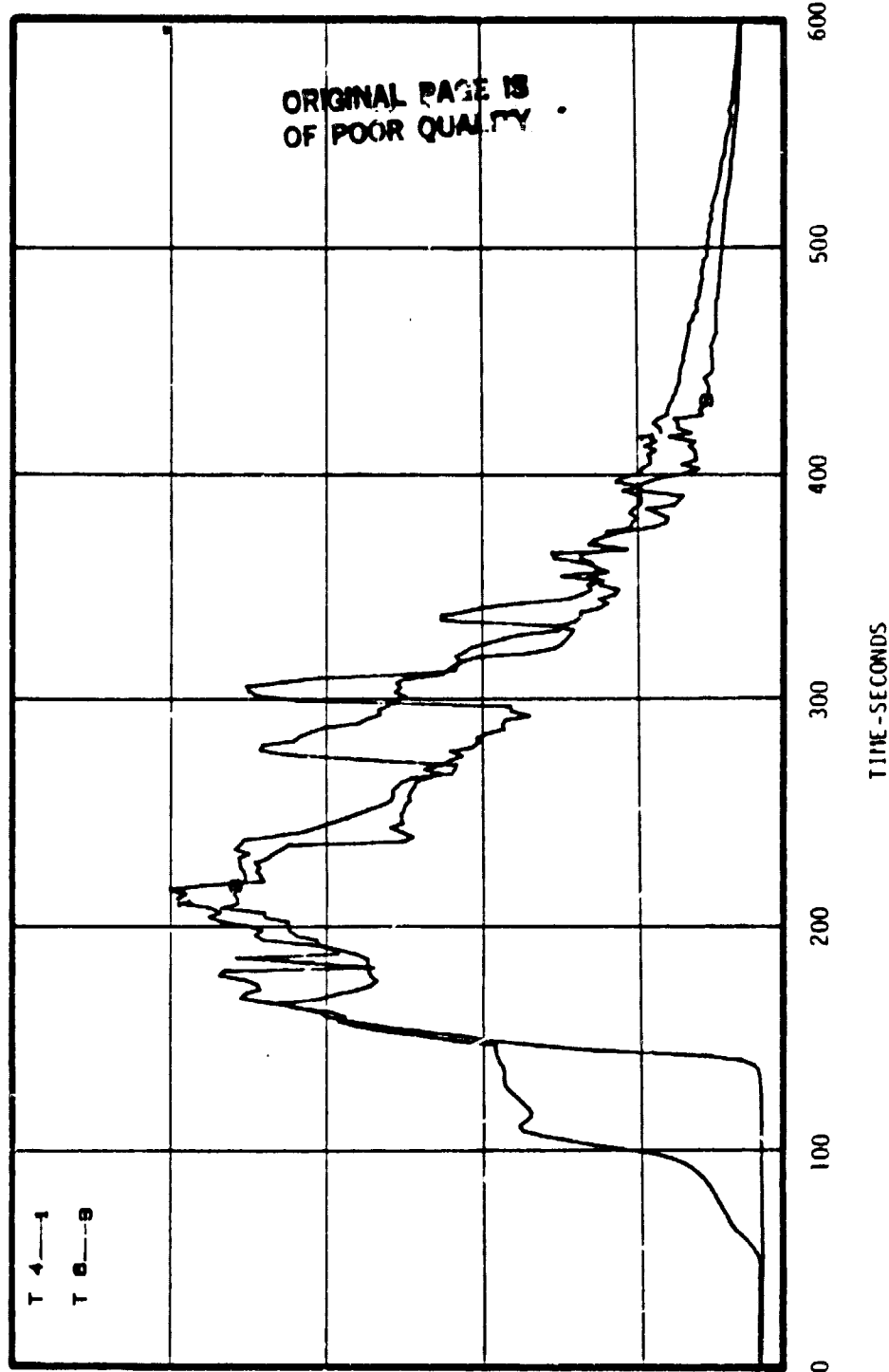
SEAT CUSHION TEMPERATURES

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CELSIUS



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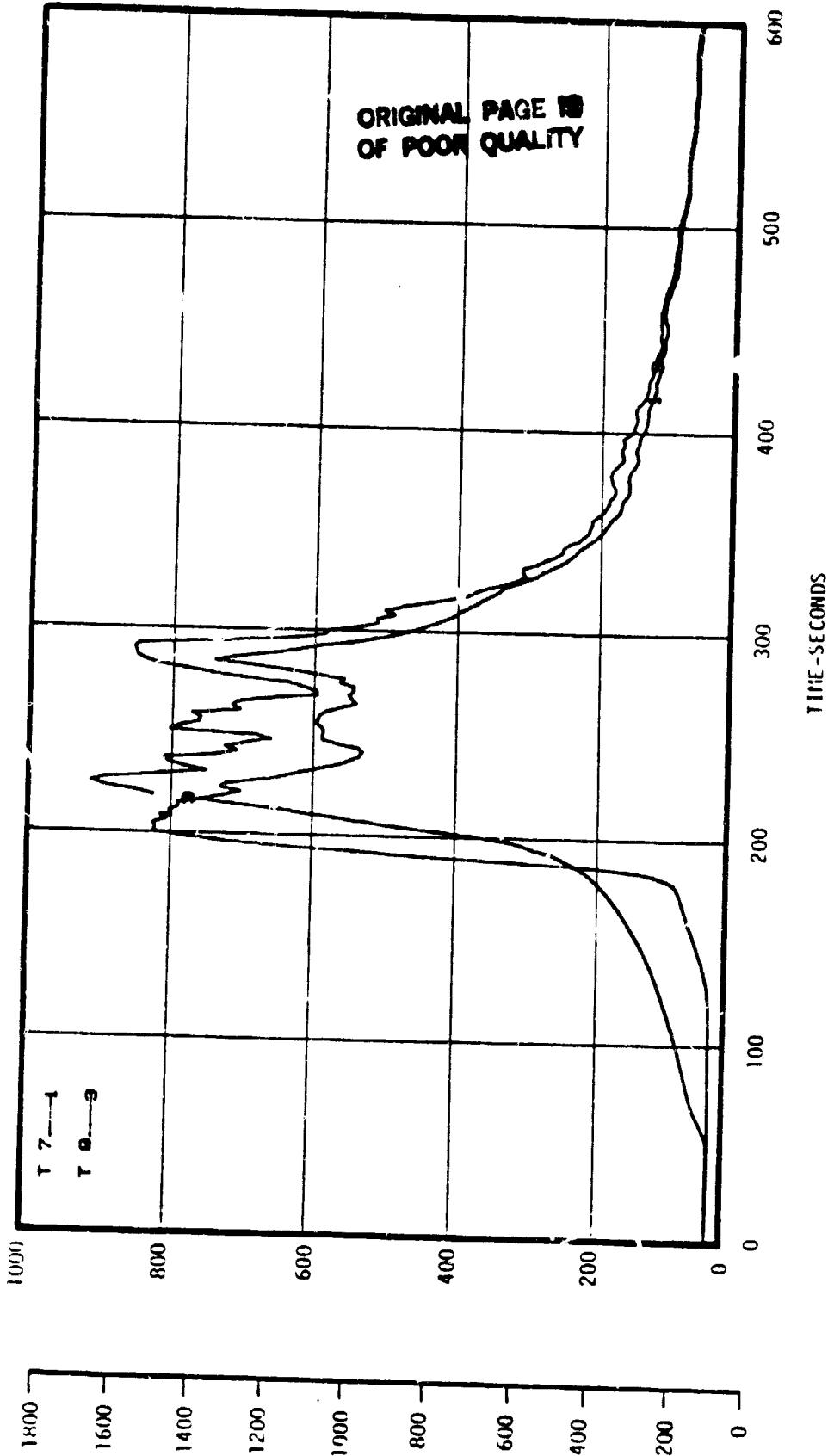
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NASA-AMES FULL SCALE CUSHION BURN TEST NUMBER 1

CUSHION CONSTRUCTION NUMBER 1. B

SEAT CUSHION TEMPERATURES

TEMPERATURE IN CELSIUS



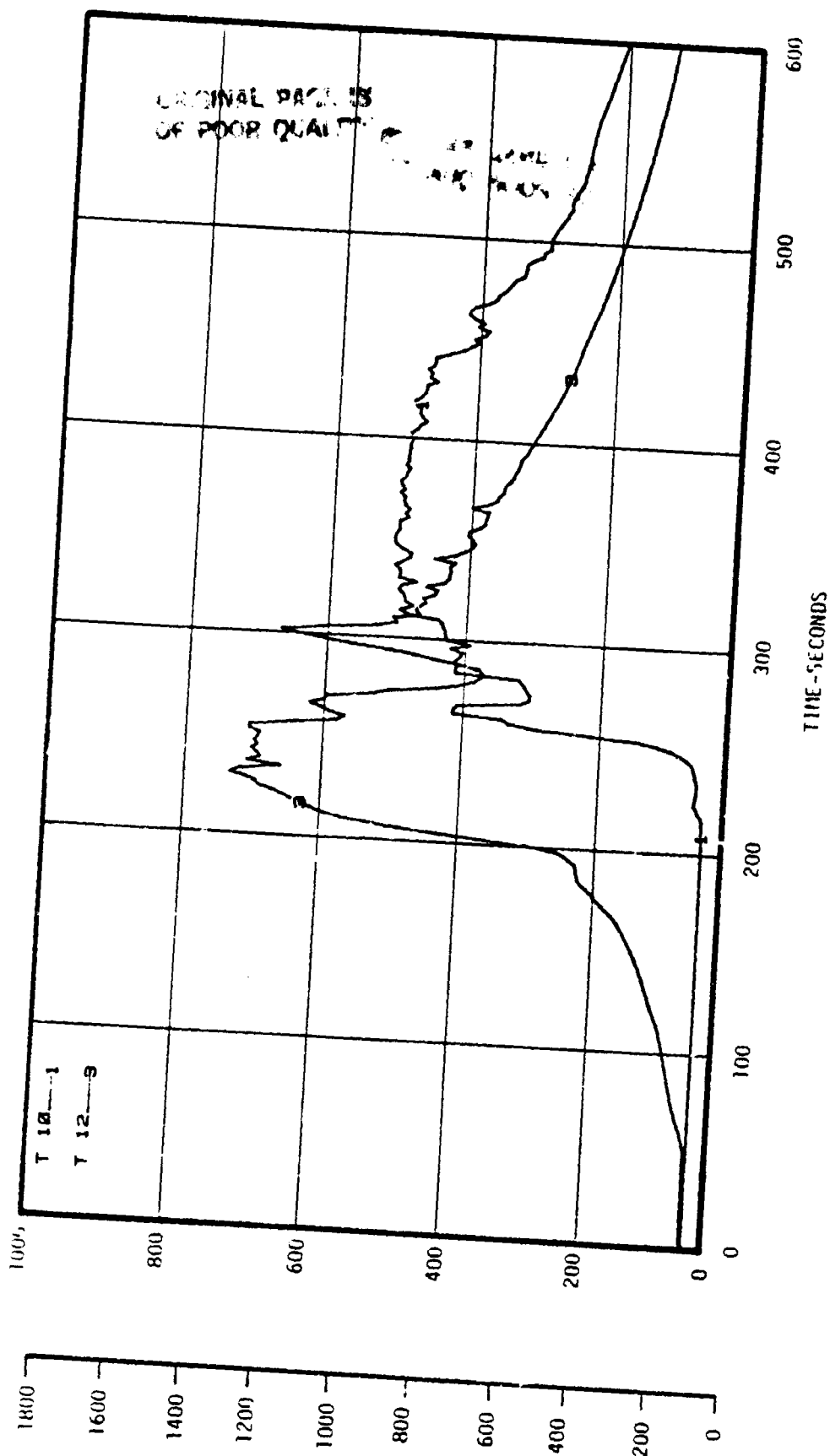
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CUSHION CONSTRUCTION NUMBER 1.0

SEAT CUSHION TEMPERATURES

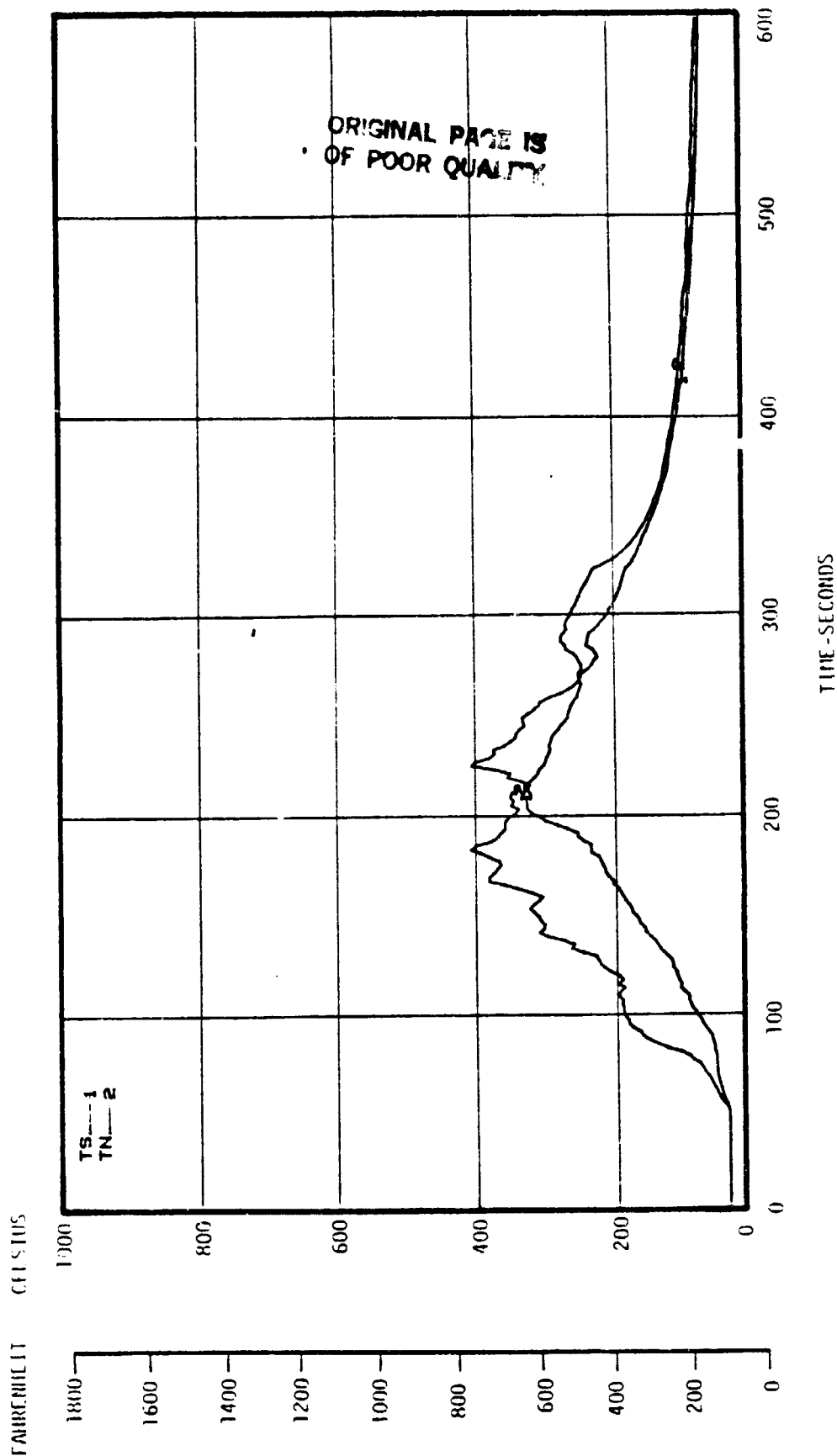
FAHRENHEIT CELSIUS



DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/05/02 08, 98
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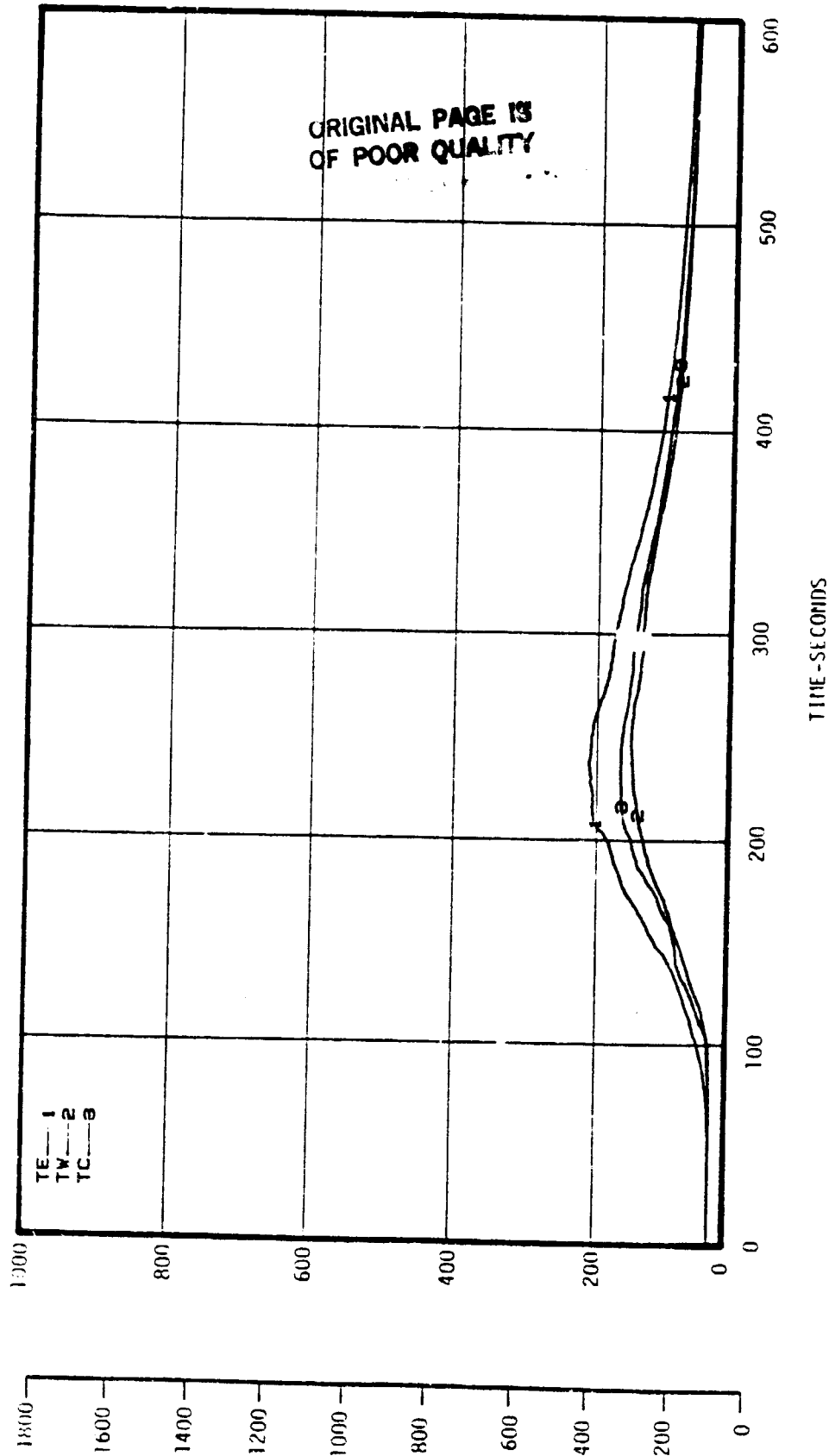
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CUSHION CONSTRUCTION NUMBER 1.0

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FAHRENHEIT CELSIUS

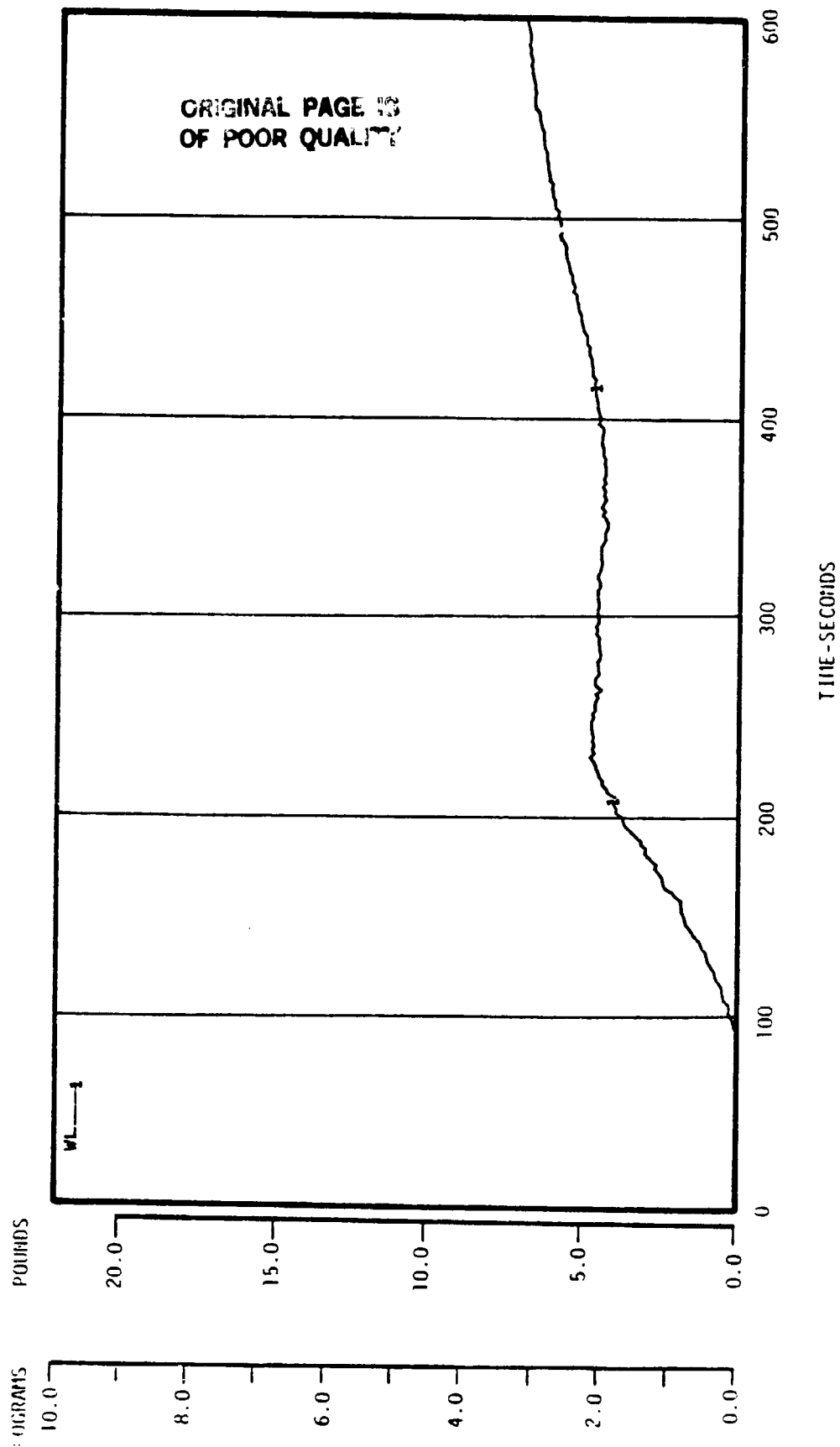


DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/05/82 08.38

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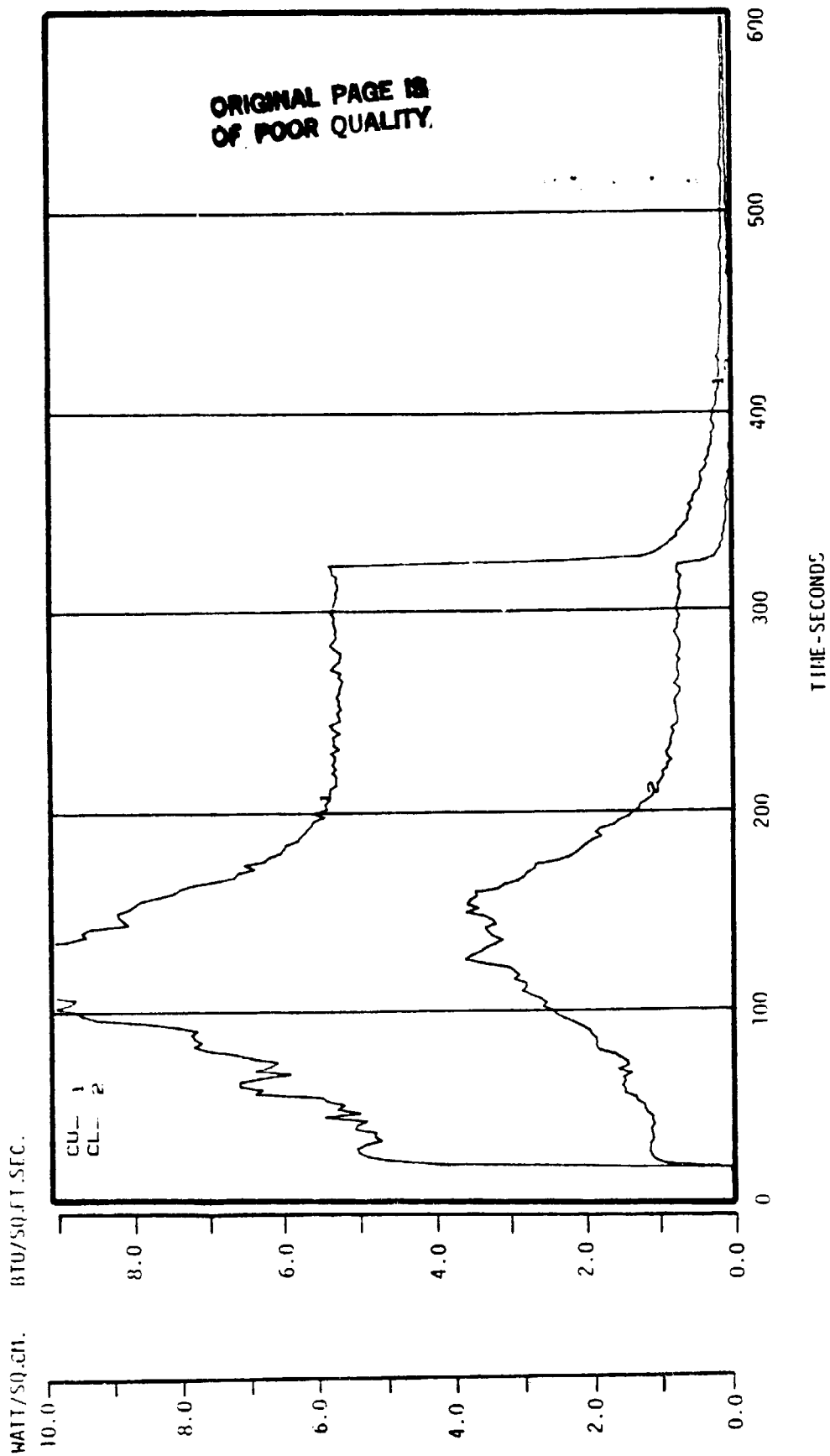
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WEIGHT LOSS



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 CUSHION CONSTRUCTION NUMBER 1.0

HEAT FLUX



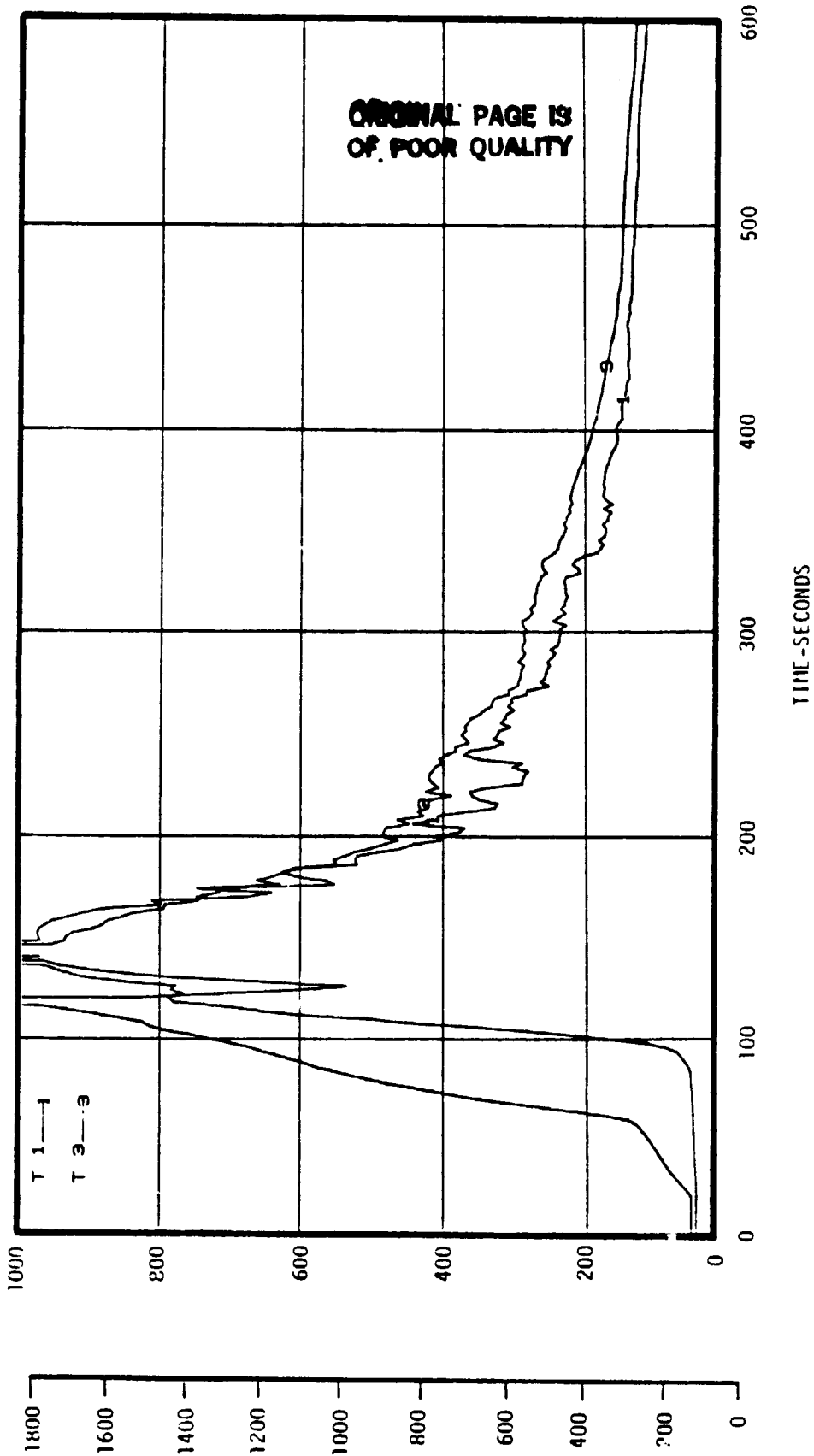
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CUSHION CONSTRUCTION NUMBER 1-0

SEAT CUSHION TEMPERATURES

FAHRENHEIT CELSIUS



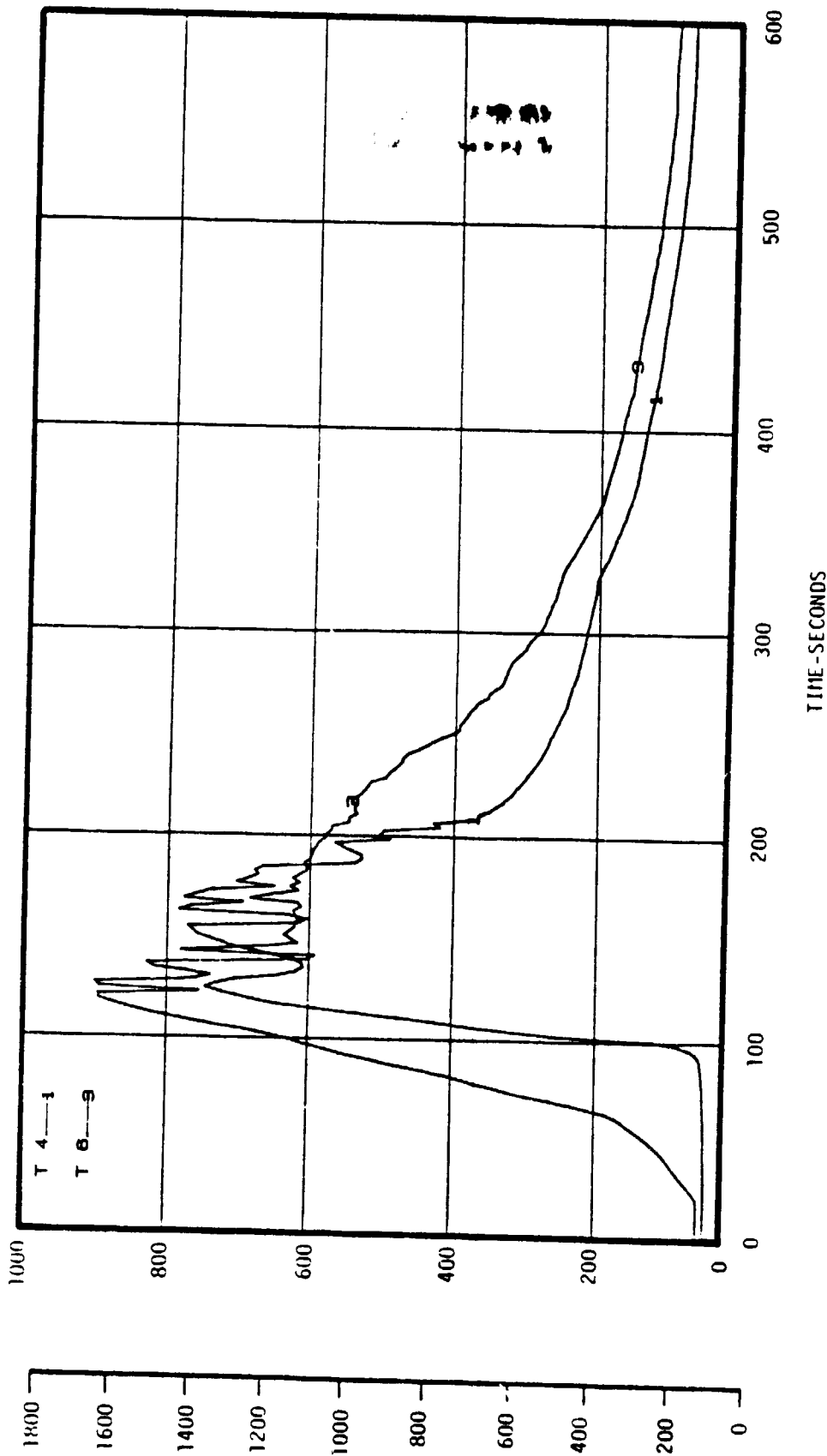
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CUSHION CONSTRUCTION NUMBER 1.0

SEAT CUSHION TEMPERATURES

FAHRENHEIT CELSIUS

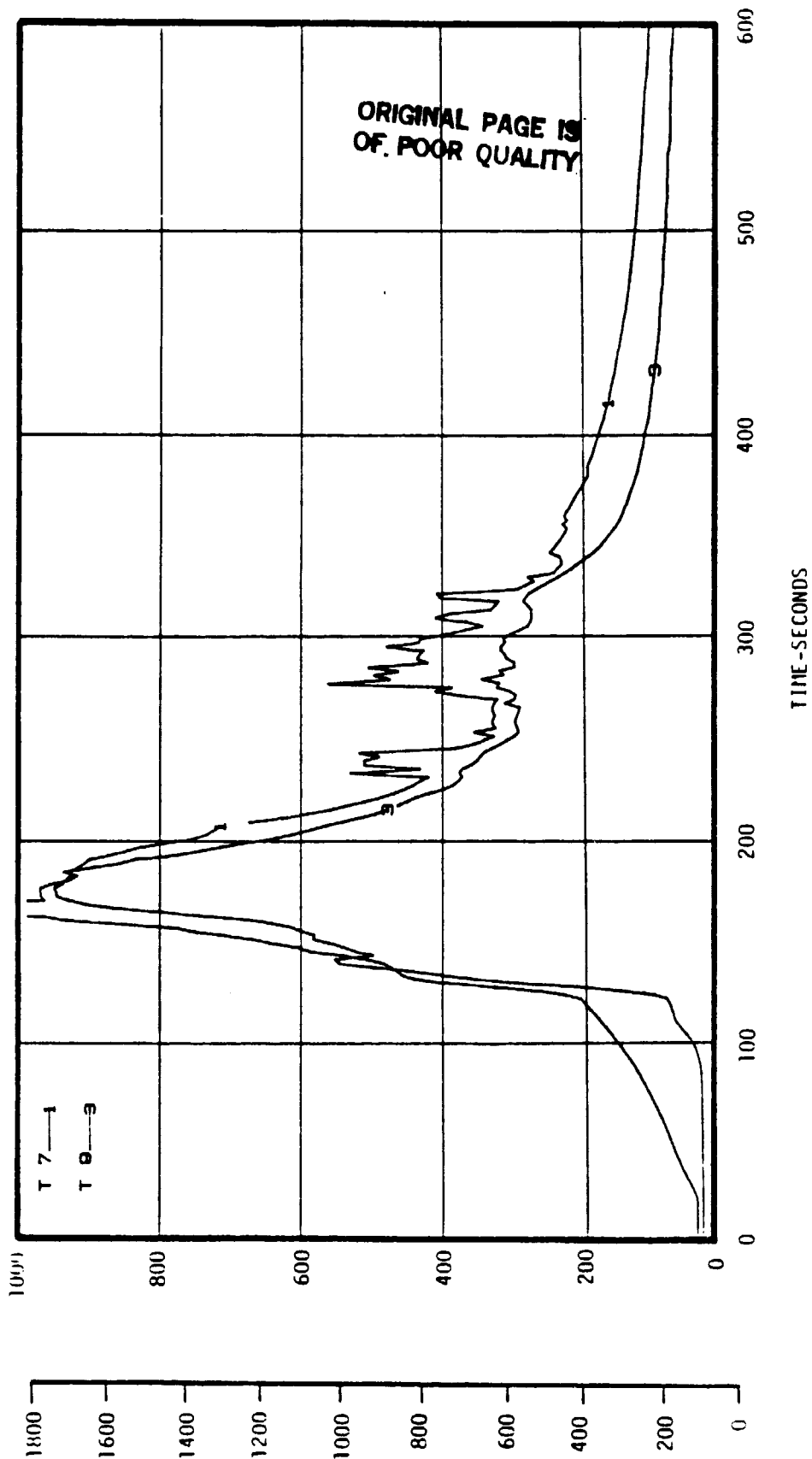


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DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/10/82 11.11
NASA-AMES FULL SCALE CUSHION BURN TEST NUMBER 17
CUSHION CONSTRUCTION NUMBER 1.0

SEAT CUSHION TEMPERATURES

FAHRENHEIT CELSIUS

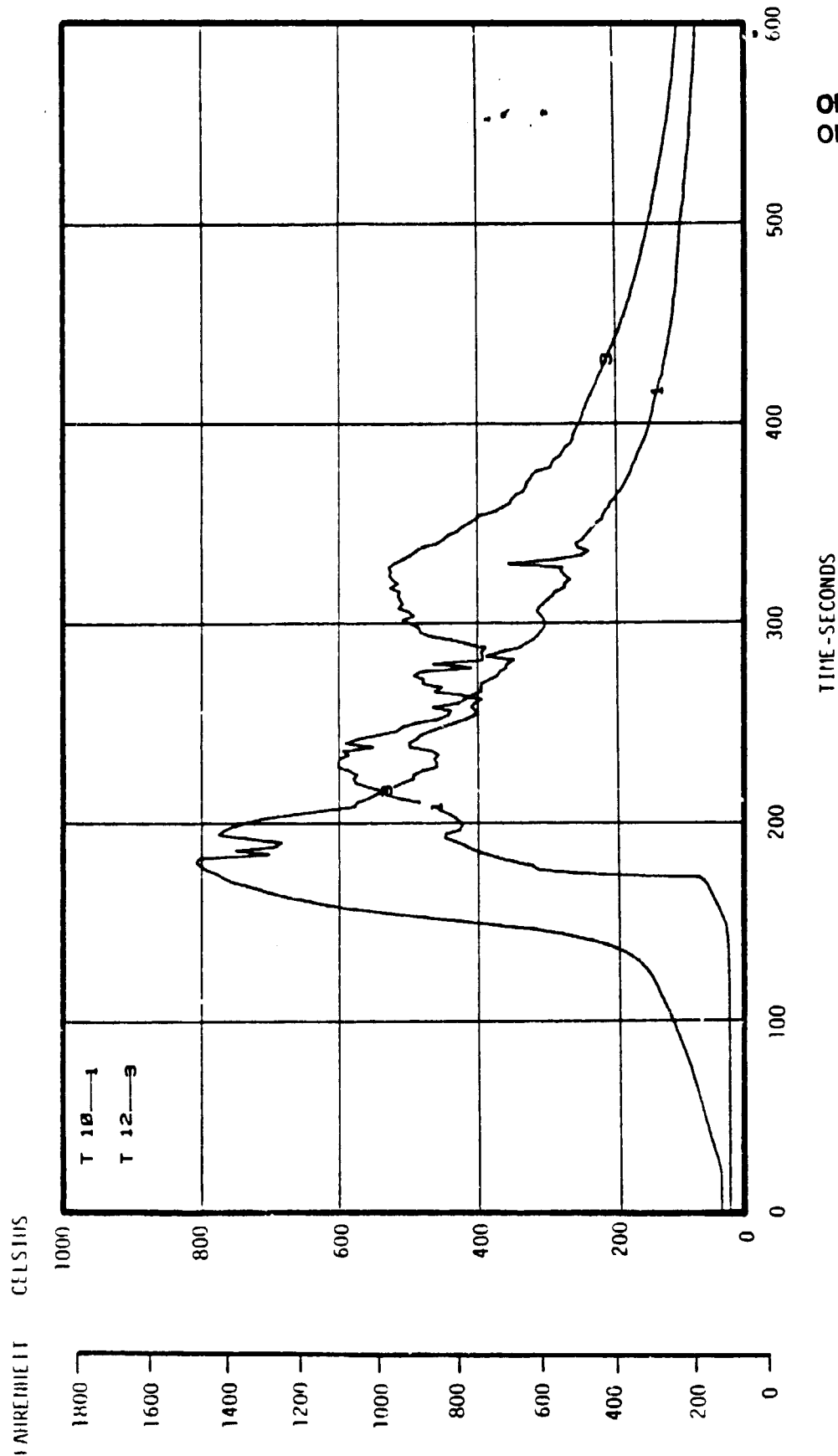


DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/18/82 11.11

NASA-AIES FULL SCALE CUSHION BURN TEST NUMBER 17

CUSHION CONSTRUCTION NUMBER 1.0

SEAT CUSHION TEMPERATURES



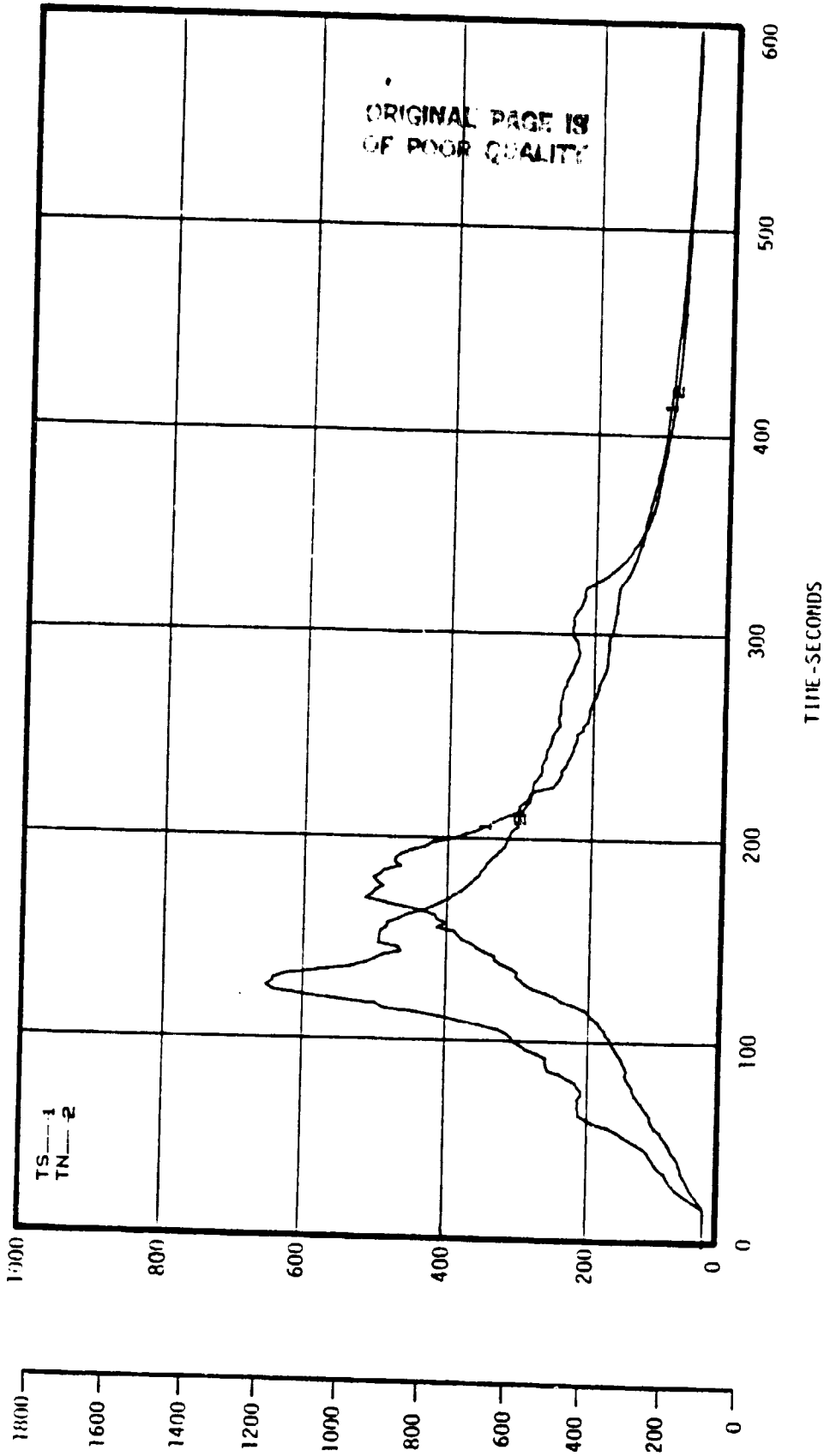
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CUSHION CONSTRUCTION NUMBER 1.0

CEILING TEMPERATURE

CELSIUS

FAHRENHEIT



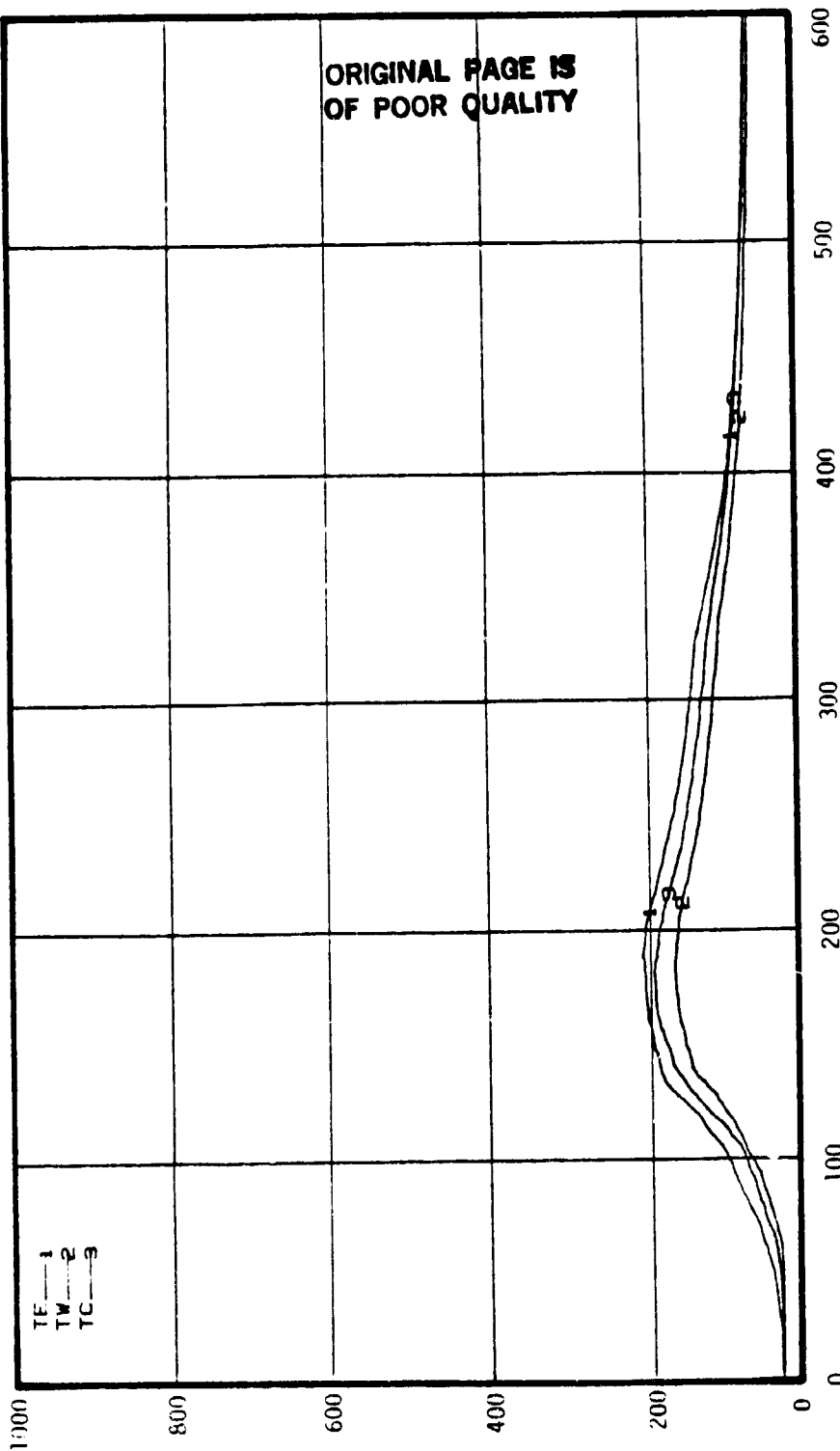
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NASA-MES FULL SCALE CUSHION BURN TEST NUMBER 17
CUSHION CONSTRUCTION NUMBER 1.0

CEILING TEMPERATURE

FAHRENHEIT CELSIUS

1800
1600
1400
1200
1000
800
600
400
200
0

TE 1
TW 2
TC 3

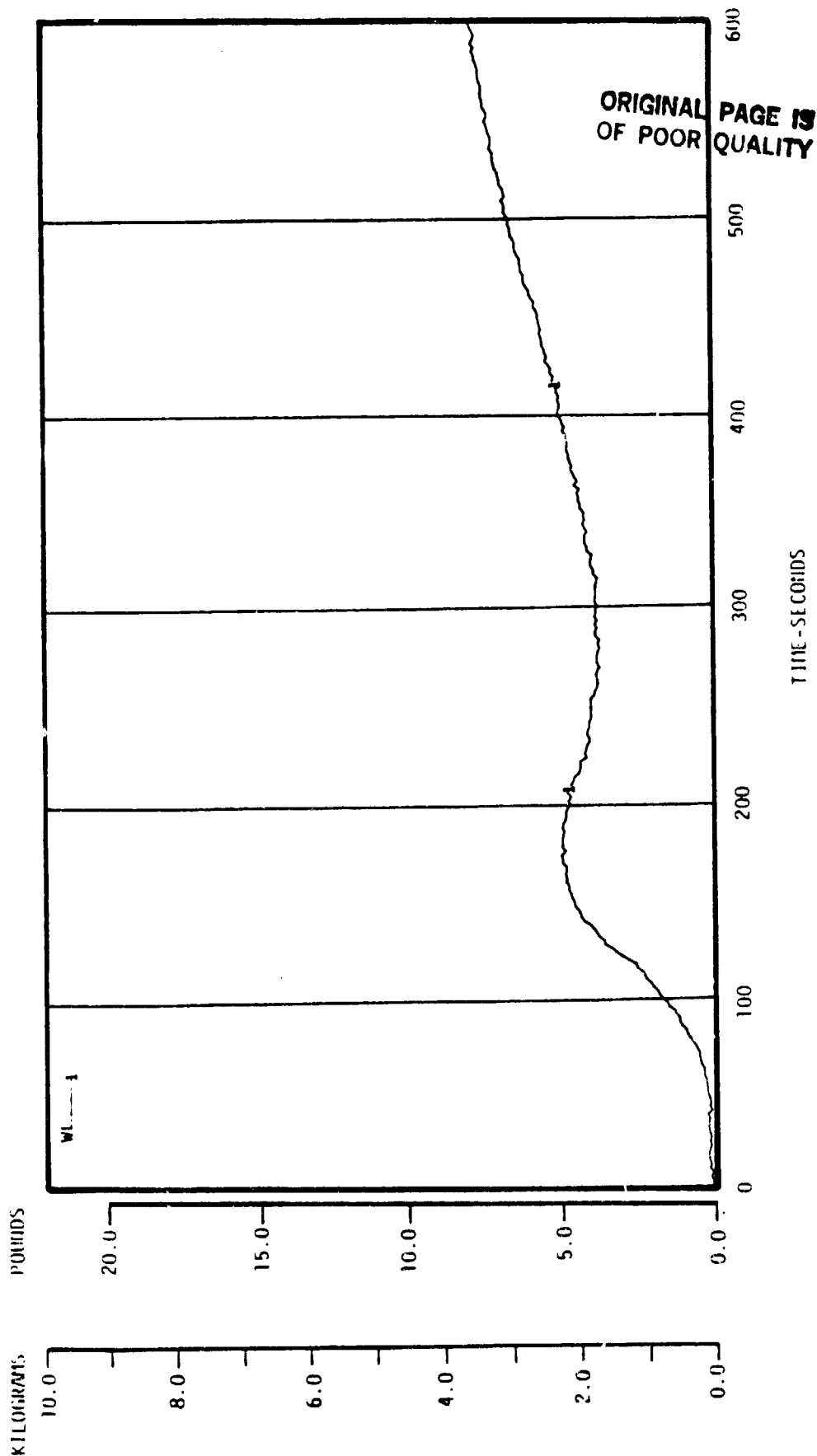


DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/16/82 11.11

NASA-AMES FULL SCALE CUSHION BURN TEST NUMBER 17

CUSHION CONSTRUCTION NUMBER 1.0

WEIGHT LOSS

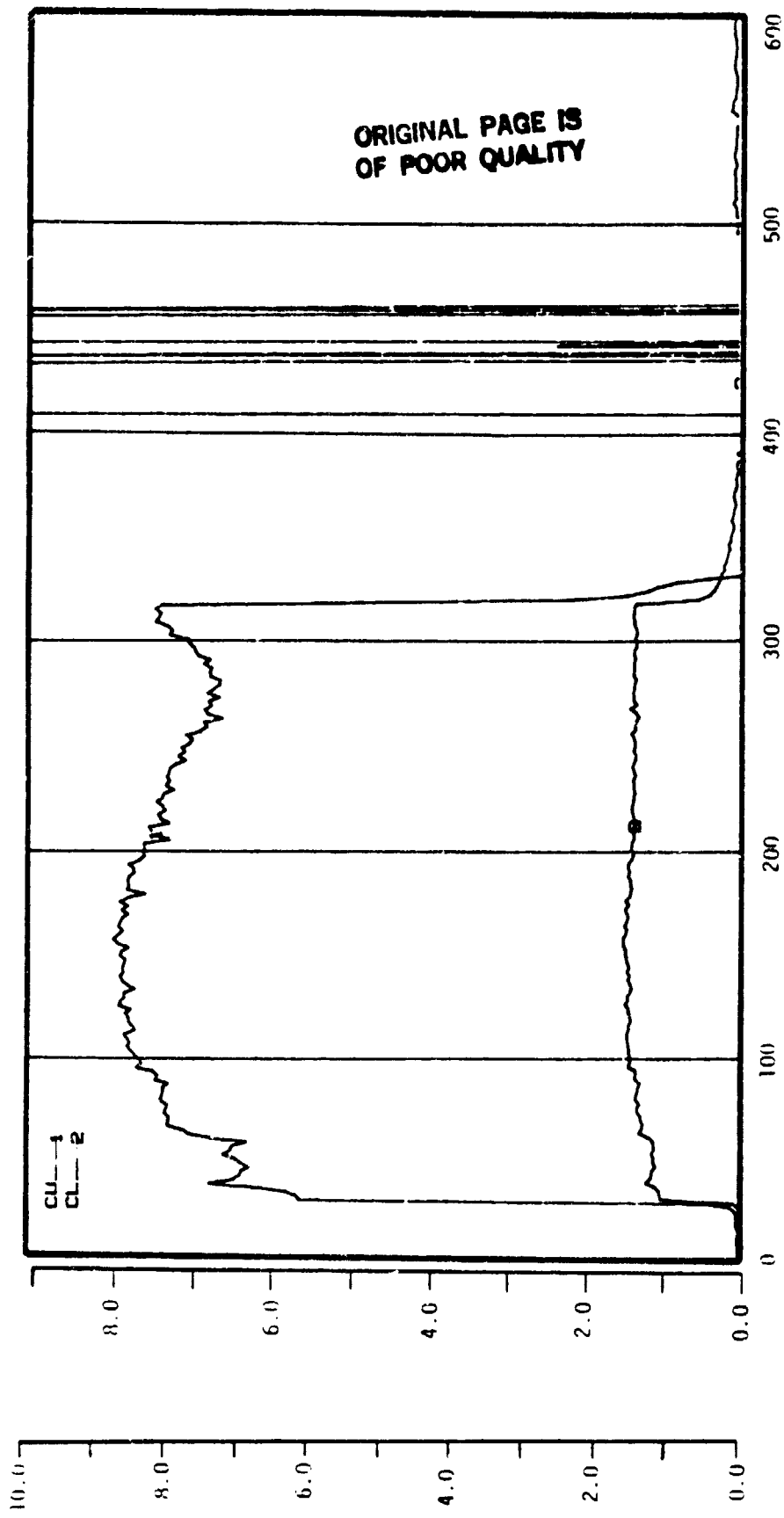


DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/16/82 11.11
 NASA-AMES FULL SCALE CUSHION BURST TEST NUMBER 17
 CUSHION CONSTRUCTION NUMBER 1.0

HEAT FLUX

BTU/SQ.FT.SEC.

WATT/50 CM.



TIME - SECONDS

DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/05/02 13.43

NASA-WES FULL SCALE CUSHION BURN TEST NUMBER 2

CUSHION CONSTRUCTION NUMBER 2.0

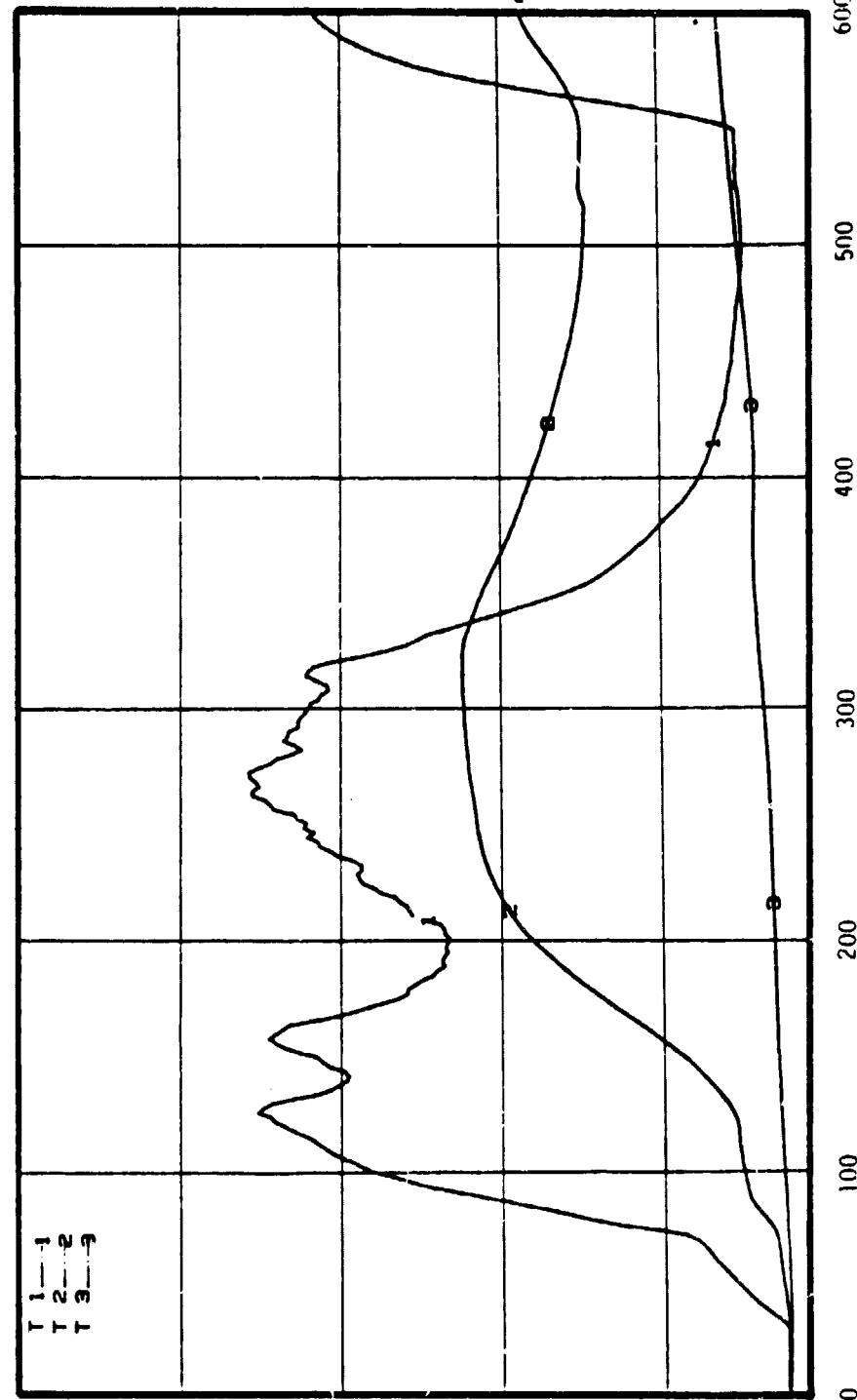
SEAT CUSHION TEMPERATURES

FAHRENHEIT CELSIUS

T 1—1
T 2—2
T 3—3

1800
1600
1400
1200
1000
800
600
400
200
0

1000
800
600
400
200
0



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TIME-SECONDS

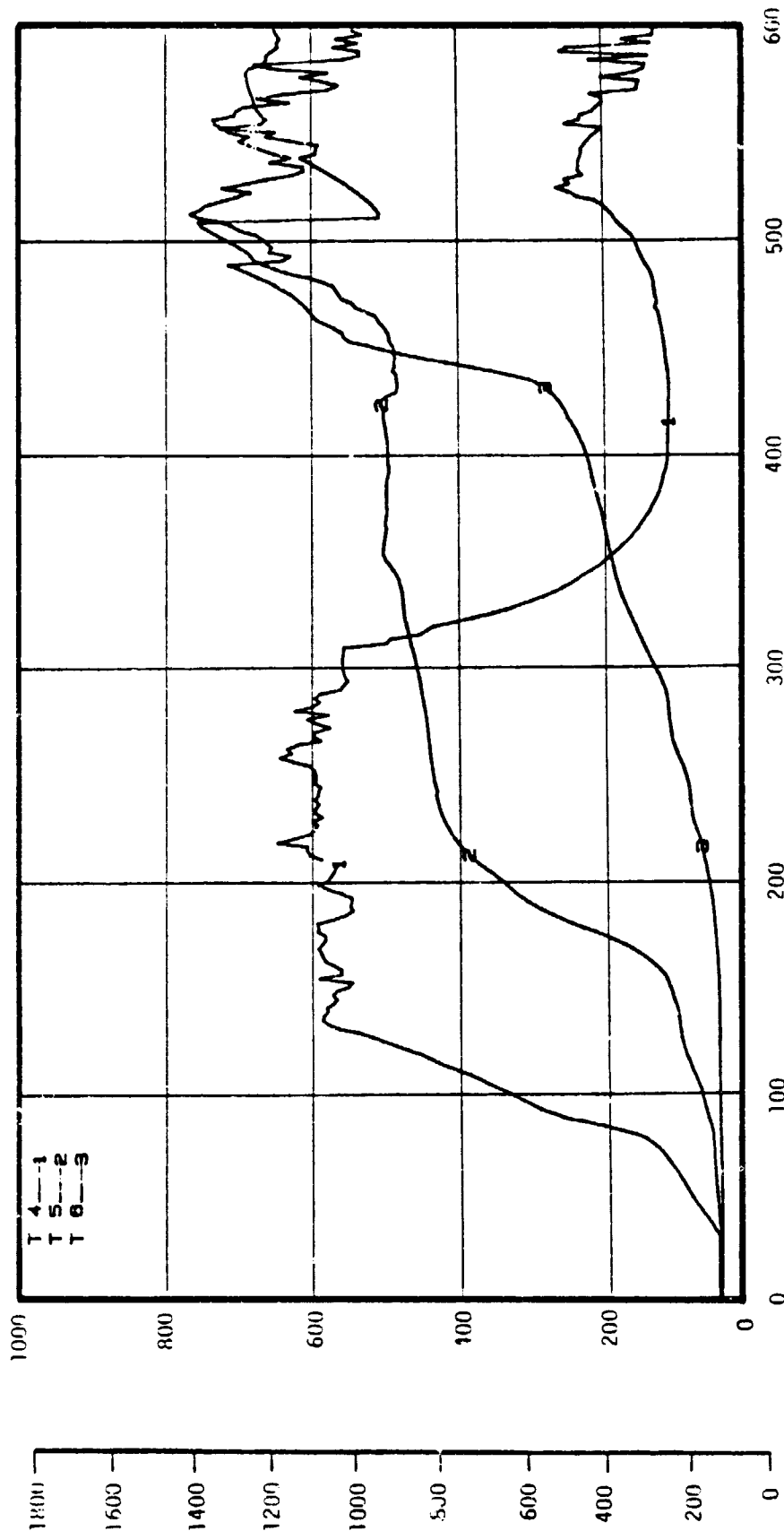
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NASA-MES FULL SCALE CUSHION BURN TEST NUMBER 2

CUSHION CONSTRUCTION NUMBER 2.0

SEAT CUSHION TEMPERATURES

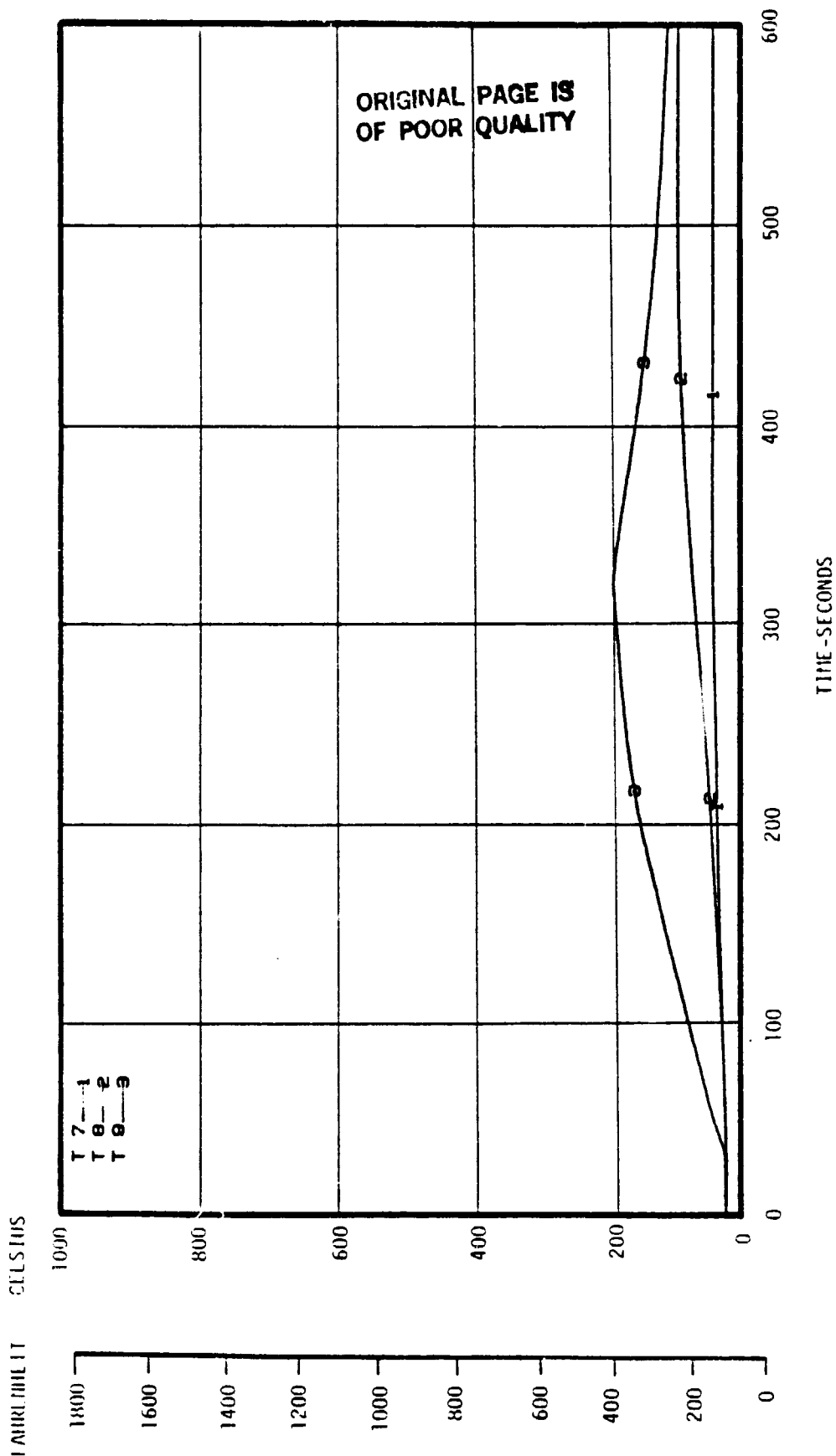
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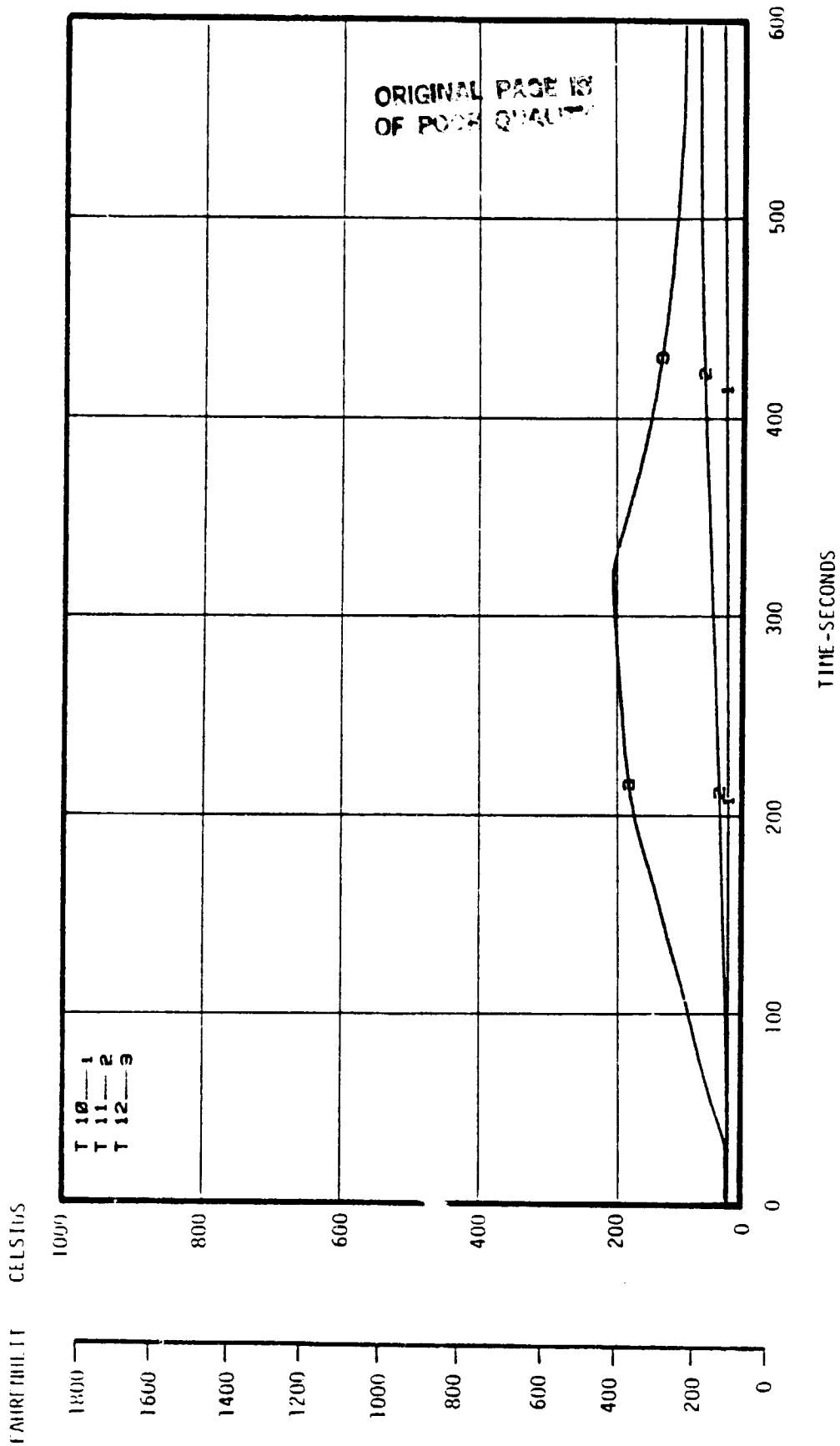
DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/05/82 13.43
NASA-AMES FULL SCALE CUSHION BURN TEST NUMBER 2
CUSHION CONSTRUCTION NUMBER 2.0

SLAT CUSHION TEMPERATURES



DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/05/02 13.43
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 CUSHION CONSTRUCTION NUMBER 2.0

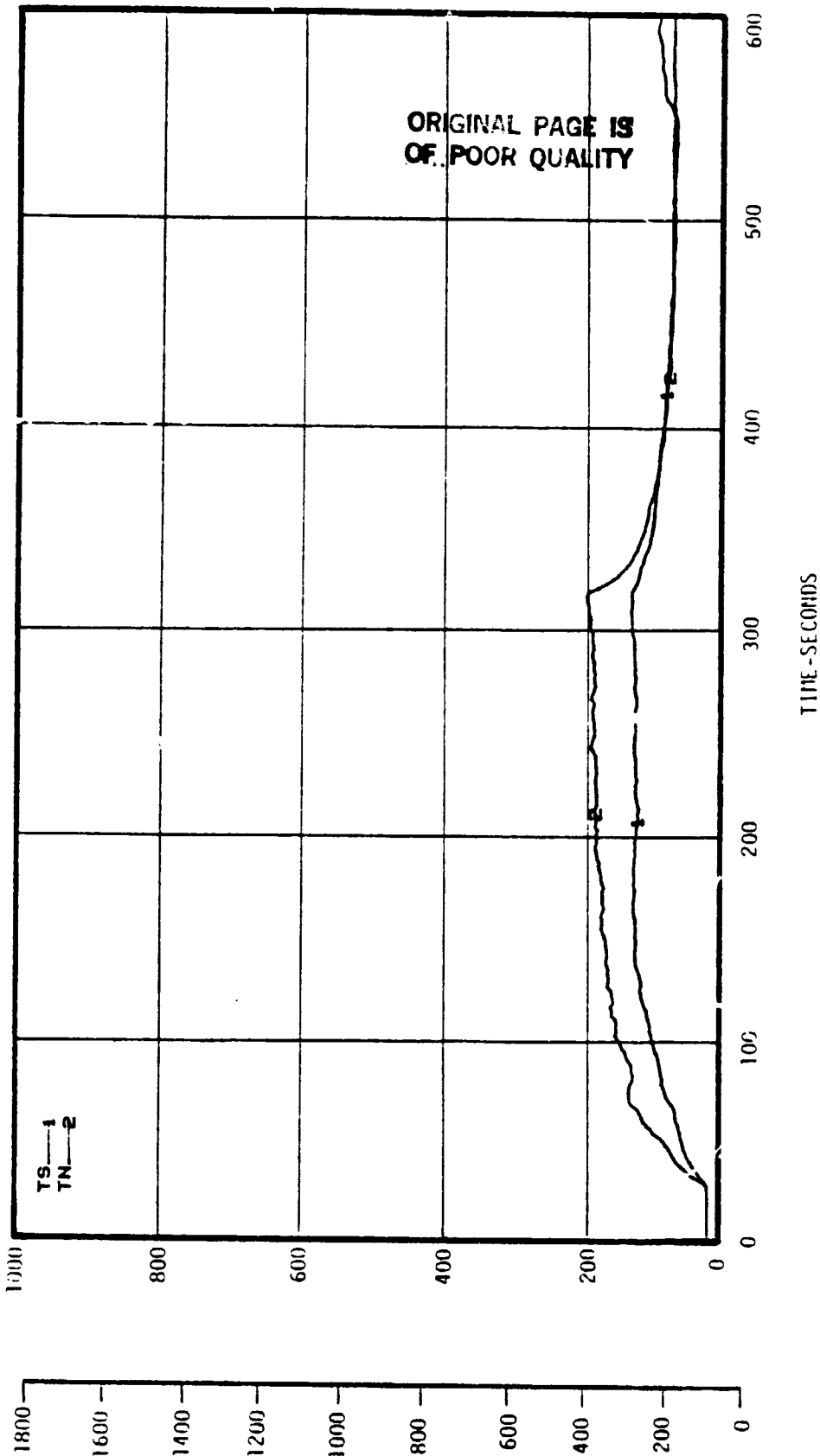
SEAT CUSHION TEMPERATURES



DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 83/85/82 13.49
 NASA-AMES FULL SCALE CUSHION BURN TEST NUMBER 2
 CUSHION CONSTRUCTION NUMBER 2.8

CEILING TEMPERATURE

FAHRENHEIT CELSIUS

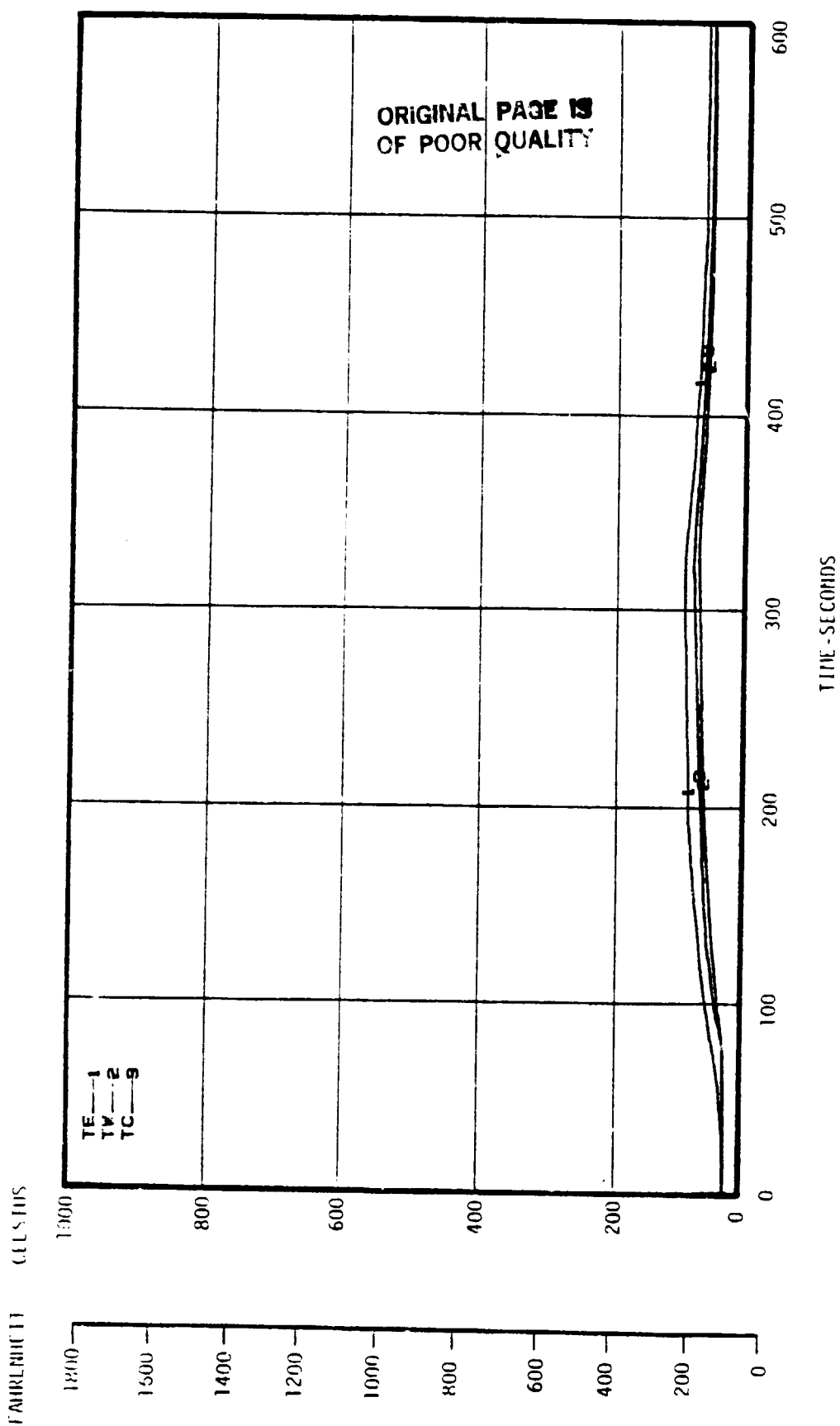


DOUGLAS AIRCRAFT CO. IN FIRE SIMULATOR 03/05/62 13.43

NASA-AIES FULL SCALE CUSHION BURN TEST NUMBER 2

CUSHION CONSTRUCTION NUMBER 2.0

CEILING TEMPERATURE

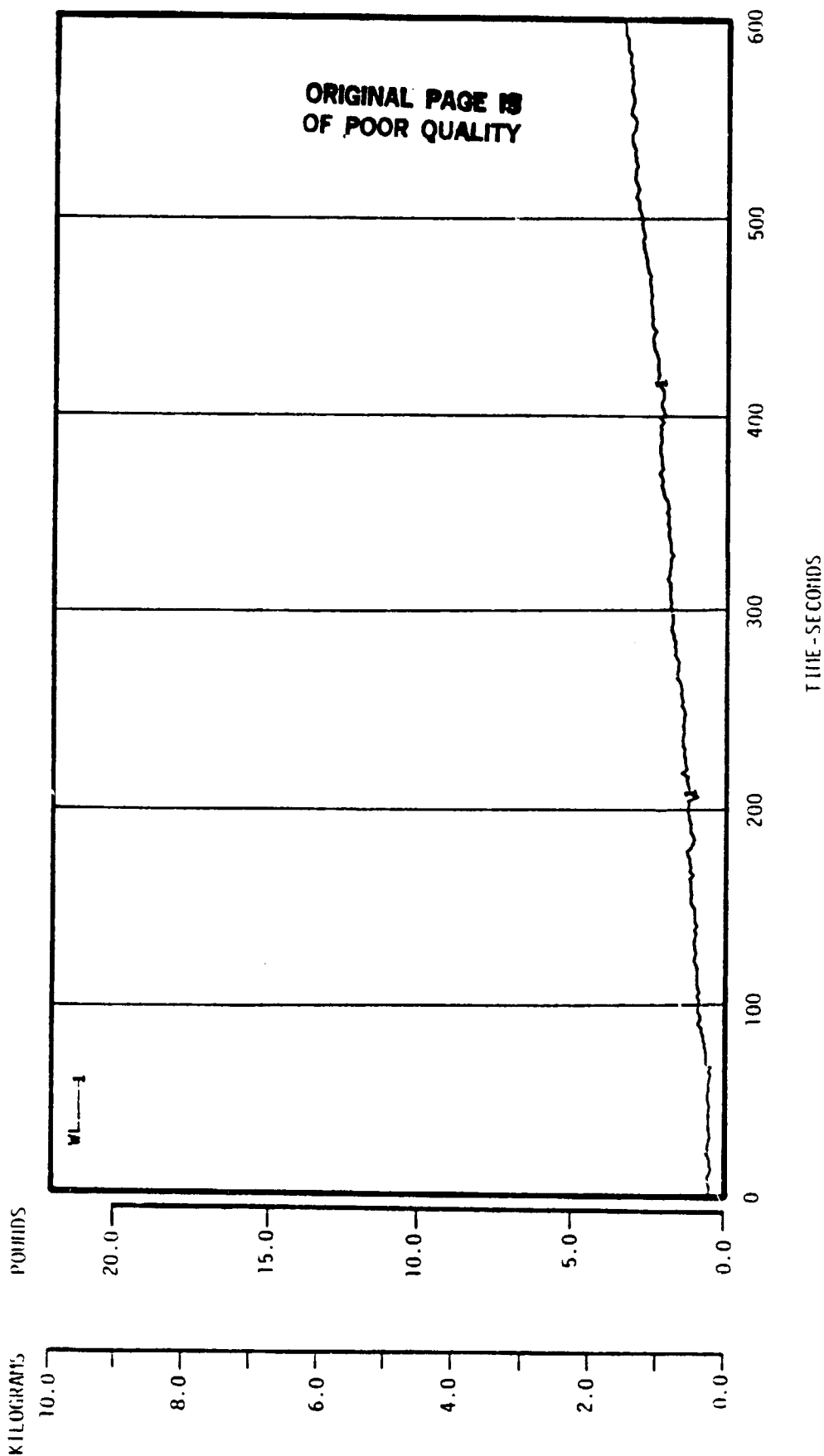


DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/05/82 13.43

HASA-NIES FULL SCALE CUSHION BURN TEST NUMBER 2

CUSHION CONSTRUCTION NUMBER 2.0

WEIGHT LOSS

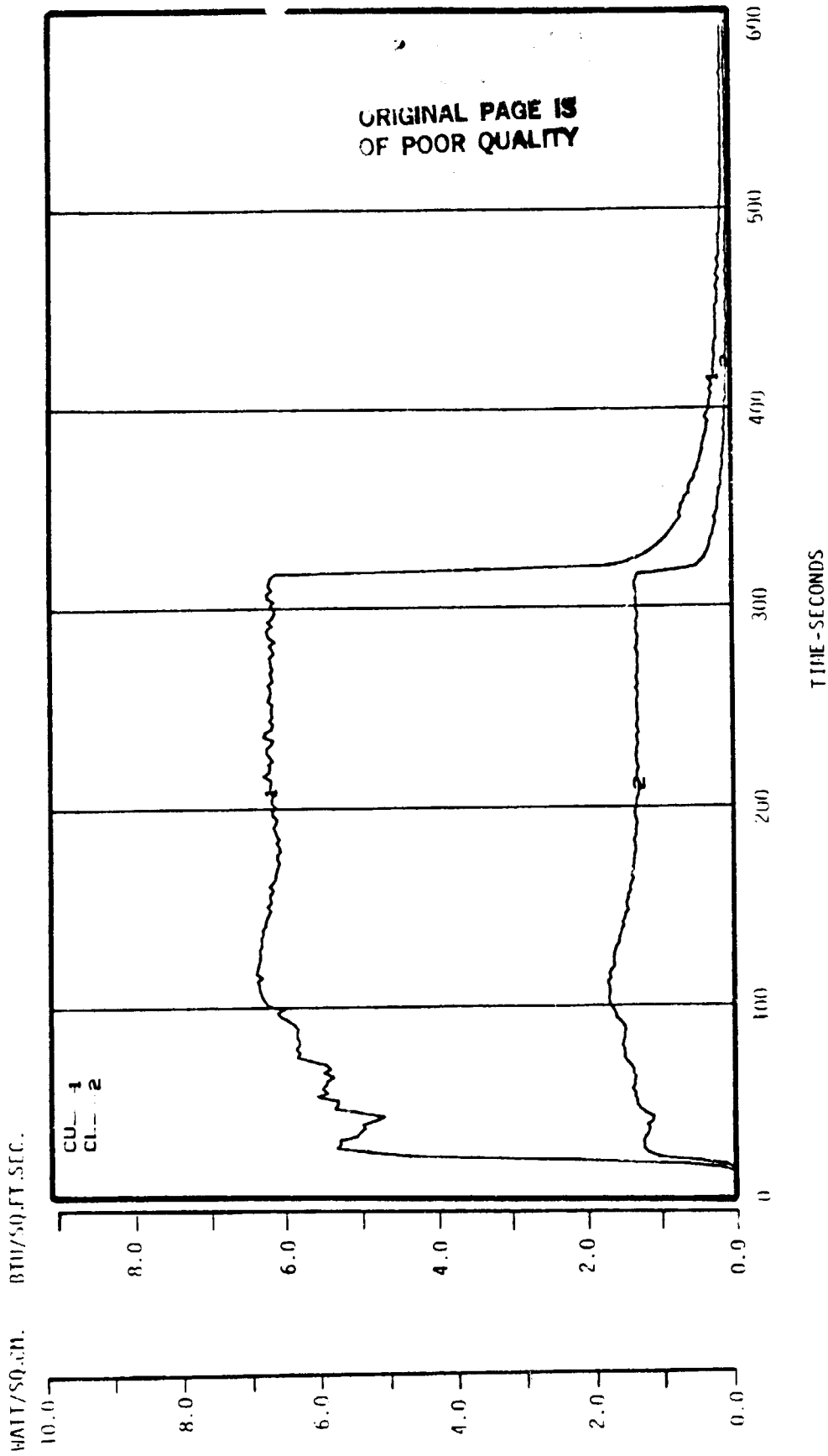


DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/05/82 19:43

NASA-AMES FULL SCALE CUSHION BURN TEST NUMBER 2

CUSHION CONSTRUCTION NUMBER 2.0

HEAT FLUX

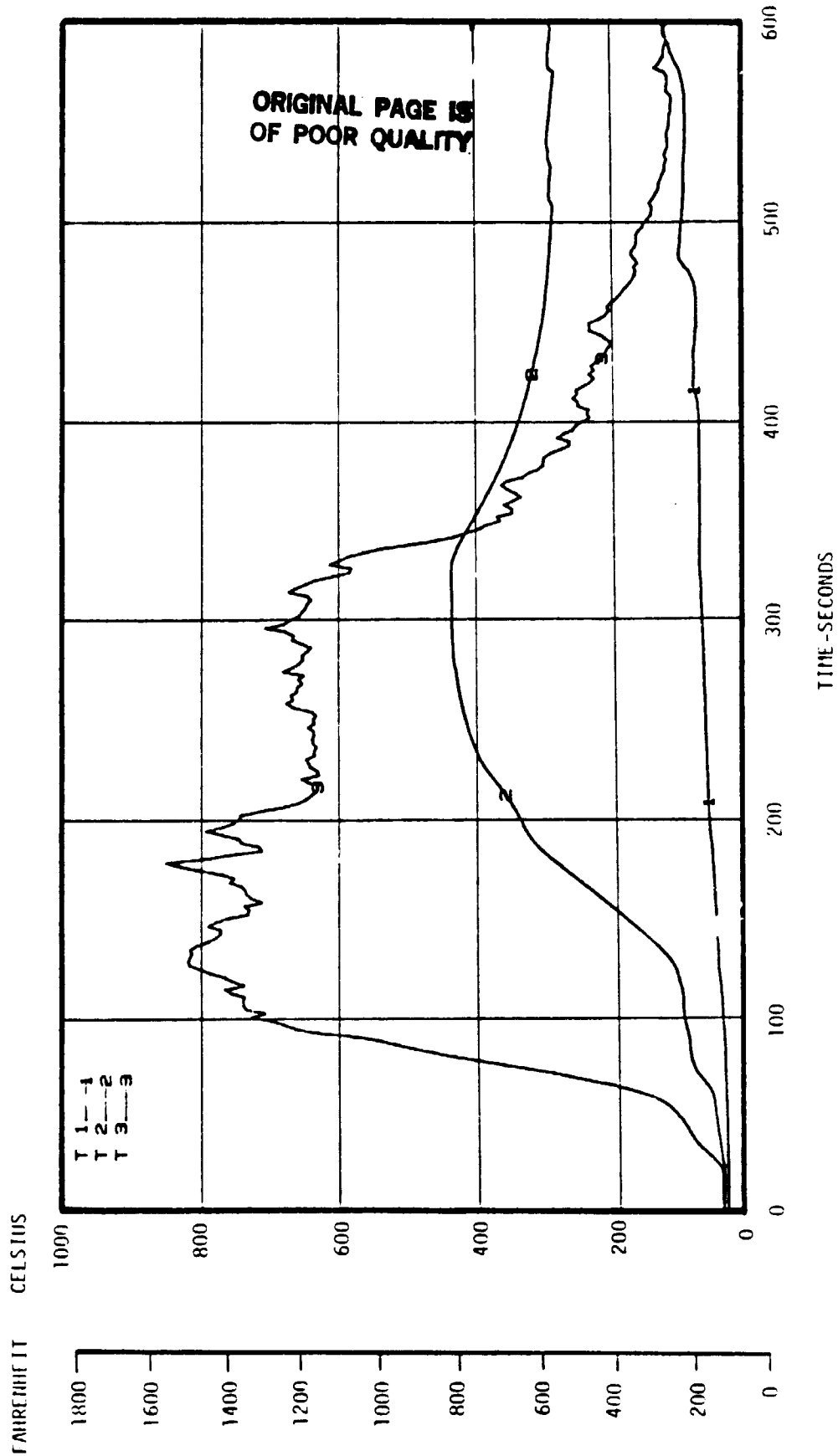


DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/00/82 10.01

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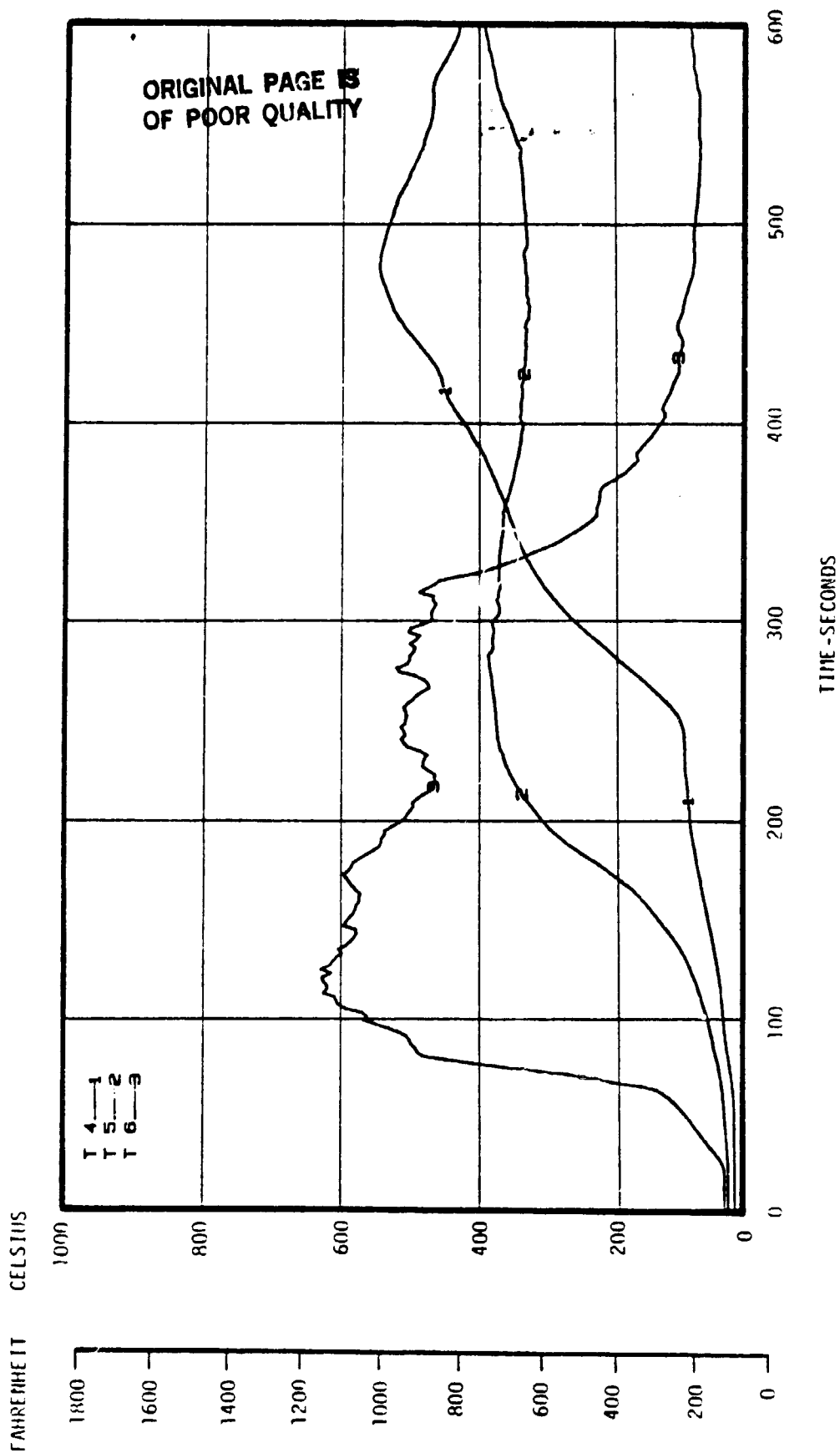
CUSHION CONSTRUCTION NUMBER 2.0

SEAT CUSHION TEMPERATURES



DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/09/82 10.01
 NASA-MES FULL SCALE CUSHION BURN TEST NUMBER 4
 CUSHION CONSTRUCTION NUMBER 2.0

SEAT CUSHION TEMPERATURES

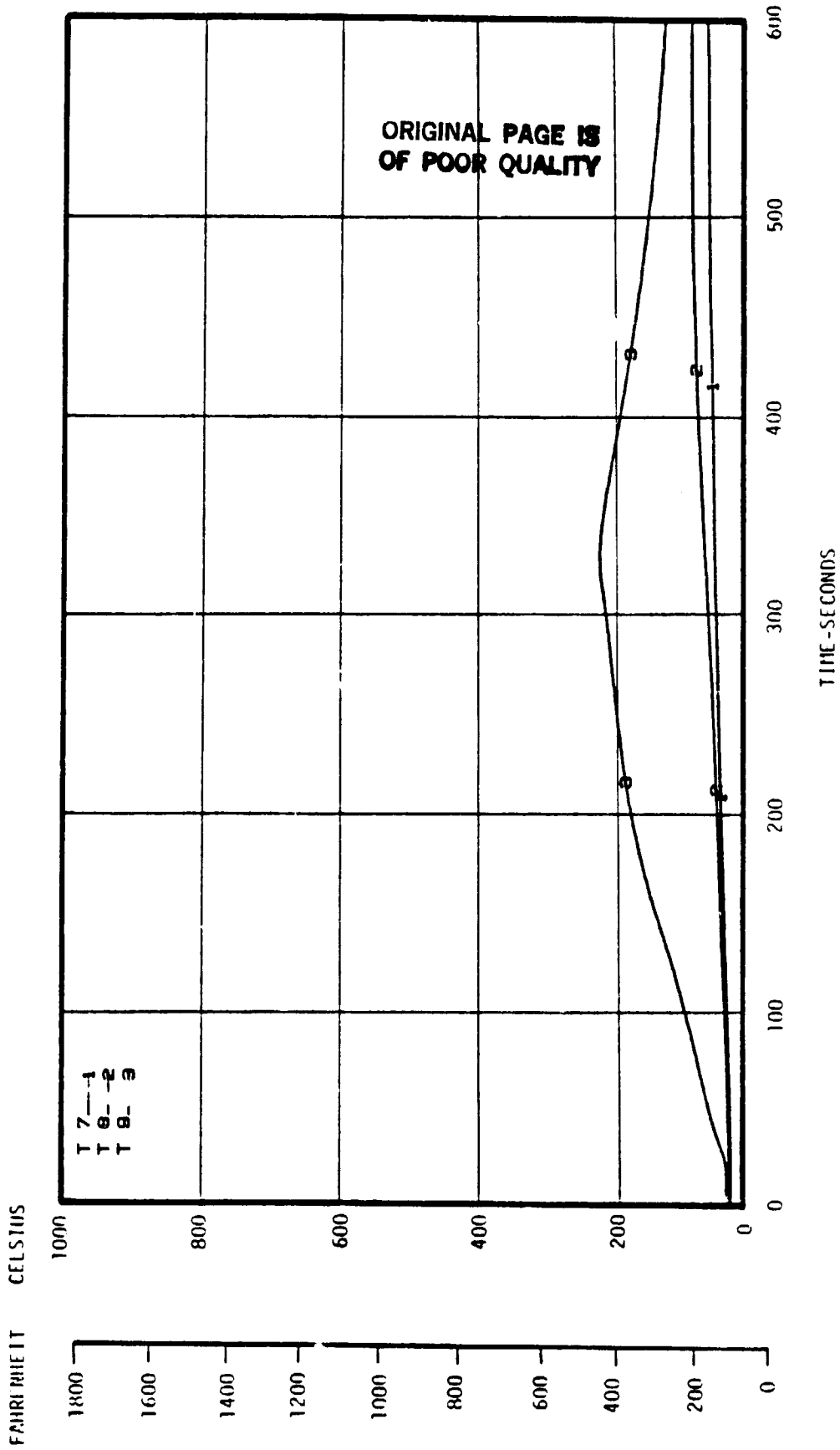


DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/08/82 10.01

NASA-AMES FULL SCALE CUSHION BURN TEST NUMBER 4

CUSHION CONSTRUCTION NUMBER 2.0

SEAT CUSHION TEMPERATURES

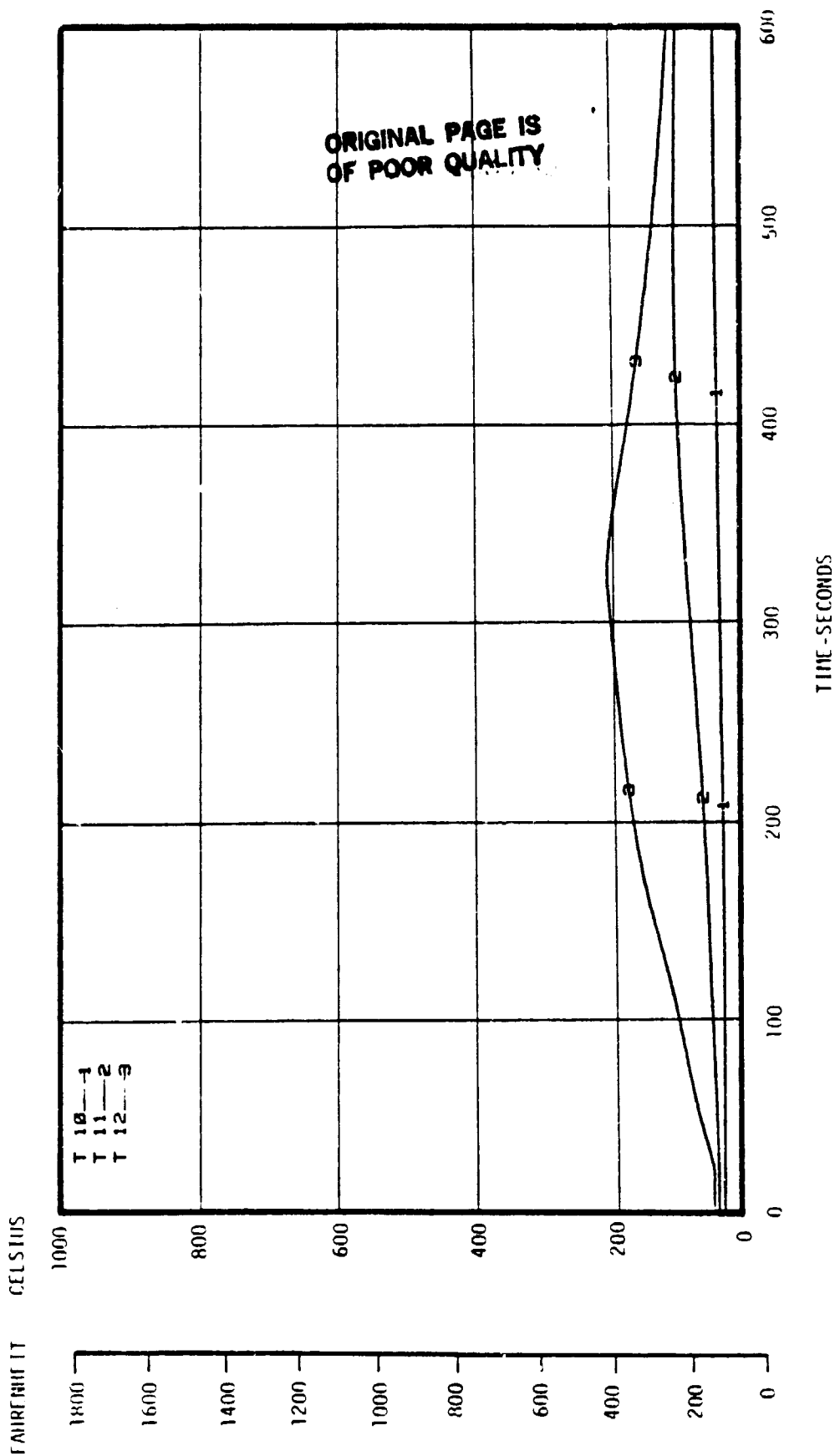


DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/08/82 10.01

NASA-AMES FULL SCALE CUSHION BURN TEST NUMBER 4

CUSHION CONSTRUCTION NUMBER 2.0

SEAT CUSHION TEMPERATURES



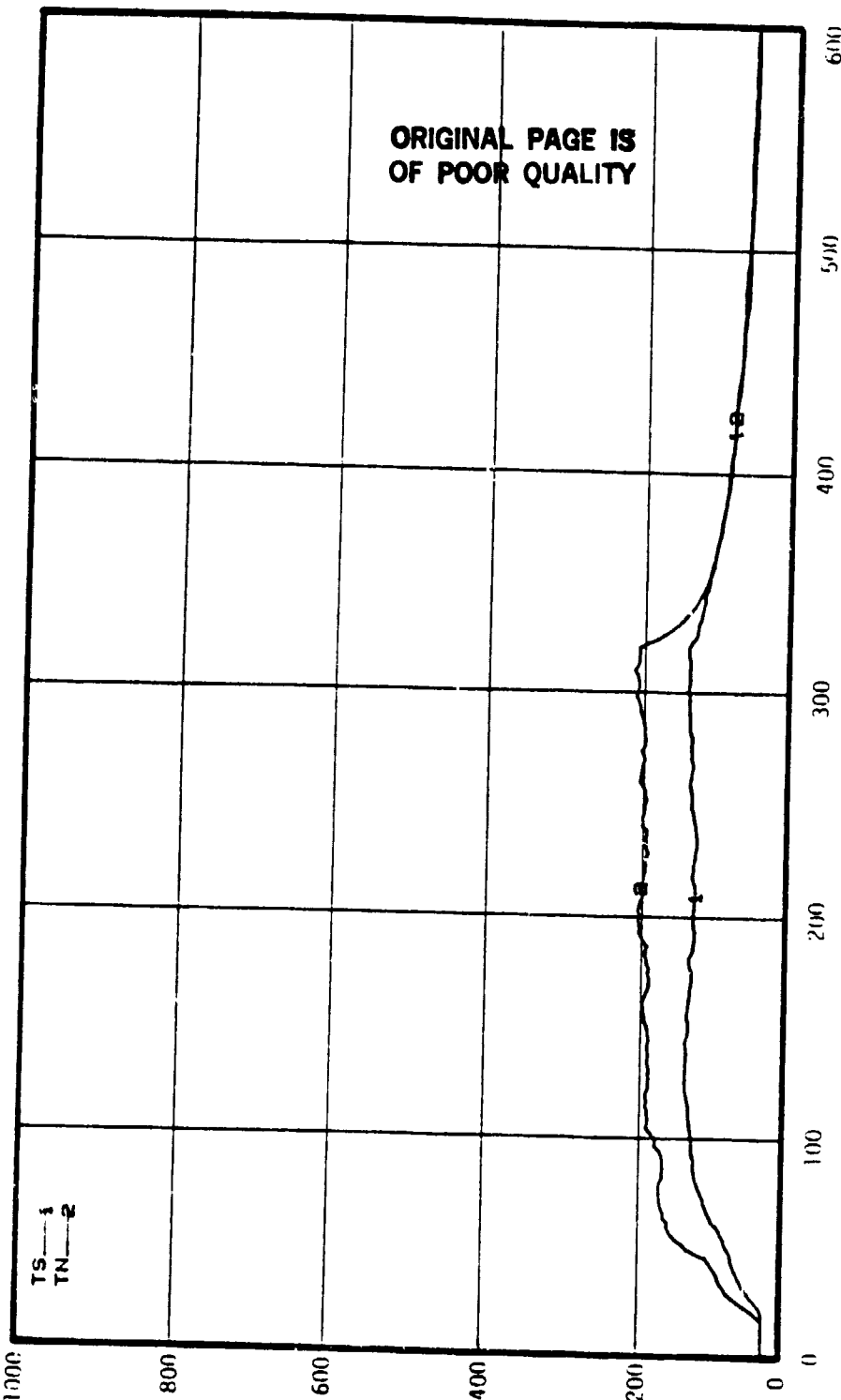
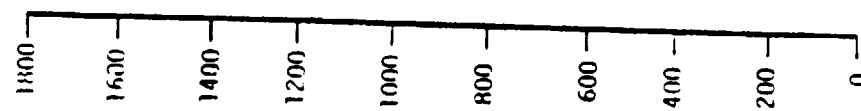
DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/00/82 10.01

NASA-WES FULL SCALE CUSHION BURP TEST NUMBER 4

CUSHION CONSTRUCTION NUMBER 2.0

CEILING TEMPERATURE

FAHRENHEIT CELSIUS



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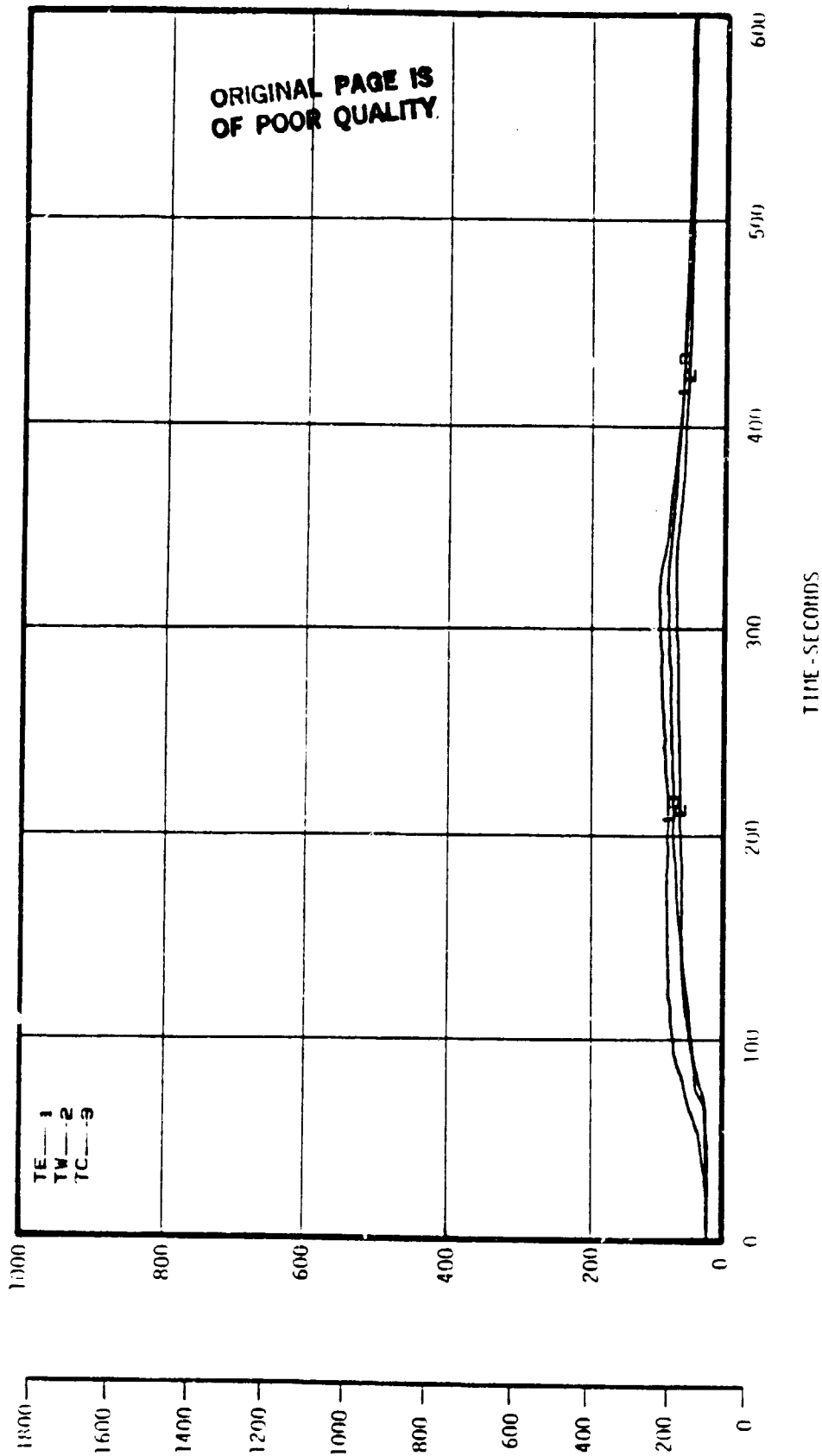
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CUSHION CONSTRUCTION NUMBER 2.0

CEILING TEMPERATURE

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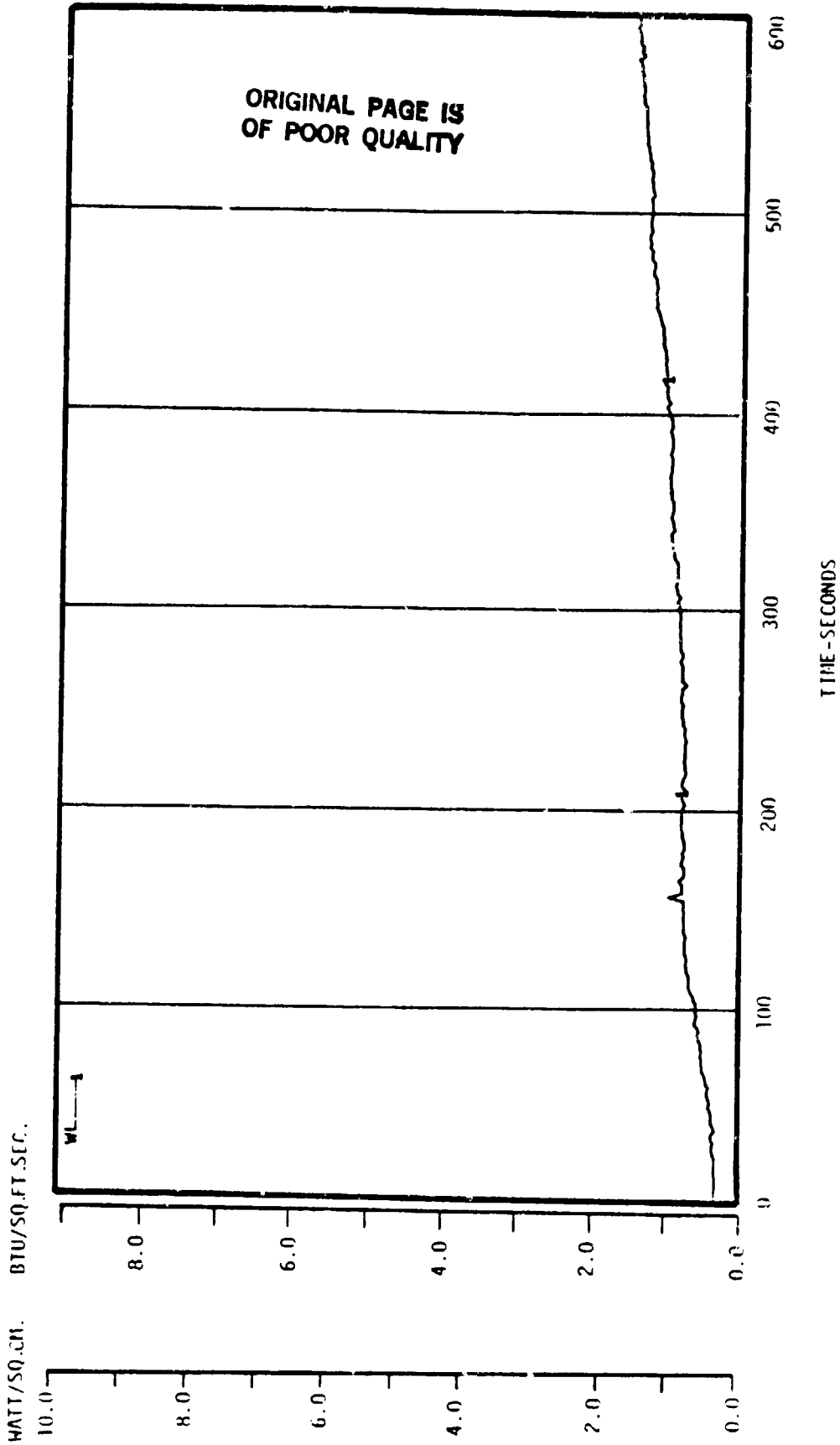


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CUSHION CONSTRUCTION NUMBER 2.0

HEAT FLUX

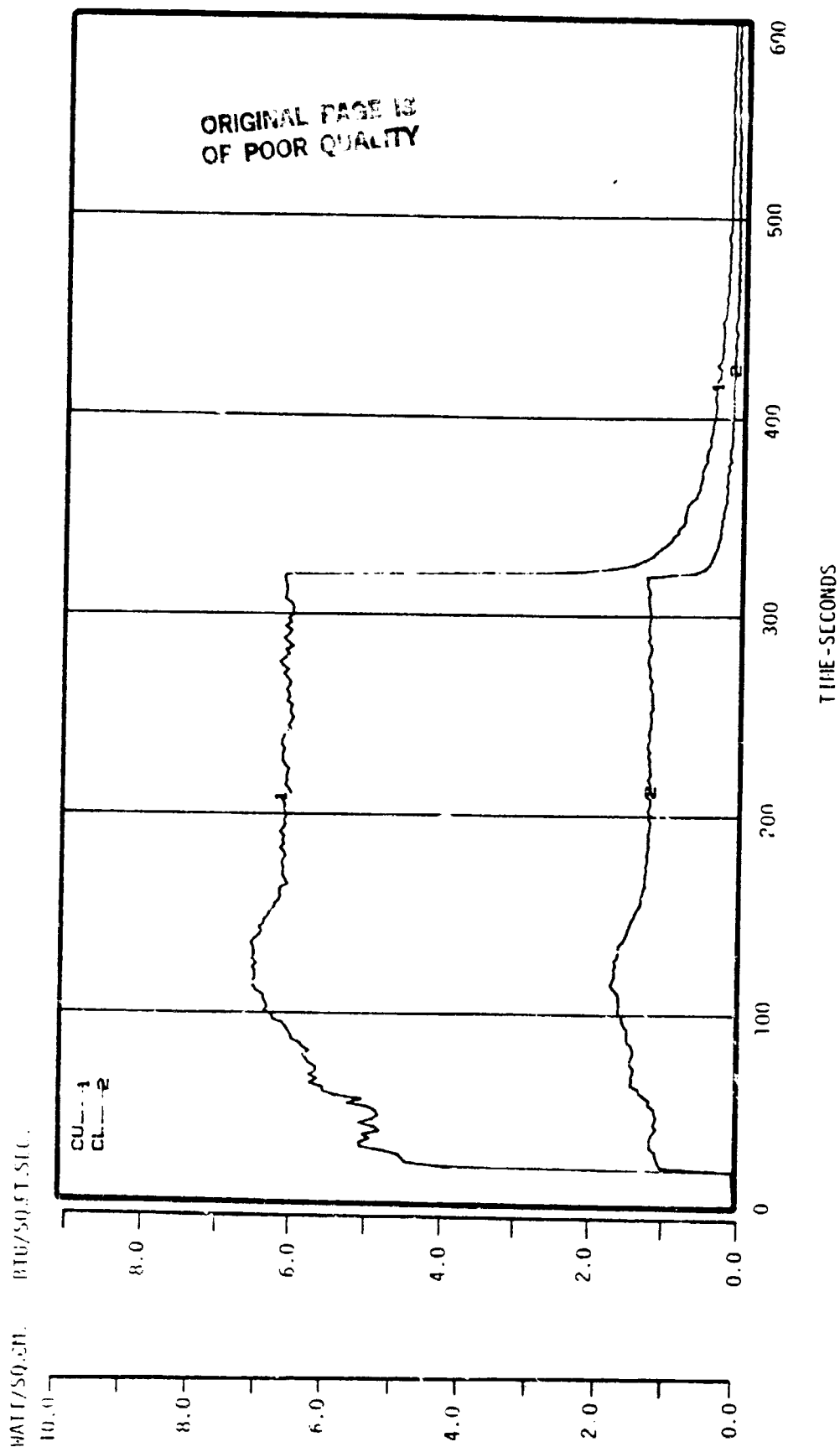


DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/02/82 10.01

NASA-MF'S FULL SCALE CUSHION BURN TEST NUMBER 4

CUSHION CONSTRUCTION NUMBER 2.0

HEAT FLUX



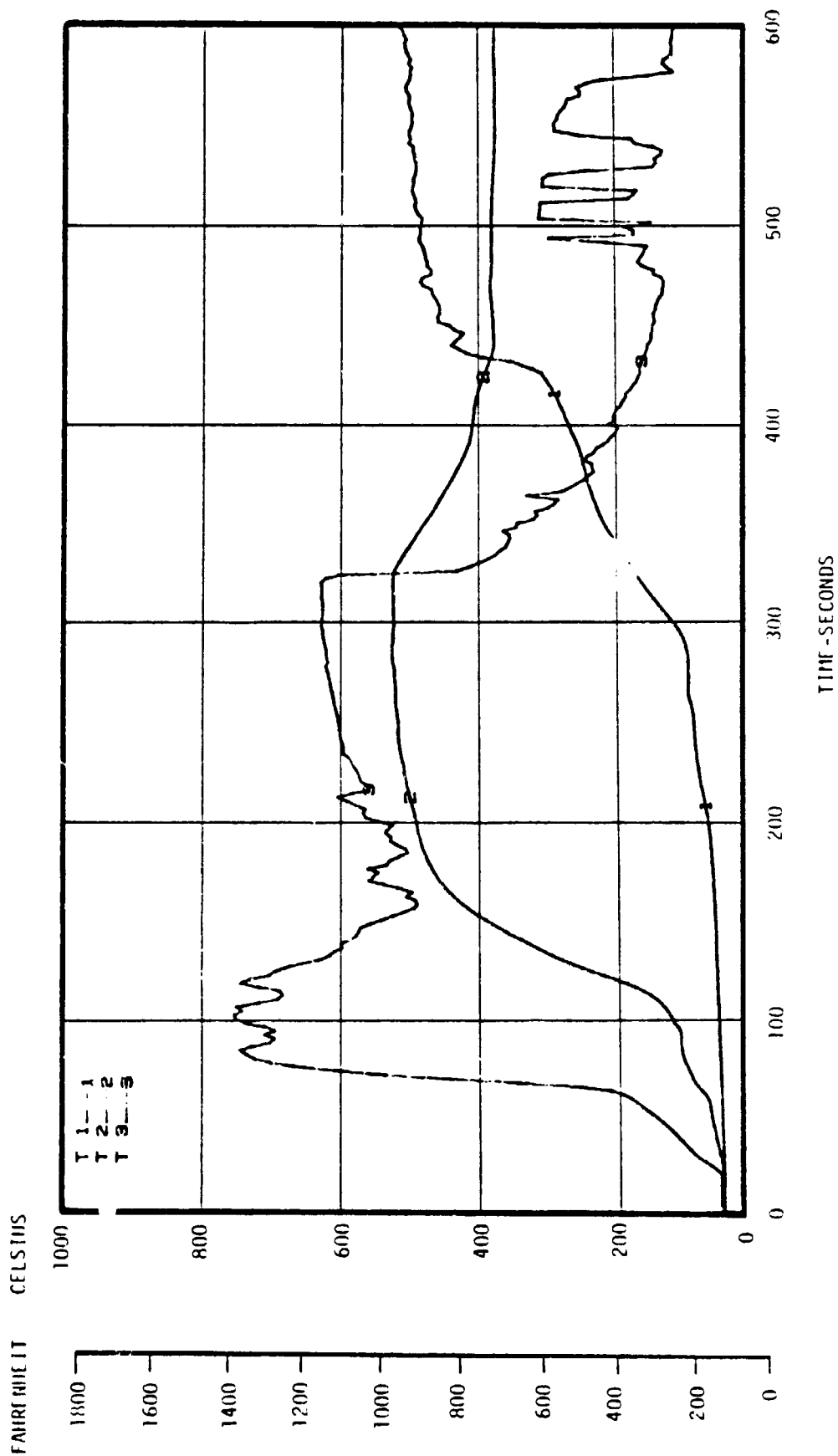
DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/12/82 10.54

NASA-NIES FULL SCALE CUSHION BURN TEST NUMBER 11

CUSHION CONSTRUCTION NUMBER 3.0

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SEAT CUSHION TEMPERATURES



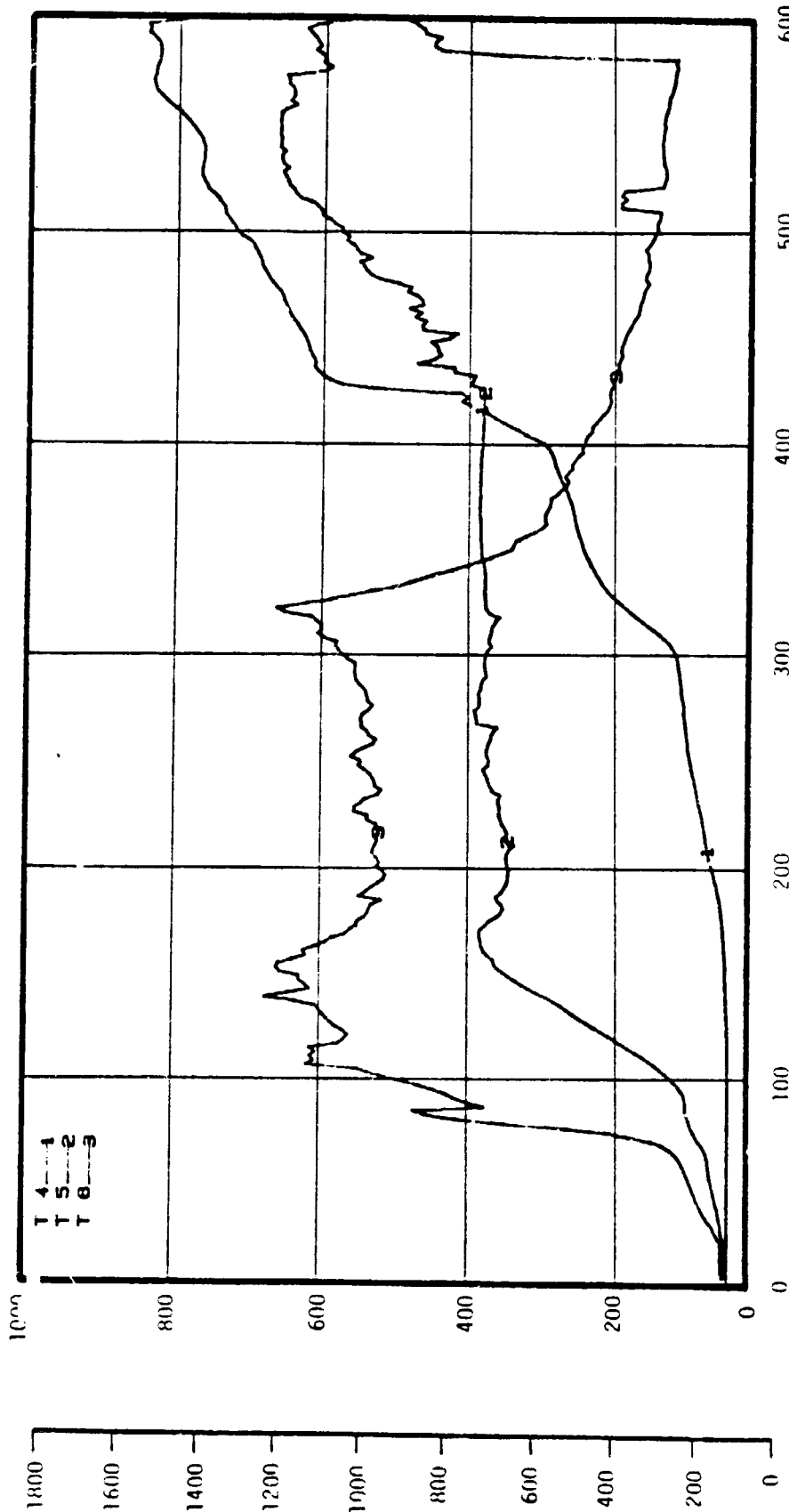
BOHIGAS AIRCRAFT CABIN FIRE SIMULATOR 09/12/82 10:54

NASA-AMES FULL SCALE CUSHION PURN TEST NUMBER 11

CUSHION CONSTRUCTION NUMBER 3.0

SEAT CUSHION TEMPERATURES

FAHRENHEIT CELSIUS



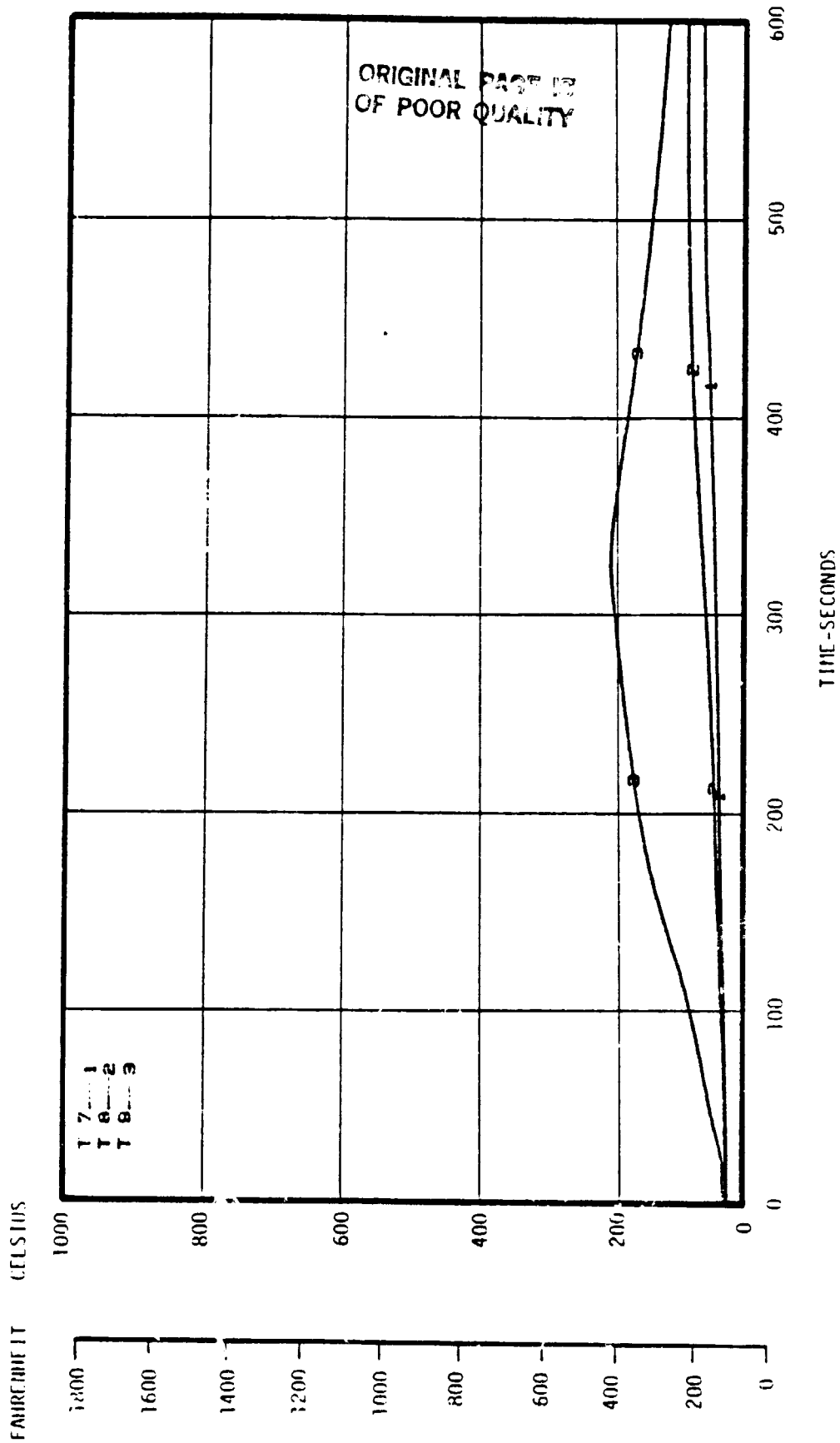
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DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 83/12/82 10.54

NASA-AMES FULL SCALE CUSHION BURN TEST NUMBER 11

CUSHION CONSTRUCTION NUMBER 3.0

SEAT CUSHION TEMPERATURES

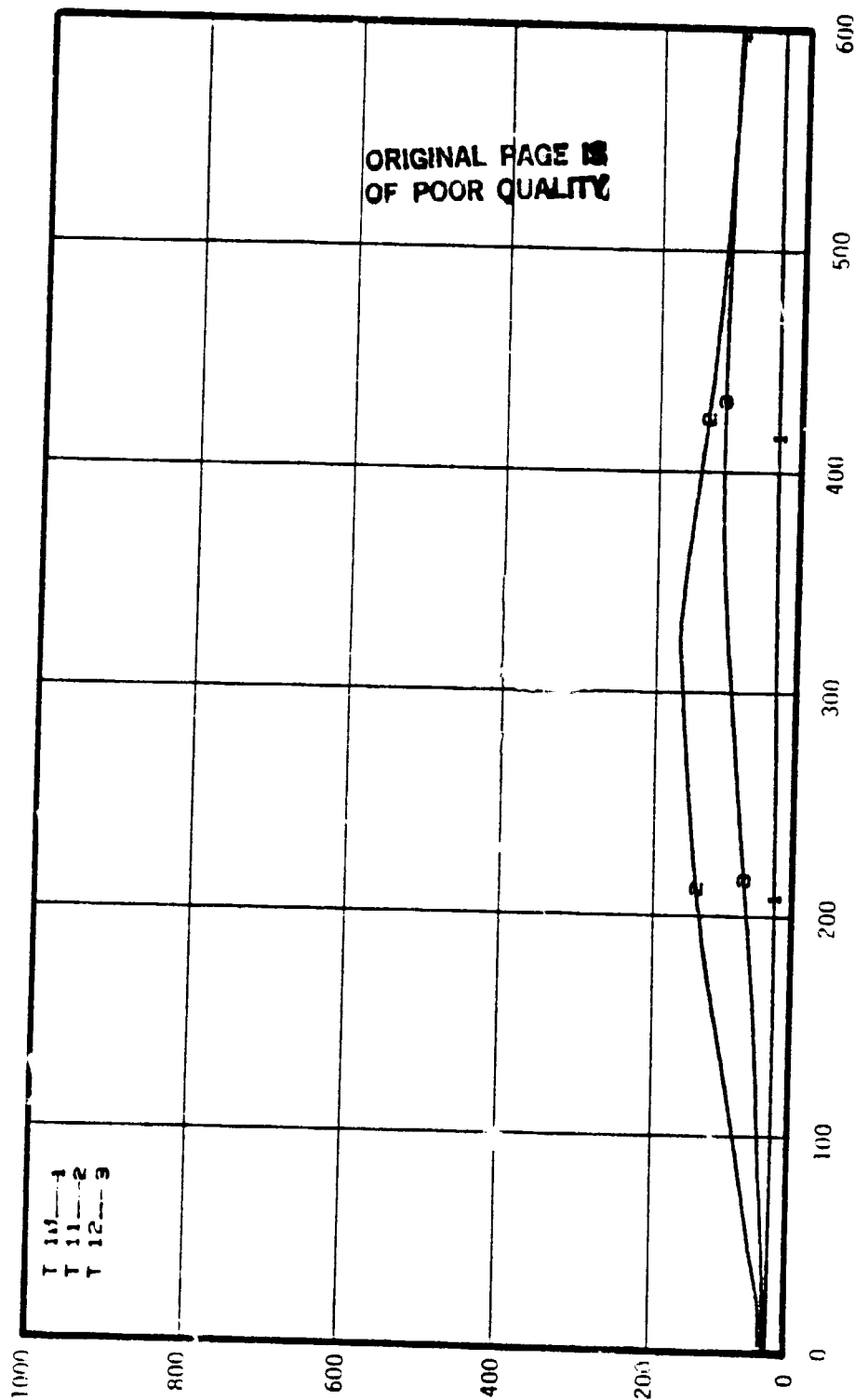
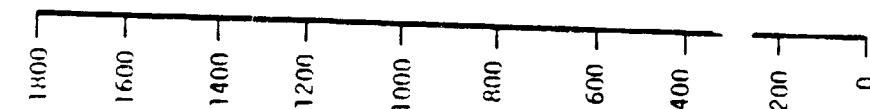


DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03.12.82 10.54
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 CUSHION CONSTRUCTION NUMBER 3.0

SEAT CUSHION TEMPERATURES

FAHRENHEIT

CELSIUS

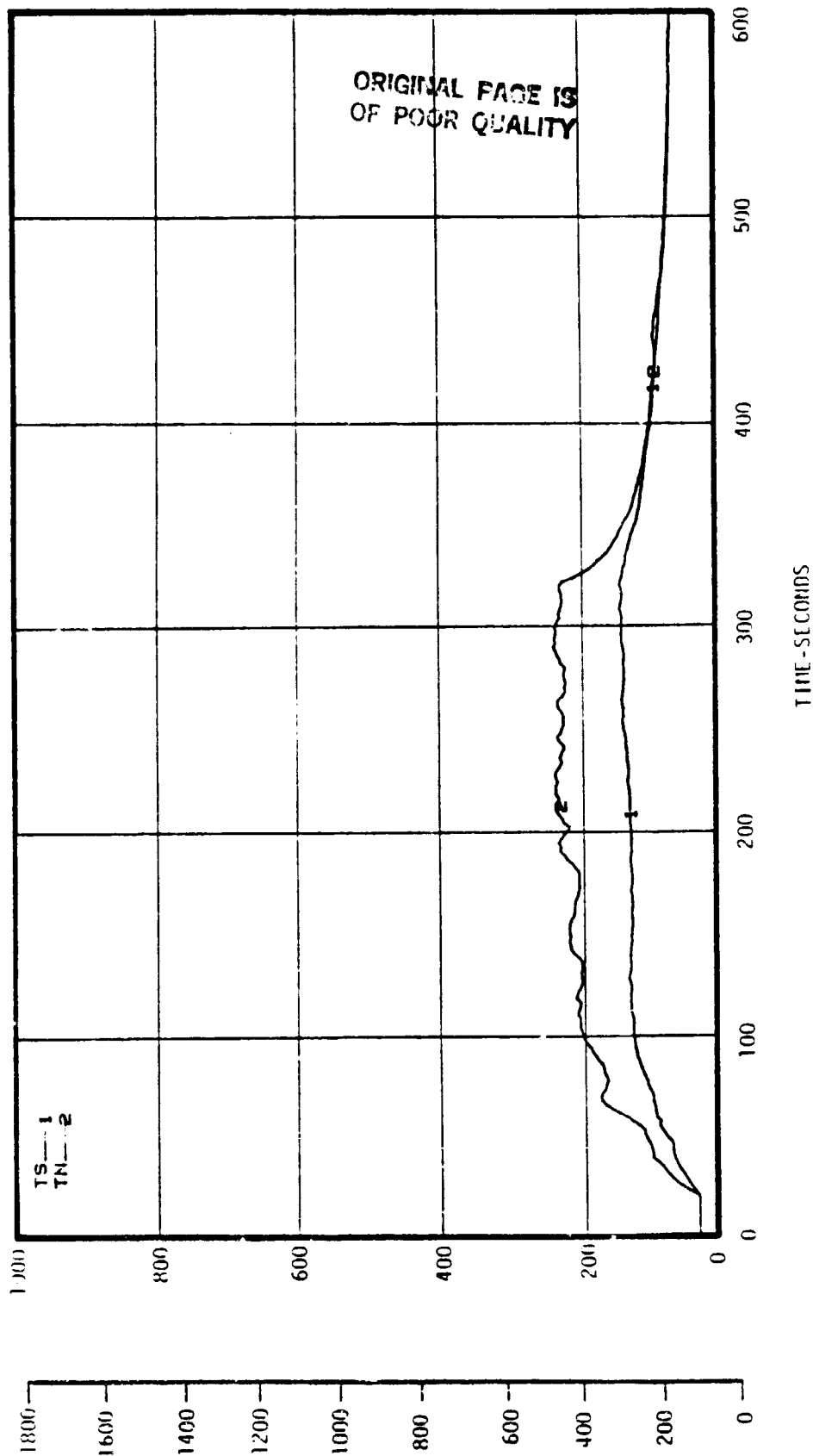


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DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/12/82 10.54
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CUSHION CONSTRUCTION NUMBER 3.0

CEILING TEMPERATURE

FAHRENHEIT CELSIUS



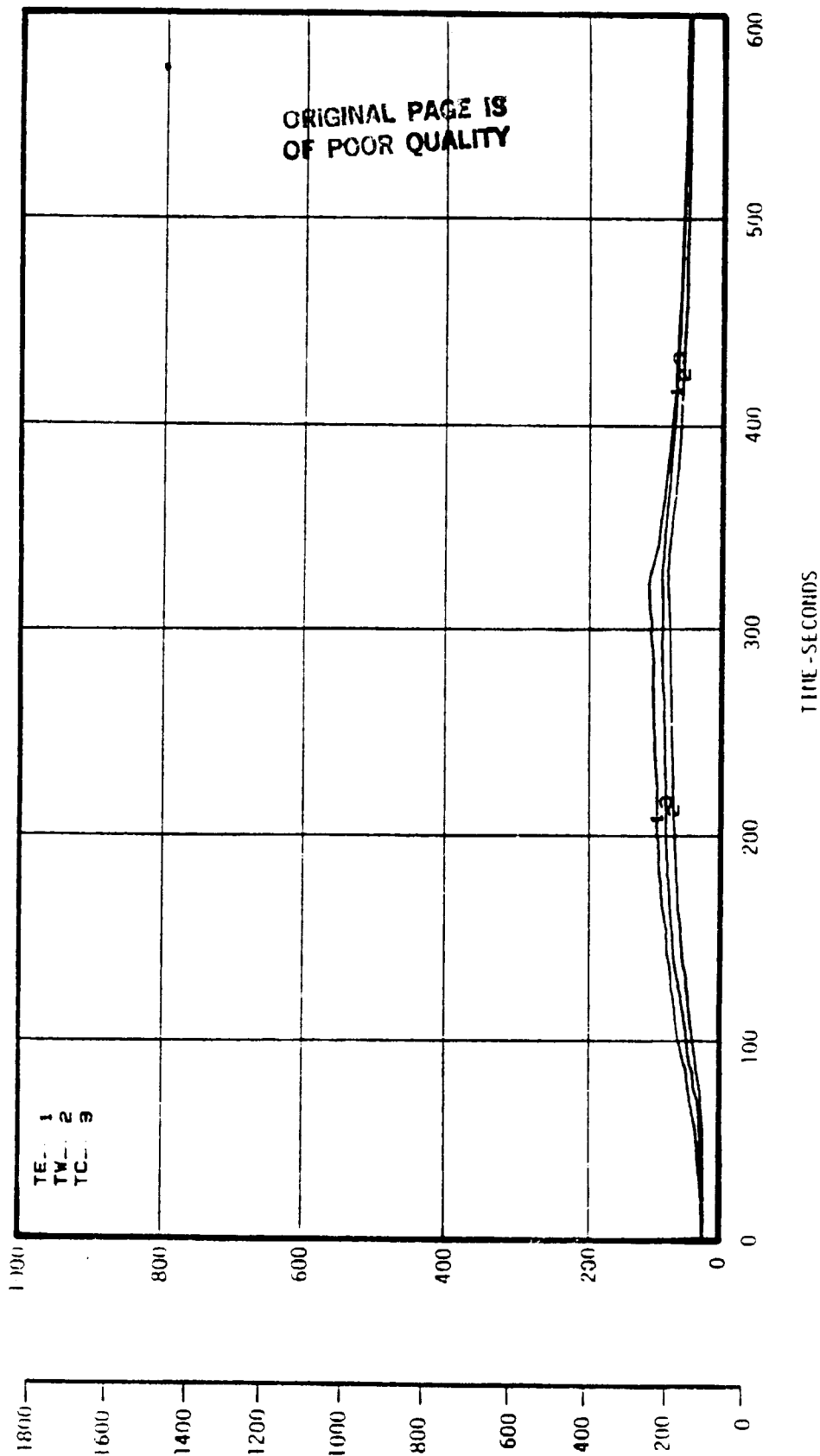
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CUSHION CONSTRUCTION NUMBER 3.0

CEILING TEMPERATURE

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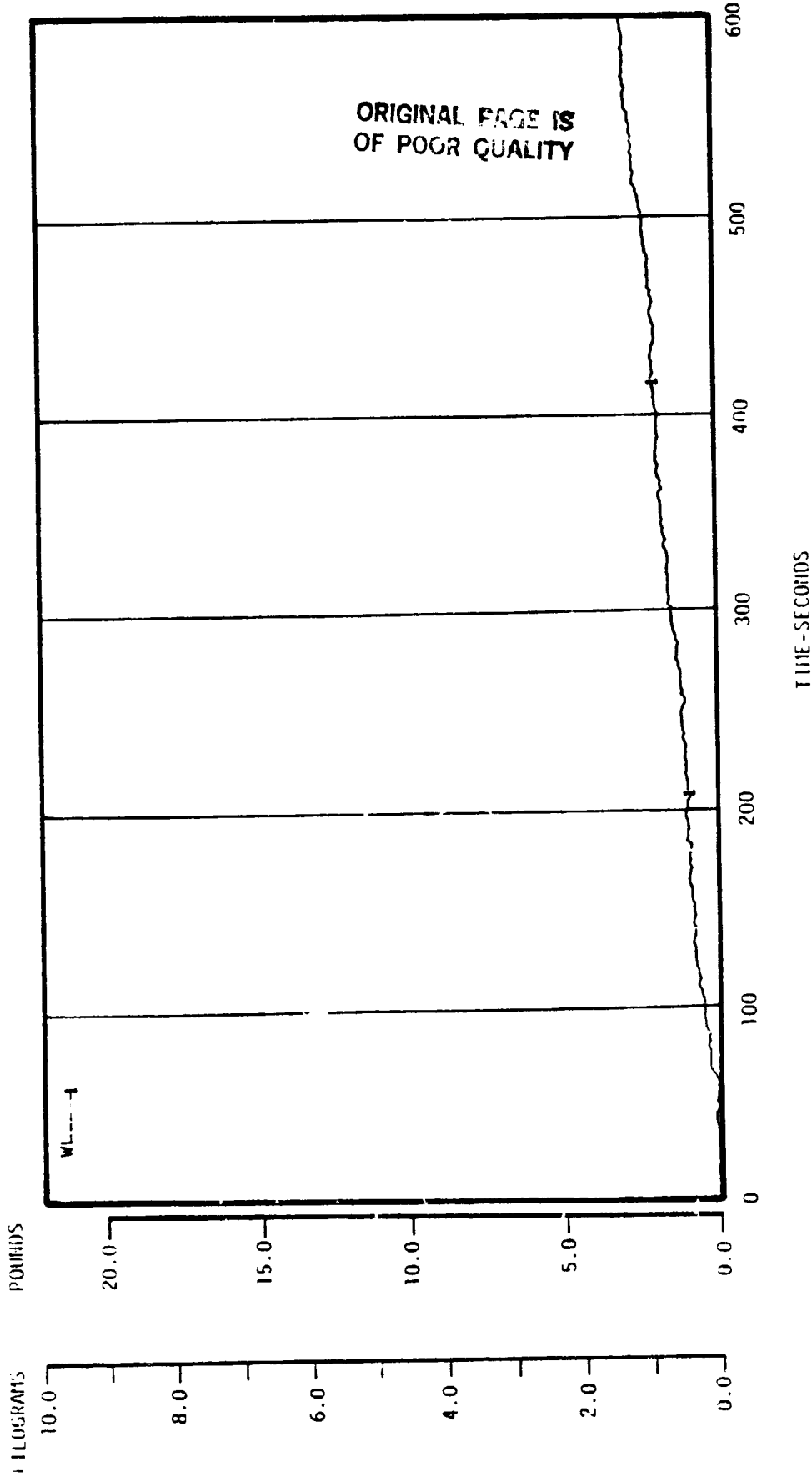


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NASA-MTS FULL SCALE CUSHION BURN TEST NUMBER 11

CUSHION CONSTRUCTION NUMBER 3.0

WEIGHT LOSS

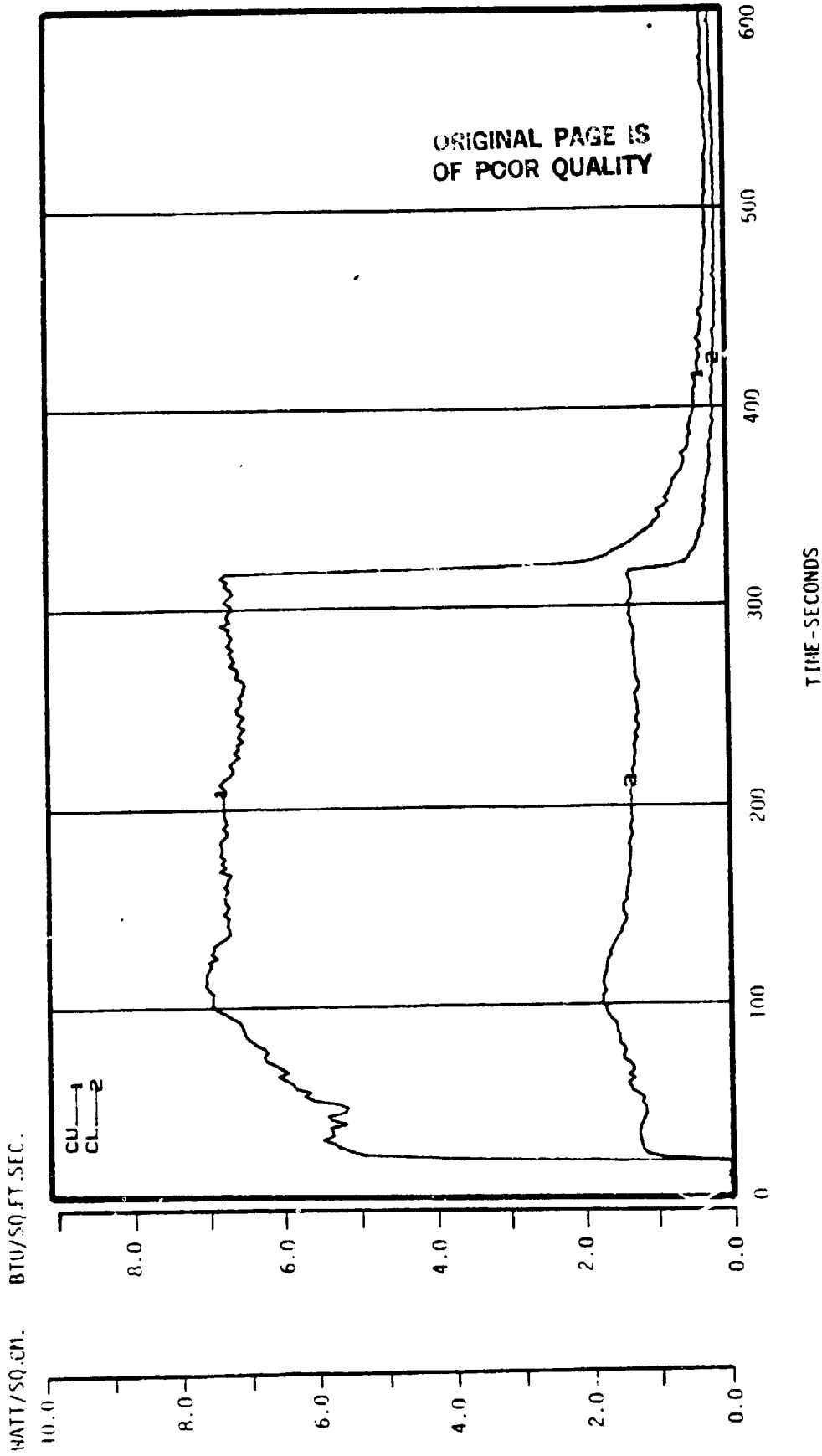


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NASA-AMES FULL SCALE CUSHION BURN TEST NUMBER 11

CUSHION CONSTRUCTION NUMBER 3.0

HEAT FLUX

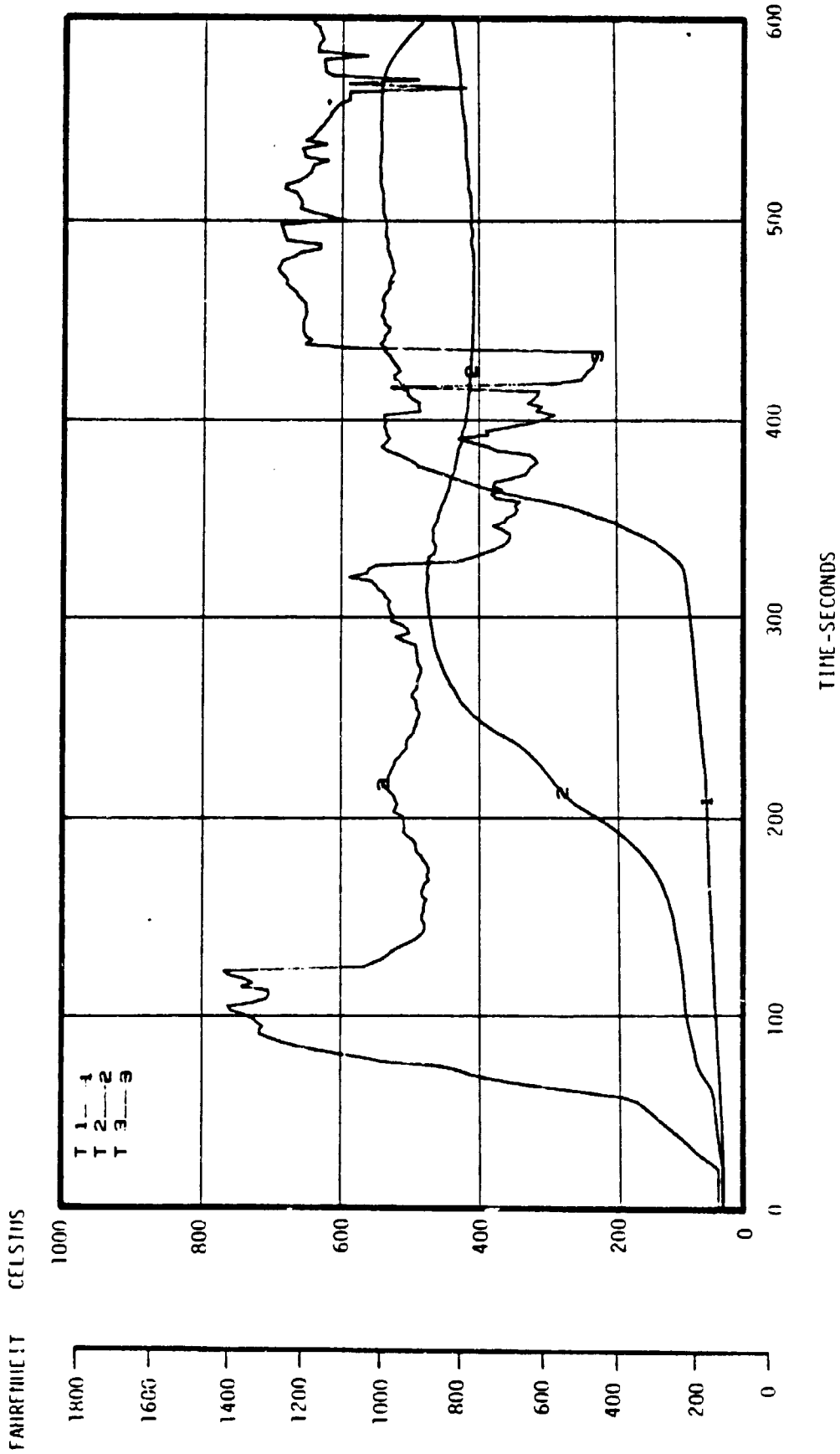


DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/12/82 13.51

NASA-NIES FULL SCALE CUSHION BURN TEST NUMBER 12

CUSHION CONSTRUCTION NUMBER 3.0

SEAT CUSHION TEMPERATURES



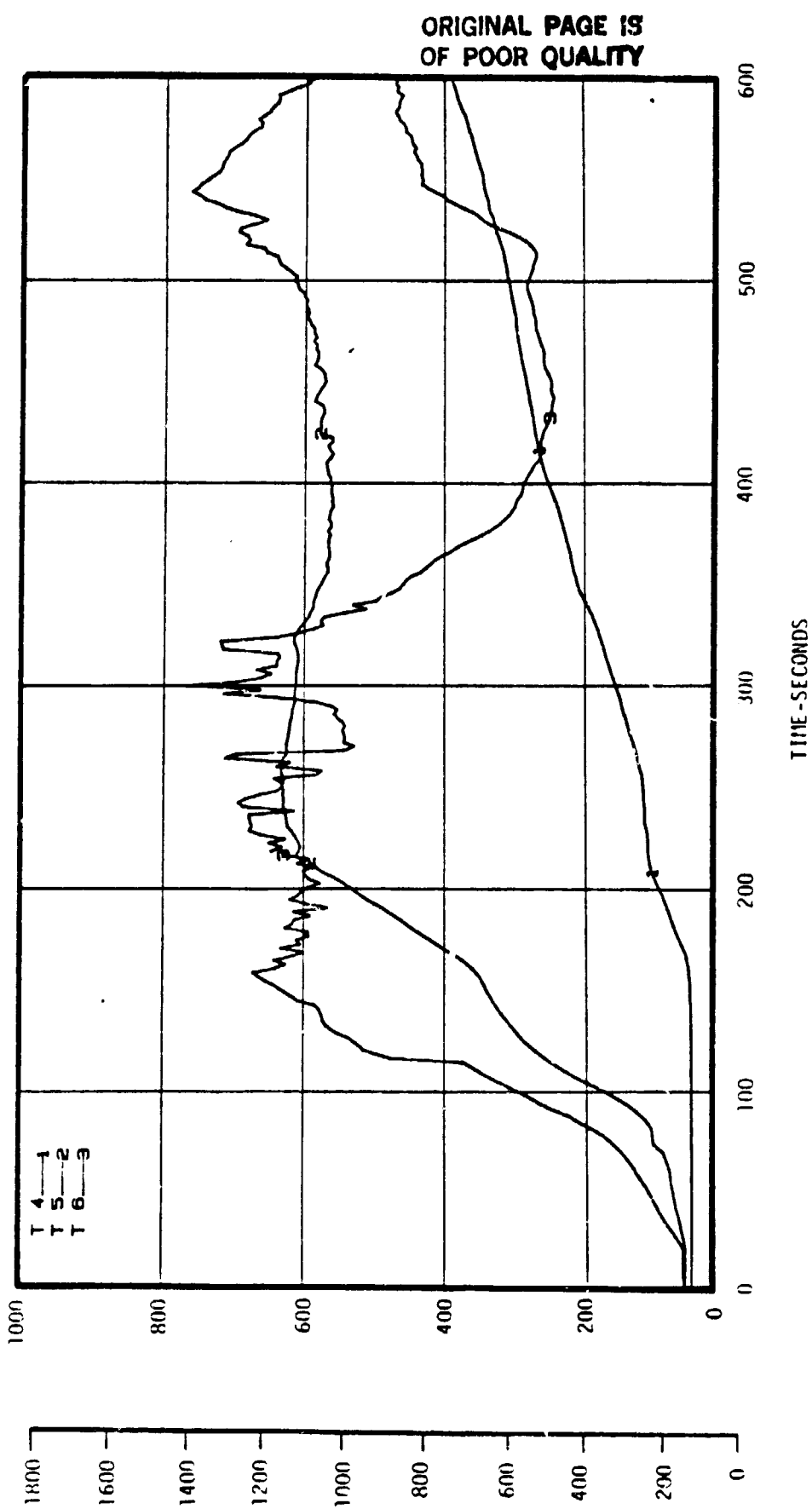
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CUSHION CONSTRUCTION NUMBER 3.0

SEAT CUSHION TEMPERATURES

FAHRENHEIT CELSIUS



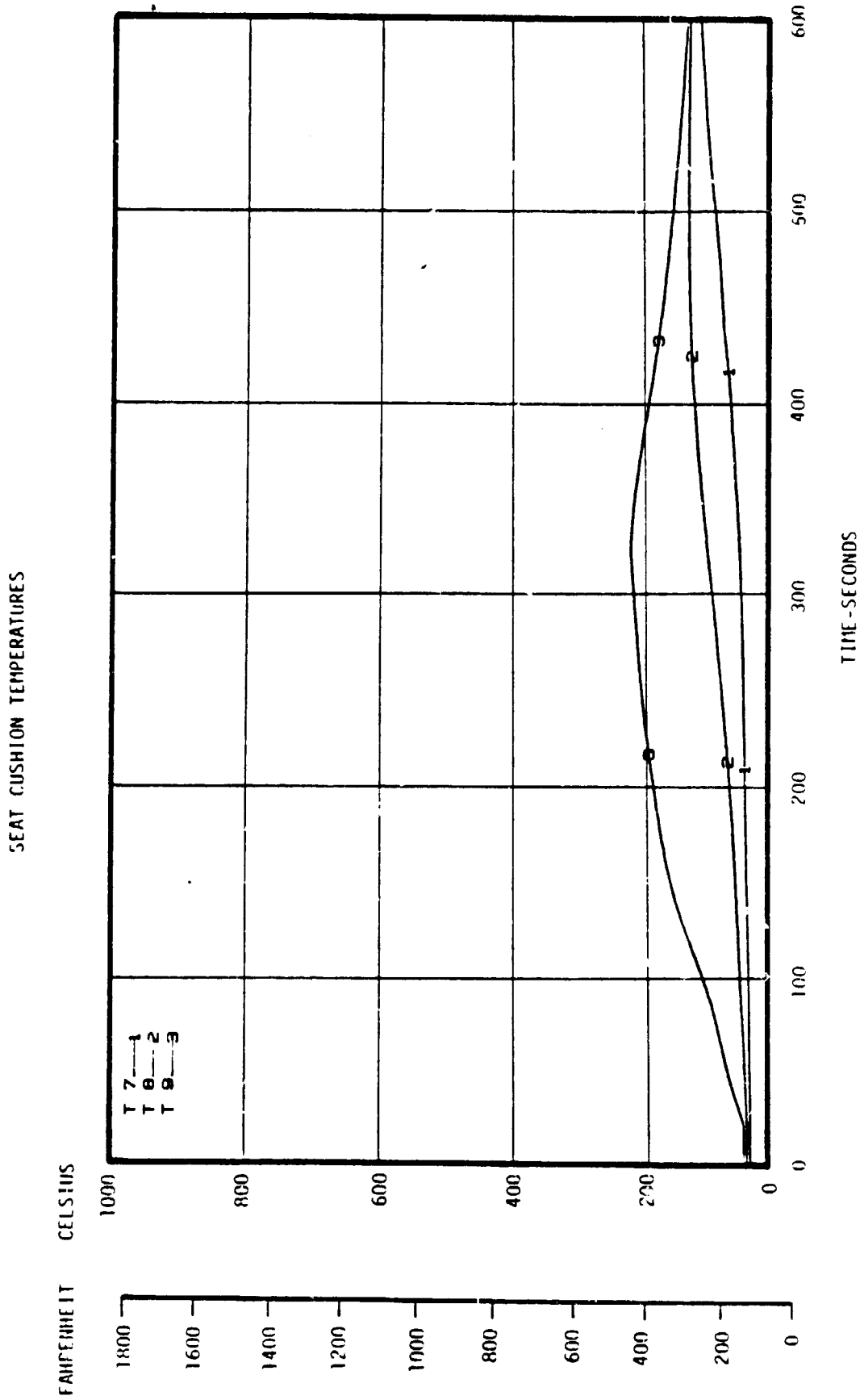
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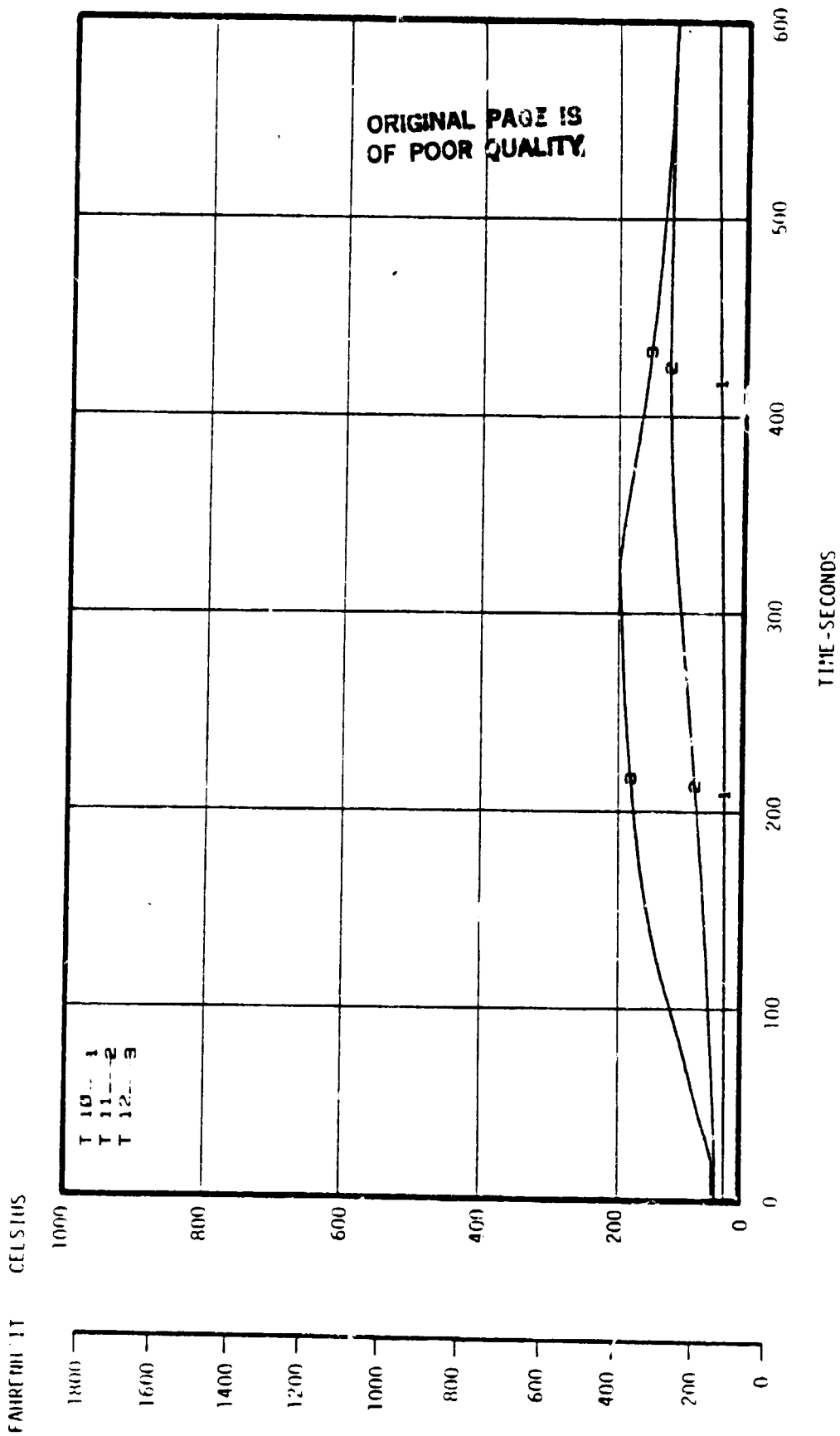


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CUSHION CONSTRUCTION NUMBER 3.0

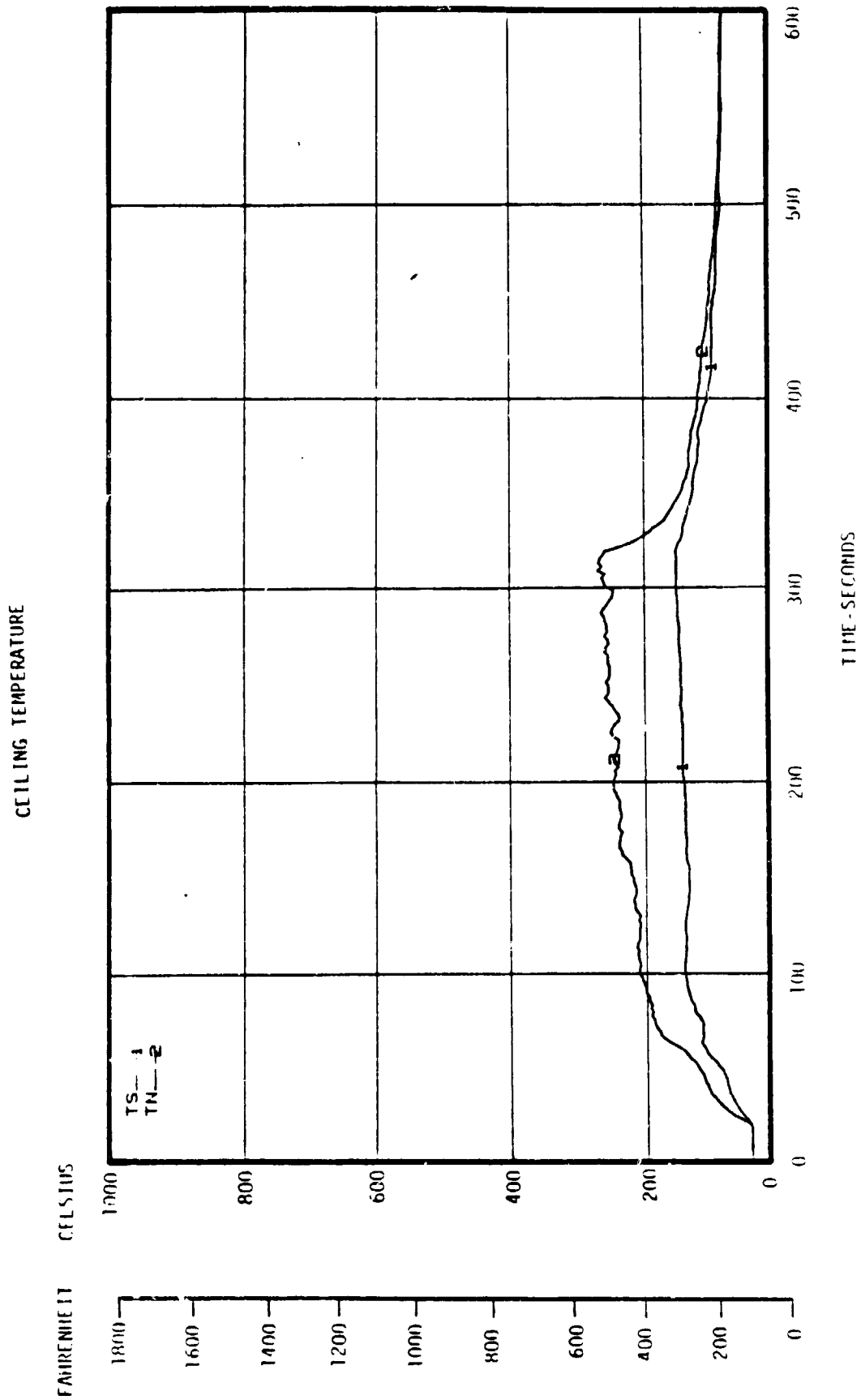
SEAT CUSHION TEMPERATURES



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CUSHION CONSTRUCTION NUMBER 3.0



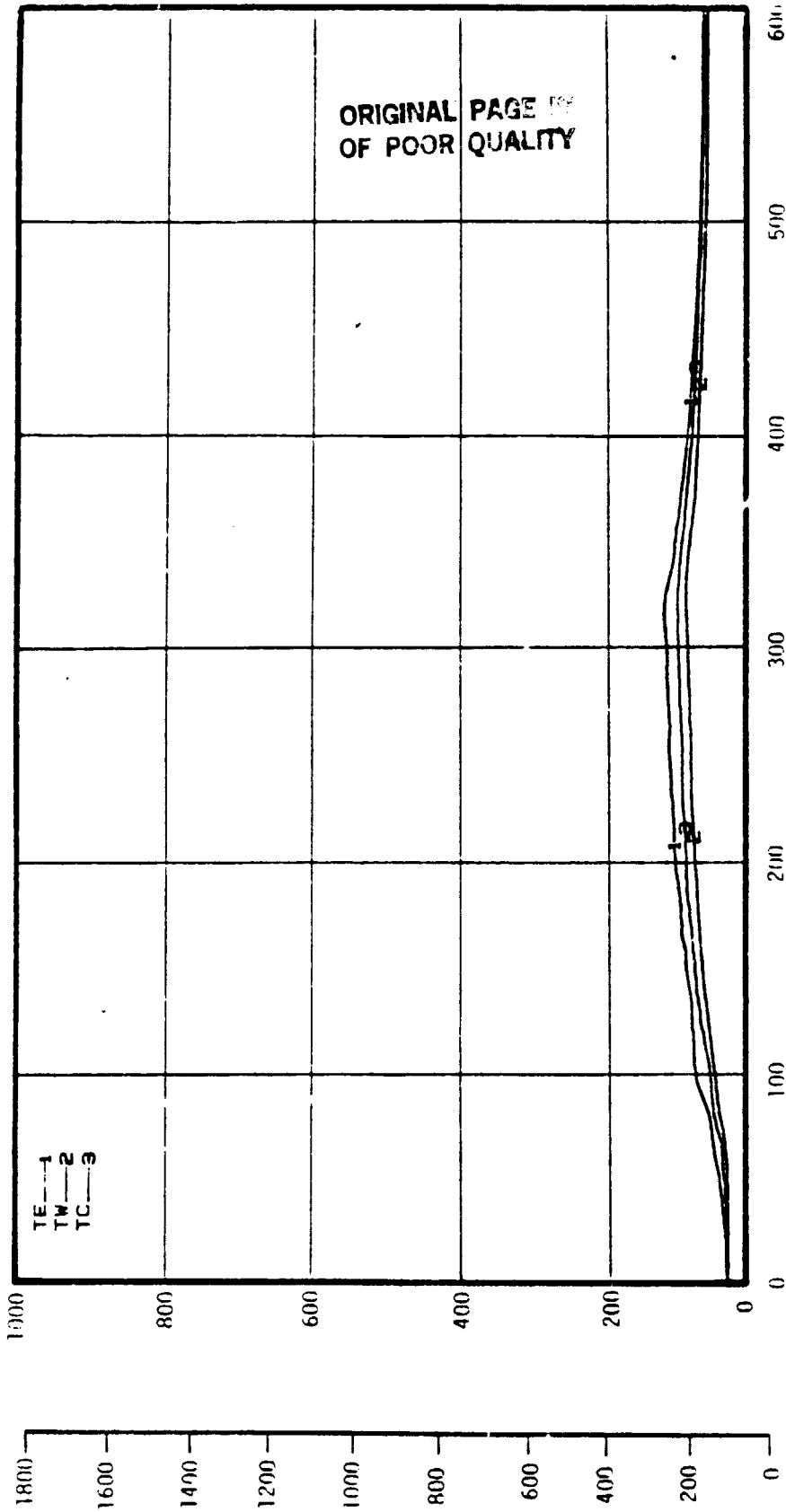
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CUSHION CONSTRUCTION NUMBER 3.0

CEILING TEMPERATURE

FAHRENHEIT CELSIUS

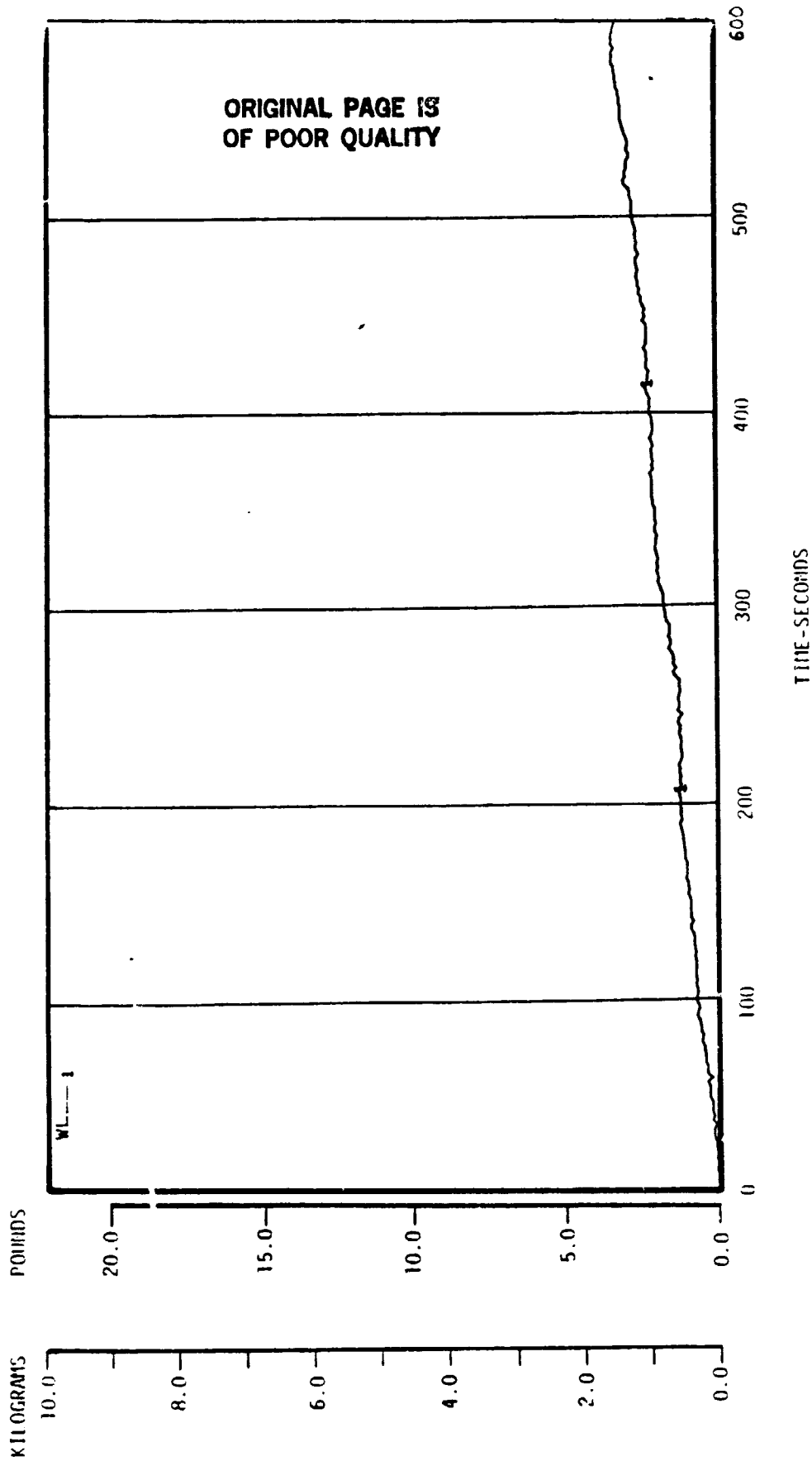


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NASA-AMES FULL SCALE CUSHION BURN TEST NUMBER 12

CUSHION CONSTRUCTION NUMBER 3.0

WEIGHT LOSS

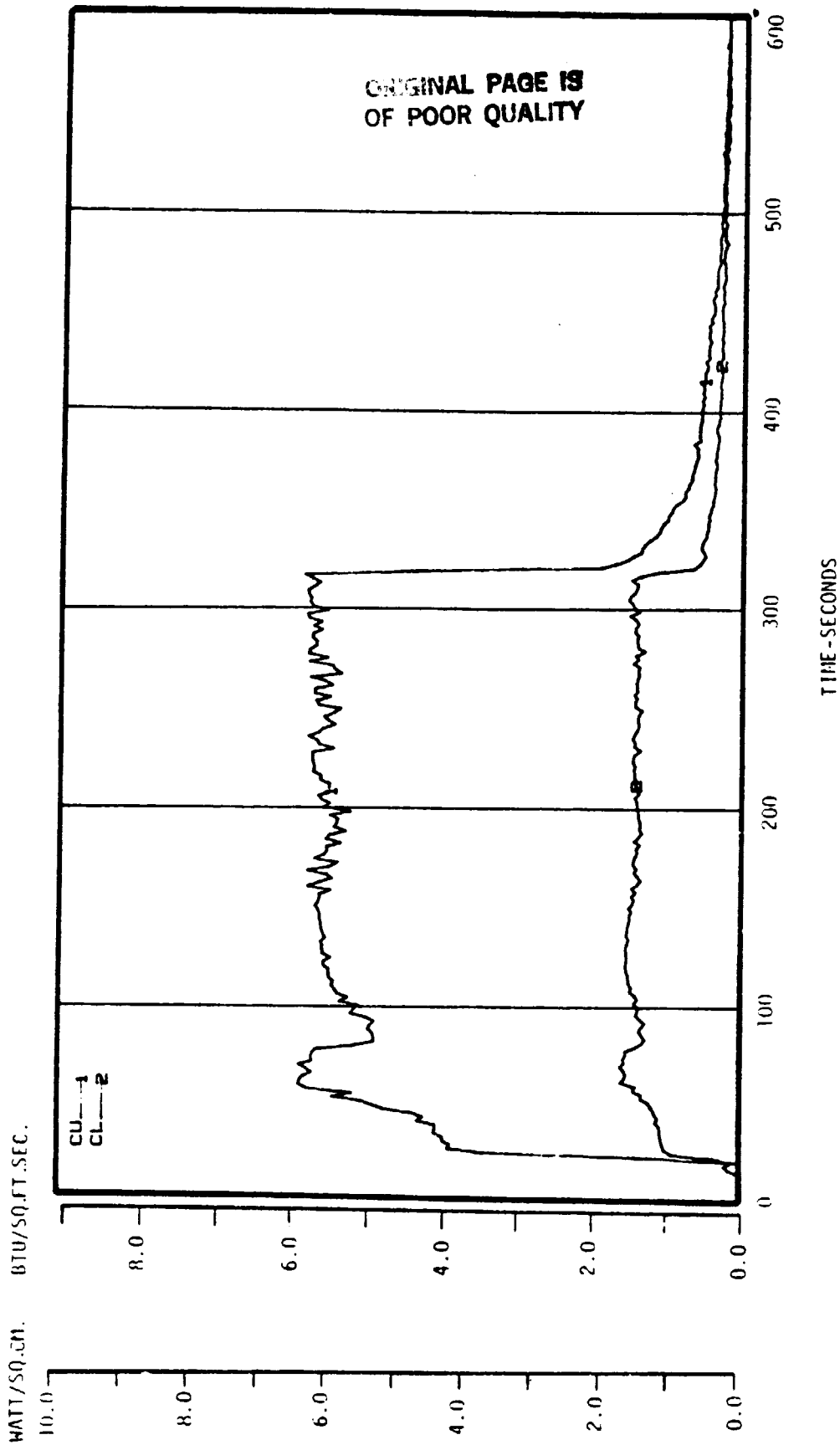


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NASA-AMES FULL SCALE CUSHION BURN TEST NUMBER 12

CUSHION CONSTRUCTION NUMBER 3.0

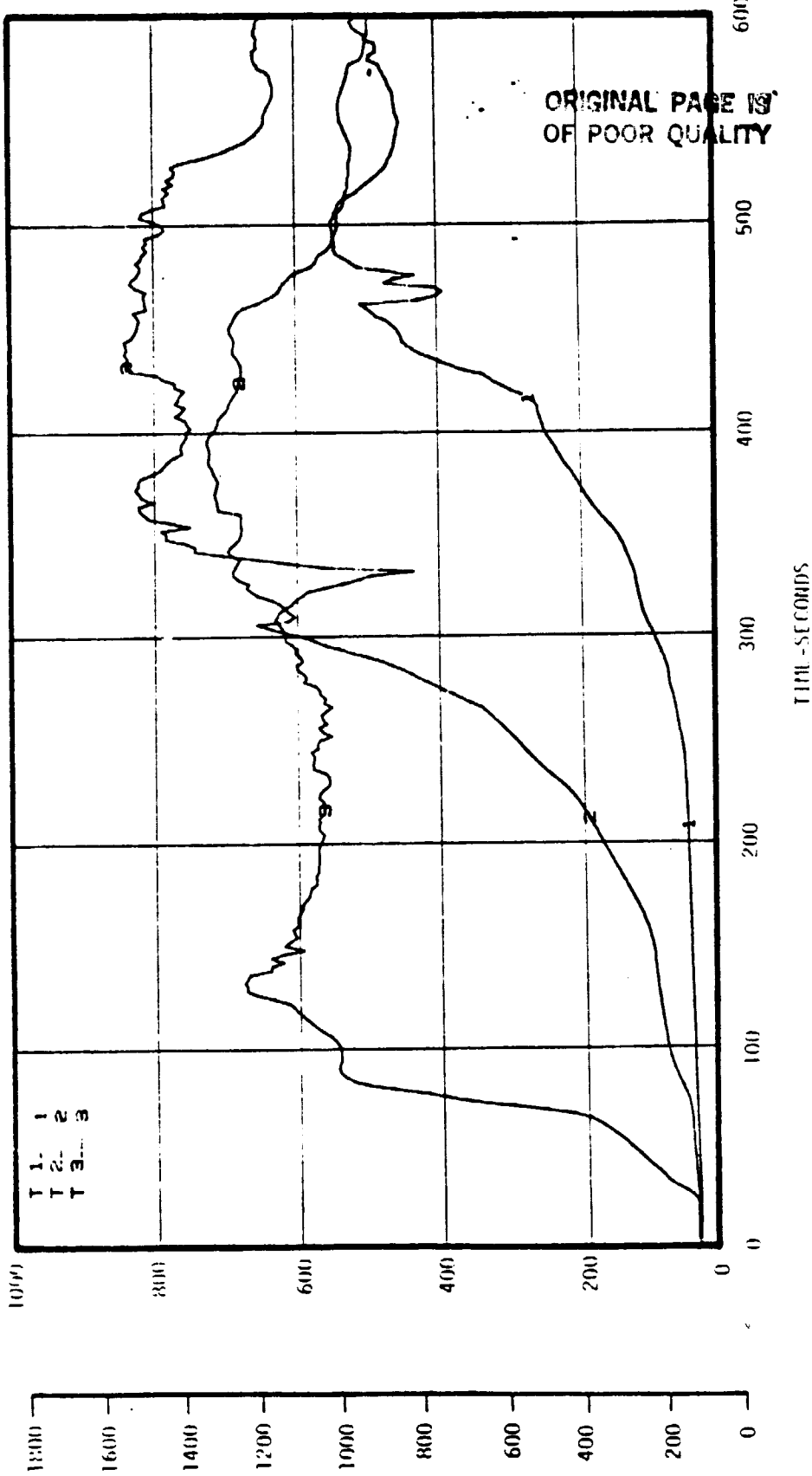
HEAT FLUX



DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/08/82 19.27
NASA-AMES FULL SCALE CUSHION BURN TEST NUMBER 3
CUSHION CONSTRUCTION NUMBER 4.0

SEAT CUSHION TEMPERATURES

FAHRENHEIT CELSIUS



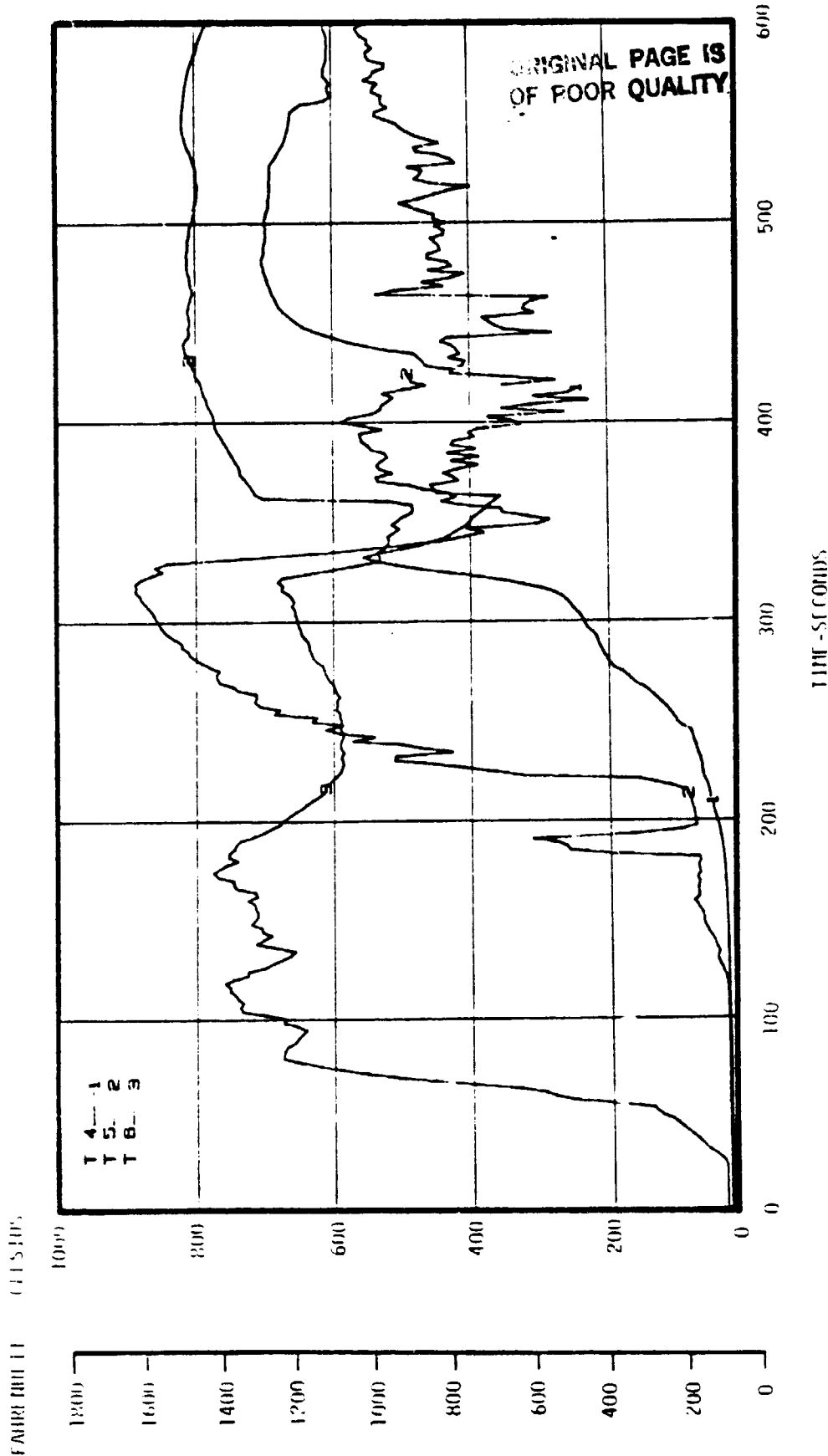
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DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/08/82 13.27

RAYA-MIE'S FULL SEAT CUSHION BURST TEST NUMBER 3

CUSHION CONSTRUCTION NUMBER 4.0

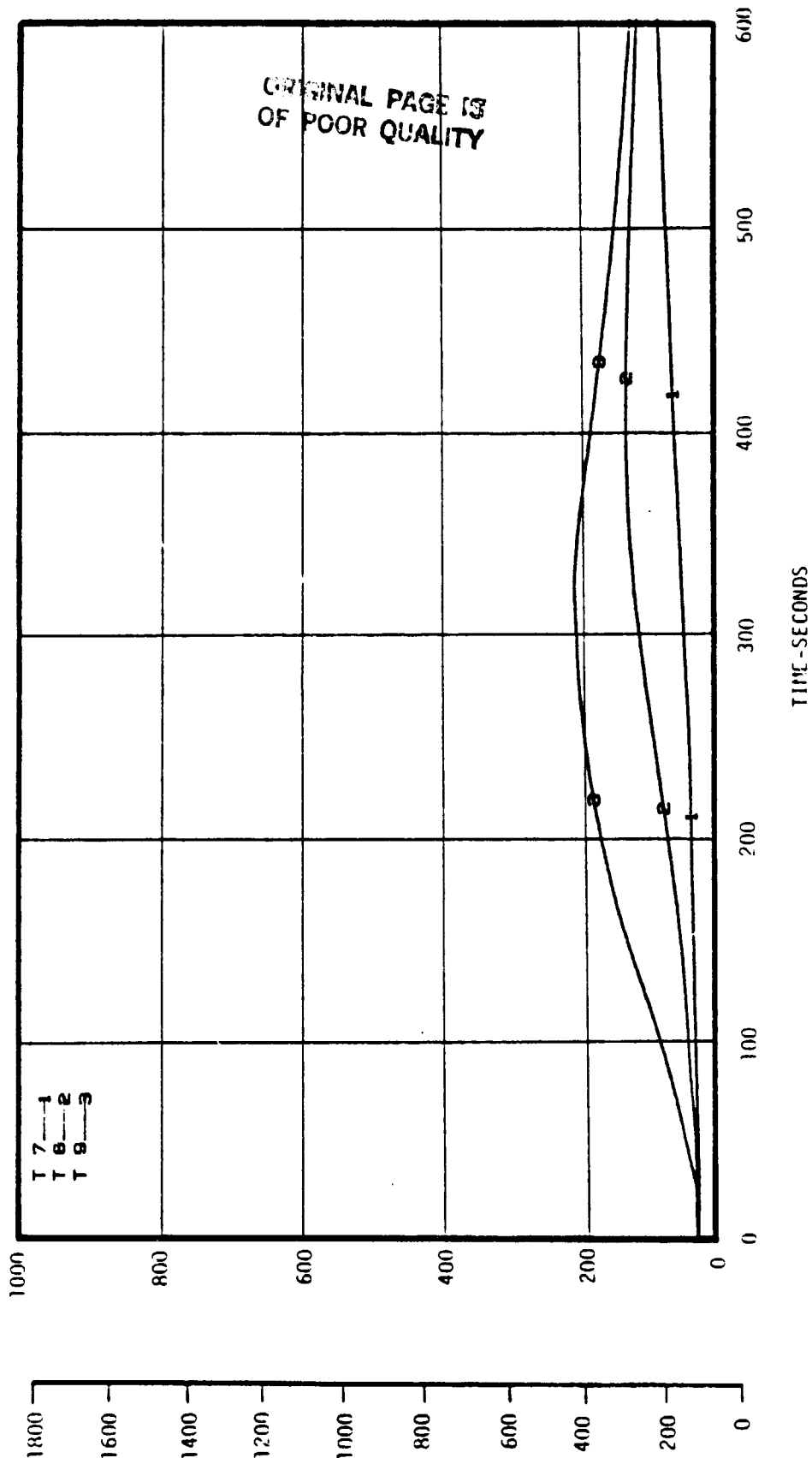
HEAT CUSHION TEMPERATURES



DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/08/82 13.27
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 CUSHION CONSTRUCTION NUMBER 4.0

SEAT CUSHION TEMPERATURES

FAHRENHEIT CELSIUS

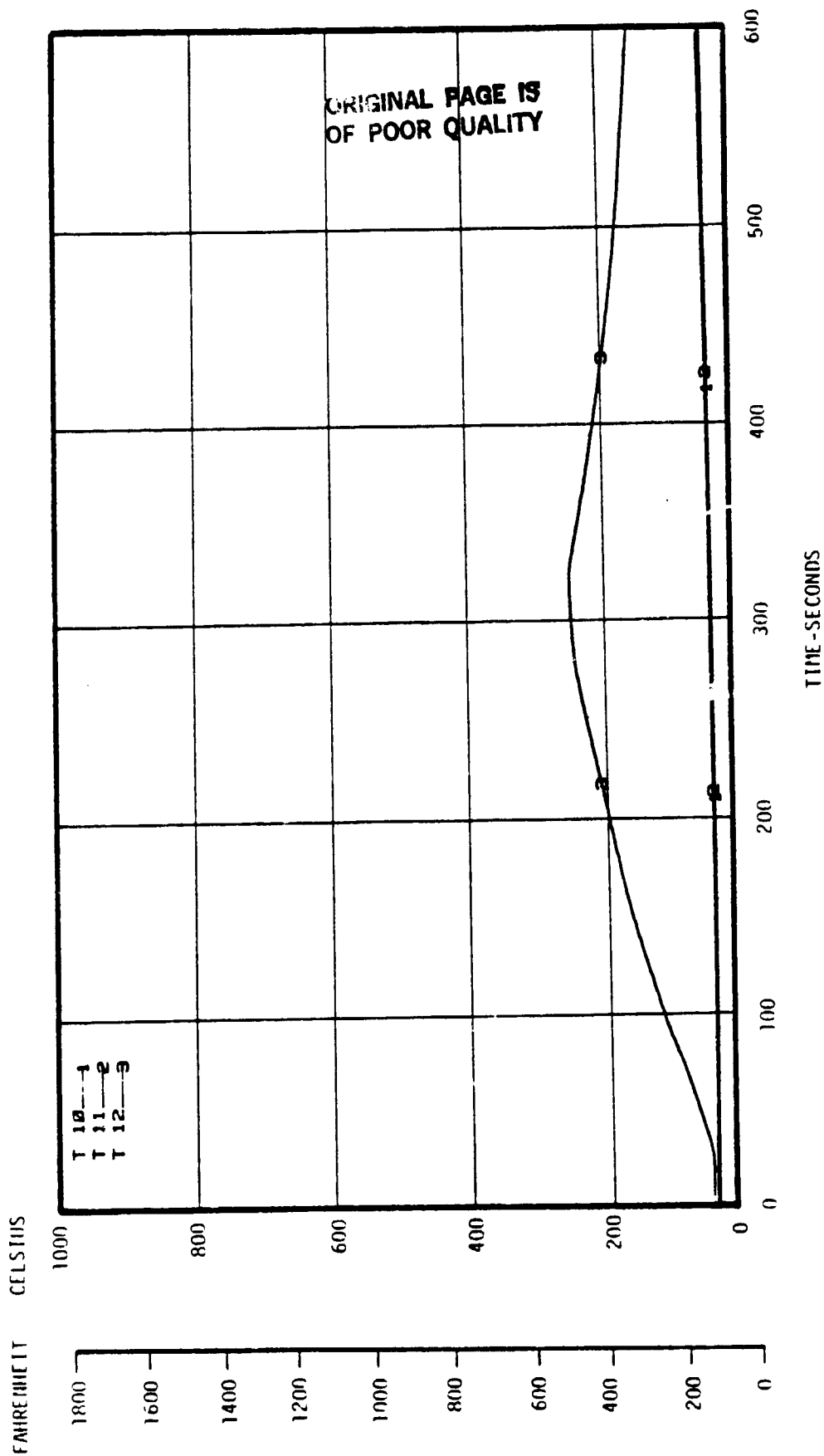


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NASA-AMES FULL SCALE CUSHION BURN TEST NUMBER 3

CUSHION CONSTRUCTION NUMBER 4.0

SEAT CUSHION TEMPERATURES



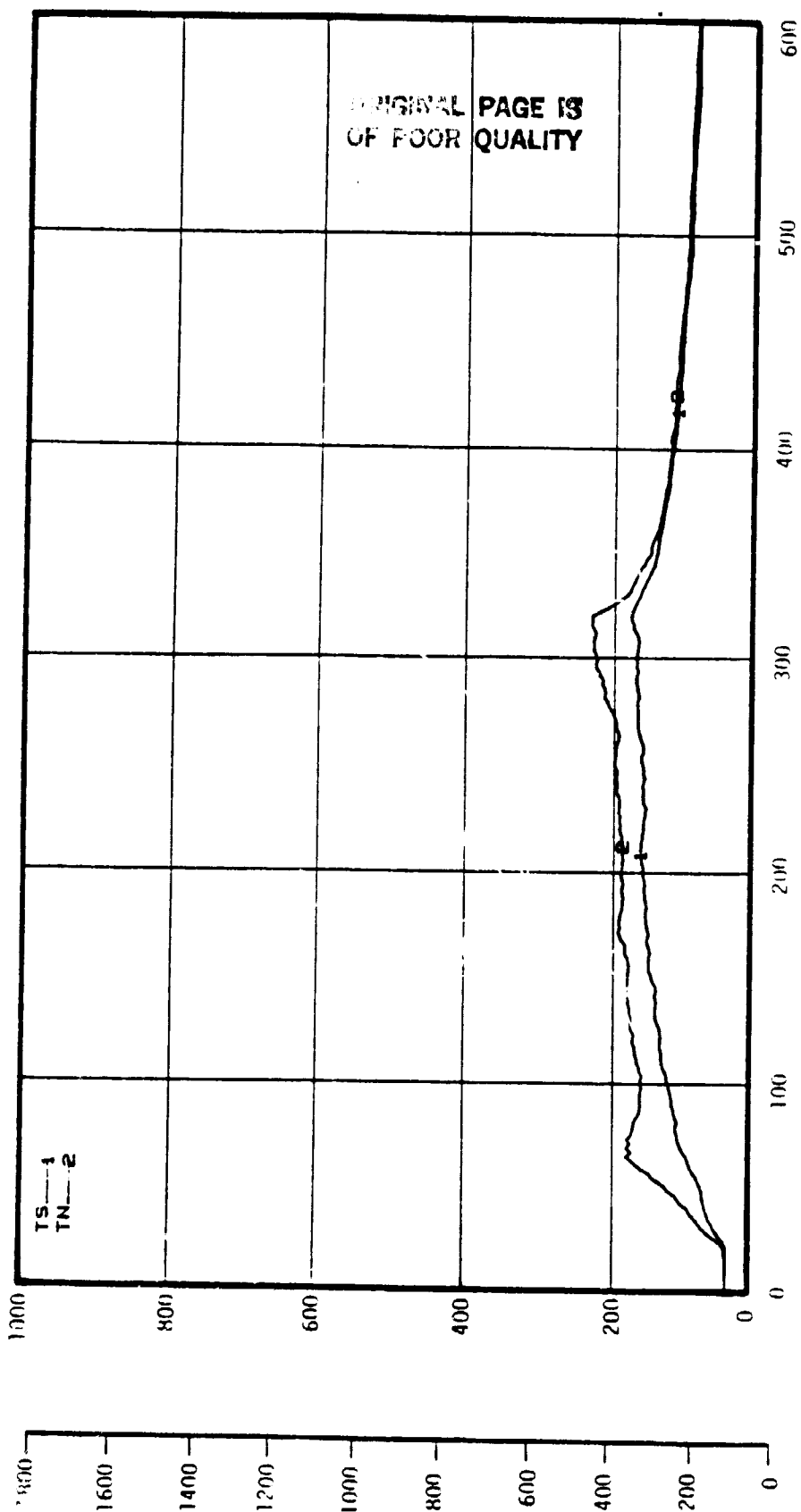
DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/00/82 19.27

NASA-AMES FULL SCALE CUSHION BURN TEST NUMBER 3

CUSHION CONSTRUCTION NUMBER 4.0

CEILING TEMPERATURE

FAHRENHEIT CELSIUS



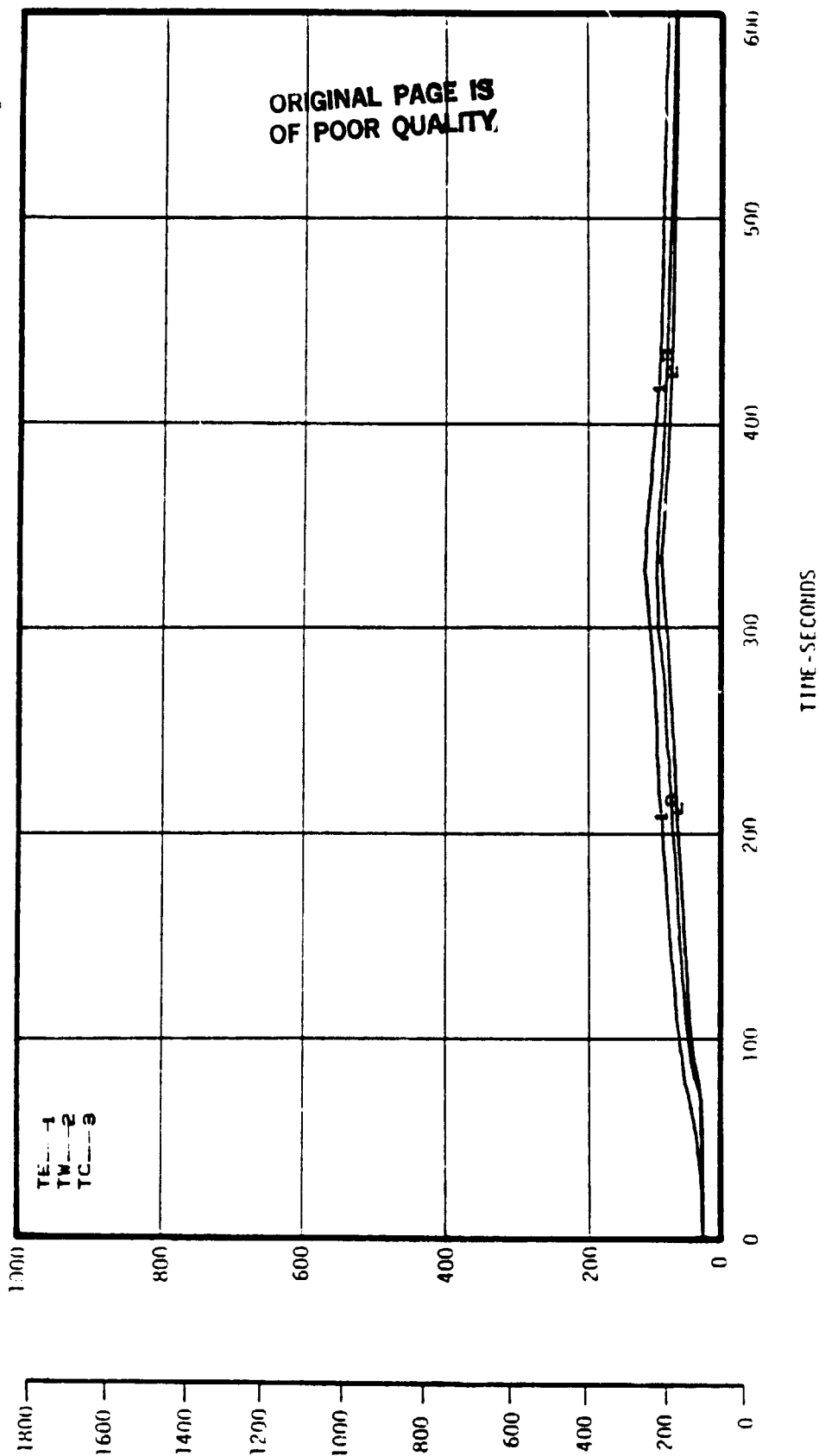
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NASA-MES FULL SCALE CUSHION BURN TEST NUMBER 3

CUSHION CONSTRUCTION NUMBER 4.0

CEILING TEMPERATURE

FAHRENHEIT CELSIUS



DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/08/82 13.27

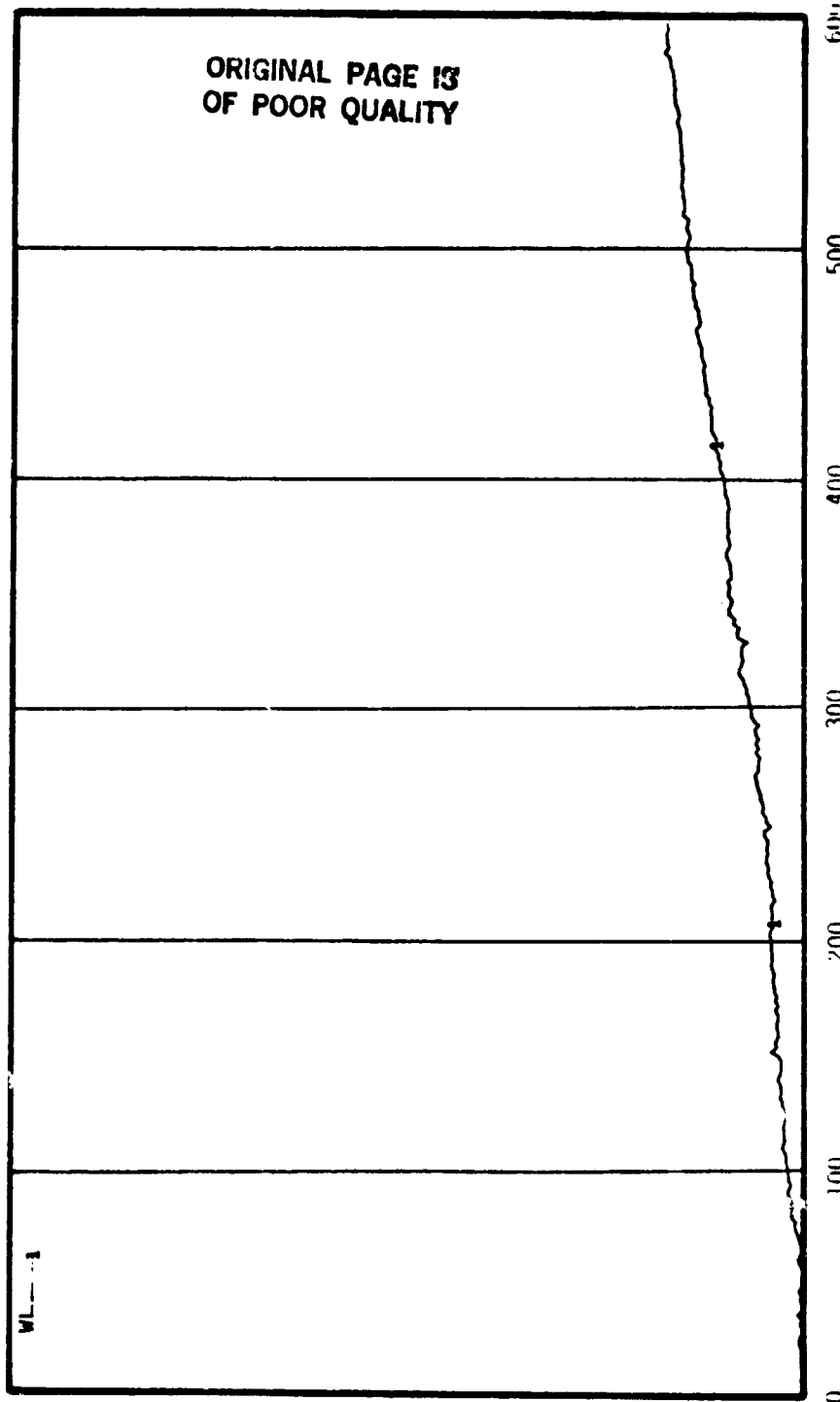
NASA-MES FULL SCALE CUSHION BURN TEST NUMBER 3

CUSHION CONSTRUCTION NUMBER 4.0

WEIGHT LOSS

KILOGRAMS
10.0
8.0
6.0
4.0
2.0
0.0

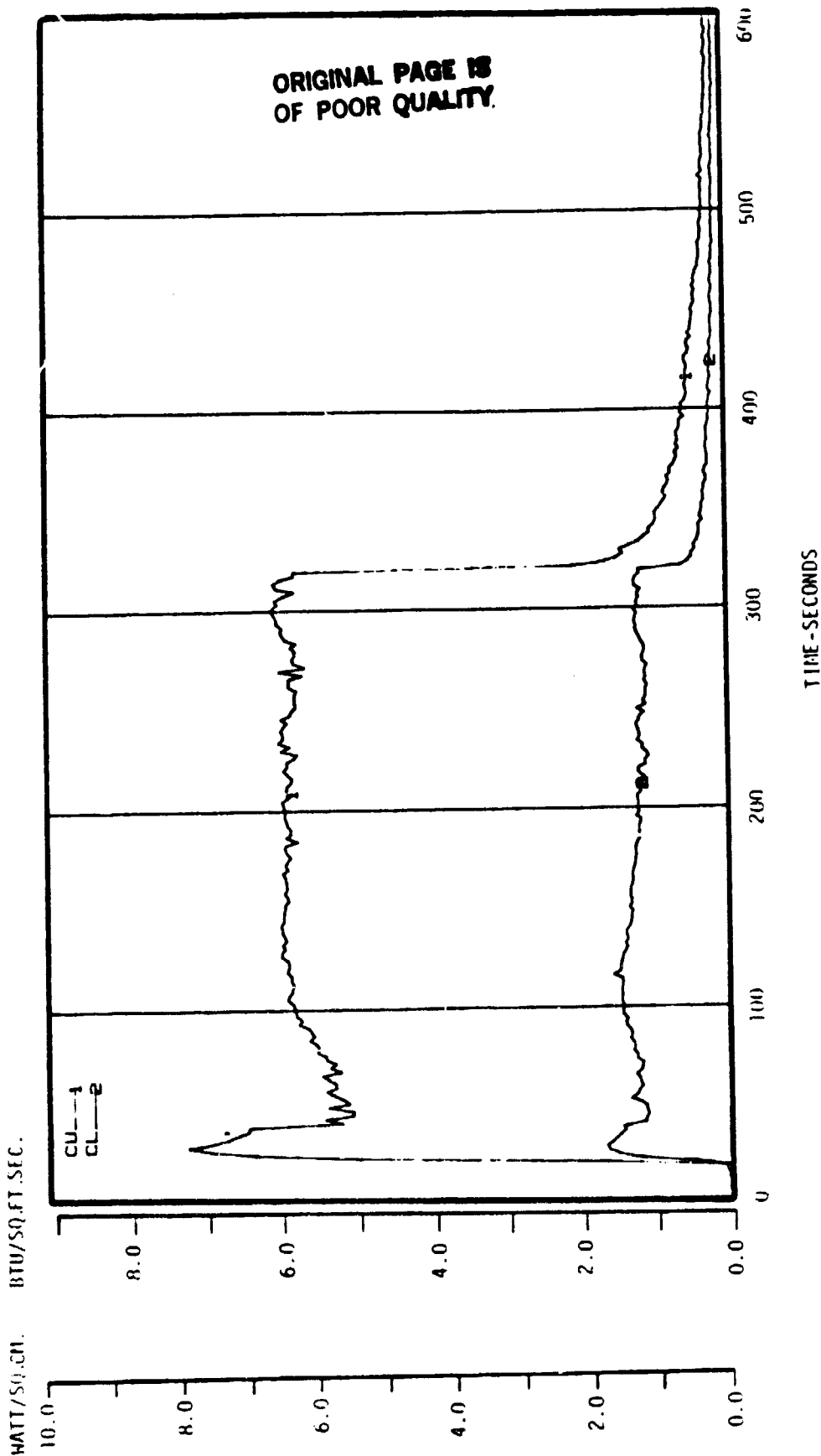
POUNDS
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15.0
10.0
5.0
0.0



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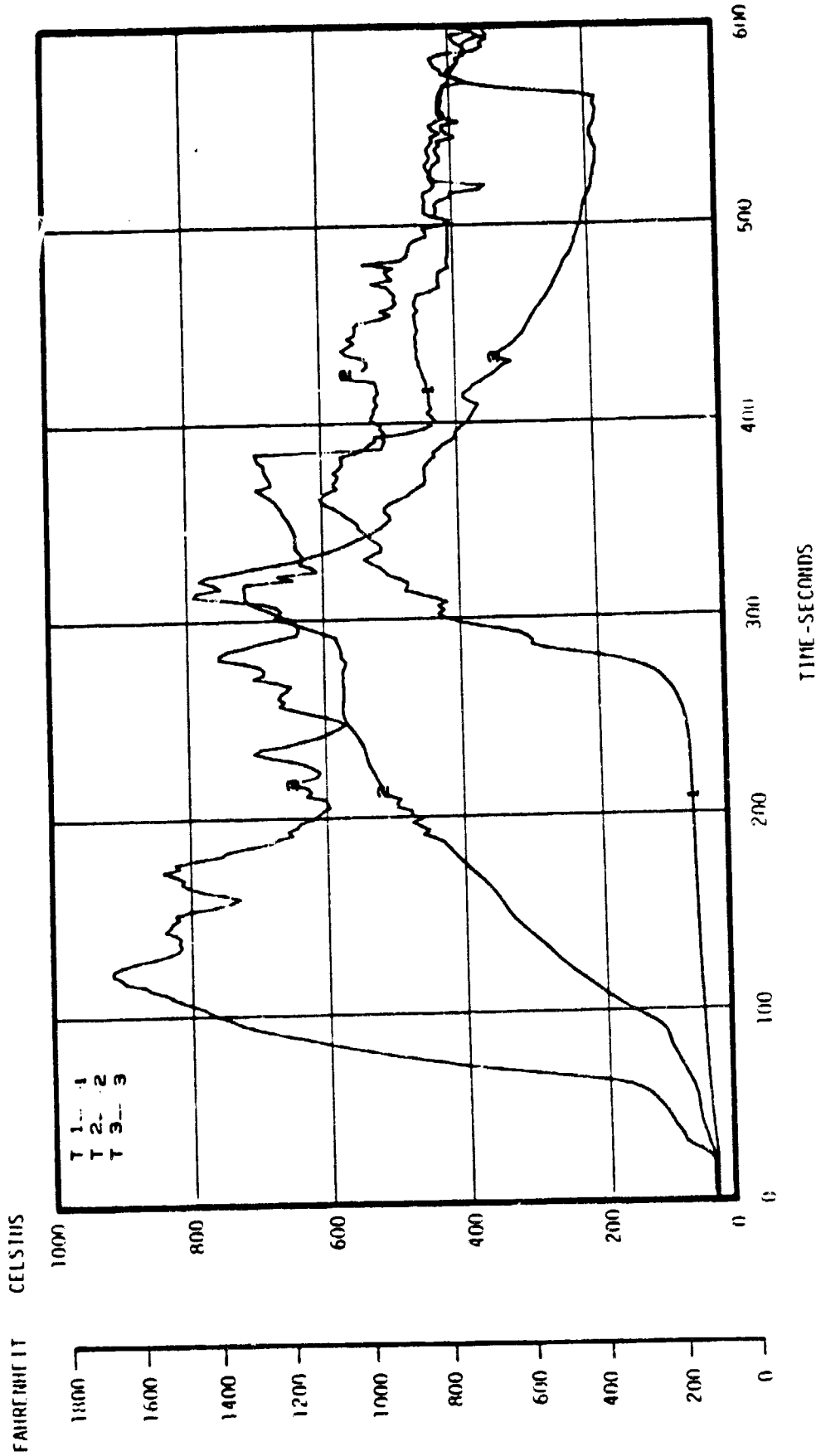
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NASA-AMES FULL SCALE CUSHION BURN TEST NUMBER 3
CUSHION CONSTRUCTION NUMBER 4.0

HEAT FLUX



DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/12/82 09.02
 NASA-AMES FULL SCALE CUSHION BURN TEST NUMBER 10
 CUSHION CONSTRUCTION NUMBER 4.0

SEAT CUSHION TEMPERATURES



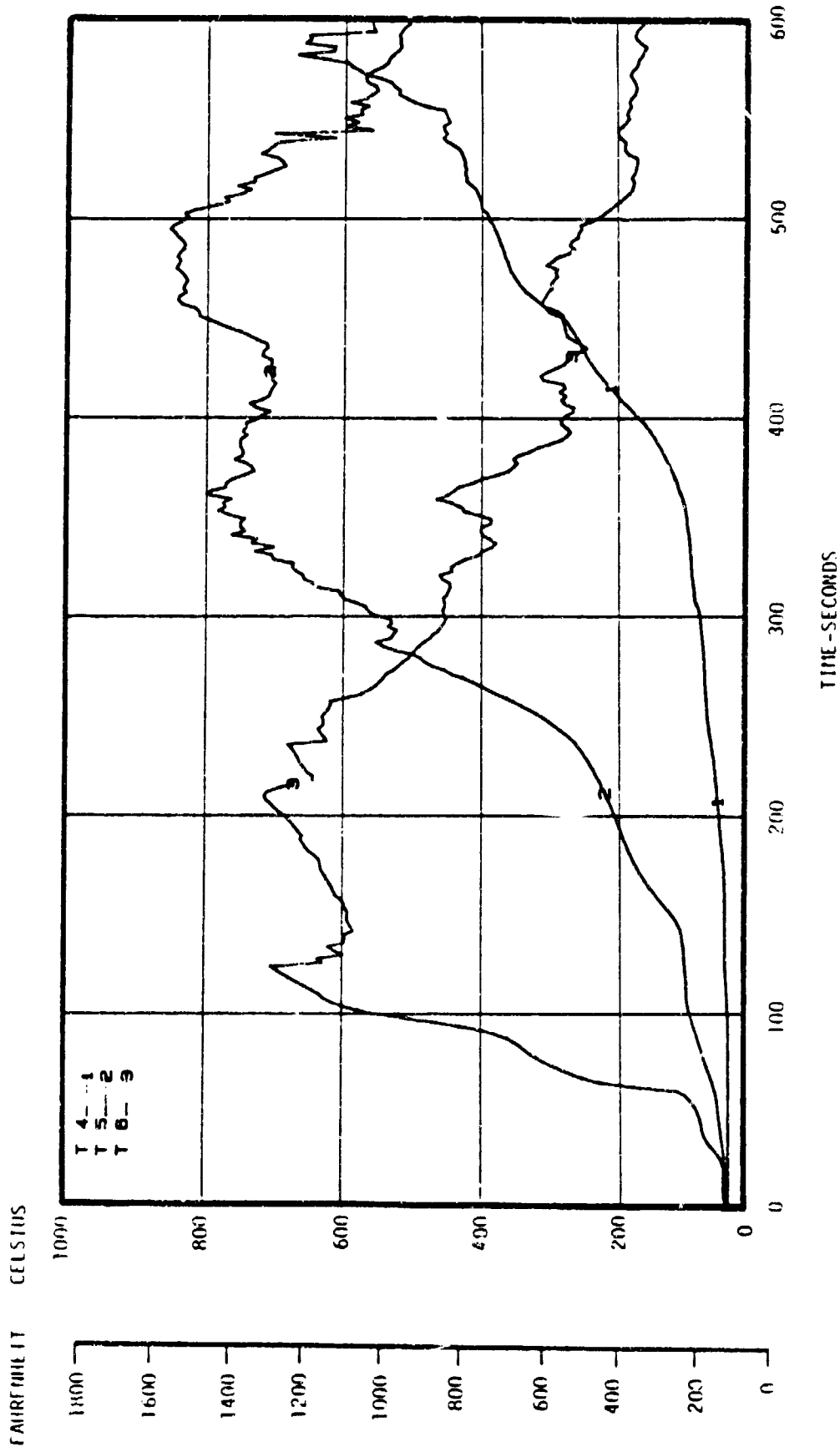
DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/12/82 00.02

NASA-AMES FULL SCALE CUSHION BURN TEST NUMBER 10

CUSHION CONSTRUCTION NUMBER 4.0

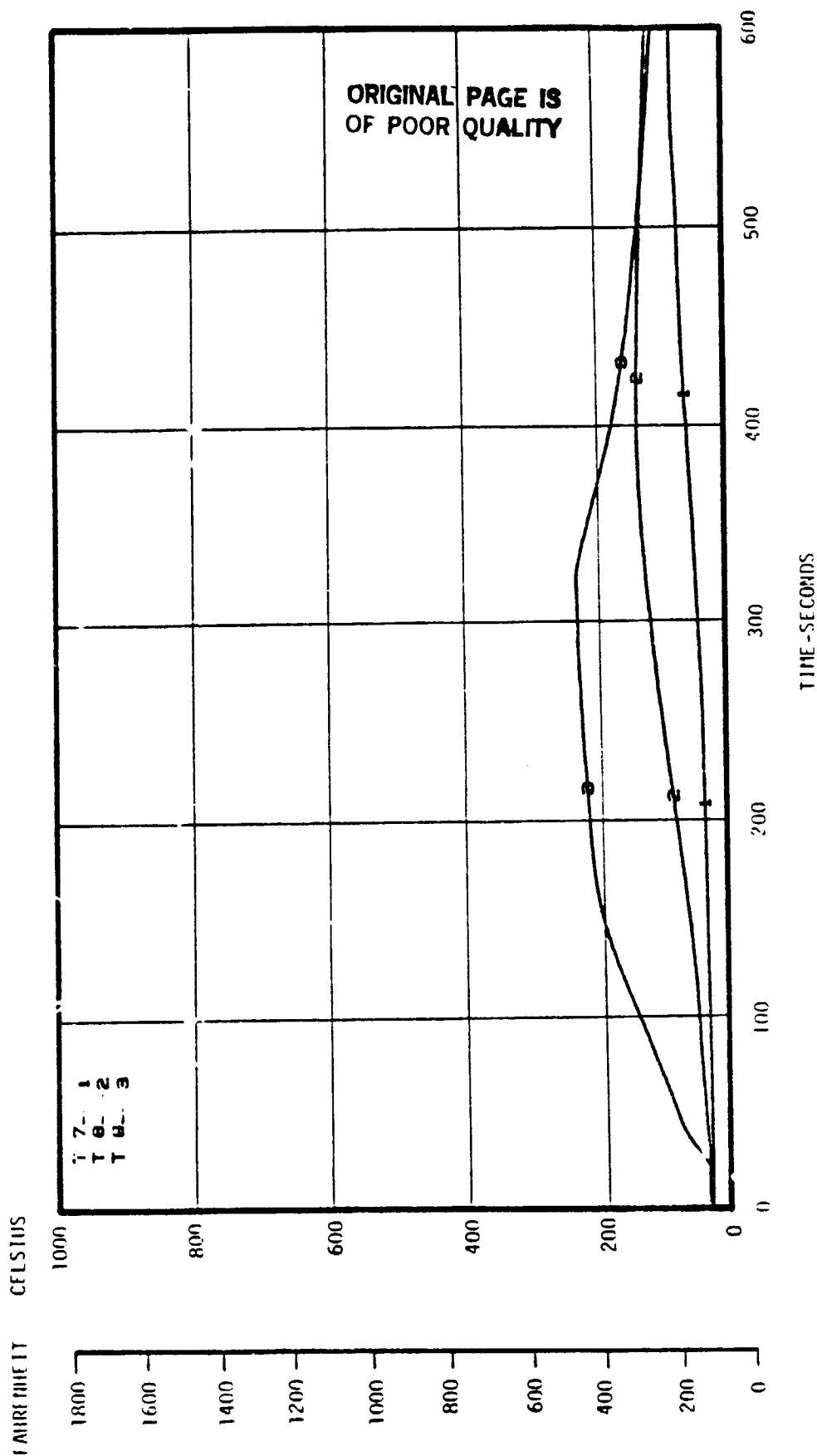
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SEAT CUSHION TEMPERATURES



DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/12/82 00.02
NASA-AMES FULL SCALE CUSHION BURN TEST NUMBER 10
CUSHION CONSTRUCTION NUMBER 4.0

SEAT CUSHION TEMPERATURES



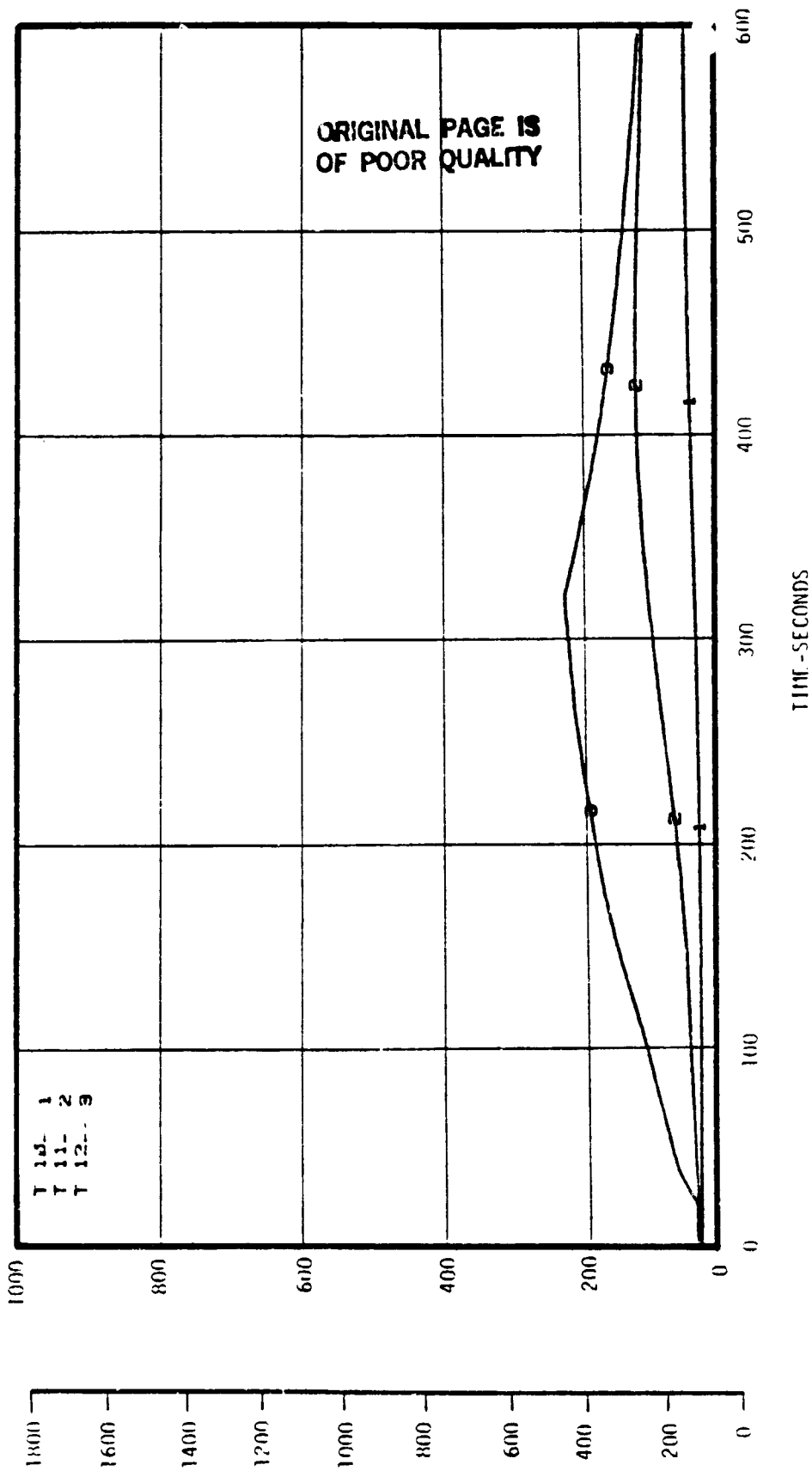
DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/12/62 00.02

NASA-AMES FULL SCALE CUSHION BURN TEST NUMBER 10

CUSHION CONSTRUCTION NUMBER 4.0

SEAT CUSHION TEMPERATURES

FAHRENHEIT CELSIUS



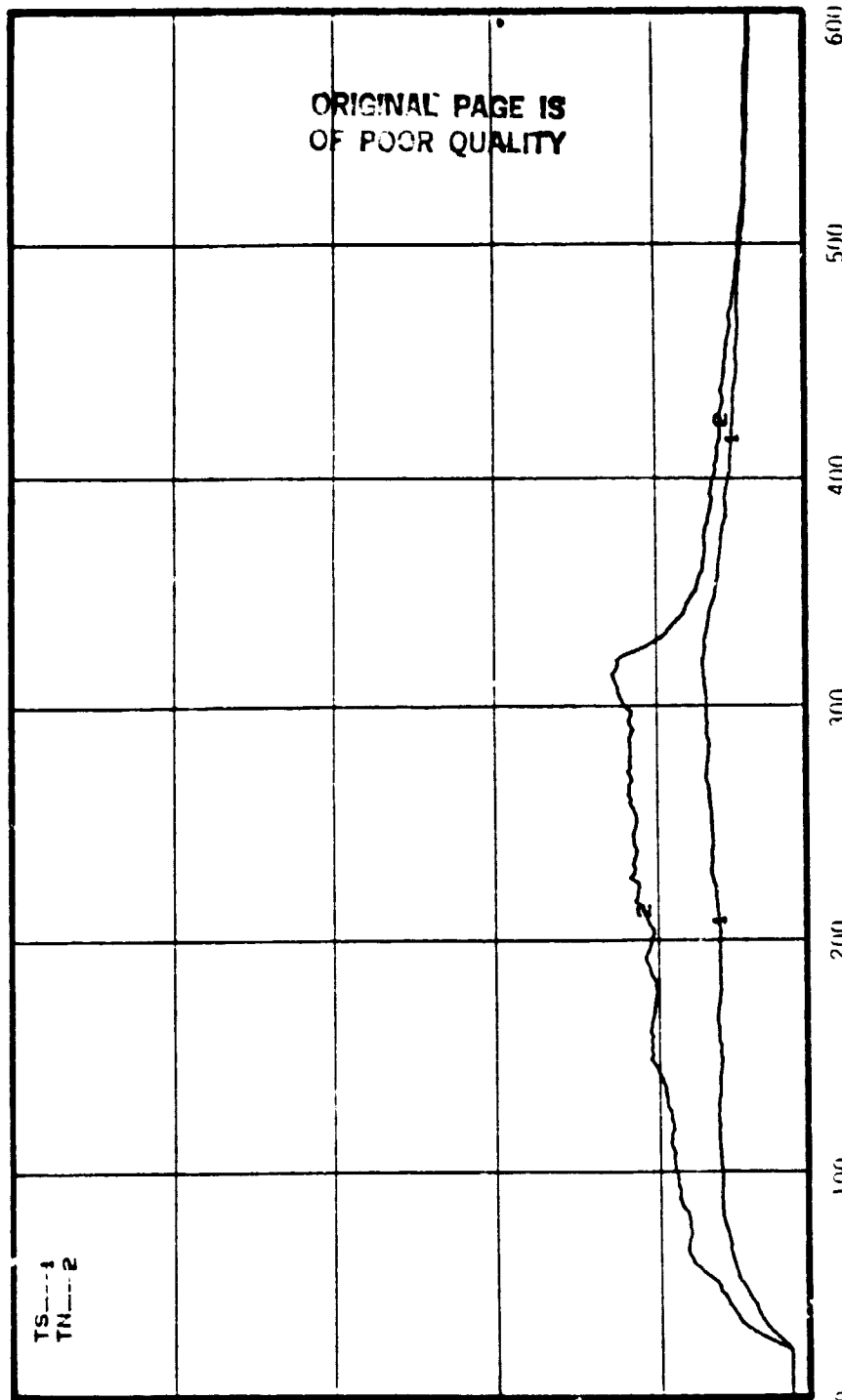
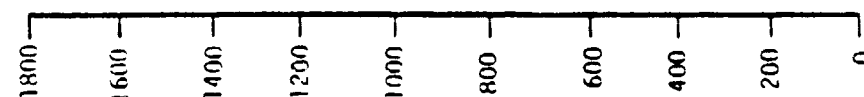
DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/12/82 00.02

NASA-AMES FULL SCALE CUSHION BURN TEST NUMBER 10

CUSHION CONSTRUCTION NUMBER 4.0

CEILING TEMPERATURE

FAHRENHEIT CELSIUS



TIME-SECONDS

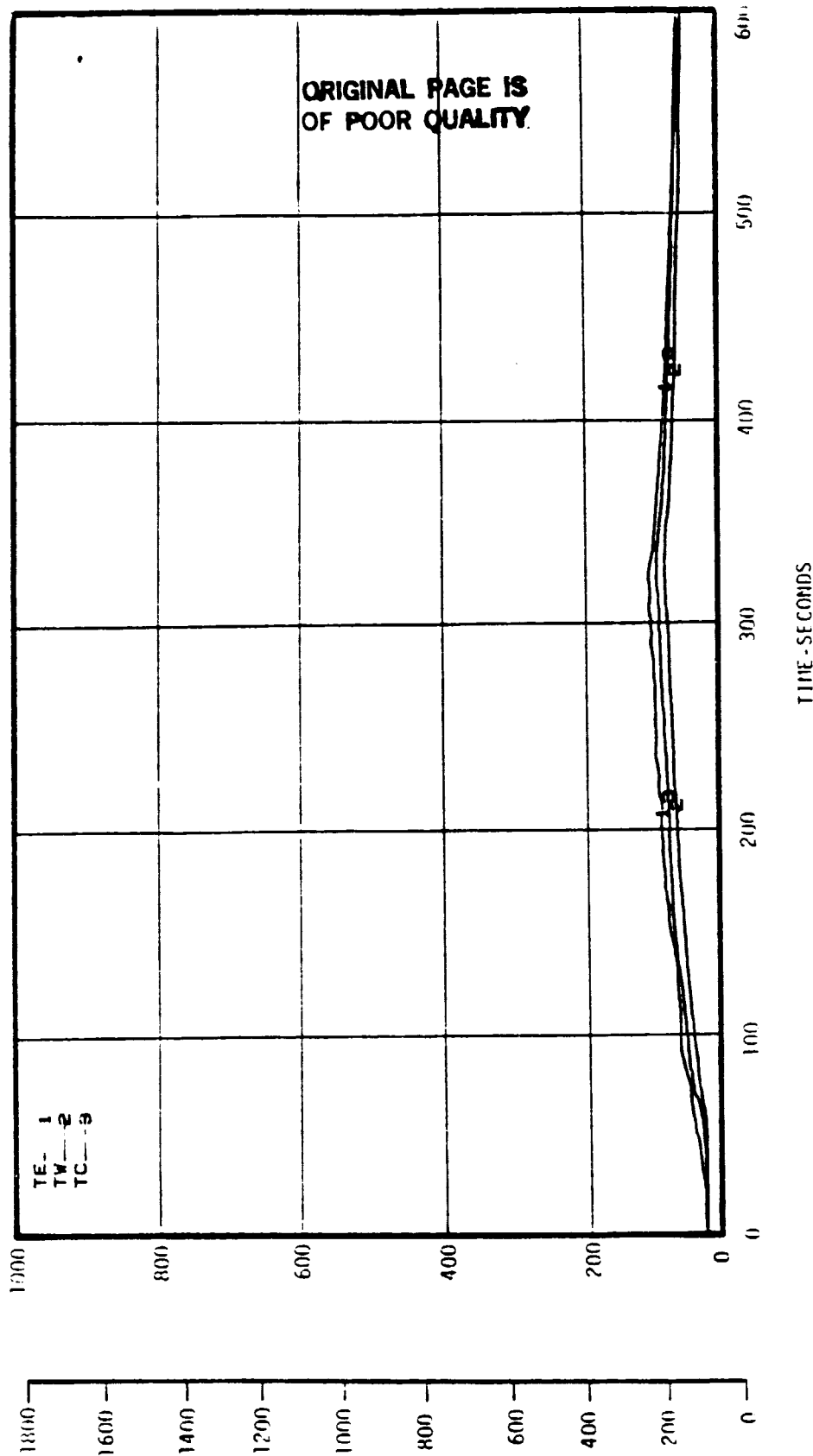
DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/12/82 09.02

NASA-MF'S FULL SCALE CUSHION BURN TEST NUMBER 10

CUSHION CONSTRUCTION NUMBER 4.0

CEILING TEMPERATURE

FAHRENHEIT CELSIUS

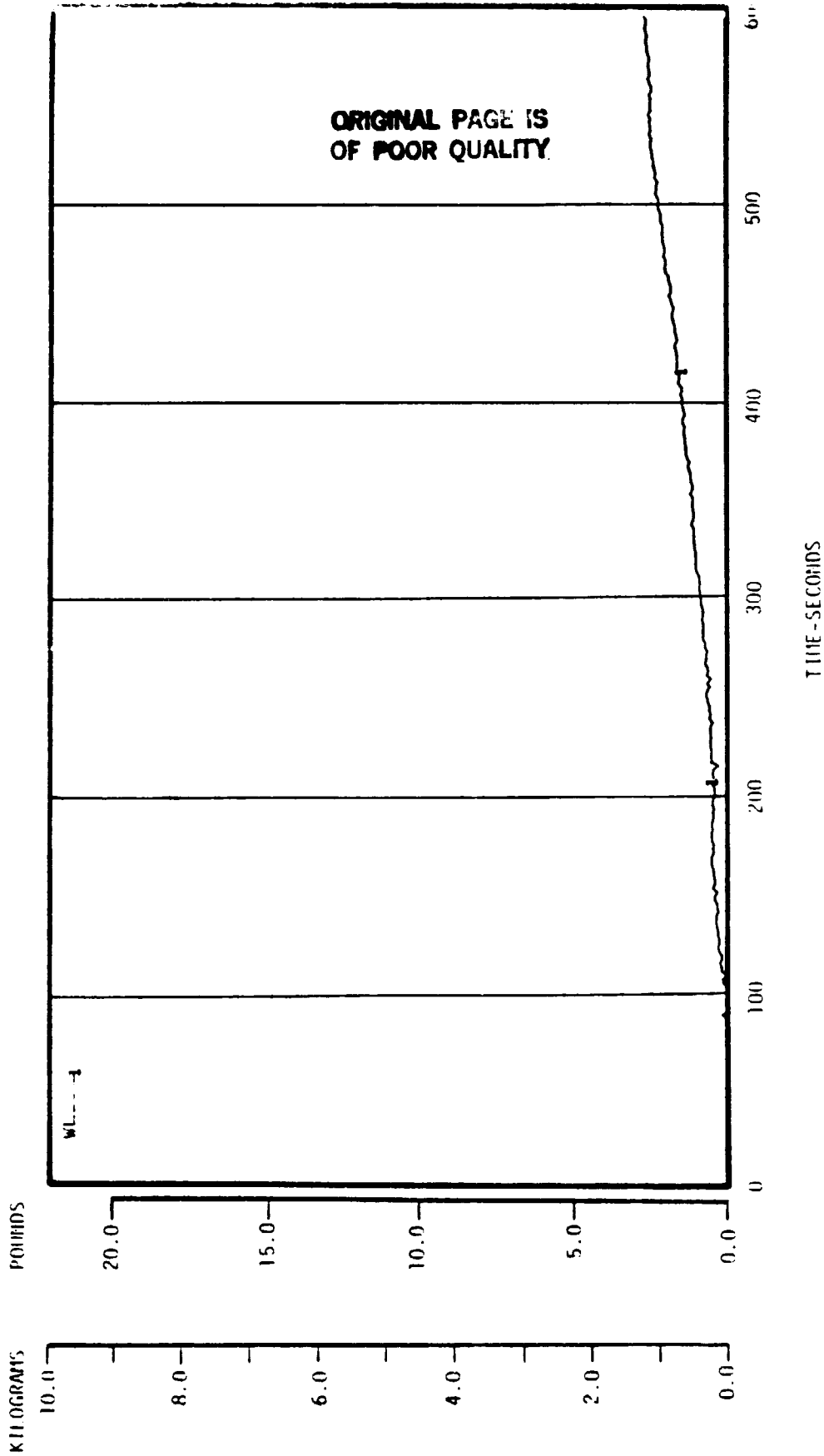


DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/12/82 00.02

NASA-AIES FULL SCALE CUSHION BURST TEST NUMBER 10

CUSHION CONSTRUCTION NUMBER 4.0

WEIGHT LOSS



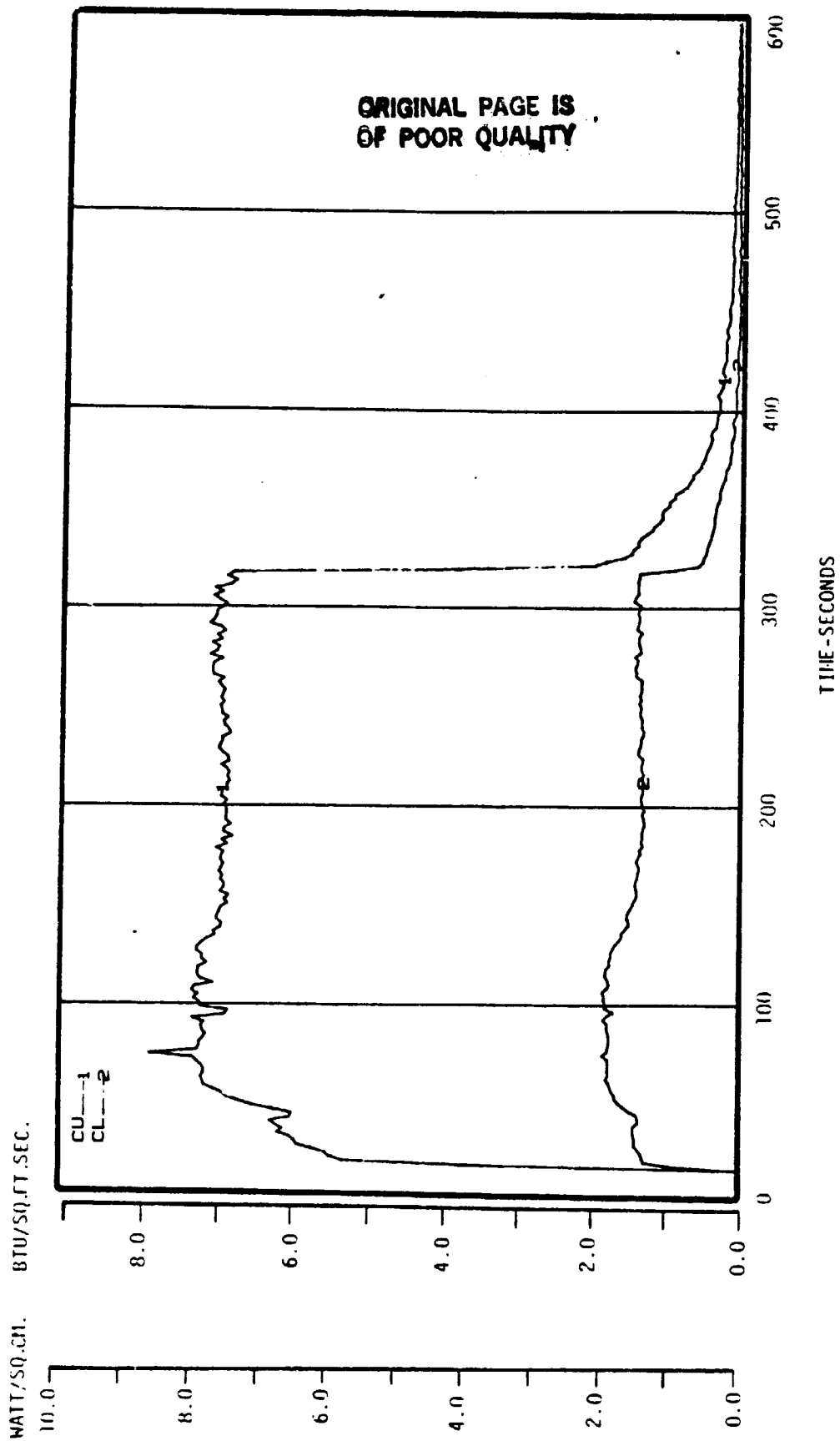
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OF POOR QUALITY

DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/12/82 09.02

NASA-AMES FULL SCALE CUSHION BURN TEST NUMBER 10

CUSHION CONSTRUCTION NUMBER 4.0

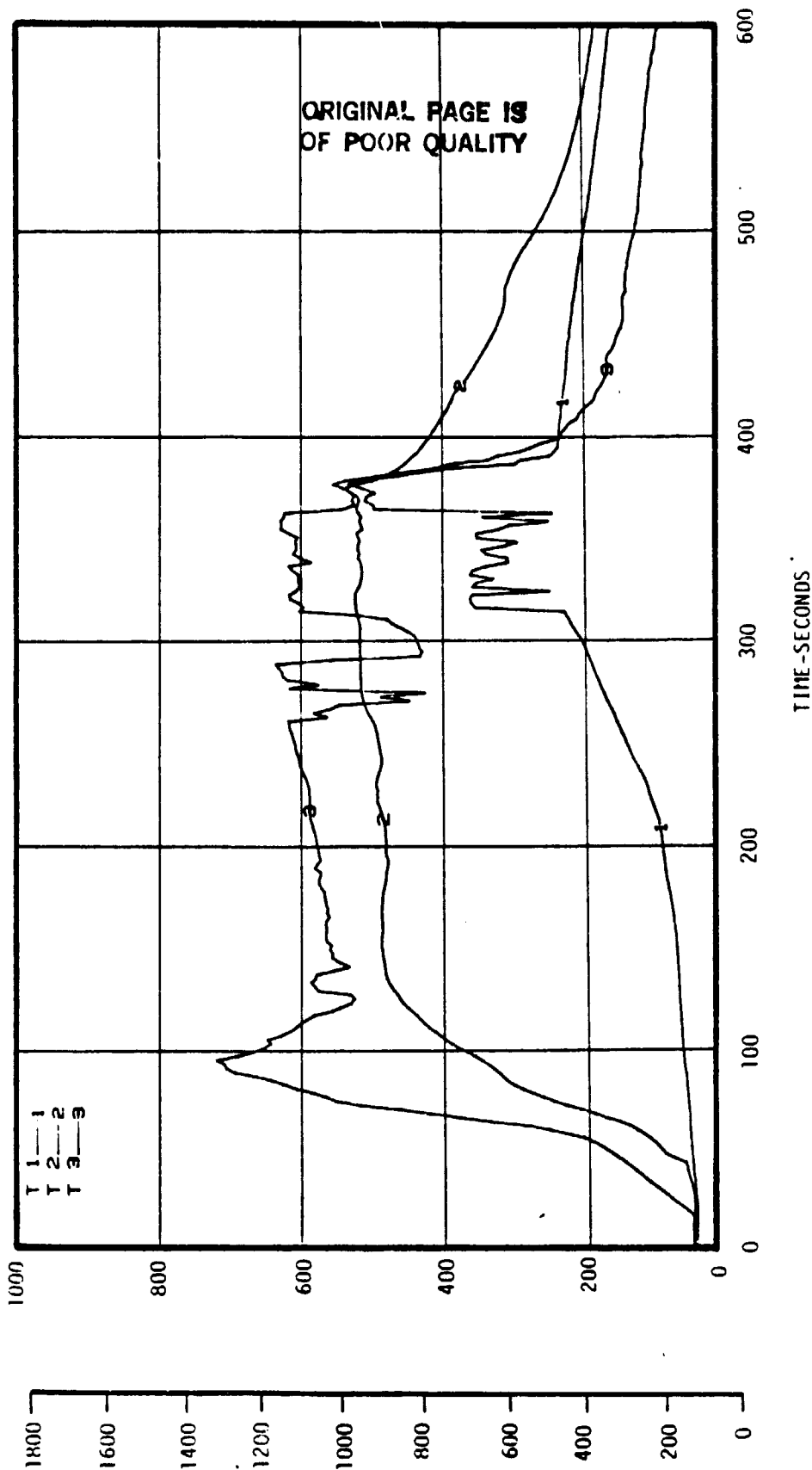
HEAT FLUX



DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/10/82 13.12
 NASA-MES. FULL SCALE CUSHION BURN TEST NUMBER 7
 CUSHION CONSTRUCTION NUMBER 5.0

SEAT CUSHION TEMPERATURES

FAHRENHEIT CELSIUS



DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 05/10/82 13.12

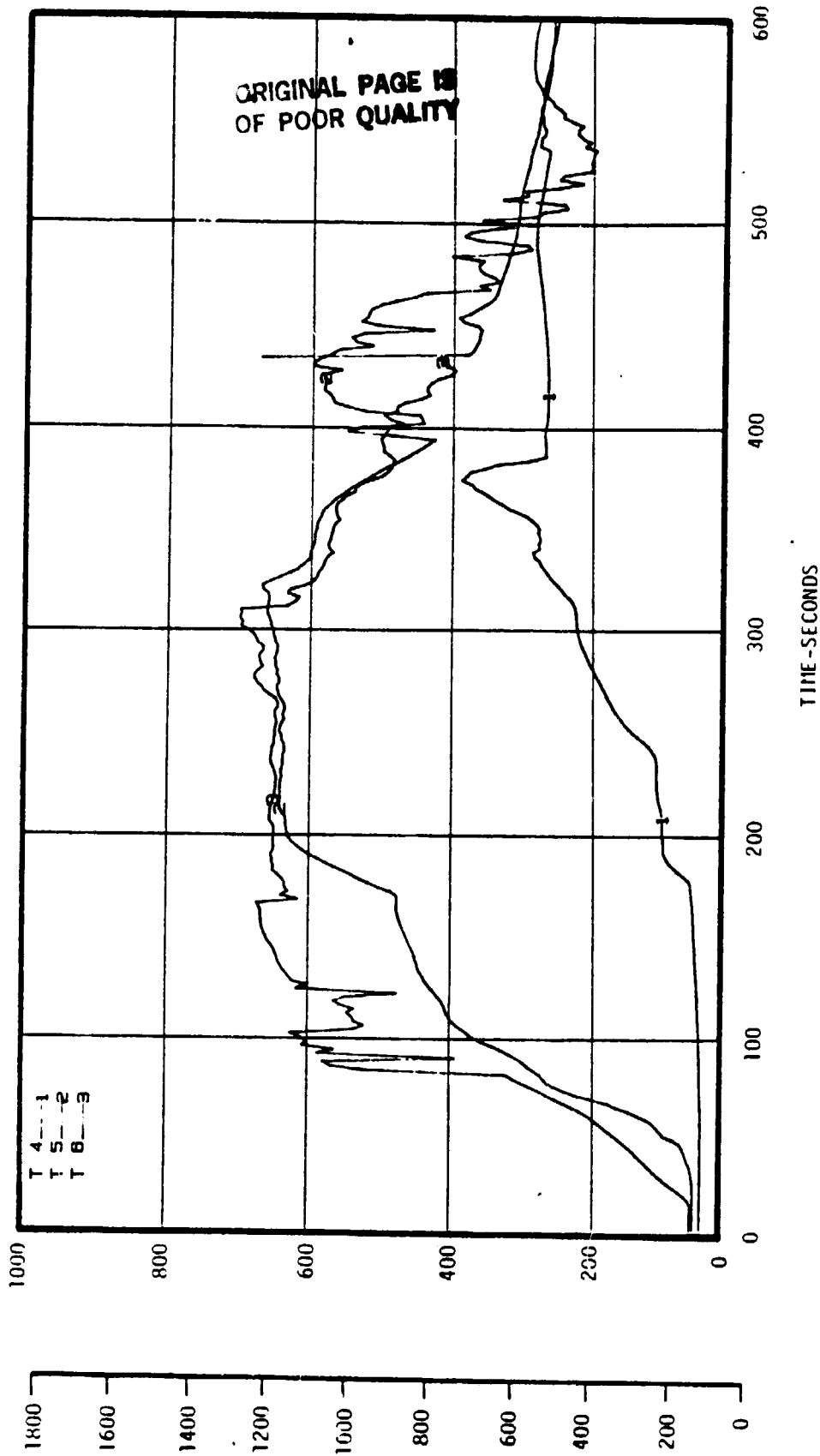
NASA-MES FULL SCALE CUSHION BURN TEST NUMBER 7

CUSHION CONSTRUCTION NUMBER 5.0

SEAT CUSHION TEMPERATURES

TEMPERATURE °F

TEMPERATURE °C

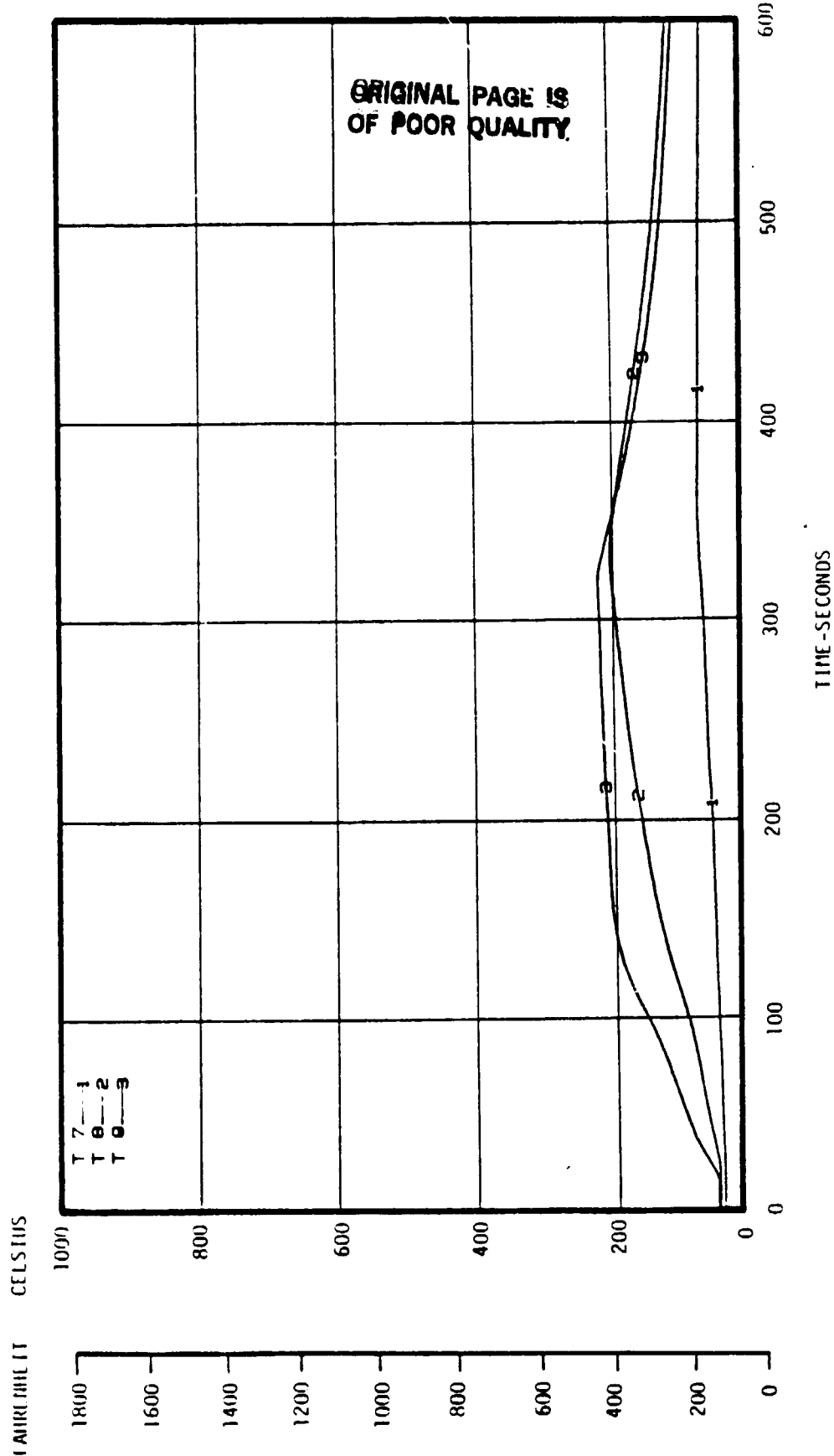


DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 13.12

NASA-AIES FULL SCALE CUSHION BURN TEST NUMBER 7

CUSHION CONSTRUCTION NUMBER 5.2

SEAT CUSHION TEMPERATURES

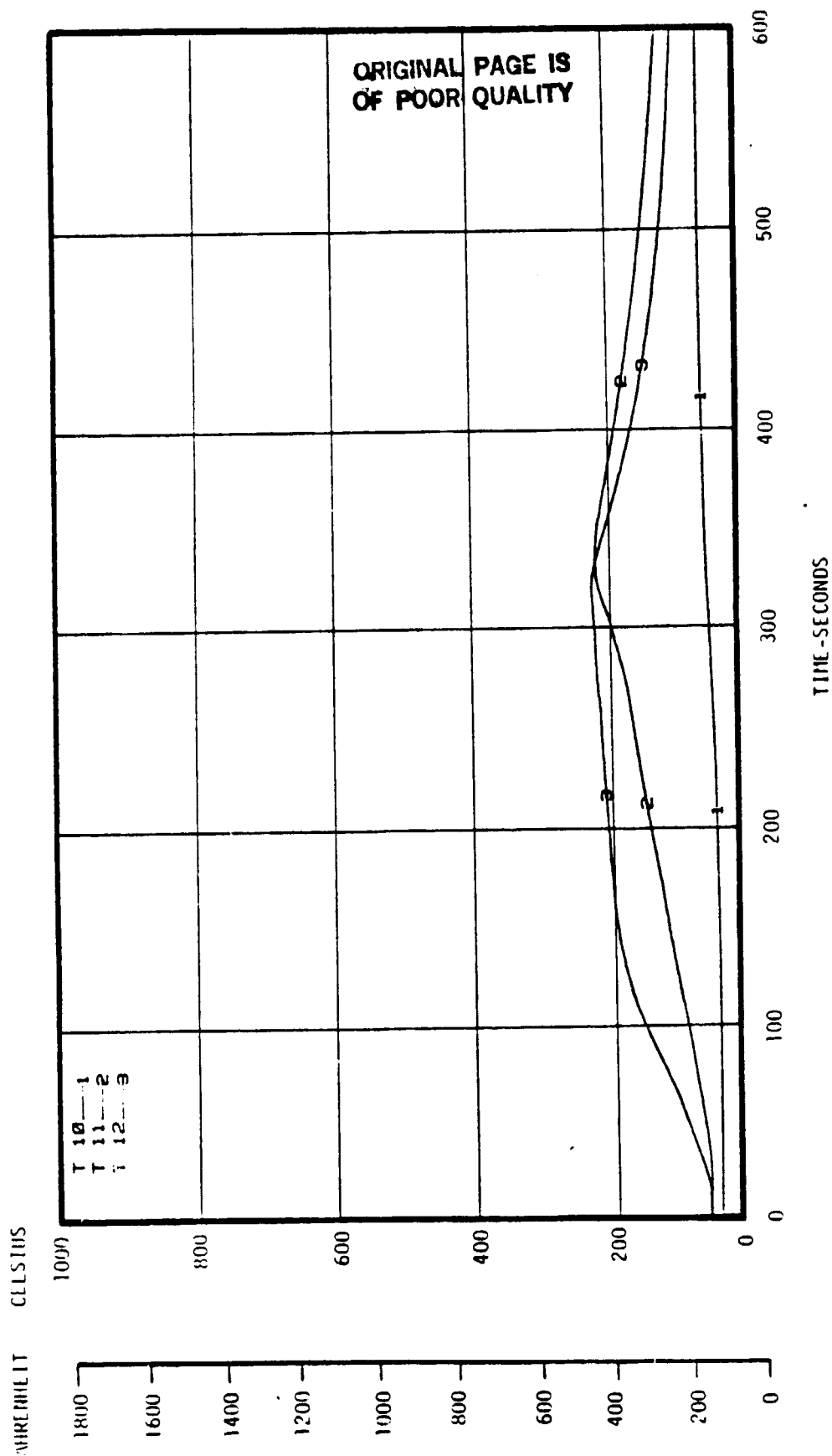


DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/10/82 13.12

NASA-AMES FULL SCALE CUSHION BURN TEST NUMBER 7

CUSHION CONSTRUCTION NUMBER 5.0

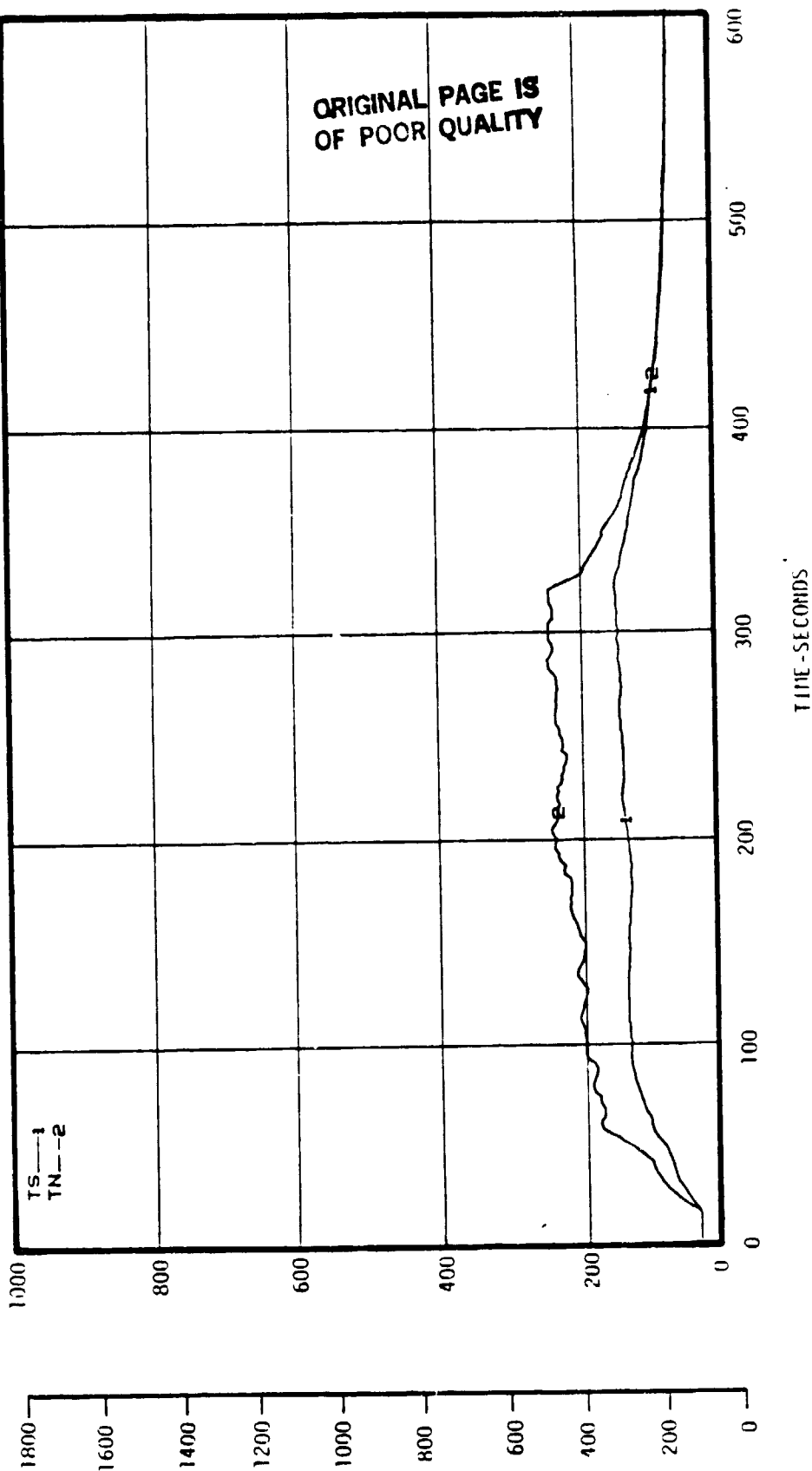
SEAT CUSHION TEMPERATURES



DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/10/82 13.12
 NASA-AMES FULL SCALE CUSHION BURN TEST NUMBER 7
 CUSHION CONSTRUCTION NUMBER 5.0

CEILING TEMPERATURE

FAHRENHEIT CELSIUS



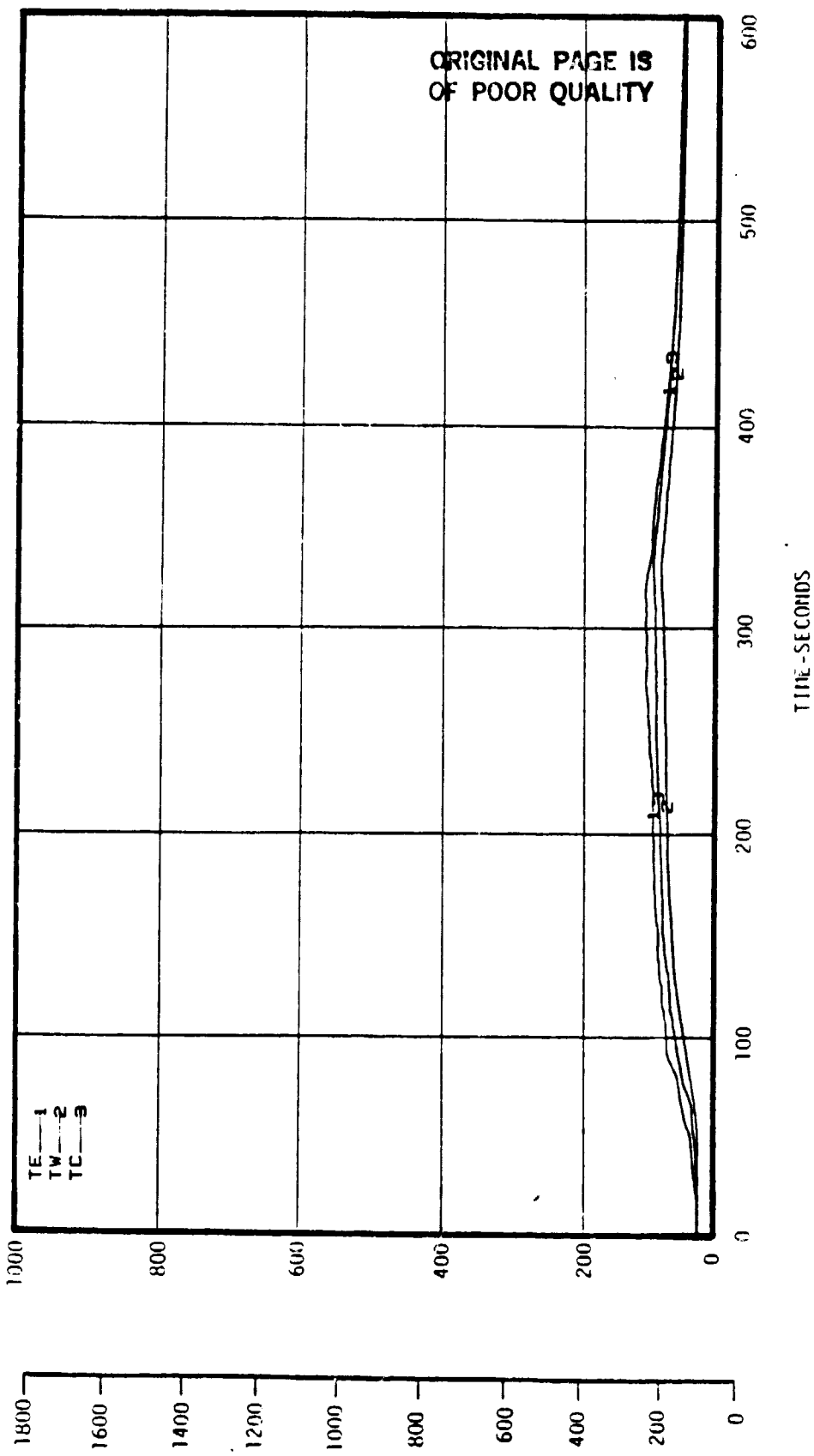
DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/10/82 13.12

NASA-AMES FULL SCALE CUSHION BURN TEST NUMBER 7

CUSHION CONSTRUCTION NUMBER 5.0

CEILING TEMPERATURE

FAHRENHEIT CELSIUS

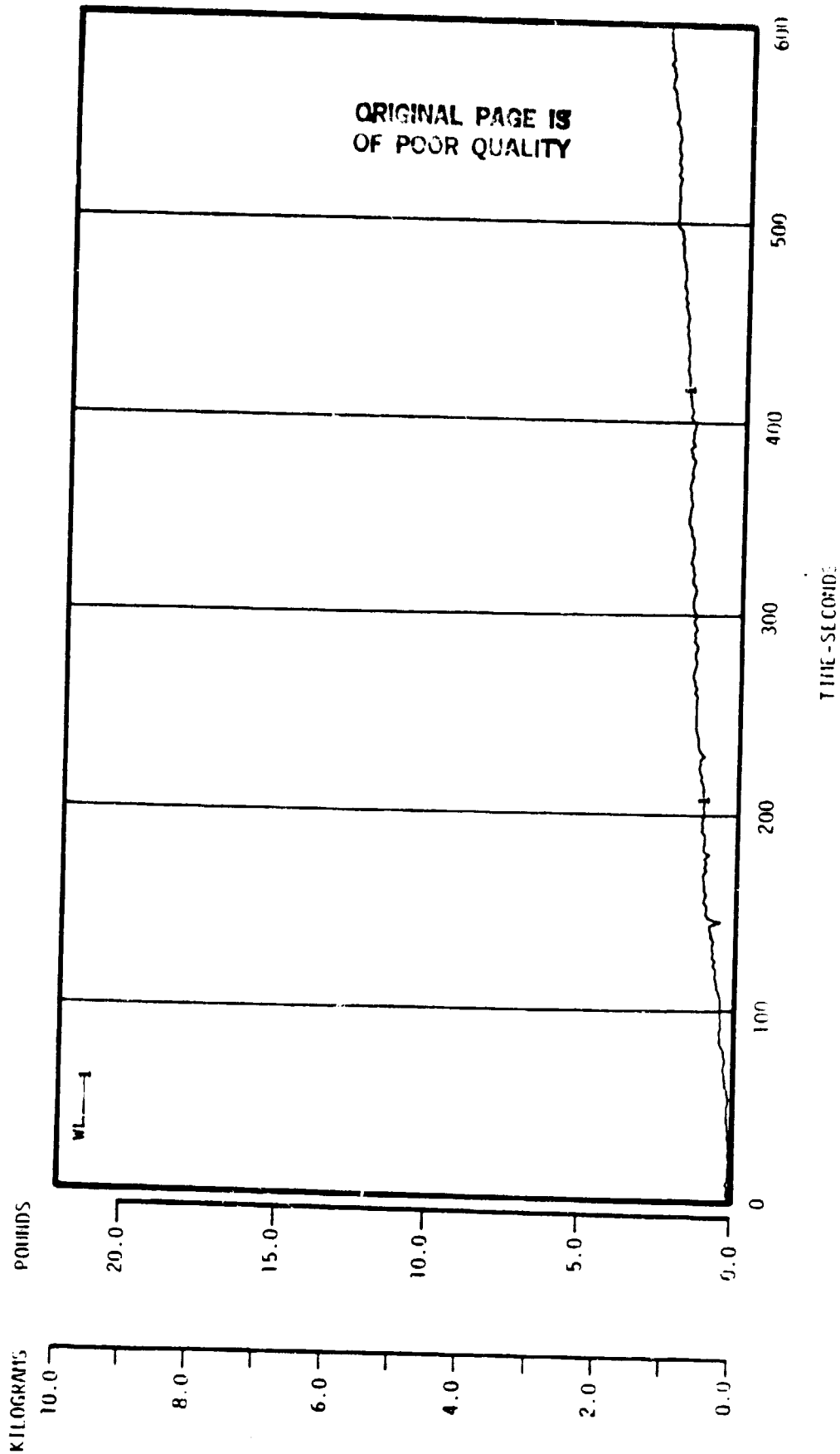


DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/10/02 13.12

NASA-AMES FULL SCALE CUSHION BURN TEST NUMBER 7

CUSHION CONSTRUCTION NUMBER 5.0

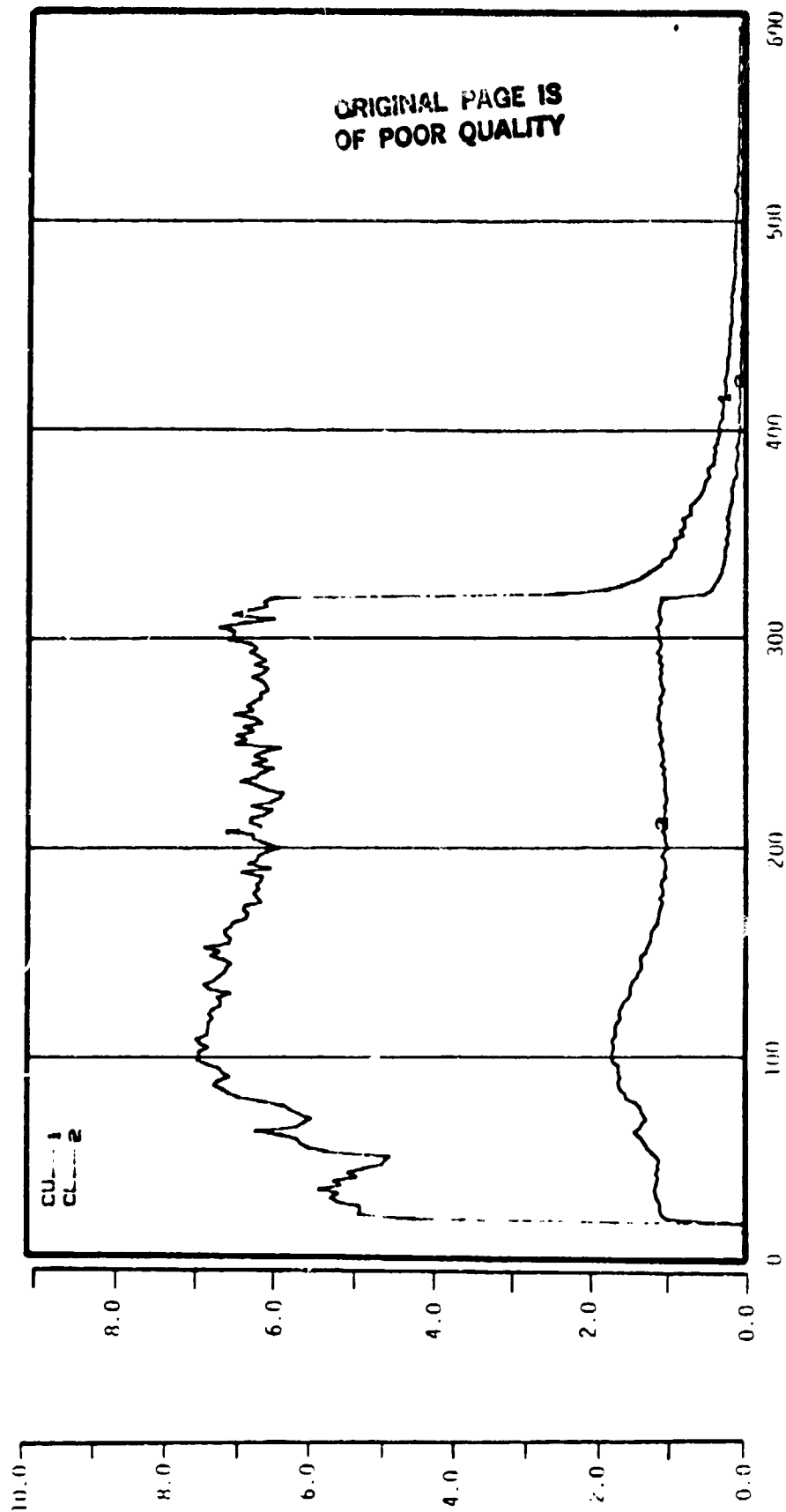
WEIGHT LOSS



DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 09/10/82 13.12
NASA-AMES FULL SCALE CUSHION BURN TEST NUMBER 7
CUSHION CONSTRUCTION NUMBER 5.0

HEAT FLUX

WATT/SQ. CM. BTU/SQ. FT. SEC.

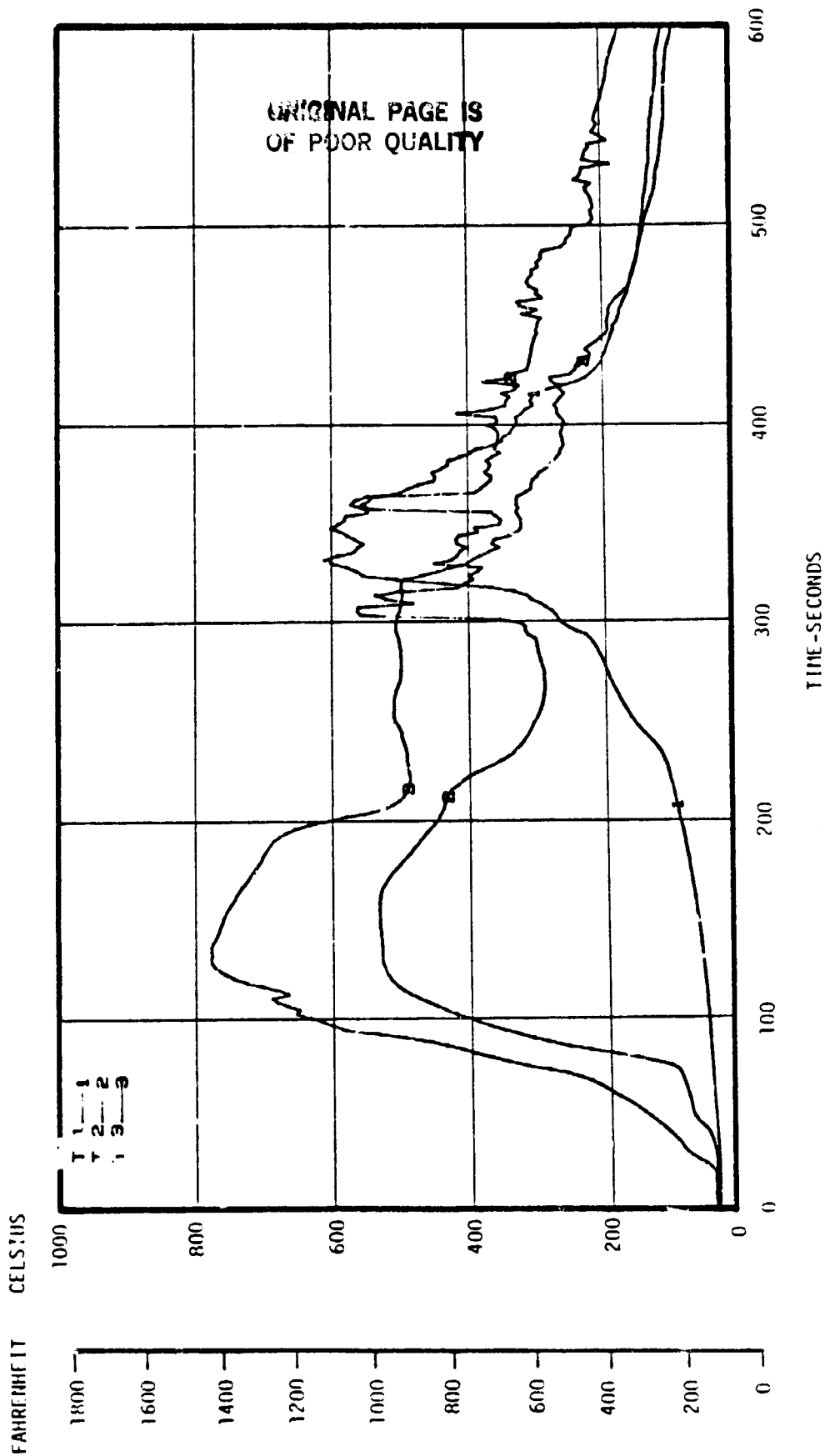


DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/15/82 00.12

NASA-AMES FULL SCALE CUSHION BURN TEST NUMBER 13

CUSHION CONSTRUCTION NUMBER 9.0

SEAT CUSHION TEMPERATURES

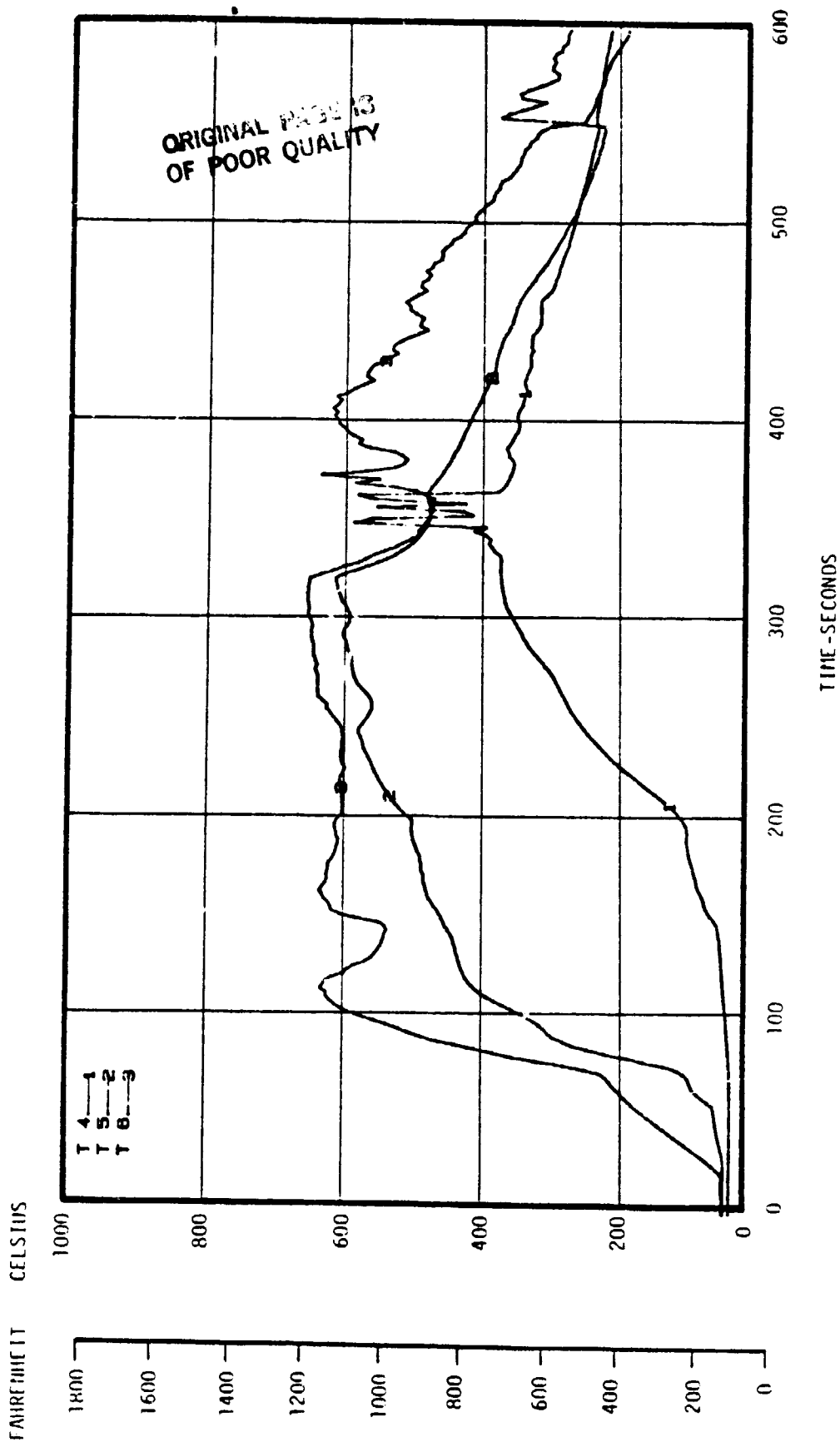


DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/13/82 00.12

NASA-AF'S FULL SCALE CUSHION BURN TEST NUMBER 13

CUSHION CONSTRUCTION NUMBER 5.0

SEAT CUSHION TEMPERATURES

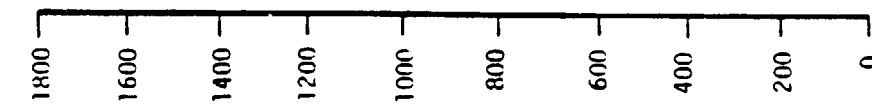


DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 09/15/82 00.12
 NASA-MES FULL SCALE CUSHION BURN TEST NUMBER 13
 CUSHION CONSTRUCTION NUMBER 5.0

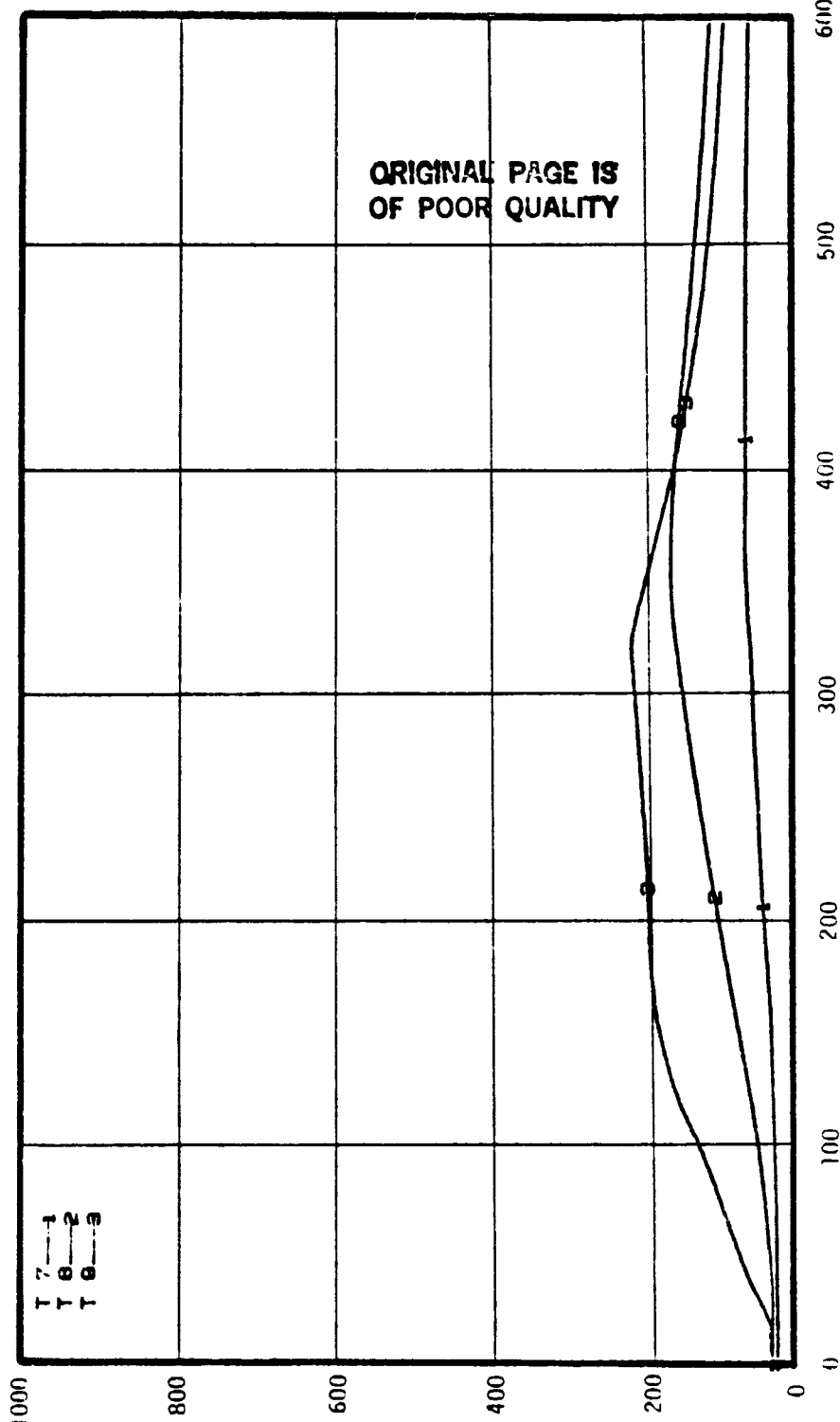
SEAT CUSHION TEMPERATURES

FAHRENHEIT

CELSIUS



T 7---1
T 8---2
T 9---3



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TIME-SECONDS

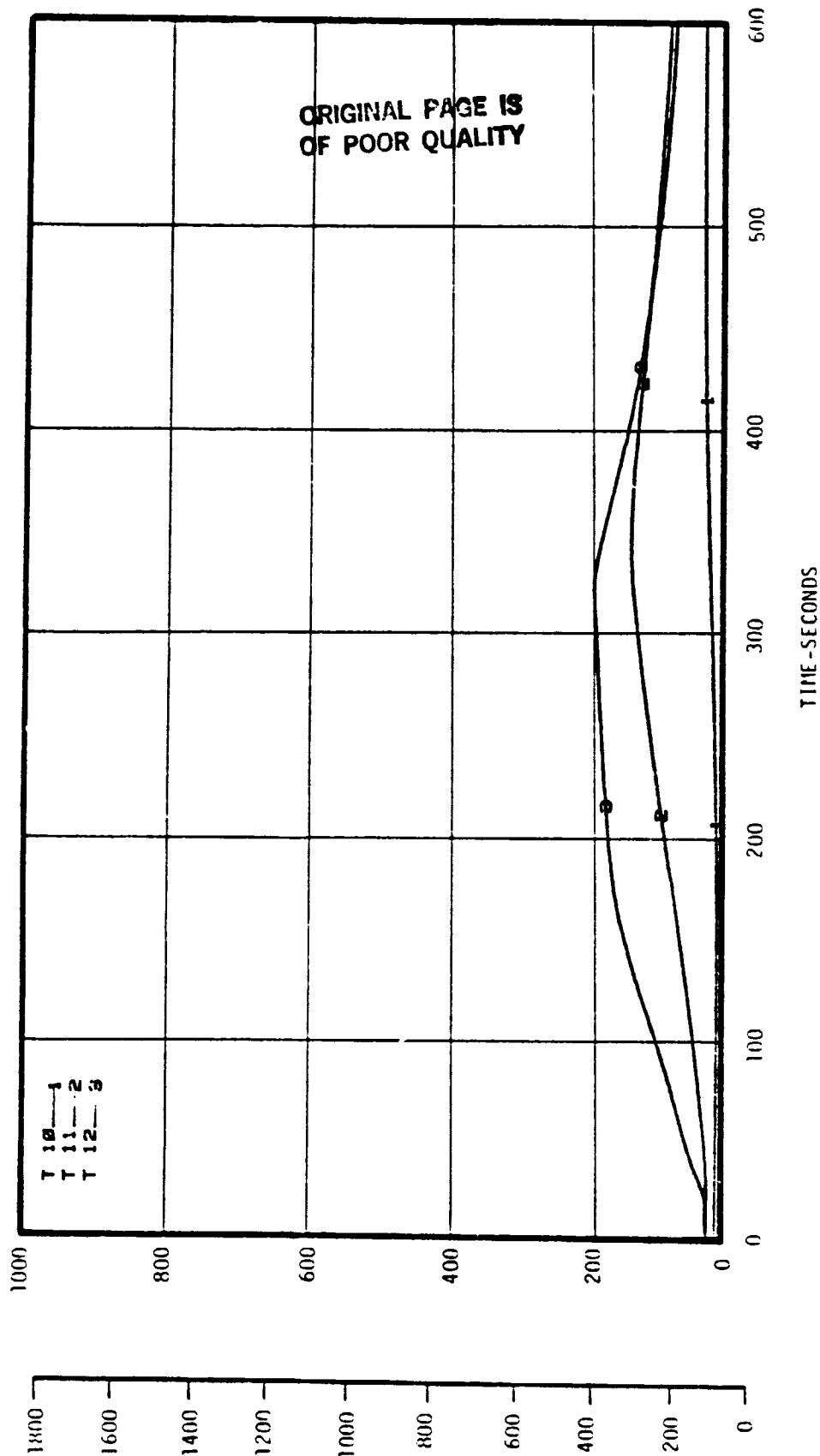
DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/15/82 06.12

NASA-AMES FULL SCALE CUSHION BURN TEST NUMBER 13

CUSHION CONSTRUCTION NUMBER 5.0

SEAT CUSHION TEMPERATURES

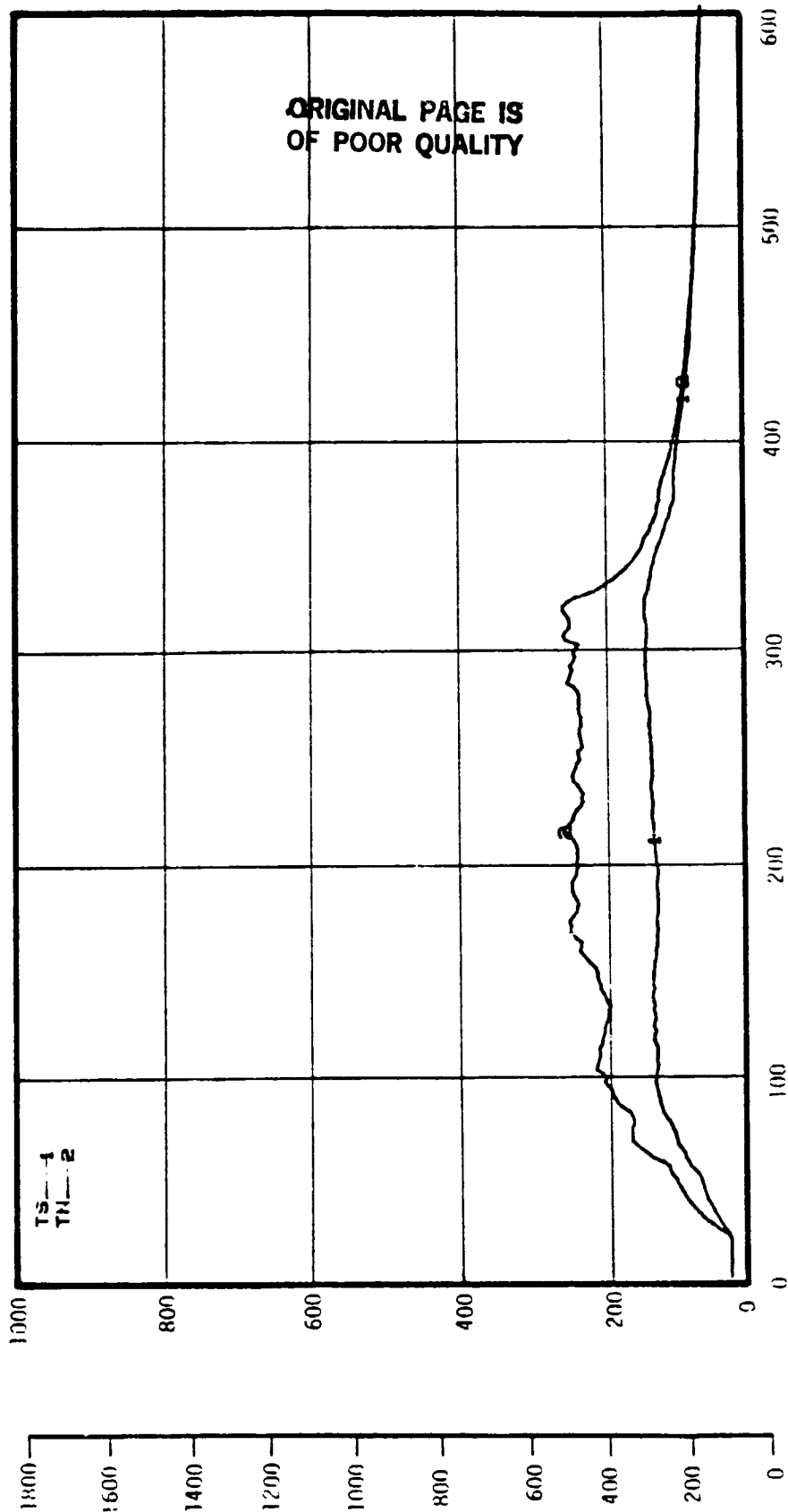
FAHRENHEIT CELSIUS



DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/15/62 00.12
 NASA-AMES FULL SCALE CUSHION BURN TEST NUMBER 13
 CUSHION CONSTRUCTION NUMBER 5.0

CEILING TEMPERATURE

FAHRENHEIT CELSIUS



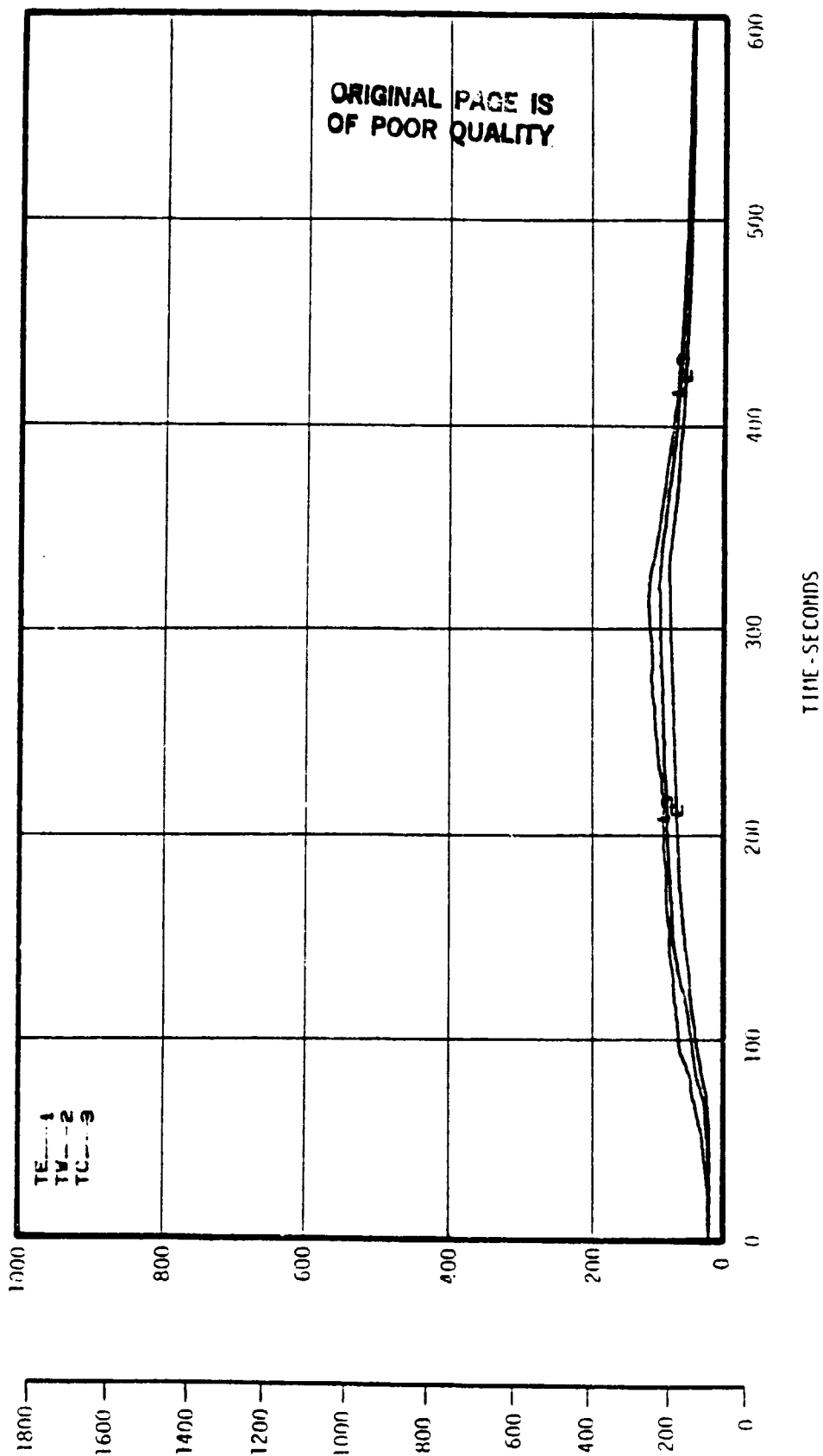
DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/15/82 09.12

NASA-AHS FULL SCALE CUSHION BURN TEST NUMBER 13

CUSHION CONSTRUCTION NUMBER 5.0

CEILING TEMPERATURE

FAHRENHEIT CELSIUS

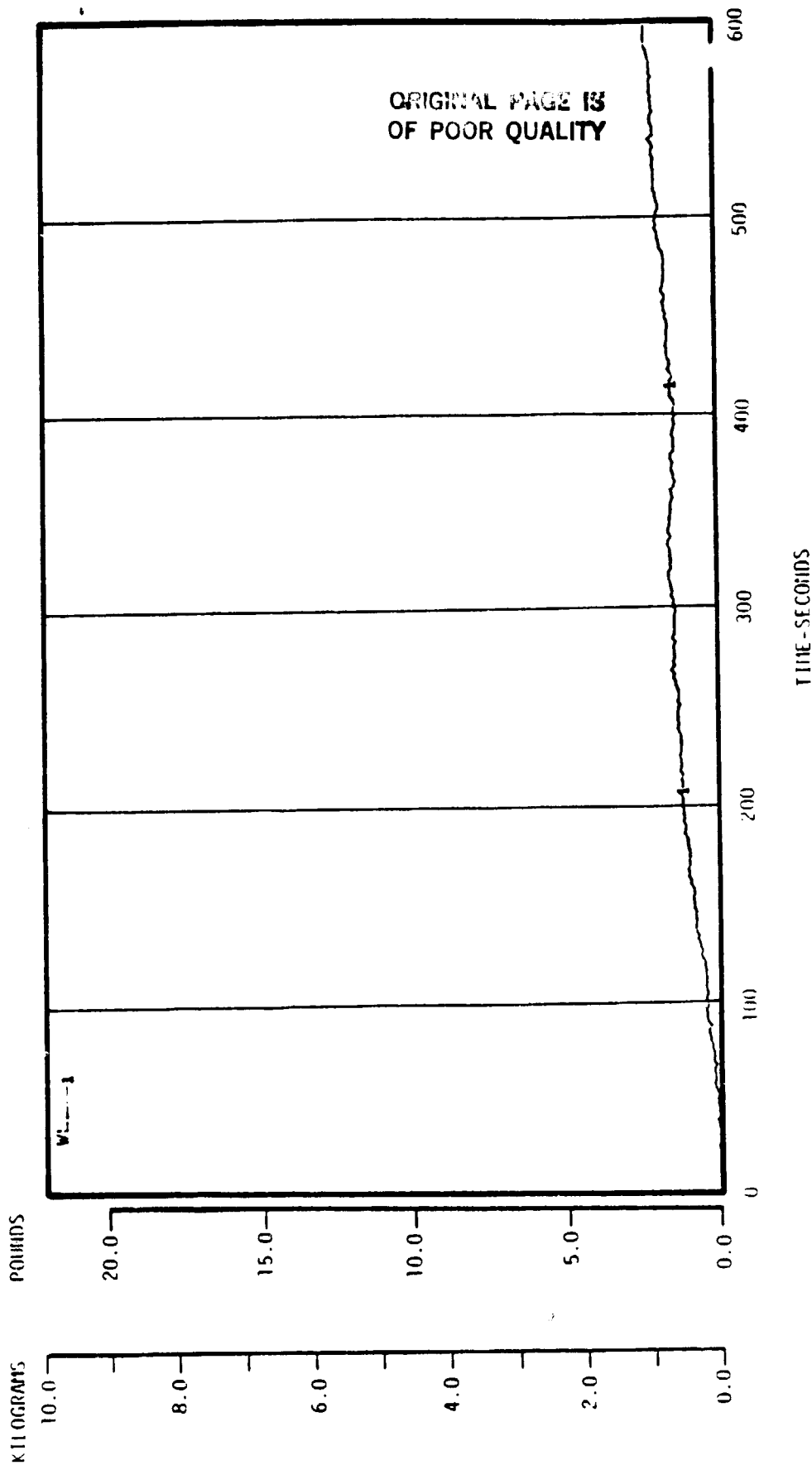


DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/15/82 00.12

NASA-AHS FULL SCALE CUSHION BURN TEST NUMBER 13

CUSHION CONSTRUCTION NUMBER 5.0

WEIGHT LOSS



ORIGINAL AS AIRCRAFT CABIN FIRE SIMULATOR 03/15/82 28.12

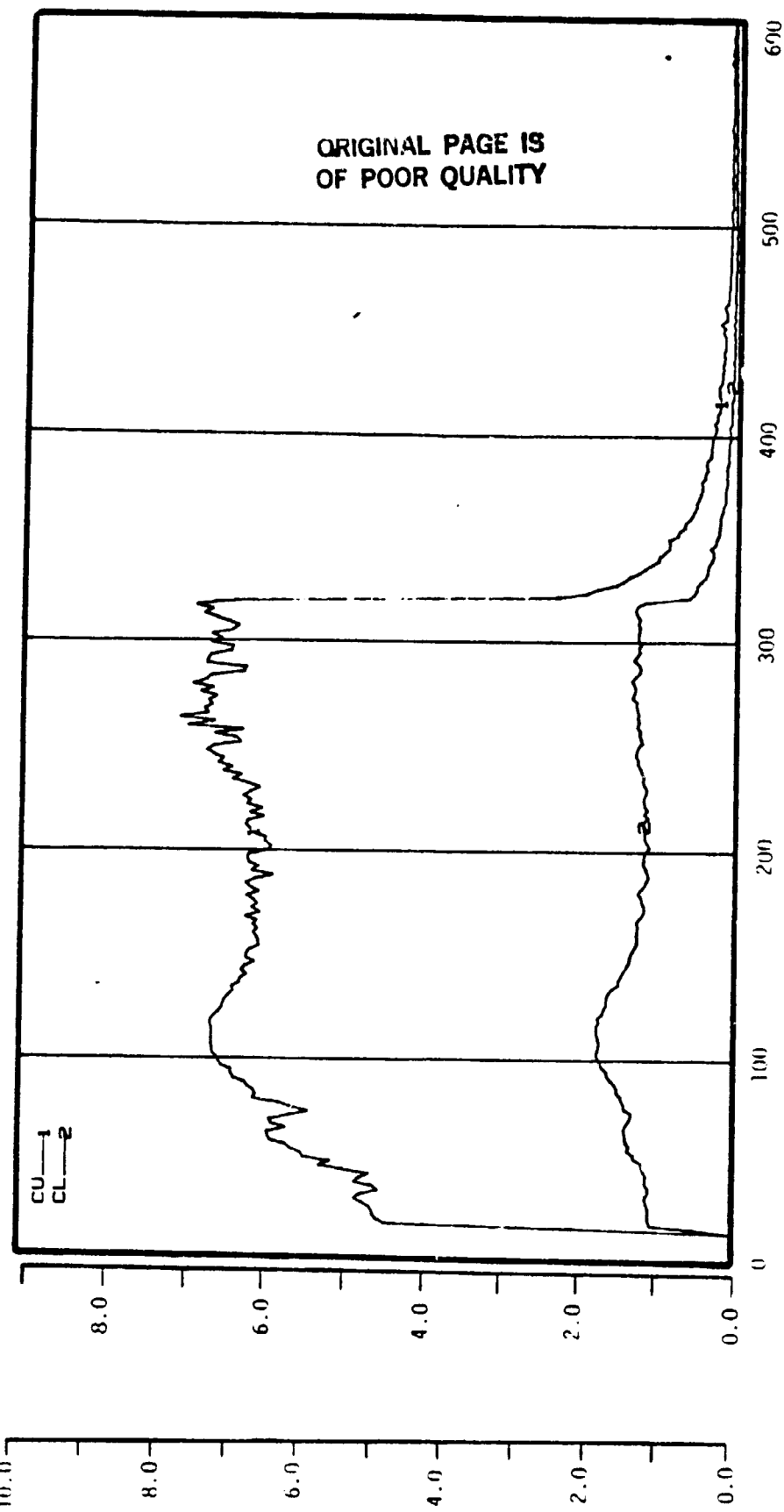
NASA-AMES FULL SCALE CUSHION BURN TEST NUMBER 13

CUSHION CONSTRUCTION NUMBER 3.0

HEAT FLUX

BTU/SQ.FT.SEC.

WATT/SQ.CM.



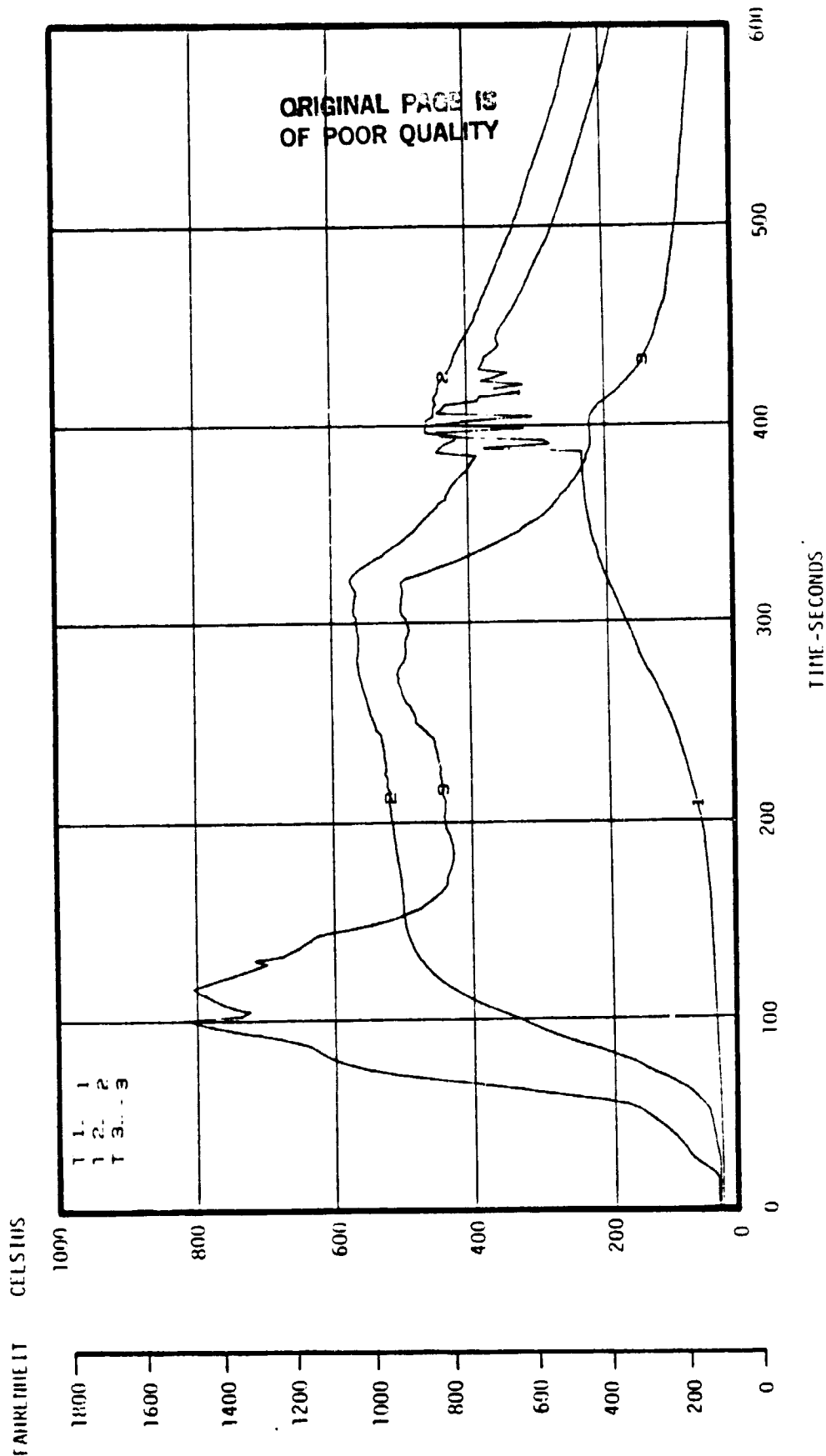
TIME-SECONDS

DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/00/82 14.28

NASA-AMES FULL SCALE CUSHION BURN TEST NUMBER 5

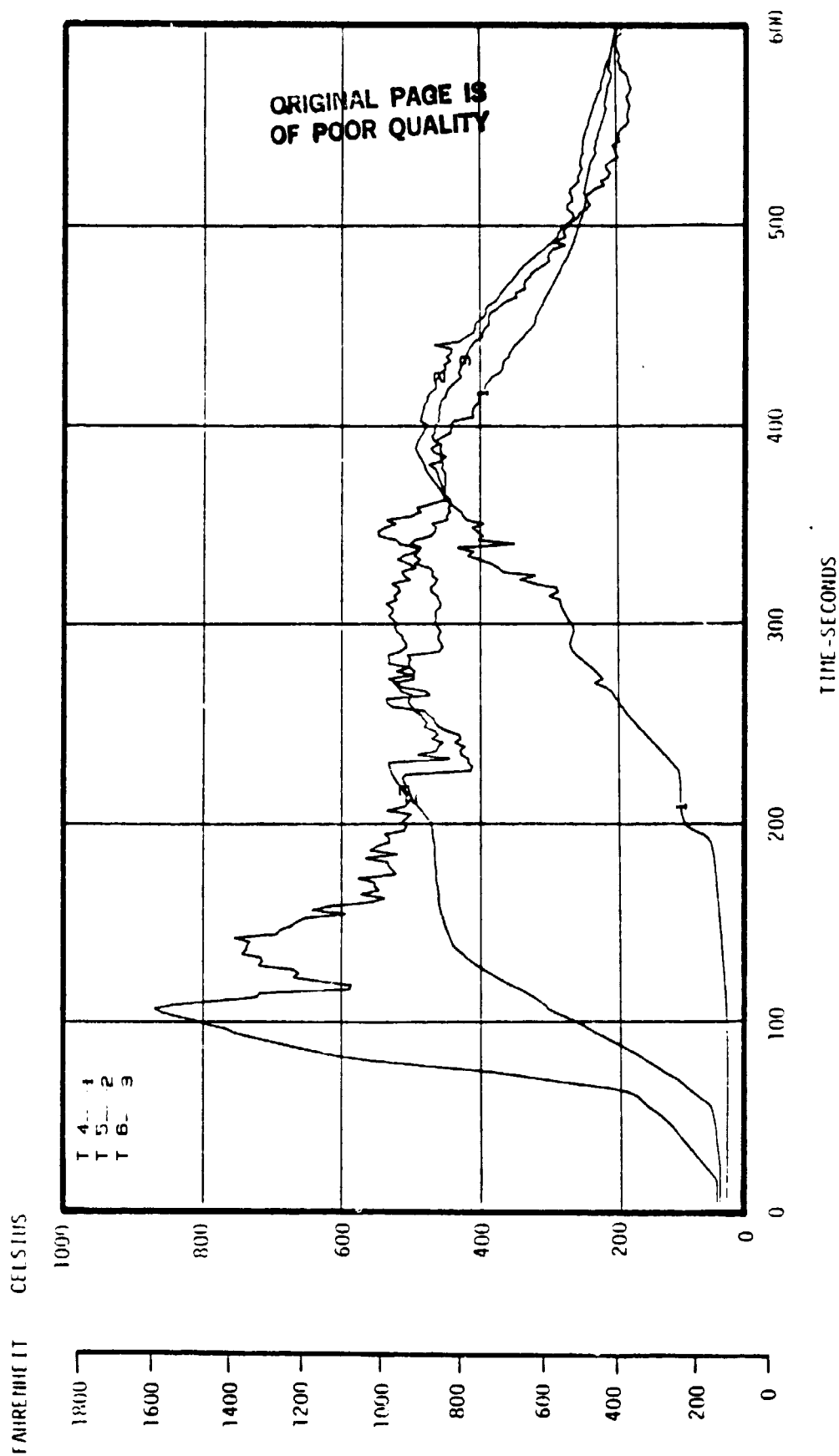
CUSHION CONSTRUCTION NUMBER 6.0

SEAT CUSHION TEMPERATURES



DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/09/82 14.28
NASA-AMES FULL SCALE CUSHION BURN TEST NUMBER 5
CUSHION CONSTRUCTION NUMBER 6.0

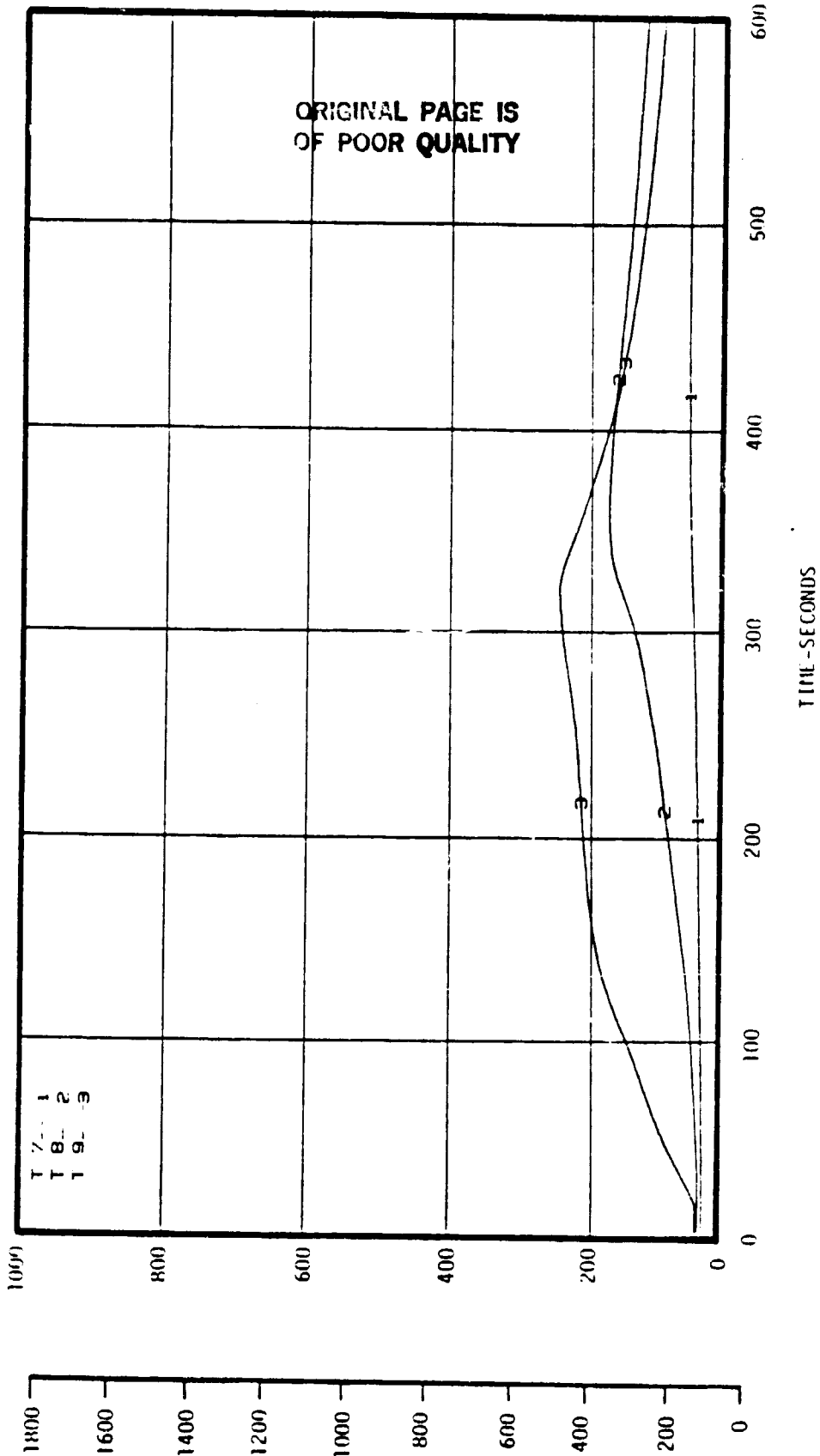
SEAT CUSHION TEMPERATURES



DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/09/82 14:28
NASA-AMES FULL SCALE CUSHION BURN TEST NUMBER 5
CUSHION CONSTRUCTION NUMBER 6.0

SEAT CUSHION TEMPERATURES

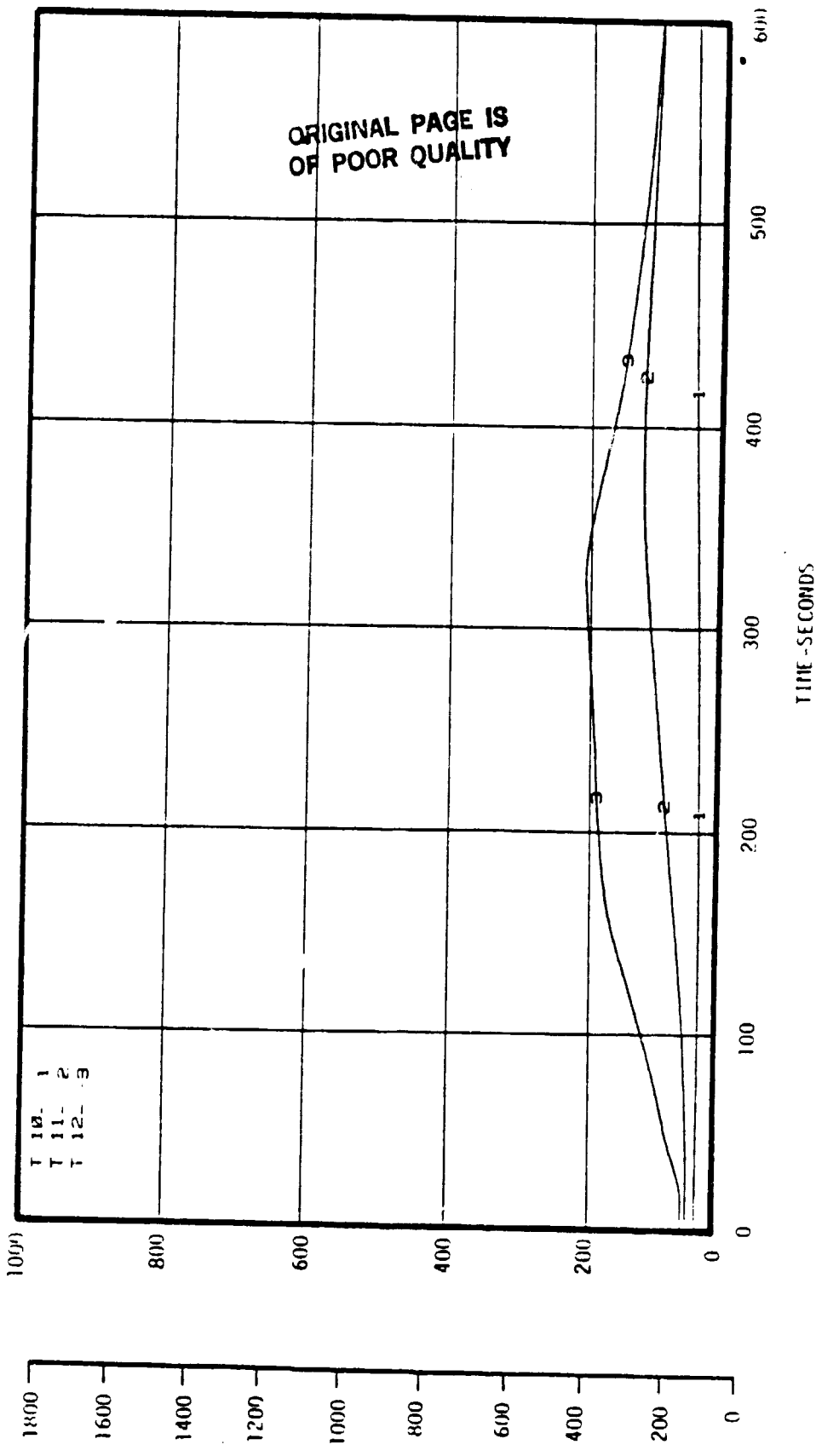
FAHRENHEIT CELSIUS



DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/09/82 14:20
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 CUSHION CONSTRUCTION NUMBER 6.0

SEAT CUSHION TEMPERATURES

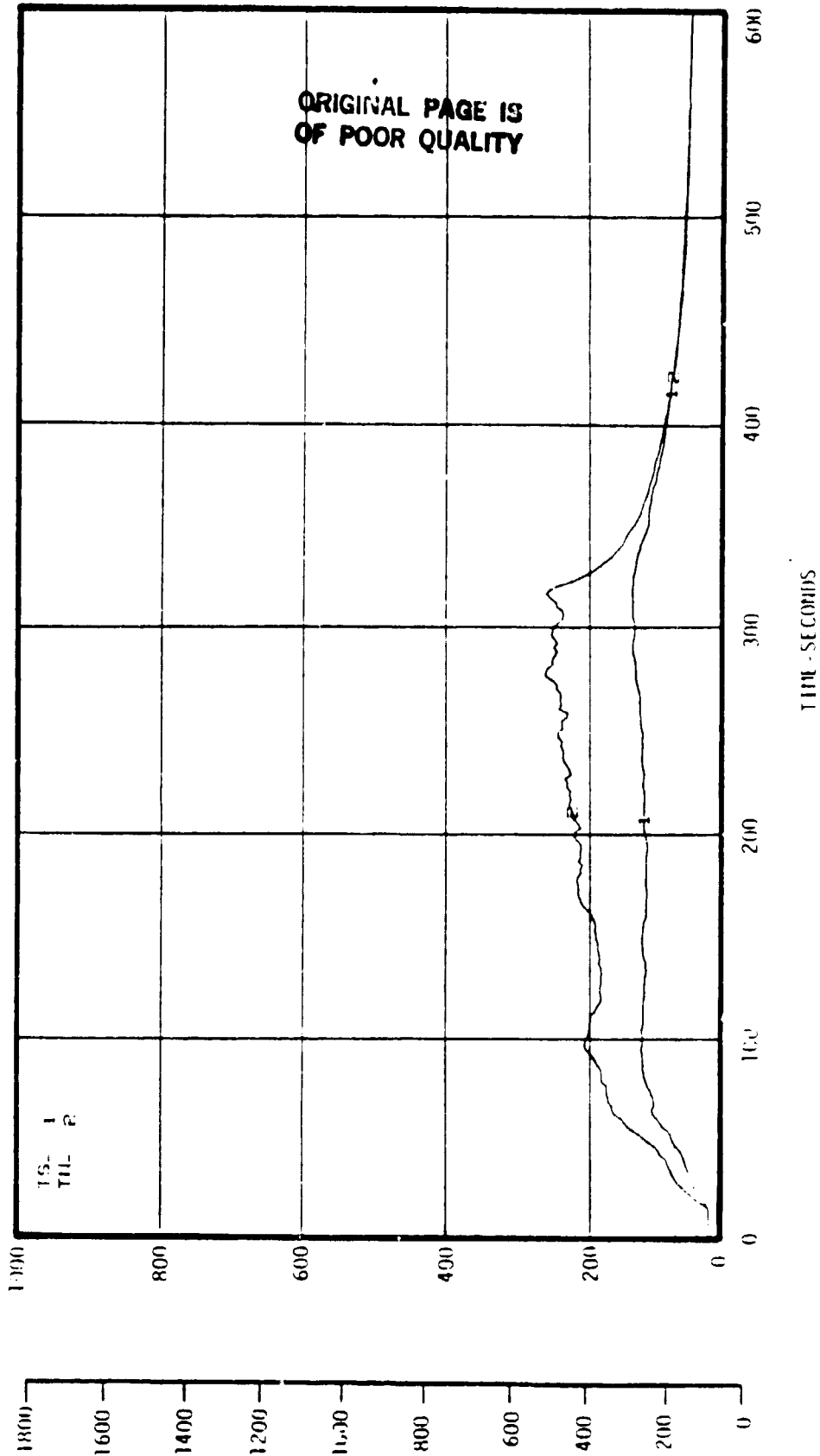
FAHRENHEIT CELSIUS



DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/09/82 14:28
 NASA-AIES FULL SCALE CUSHION BURN TEST NUMBER 5
 CUSHION CONSTRUCTION NUMBER 6.0

CEILING TEMPERATURE

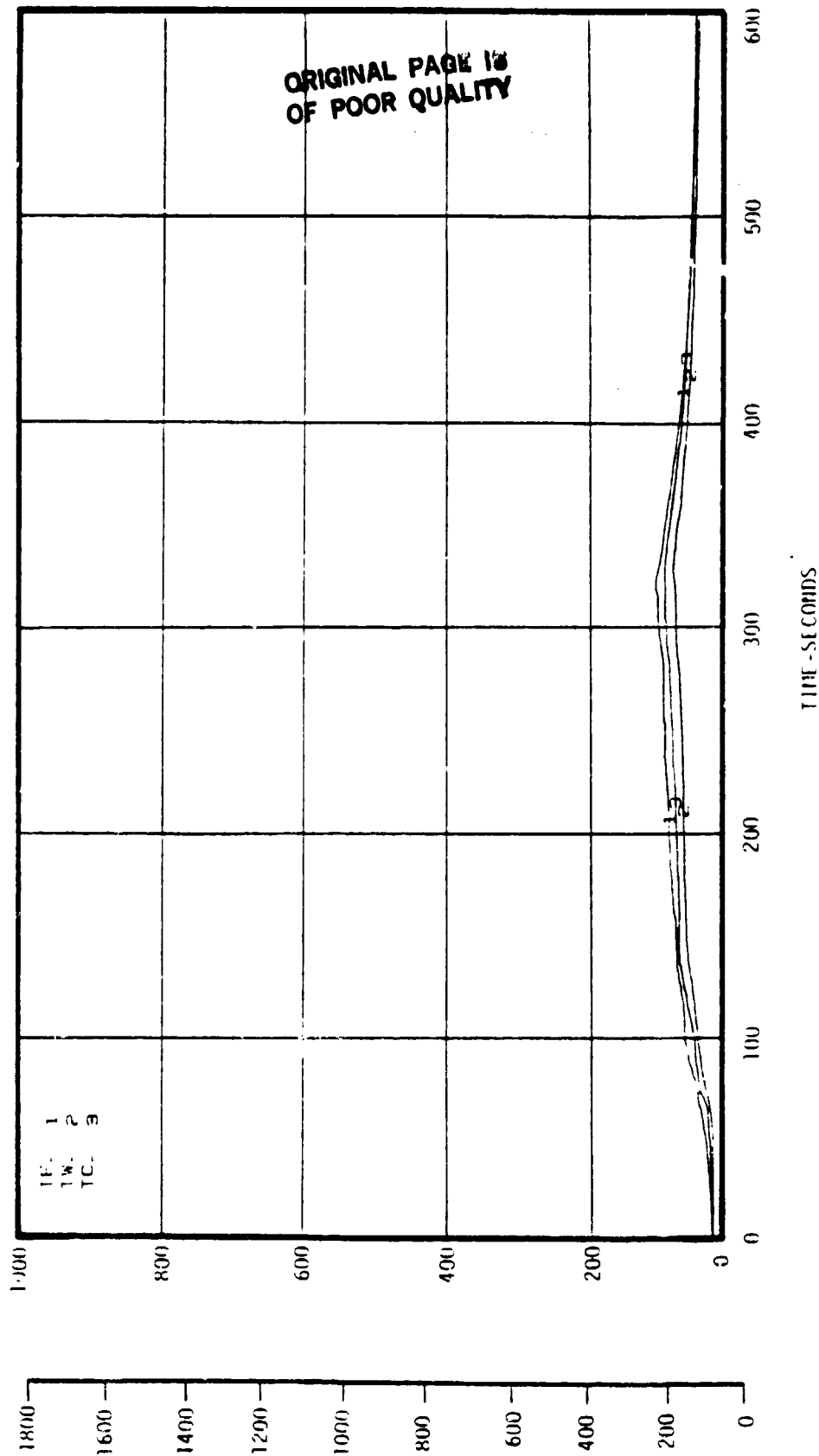
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DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/09/82 14.28
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CEILING TEMPERATURE

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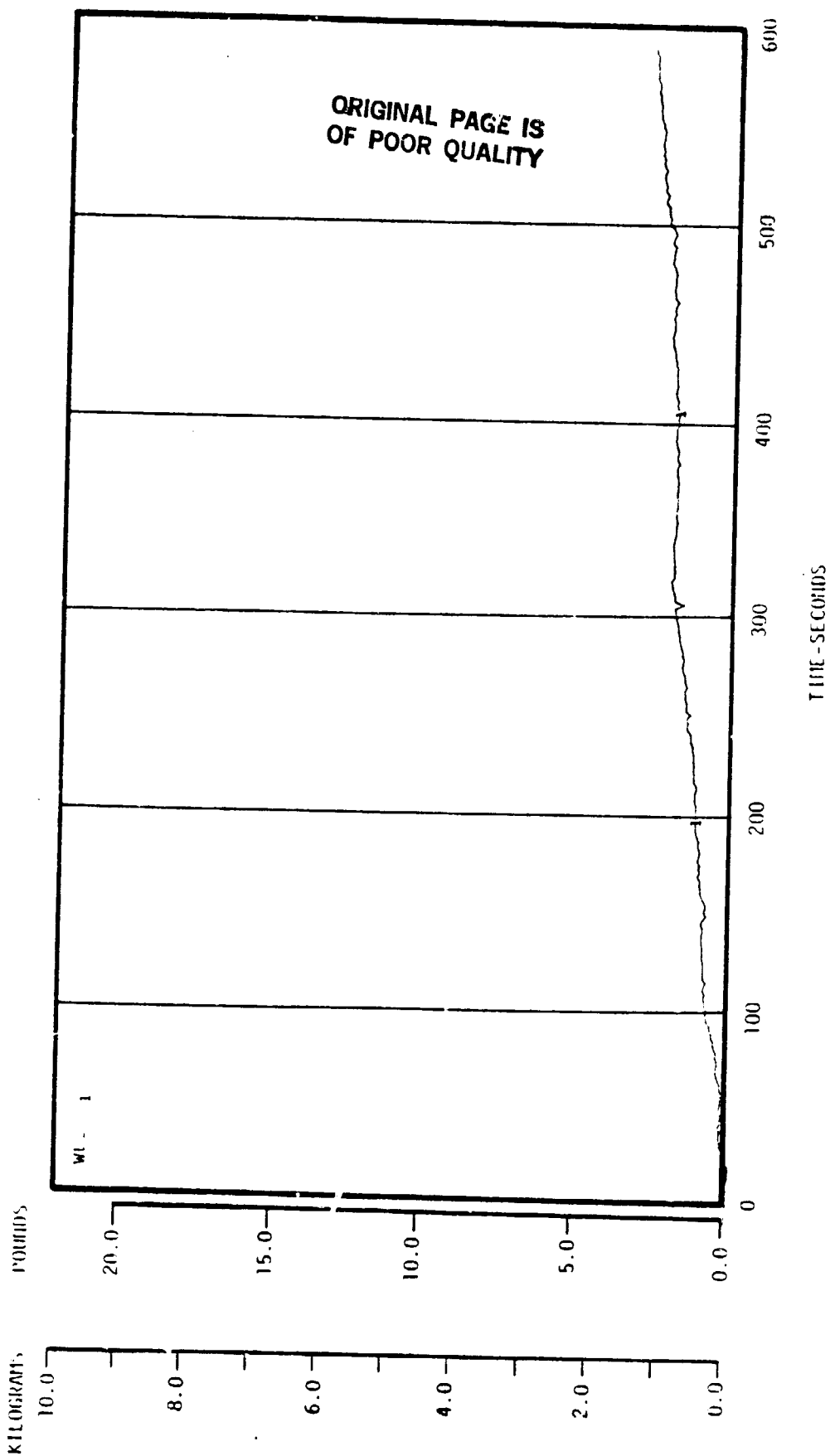


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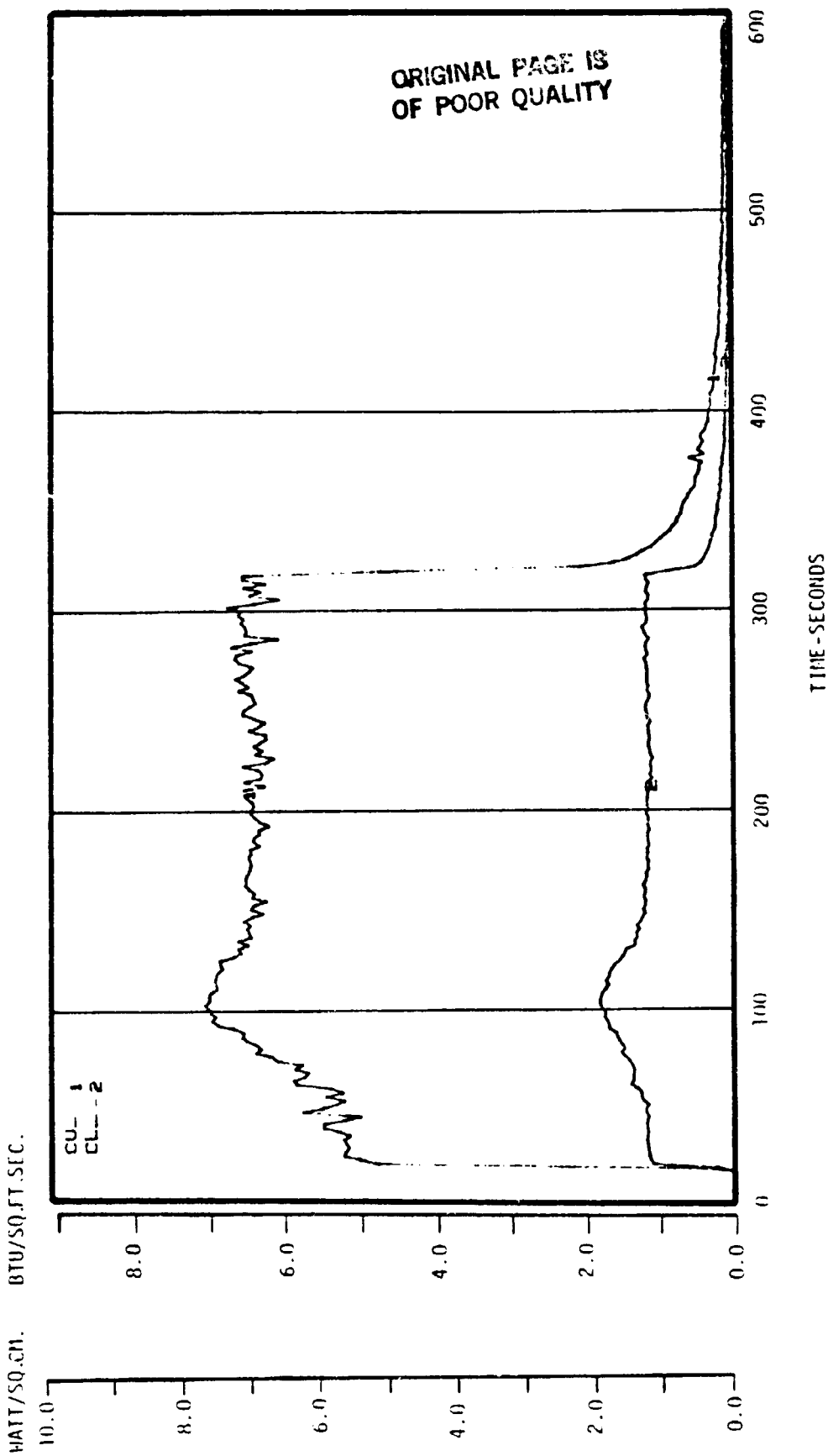
CUSHION CONSTRUCTION NUMBER 6.0

WEIGHT LOSS



DOUGLAS AIRCRAFT CABIN FIRE SIMULATION 89/09/82 14:28
NASA-AMES FULL SCALE CUSHION BURN TEST NUMBER 5
CUSHION CONSTRUCTION NUMBER 6.0

HEAT FLUX



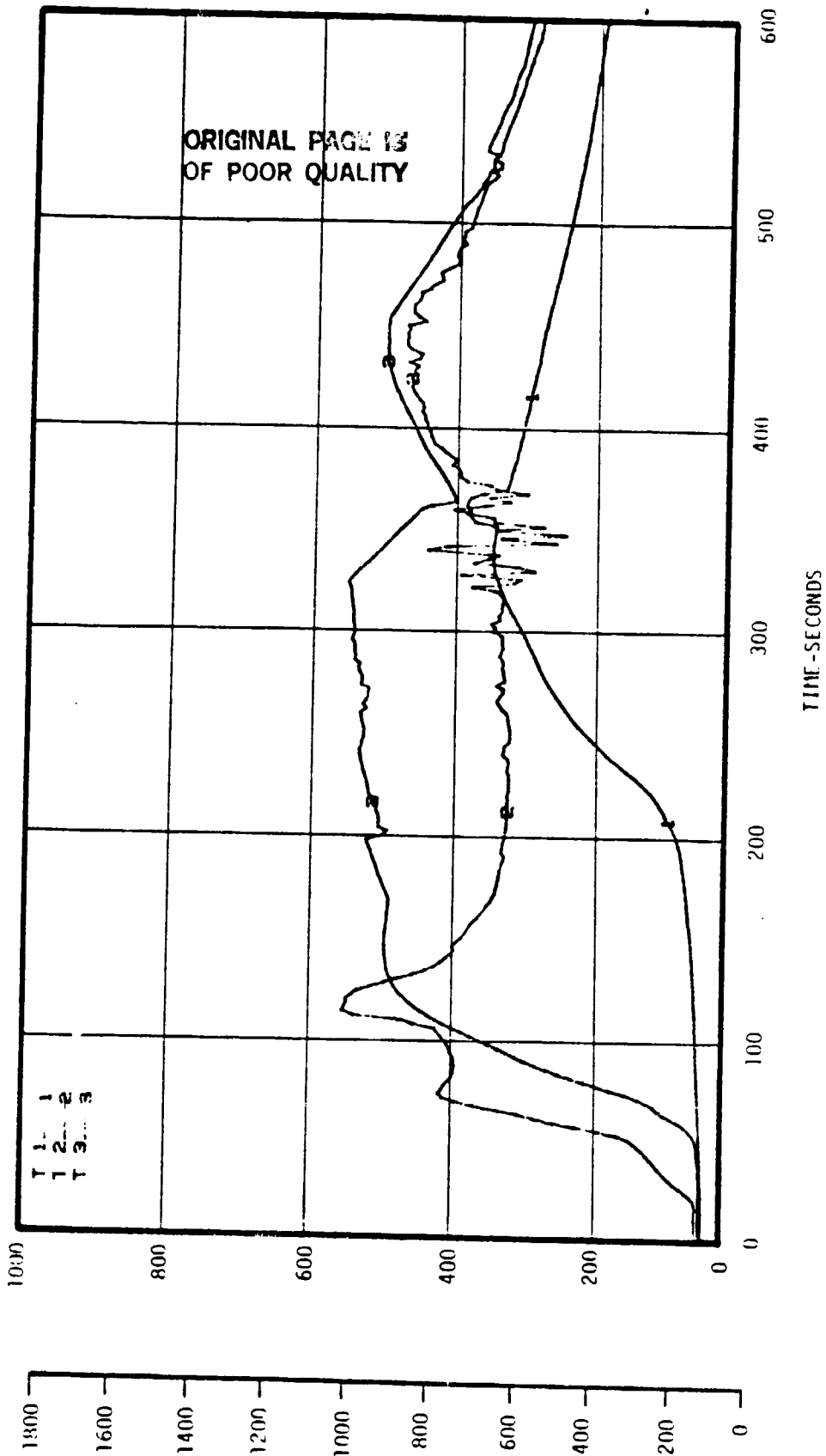
DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 23/15/62 11.43

NASA-AMES FULL SCALE CUSHION BURN TEST NUMBER 14

CUSHION CONSTRUCTION NUMBER 6.0

SEAT CUSHION TEMPERATURES

FAHRENHEIT CELSIUS

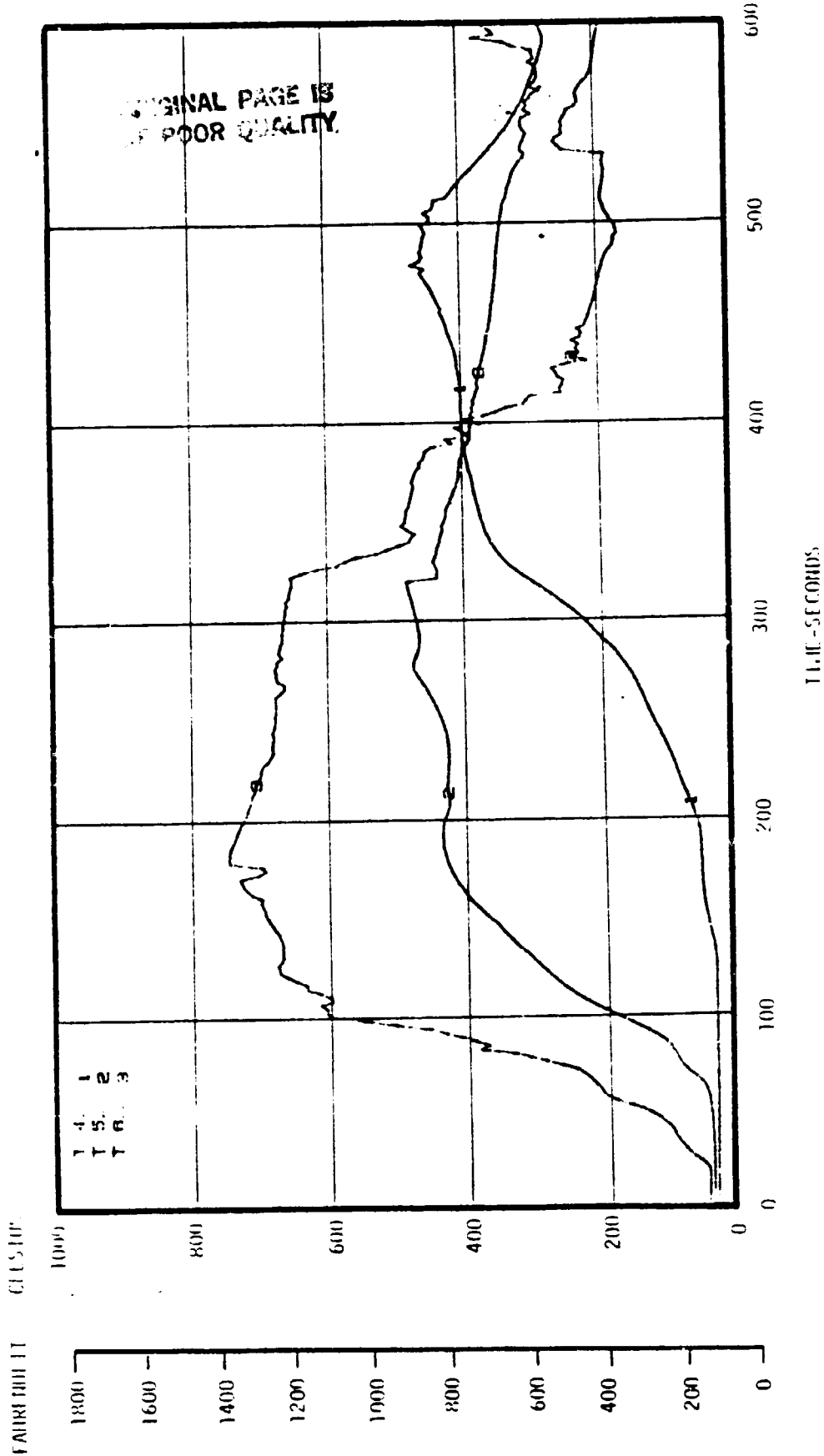


DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/15/82 11.40

NASA-AMES FULL SCALE CUSHION BURN TEST NUMBER 14

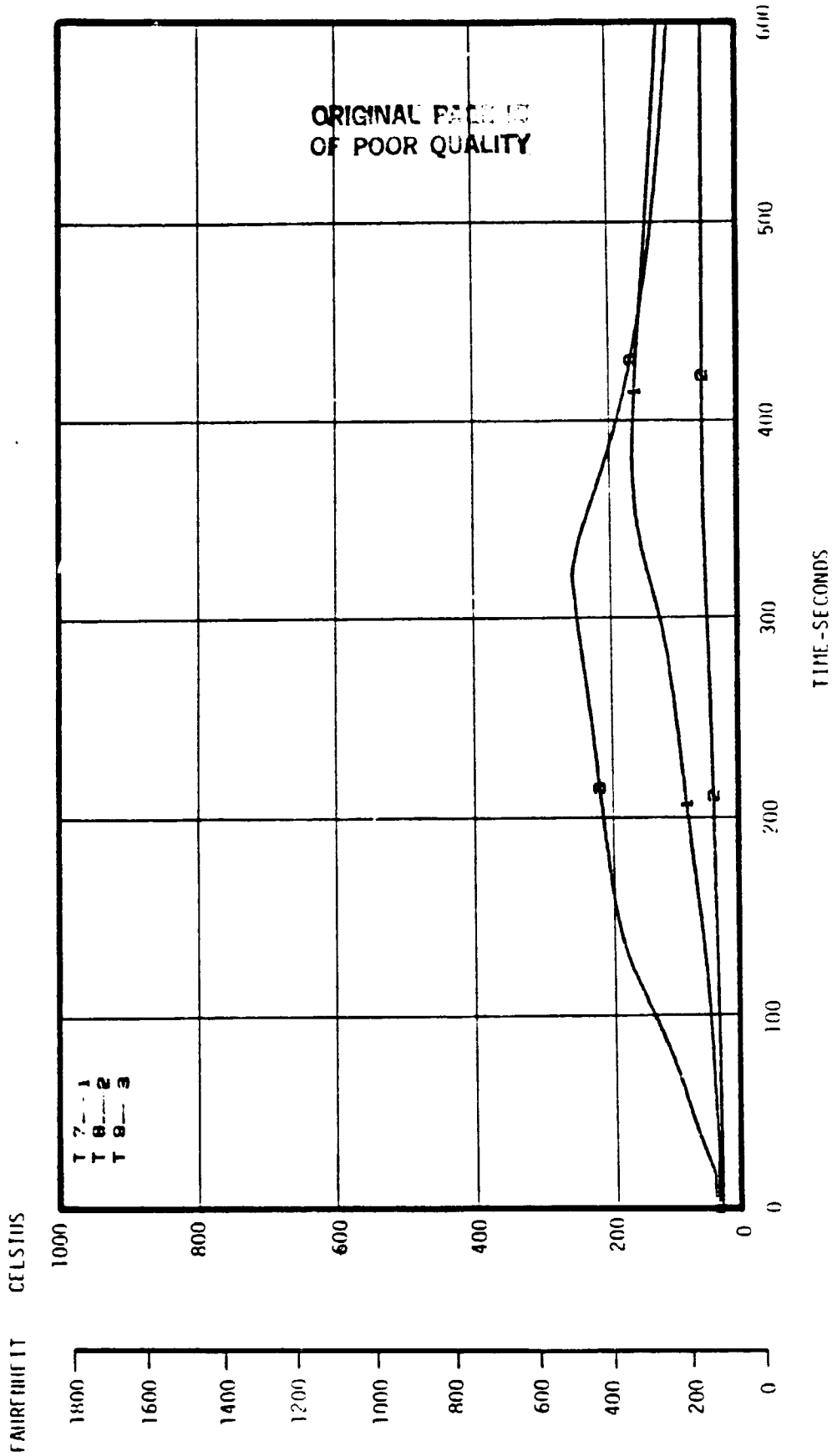
CUSHION CONSTRUCTION NUMBER 8.0

SEAT CUSHION TEMPERATURES



DOUGLAS AIRCRAFT CABIN FIRE SIM. FOR 33/15/82 11:40
 NASA-AMES FULL SCALE CUSHION BURN TEST NUMBER 14
 CUSHION CONSTRUCTION NUMBER 6.0

SEAT CUSHION TEMPERATURES

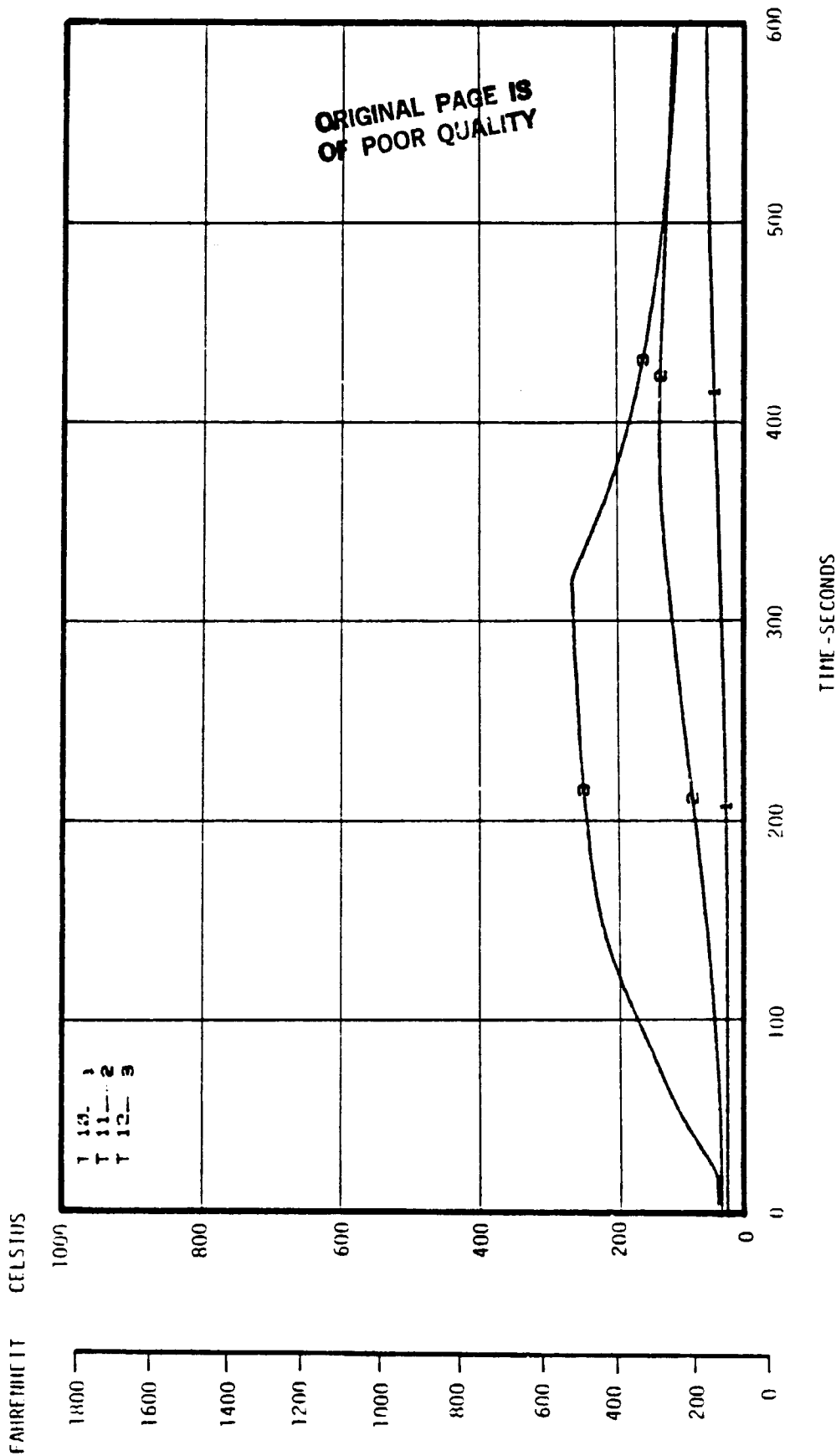


DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/15/82 11:40

NASA-AHS FULL SCALE CUSHION BURN TEST NUMBER 14

CUSHION CONSTRUCTION NUMBER 8.0

SEAT CUSHION TEMPERATURES

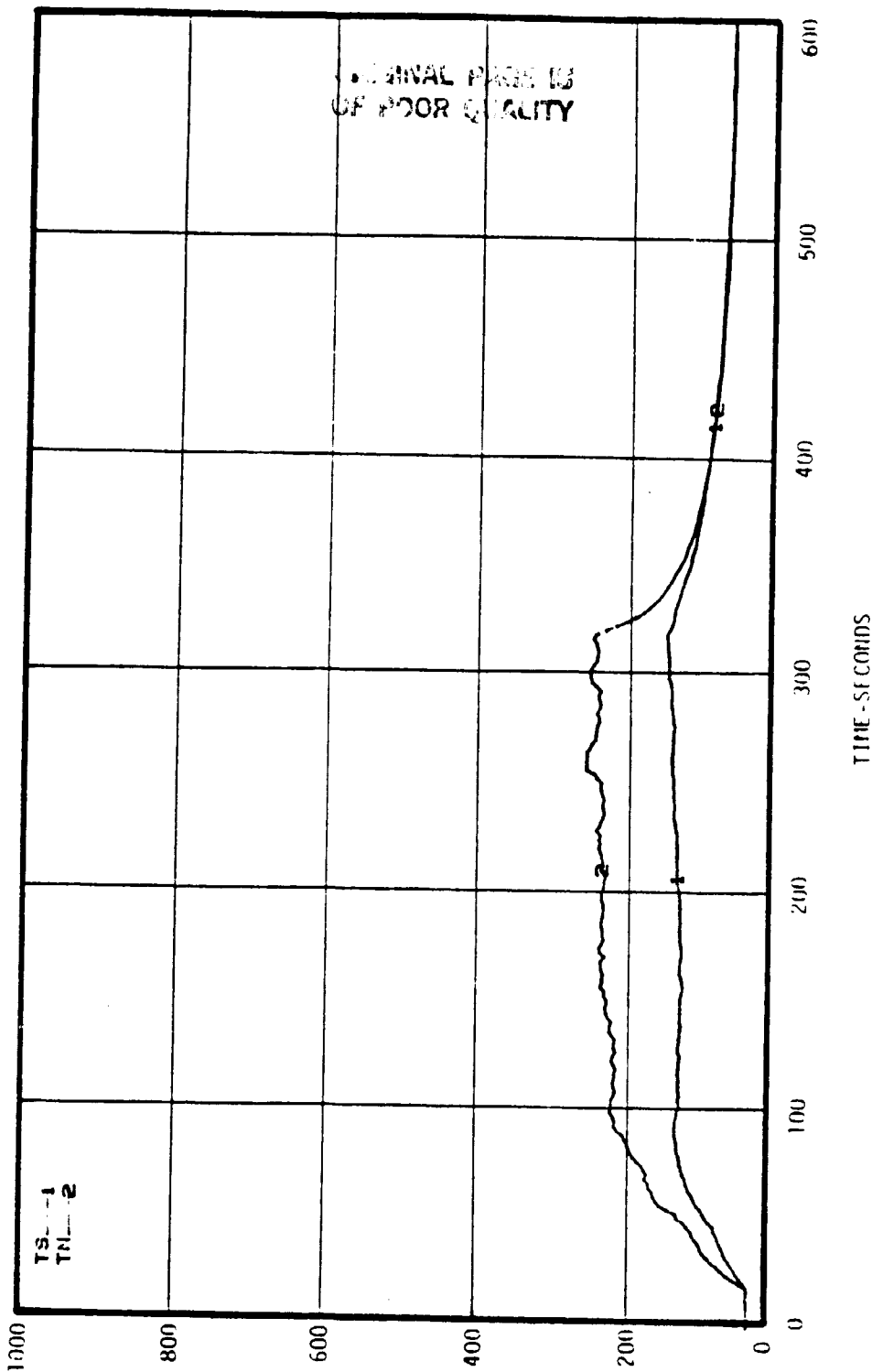
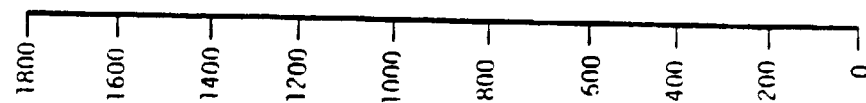


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DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 83-15-400 11-53
NASA-AMES FULL SCALE CUSHION BURN TEST NUMBER : 1
CUSHION CONSTRUCTION NUMBER 6.3

CEILING TEMPERATURE

FAHRENHEIT CELSIUS



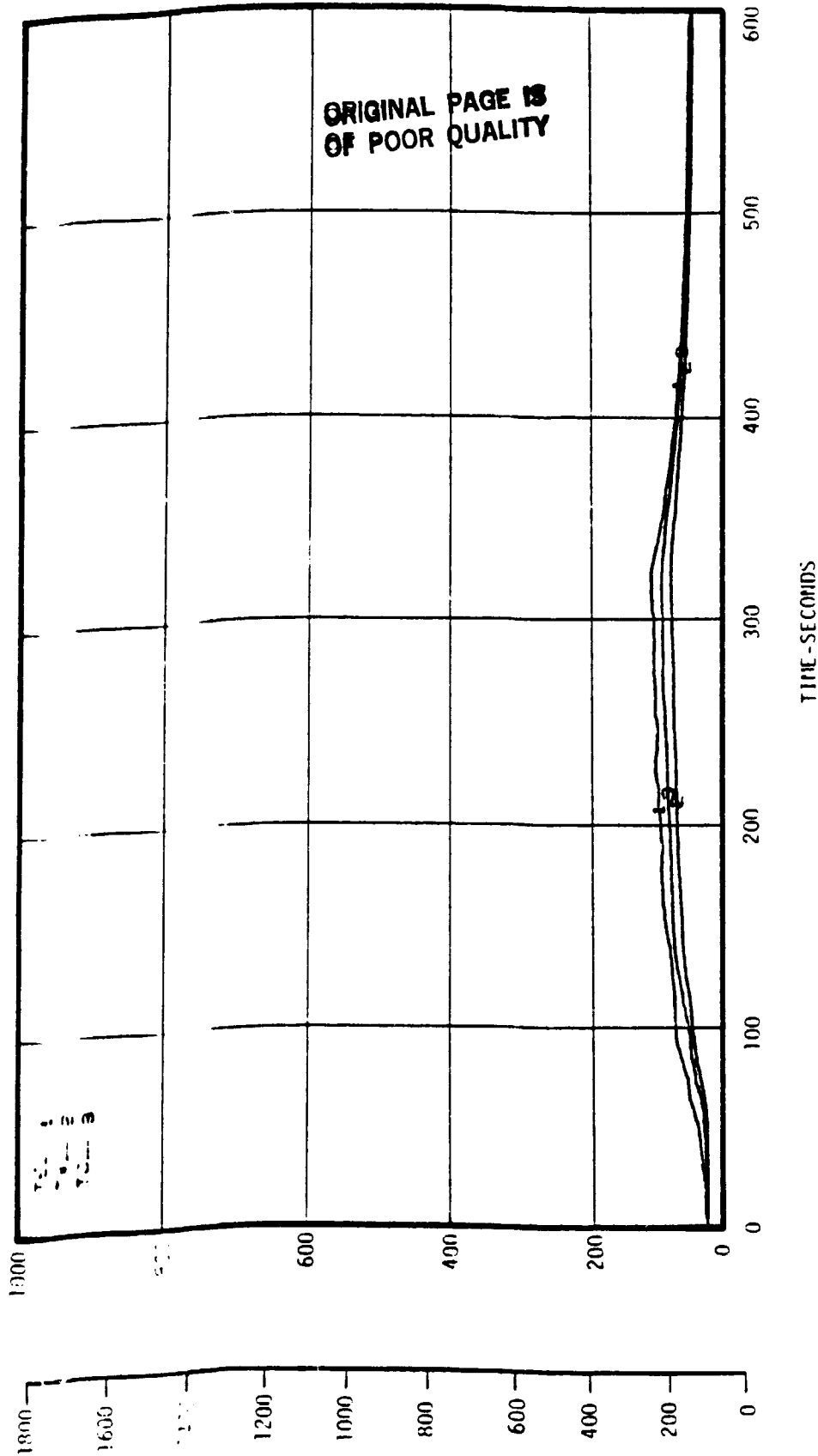
DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/15/82 11.40

NASA-AMES FULL SCALE CUSHION BURN TEST NUMBER 14

CUSHION CONSTRUCTION NUMBER 0.0

CEILING TEMPERATURE

FAHRENHEIT CELSIUS

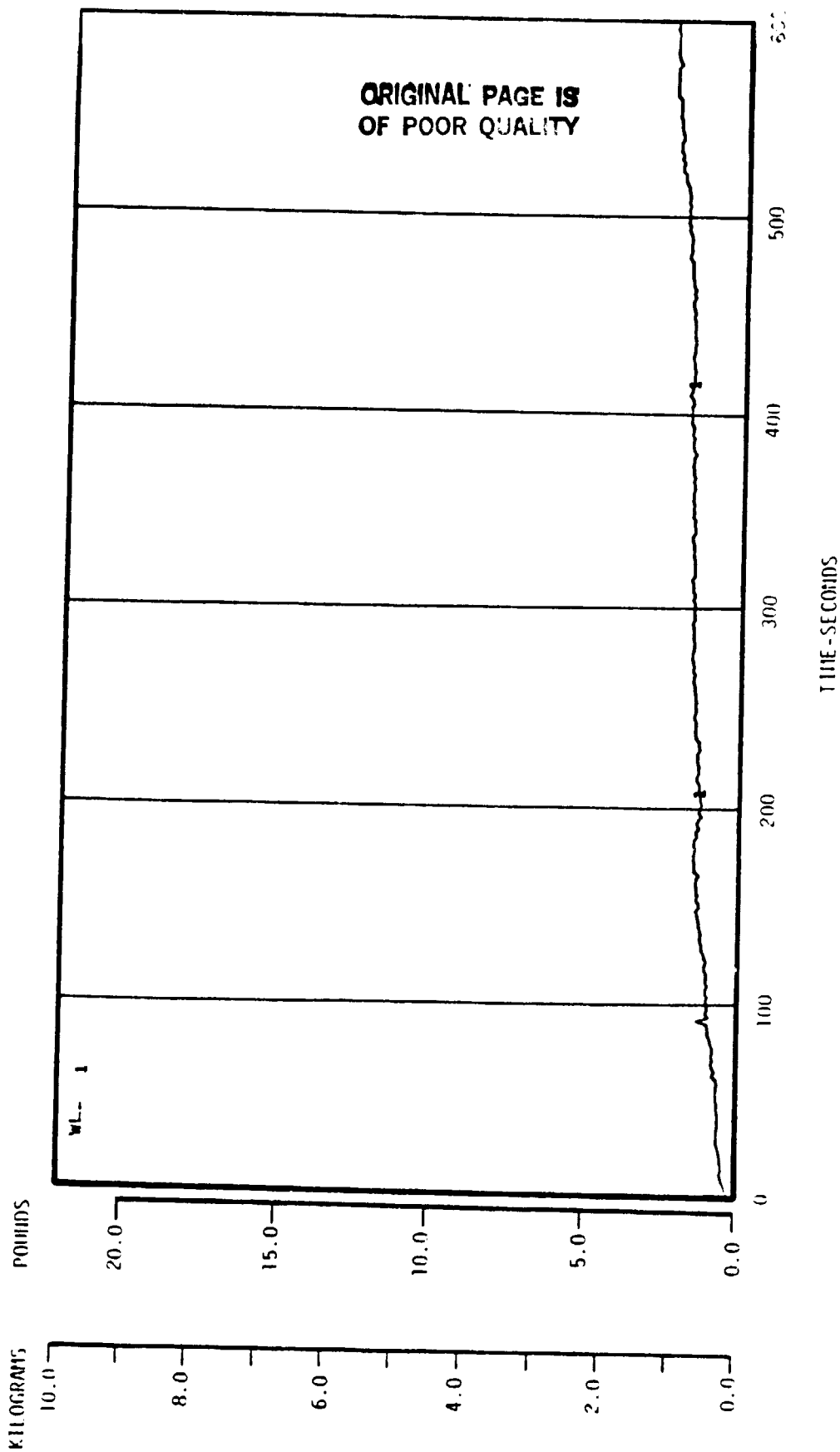


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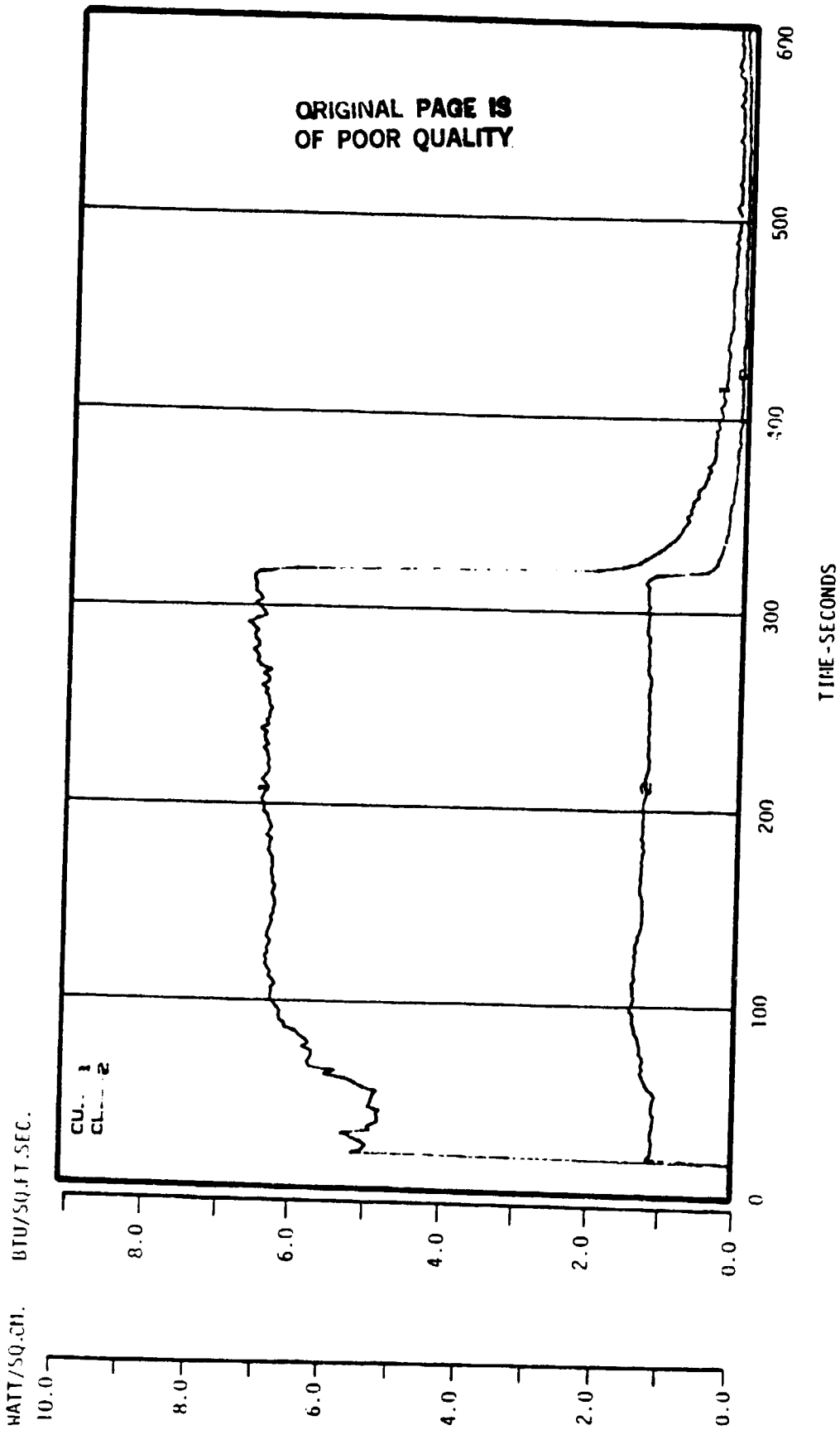
CUSHION CONSTRUCTION NUMBER 8.0

WEIGHT LOSS



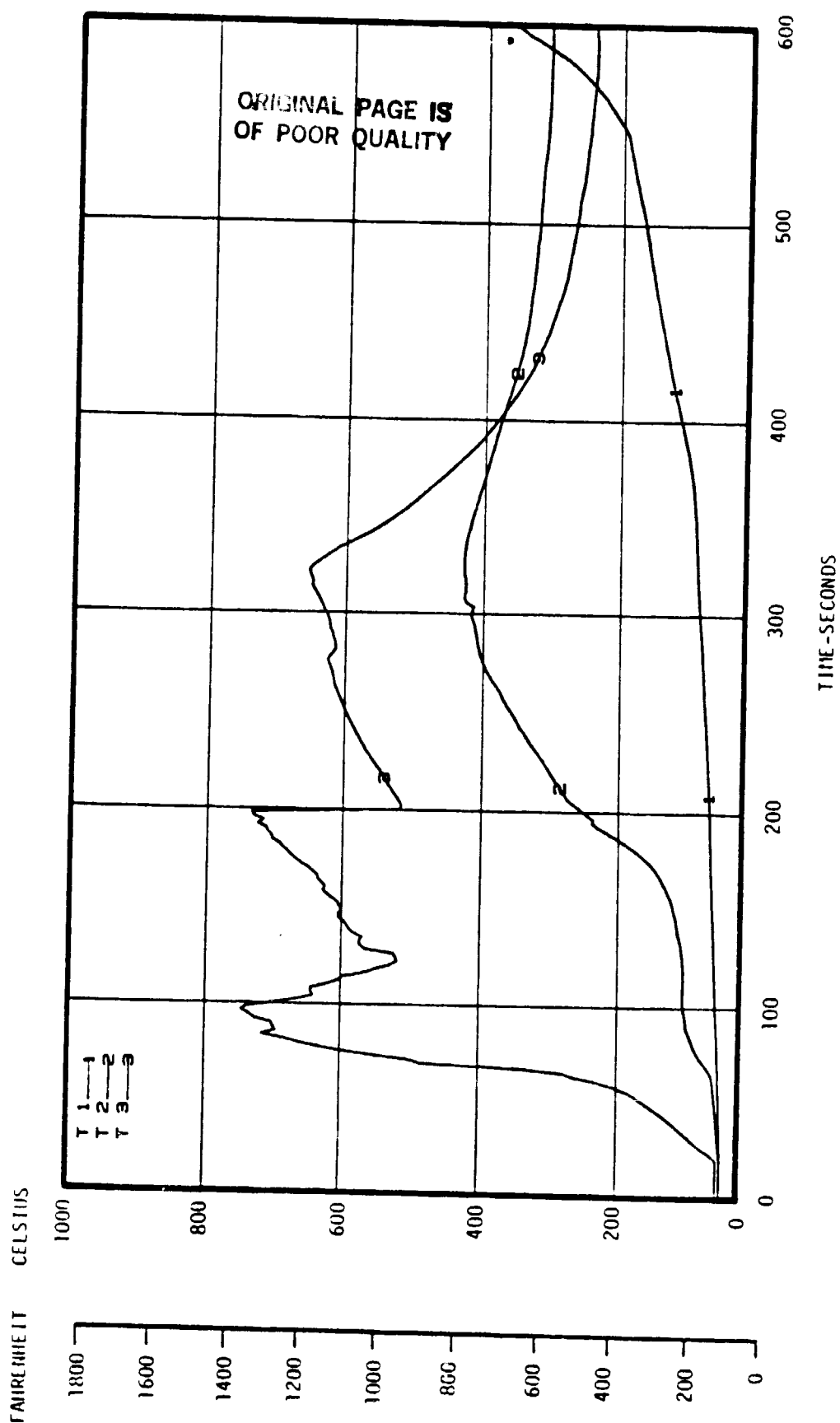
DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/15/62 11:40
 NASA-AMES FULL SCALE CUSHION BURN TEST NUMBER 14
 CUSHION CONSTRUCTION NUMBER 0.0

HEAT FLUX



DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/15/82 14, 33
NASA-MES FULL SCALE CUSHION BURN TEST NUMBER 15
CUSHION CONSTRUCTION NUMBER 7.0

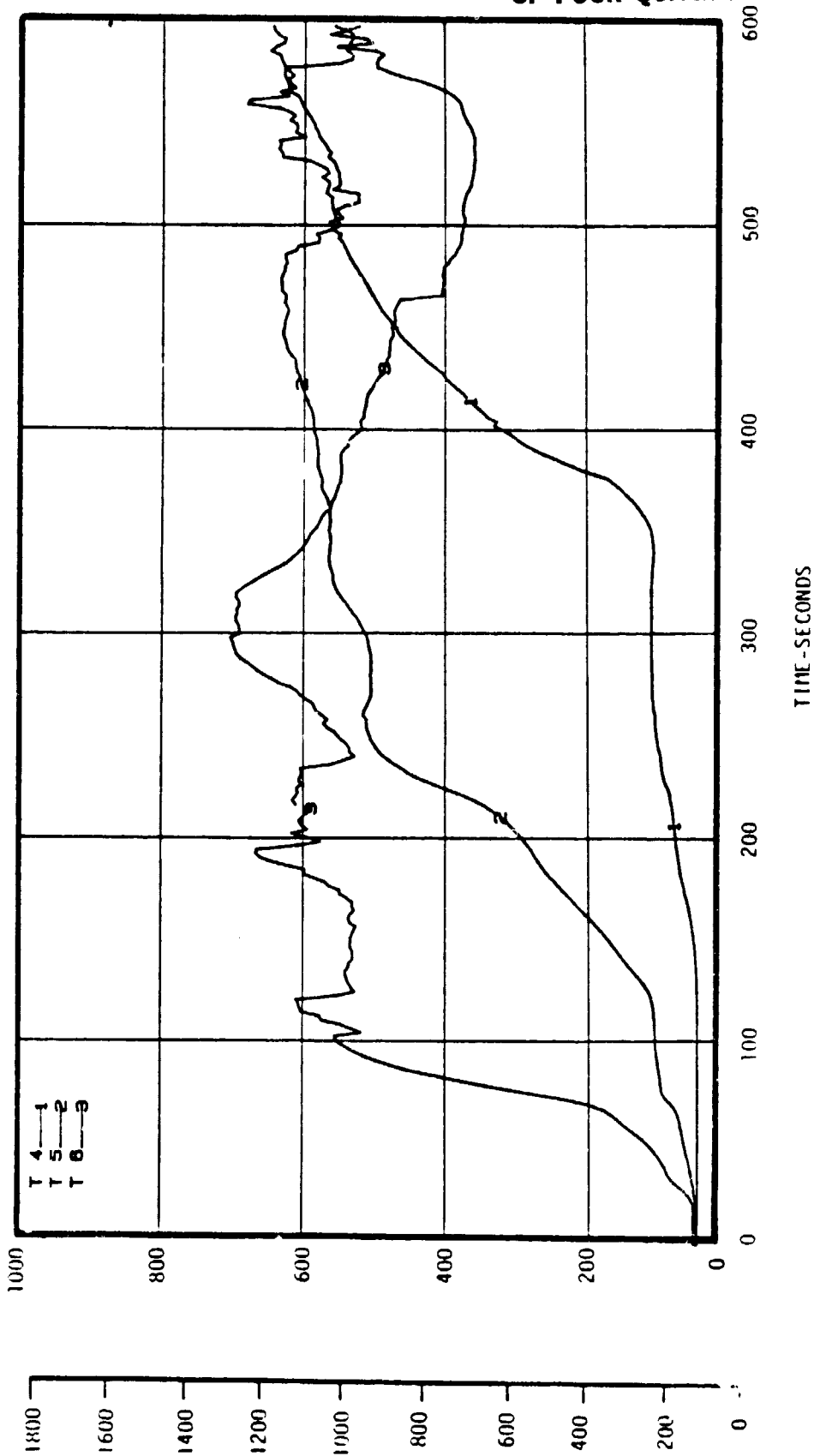
SEAT CUSHION TEMPERATURES



DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/15/02 14.33
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SEAT CUSHION TEMPERATURES

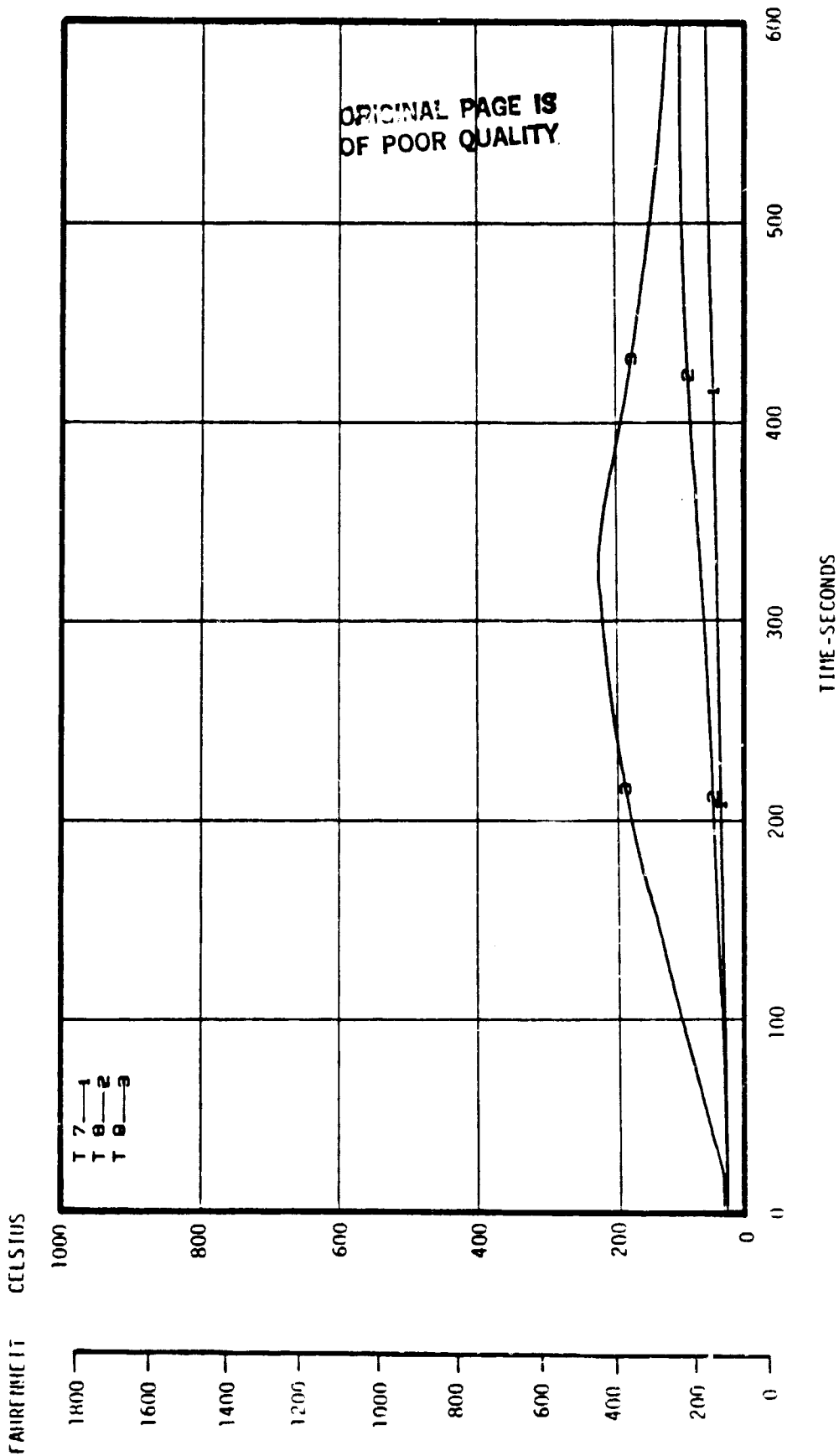
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DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/15/82 14.33
NASA-AIES FULL SCALE CUSHION BURN TEST NUMBER 15
CUSHION CONSTRUCTION NUMBER 7.0

SEAT CUSHION TEMPERATURES

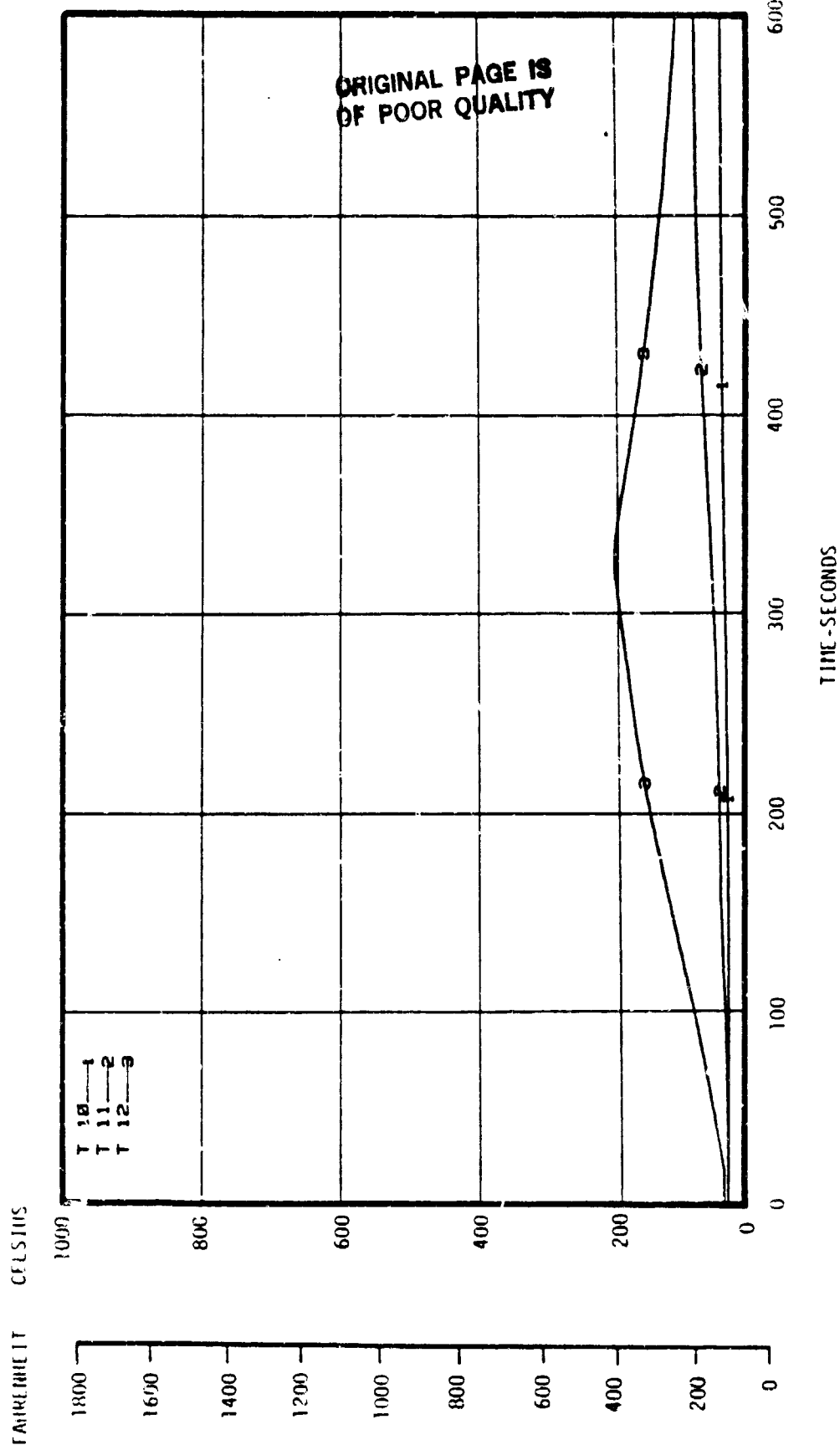


DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/15/82 14:33

NASA-AMES FULL SCALE CUSHION BURN TEST NUMBER 15

CUSHION CONSTRUCTION NUMBER 7.0

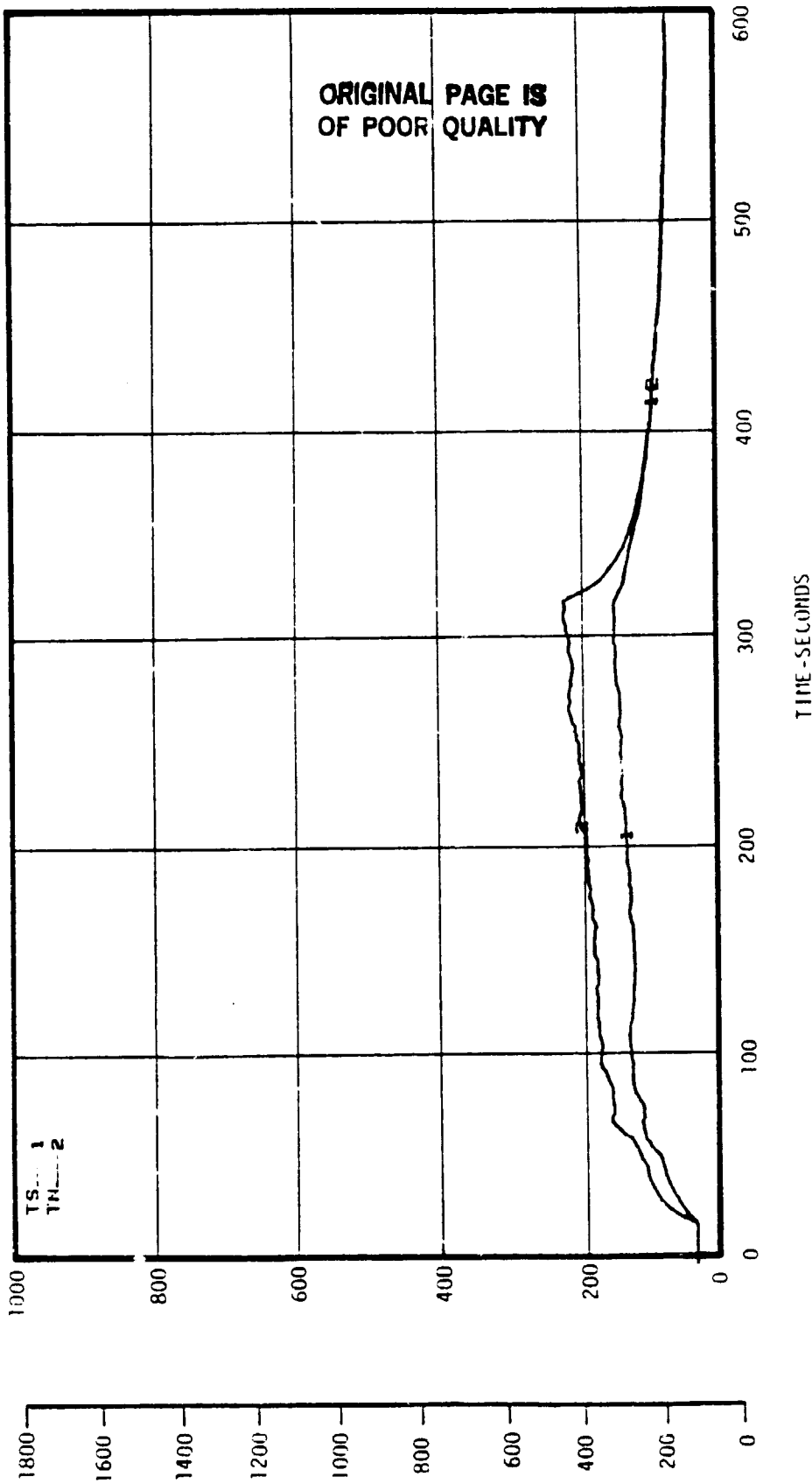
SEAT CUSHION TEMPERATURES



DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 53/15/82 14.33
 NASA-AMES FULL SCALE CUSHION BURN TEST NUMBER 15
 CUSHION CONSTRUCTION NUMBER 7.8

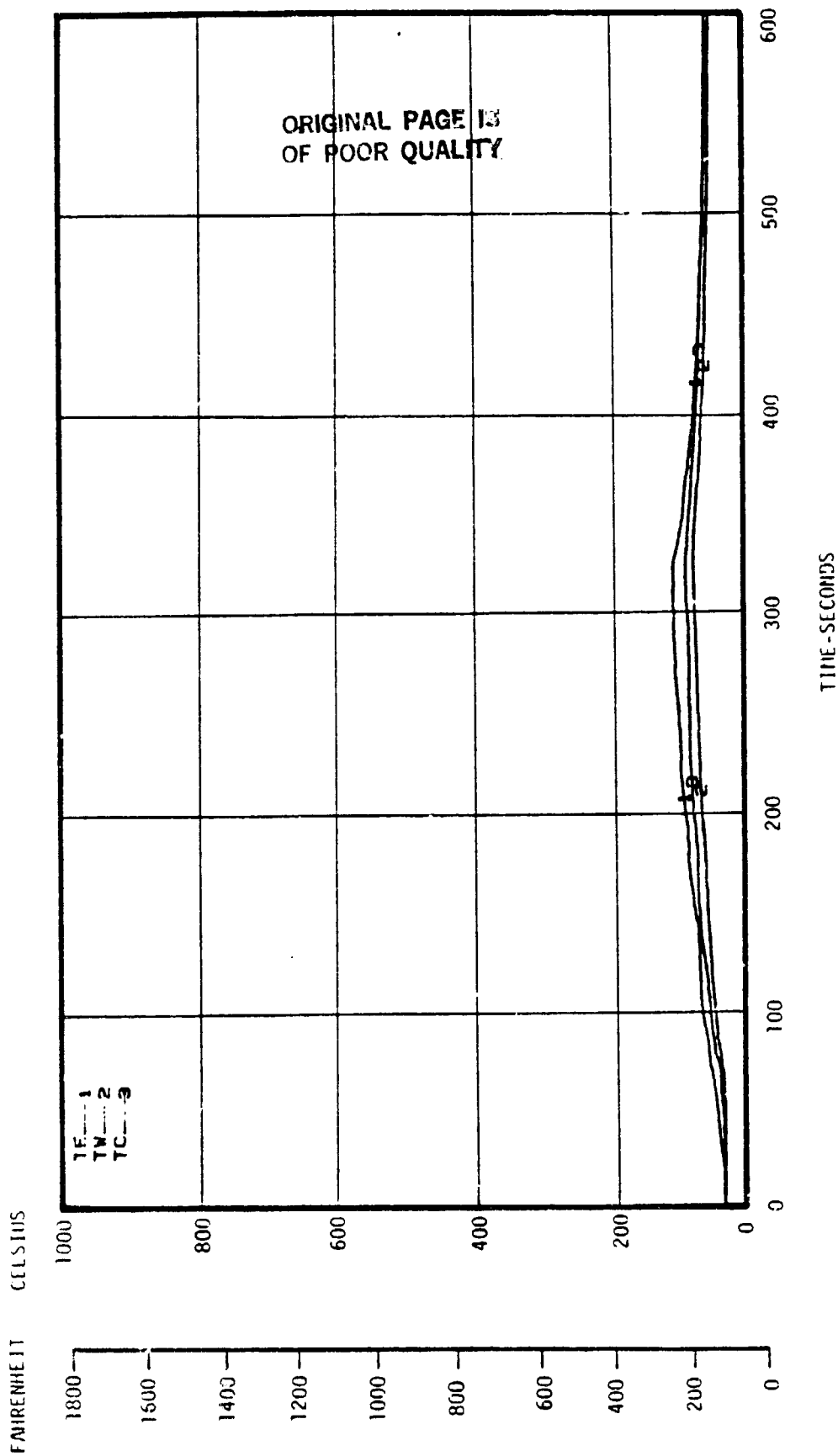
CELL 5 TEMPERATURE

FAHRENHEIT CELSIUS



DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/15/82 14.33
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 CUSHION CONSTRUCTION NUMBER 7.0

CEILING TEMPERATURE

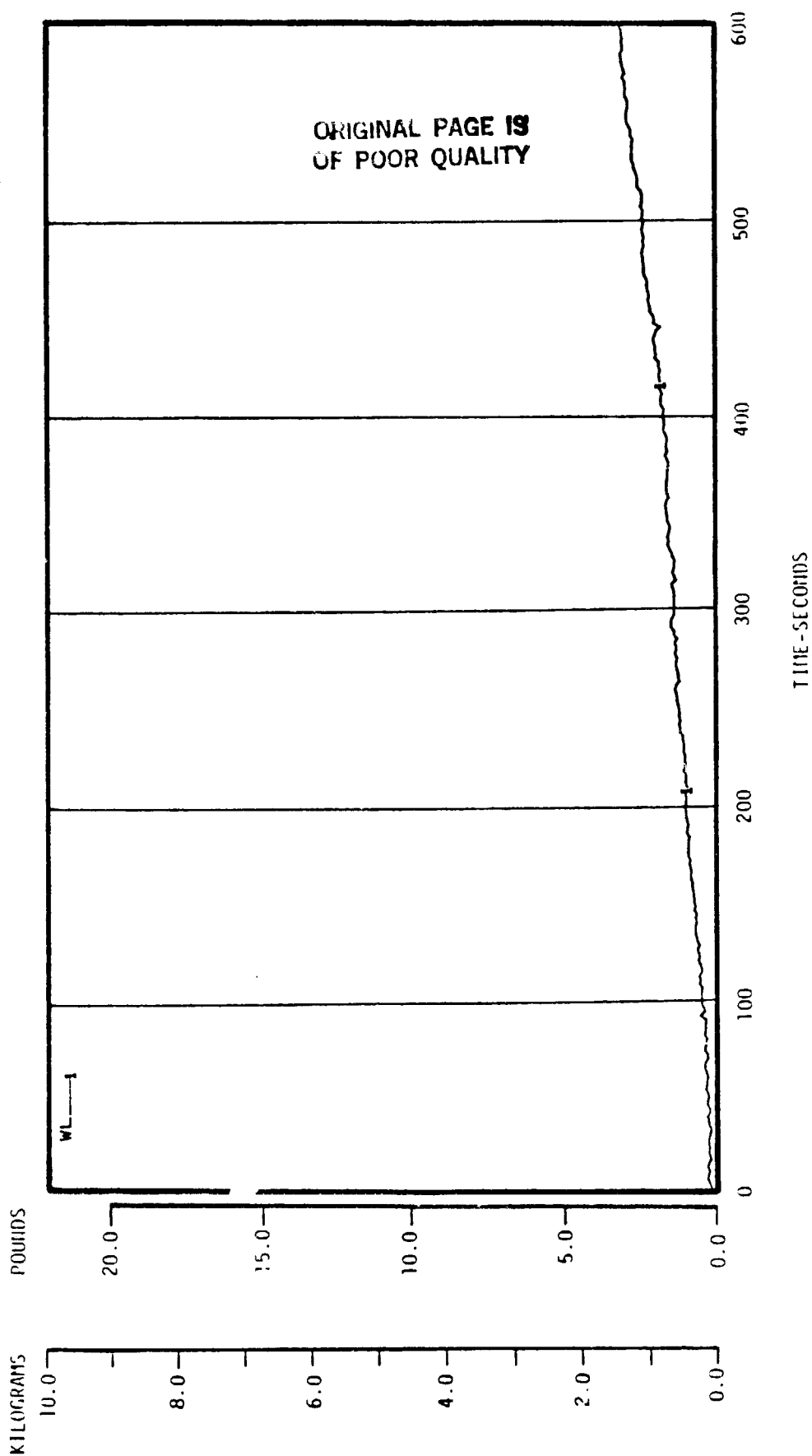


DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/15/82 14.33

NASA-AMES FULL SCALE CUSHION BURN TEST NUMBER 15

CUSHION CONSTRUCTION NUMBER 7.0

WEIGHT LOSS

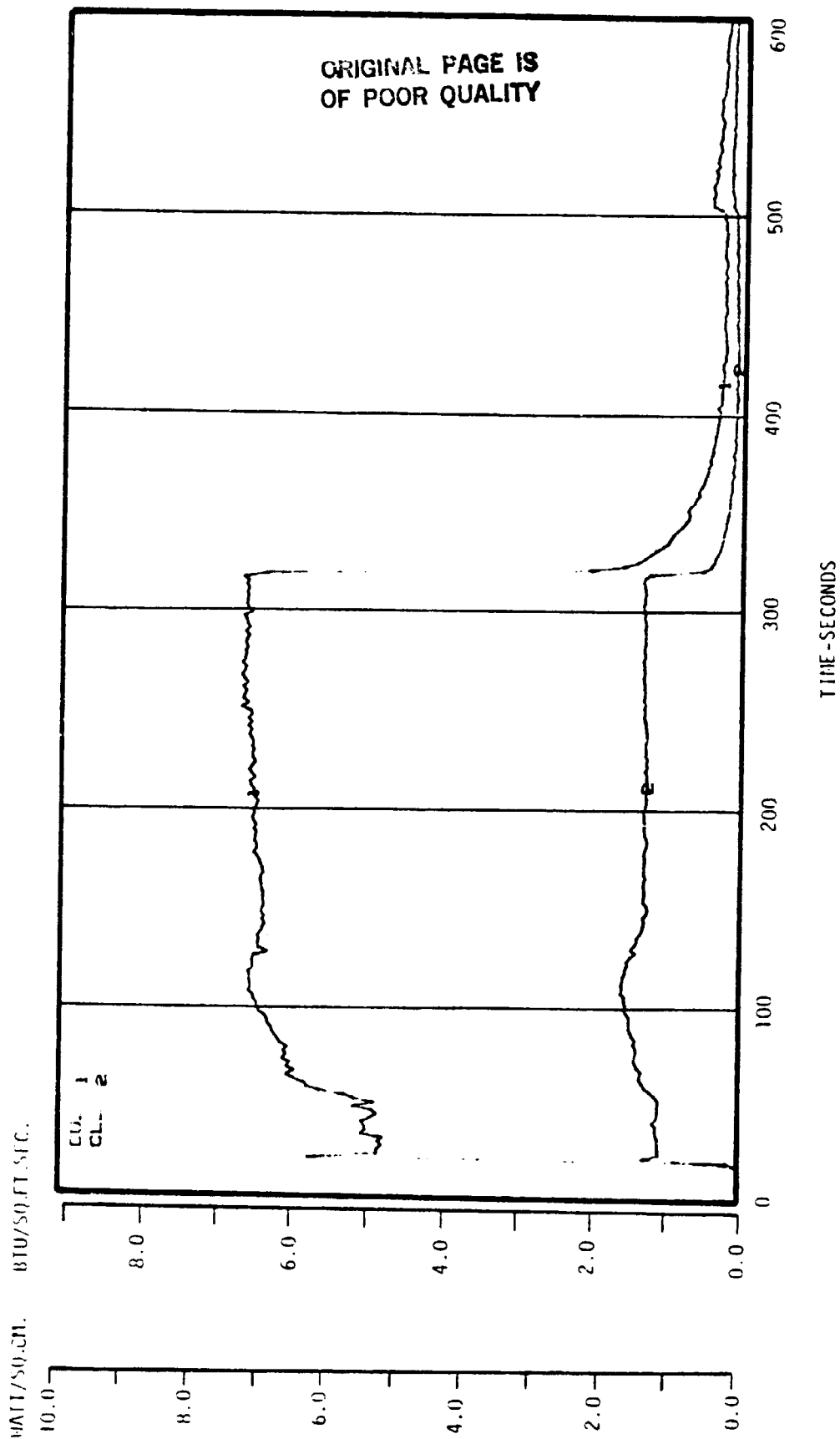


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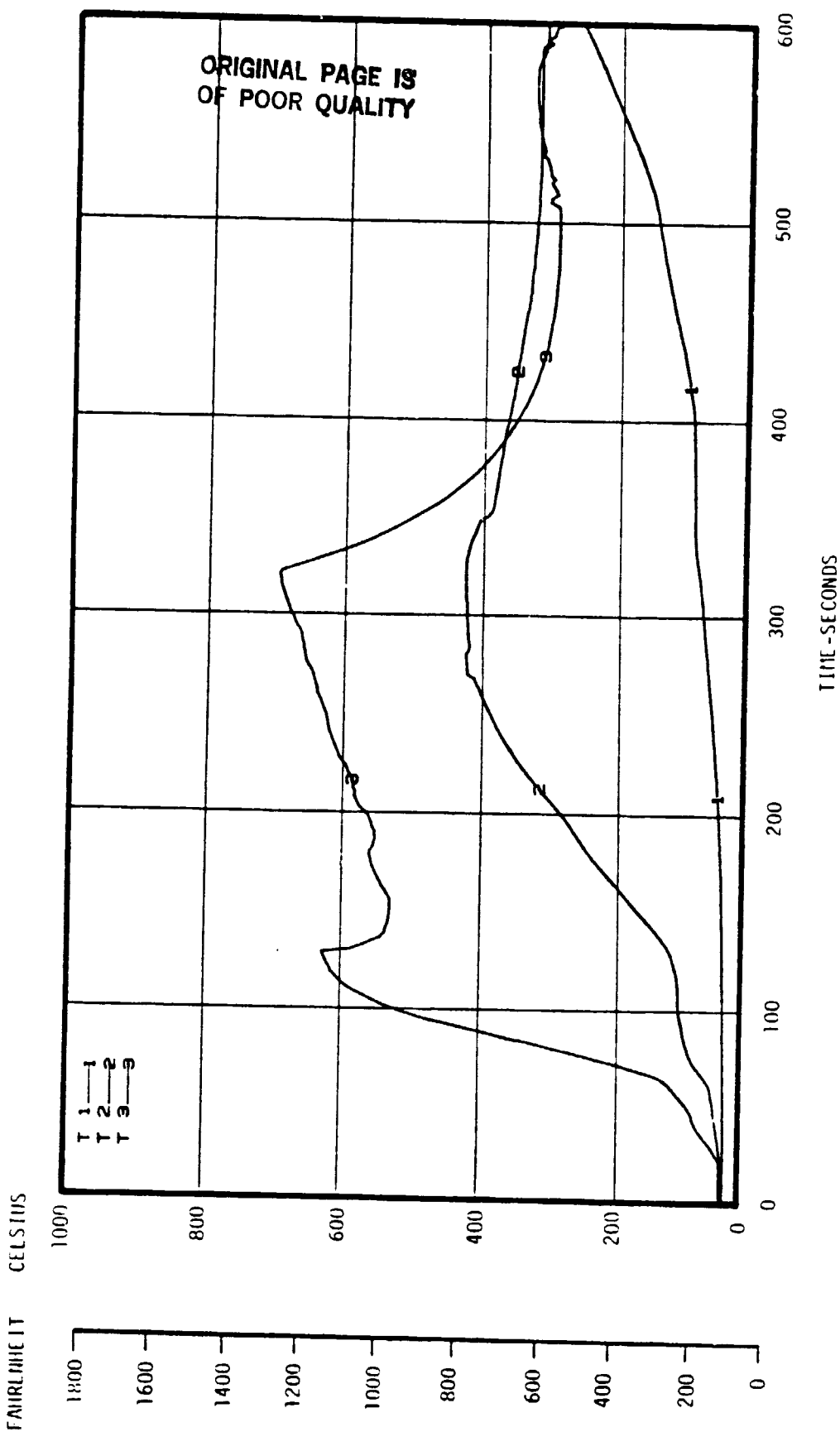
CUSHION CONSTRUCTION NUMBER 7.0

HEAT FLUX



DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 3-16-82 88, 59
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 CUSHION CONSTRUCTION NUMBER 7.0

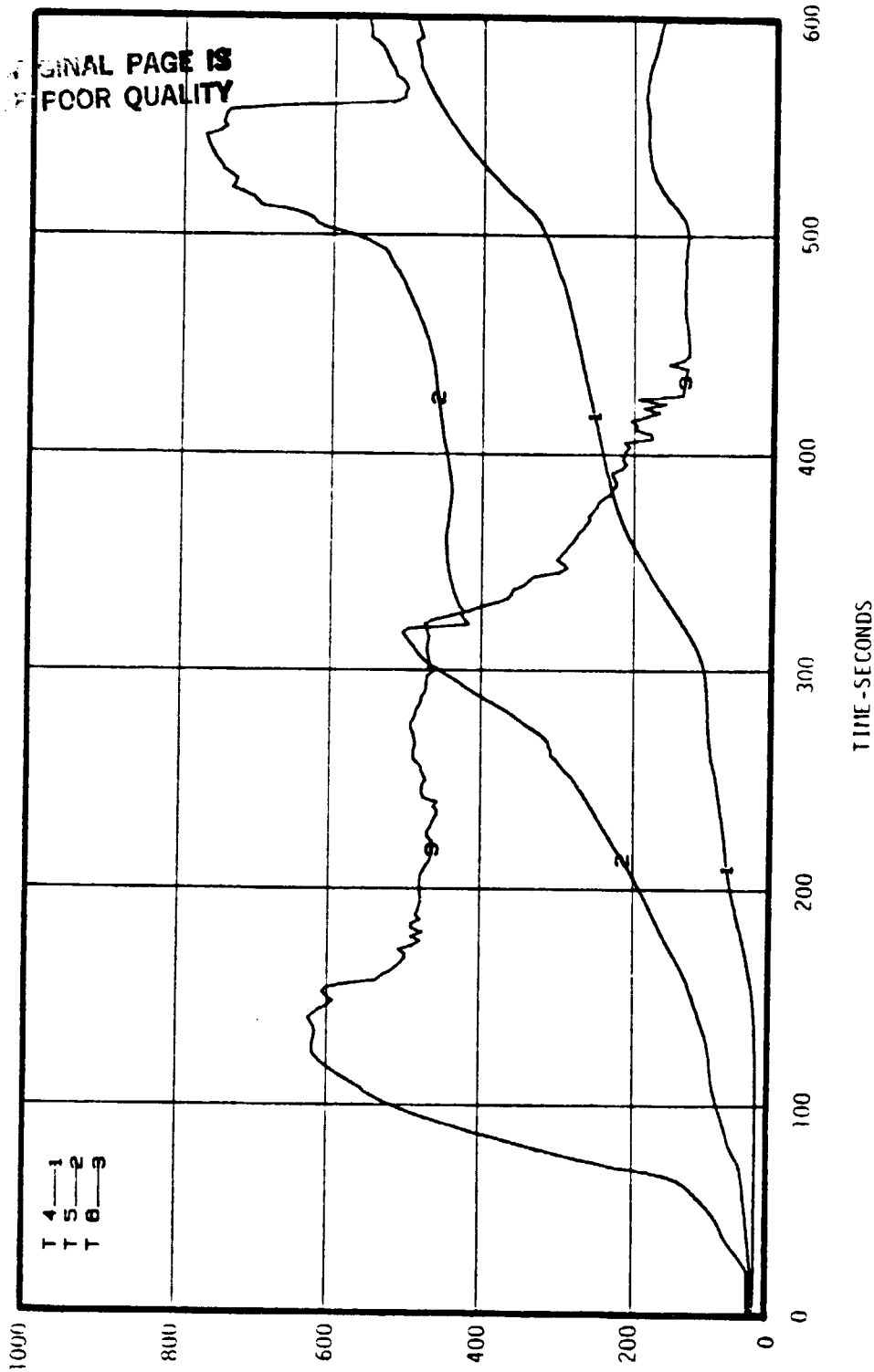
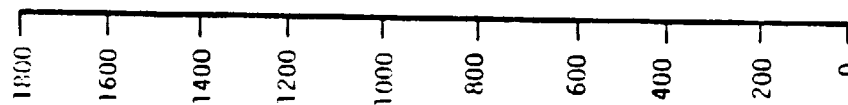
SEAT CUSHION TEMPERATURES



DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/18/82 08.50
NASA-MES FULL SCALE CUSHION BURN TEST NUMBER 18
CUSHION CONSTRUCTION NUMBER 7.0

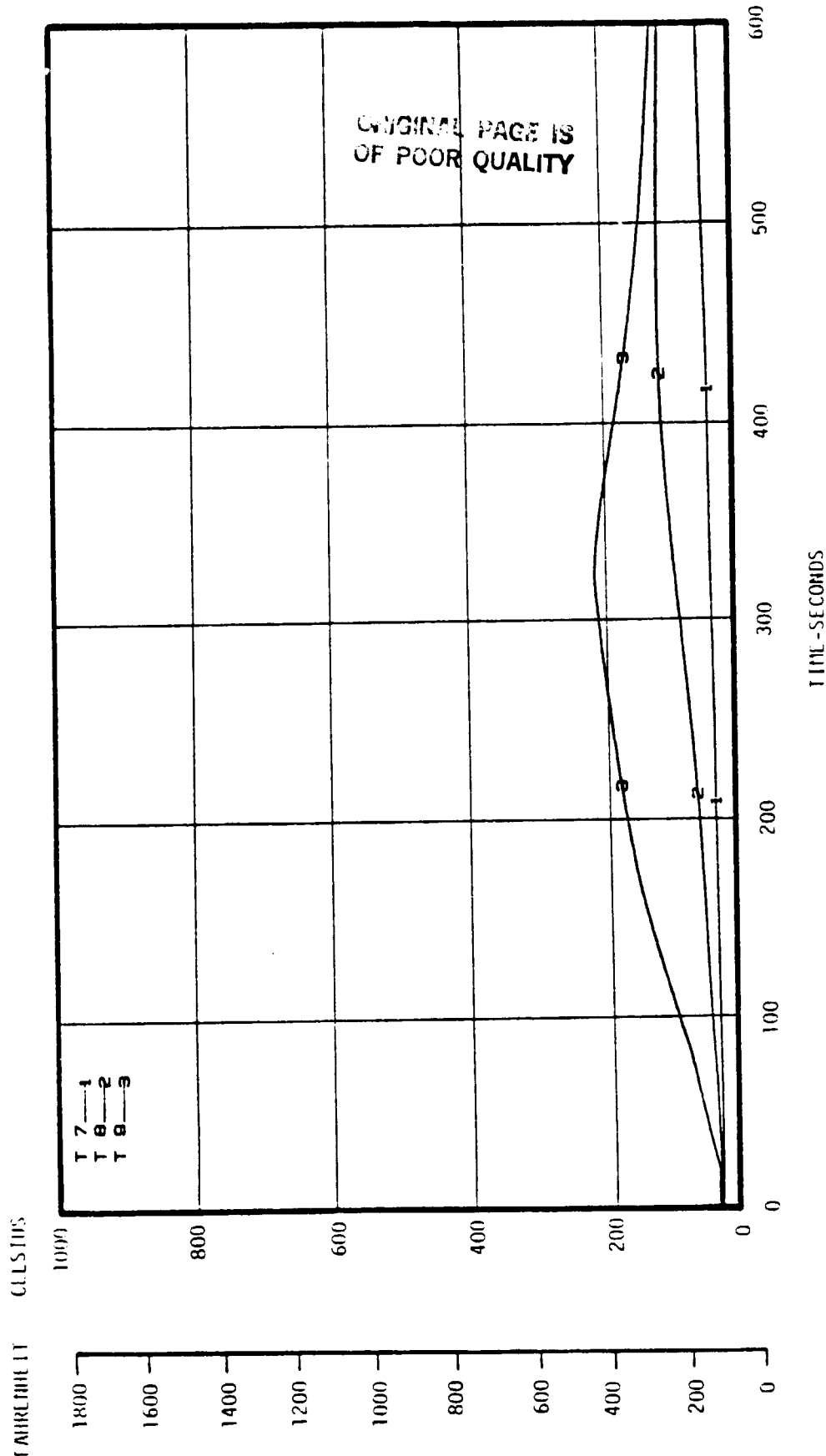
SEAT CUSHION TEMPERATURES

TEMPERATURE CELSIUS



DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/18/82 08:59
 NASA-AMES FULL SCALE CUSHION BURN TEST NUMBER 18
 CUSHION CONSTRUCTION NUMBER 7.0

SEAT CUSHION TEMPERATURES



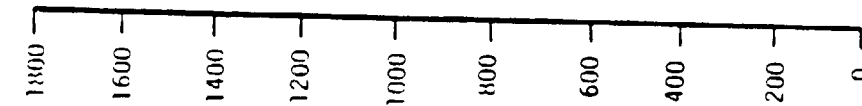
DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/18/82 08.59

NASA-AMES FULL SCALE CUSHION BURN TEST NUMBER 18

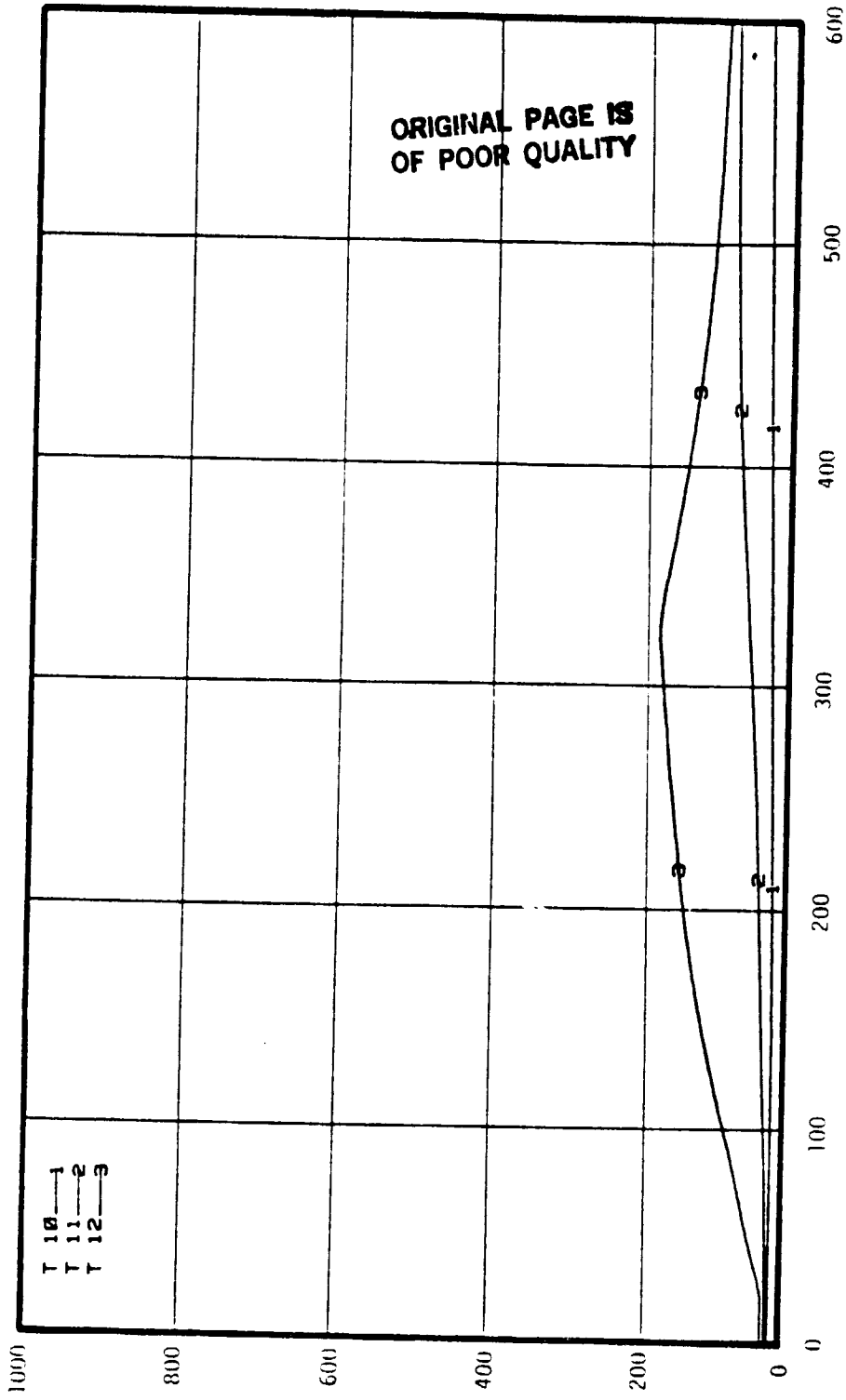
CUSHION CONSTRUCTION NUMBER 7.0

SEAT CUSHION TEMPERATURES

FAHRENHEIT CELSIUS



T 10 — 1
T 11 — 2
T 12 — 3



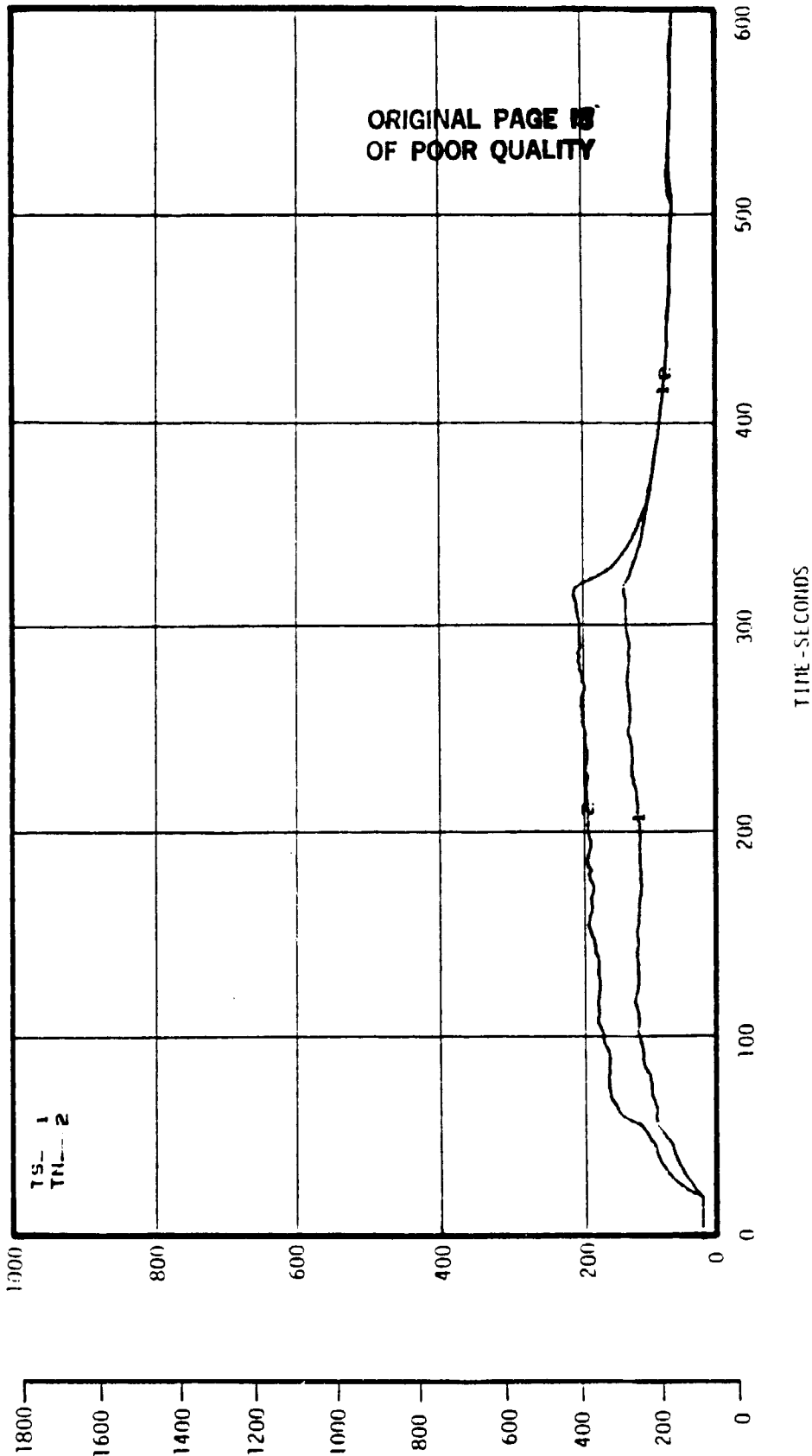
DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/18/82 08.58

NASA-MES FULL SCALE CUSHION BURN TEST NUMBER 10

CUSHION CONSTRUCTION NUMBER 7.0

CEILING TEMPERATURE

FAHRENHEIT CELSIUS



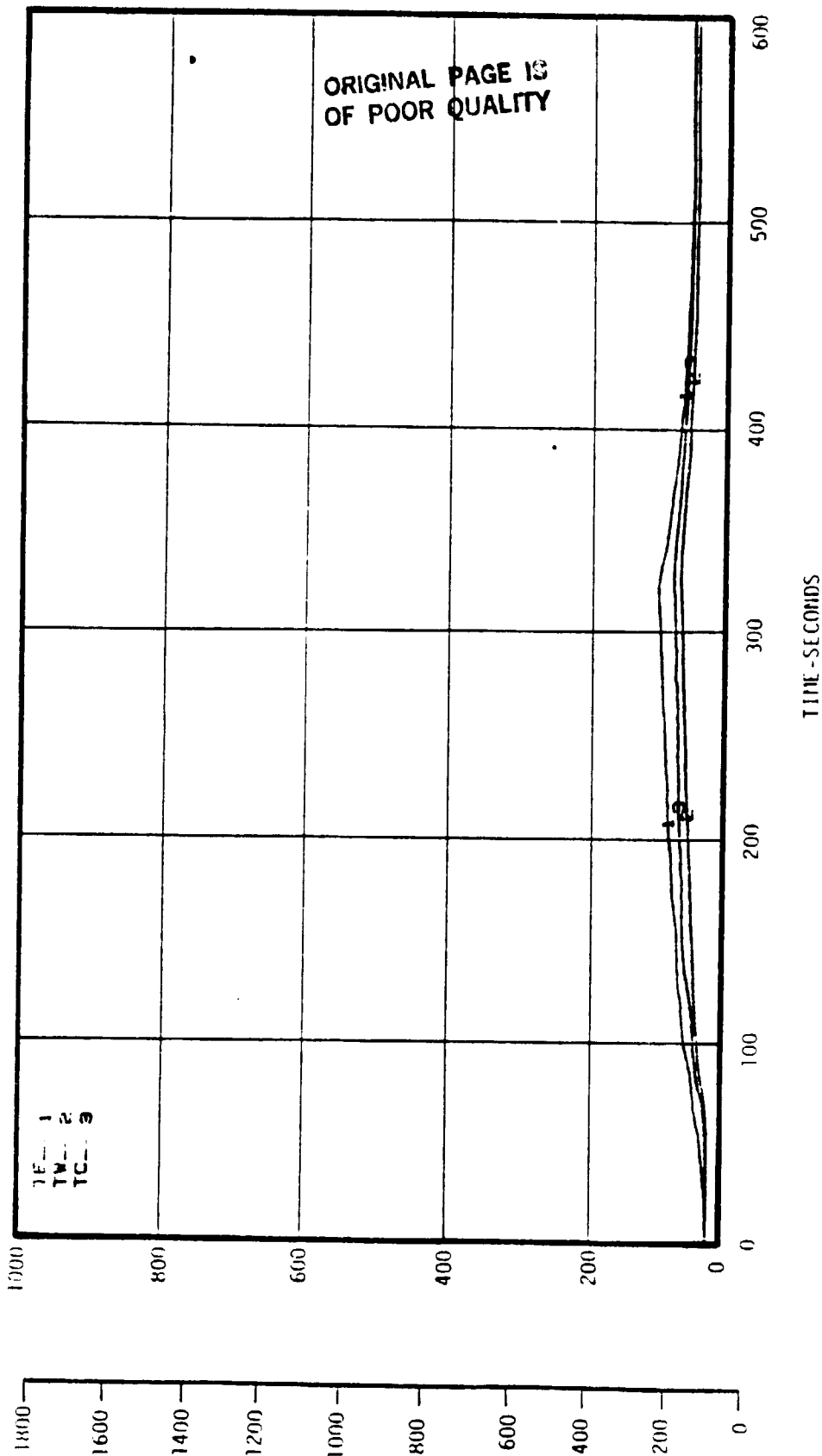
DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/16/82 04.50

NASA-MES FULL SCALE CUSHION BURN TEST NUMBER 18

CUSHION CONSTRUCTION NUMBER 7.0

CEILING TEMPERATURE

FAHRENHEIT CELSIUS

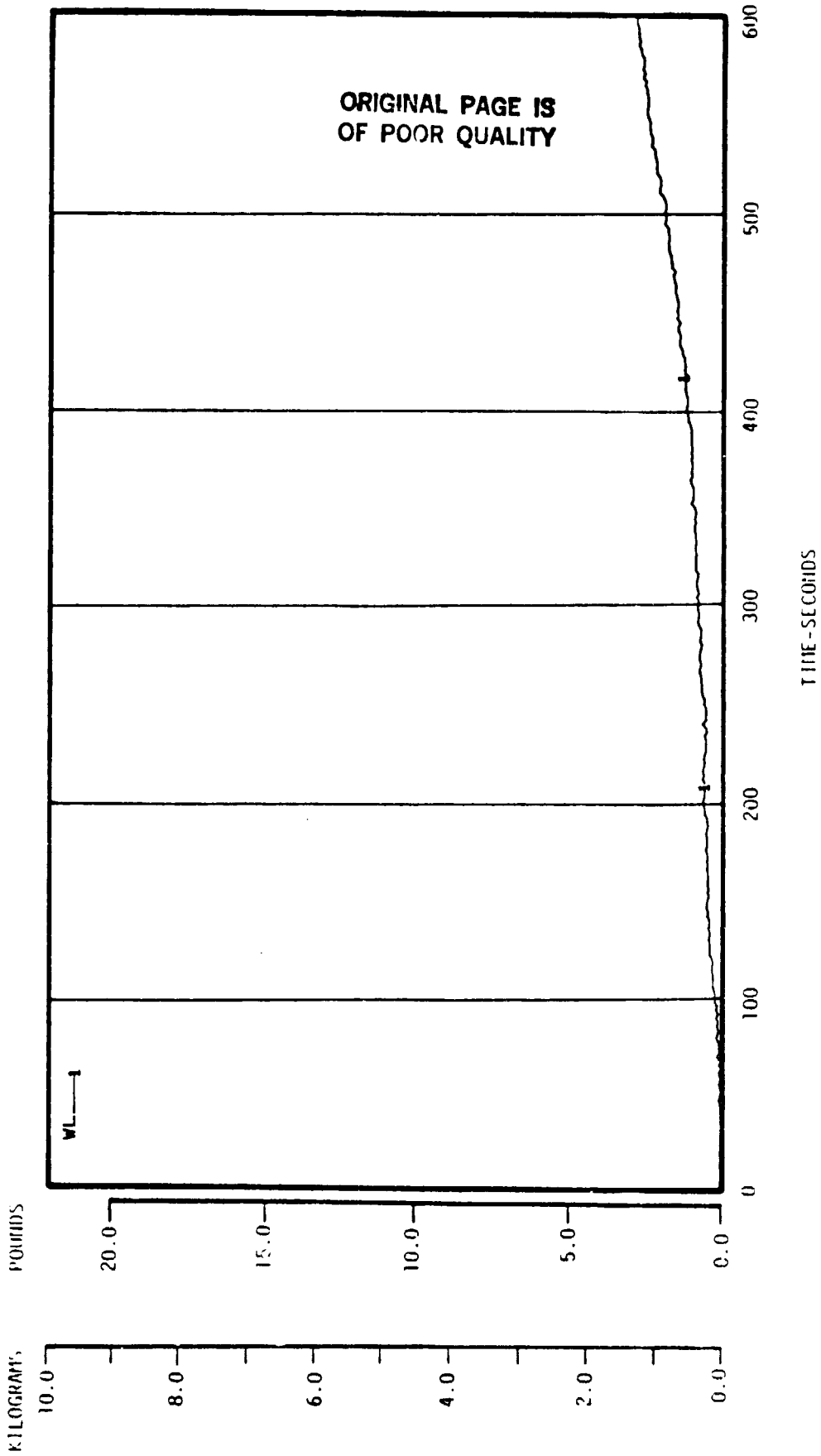


DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03-18/82 06.58

NASA-NIES FULL SCALE CUSHION BURN TEST NUMBER 10

CUSHION CONSTRUCTION NUMBER 7.0

WEIGHT LOSS

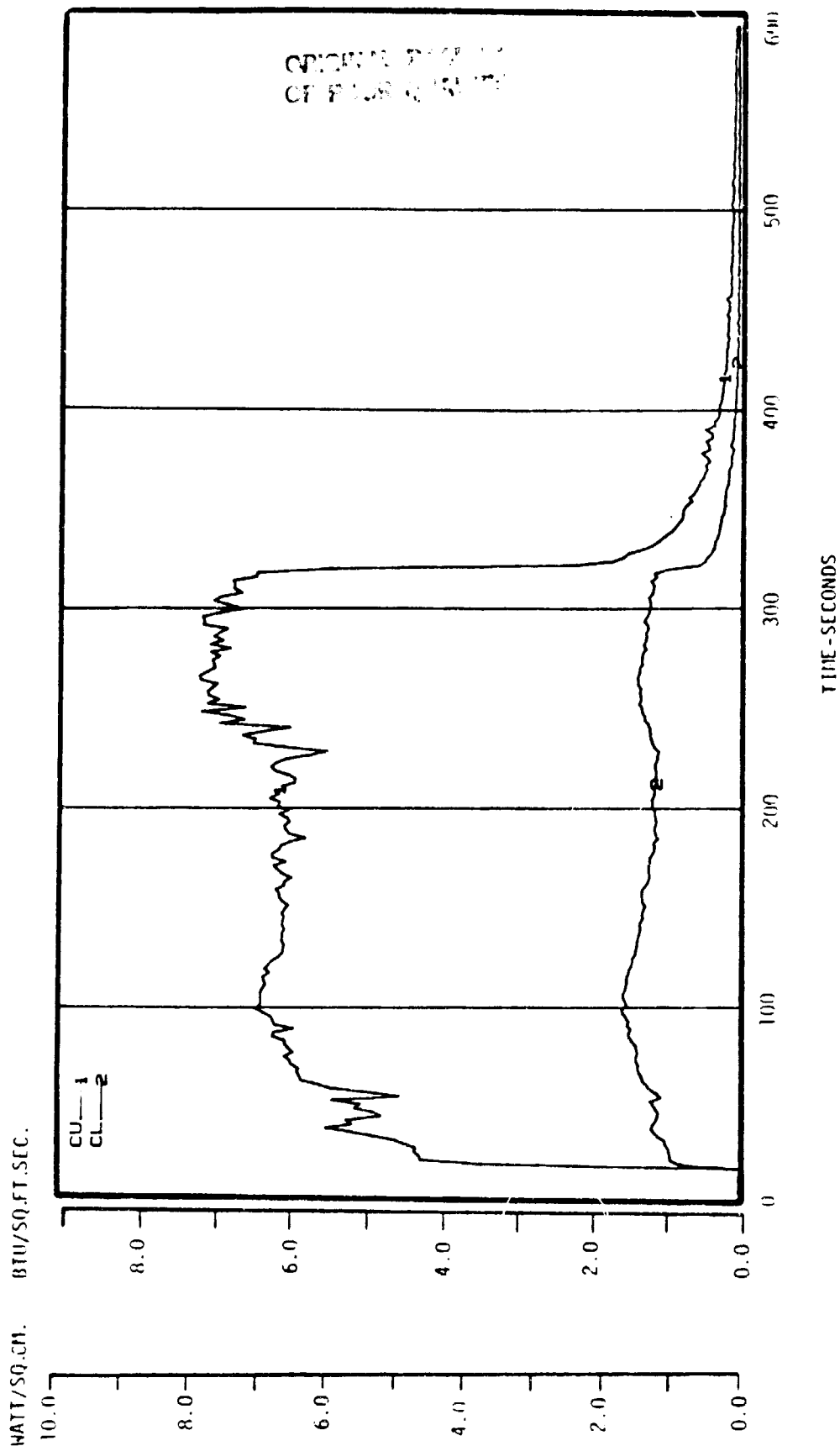


DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/18/82 08.58

NASA-AMES FULL SCALE CUSHION BURN TEST NUMBER 18

CUSHION CONSTRUCTION NUMBER 7.0

HEAT FLUX

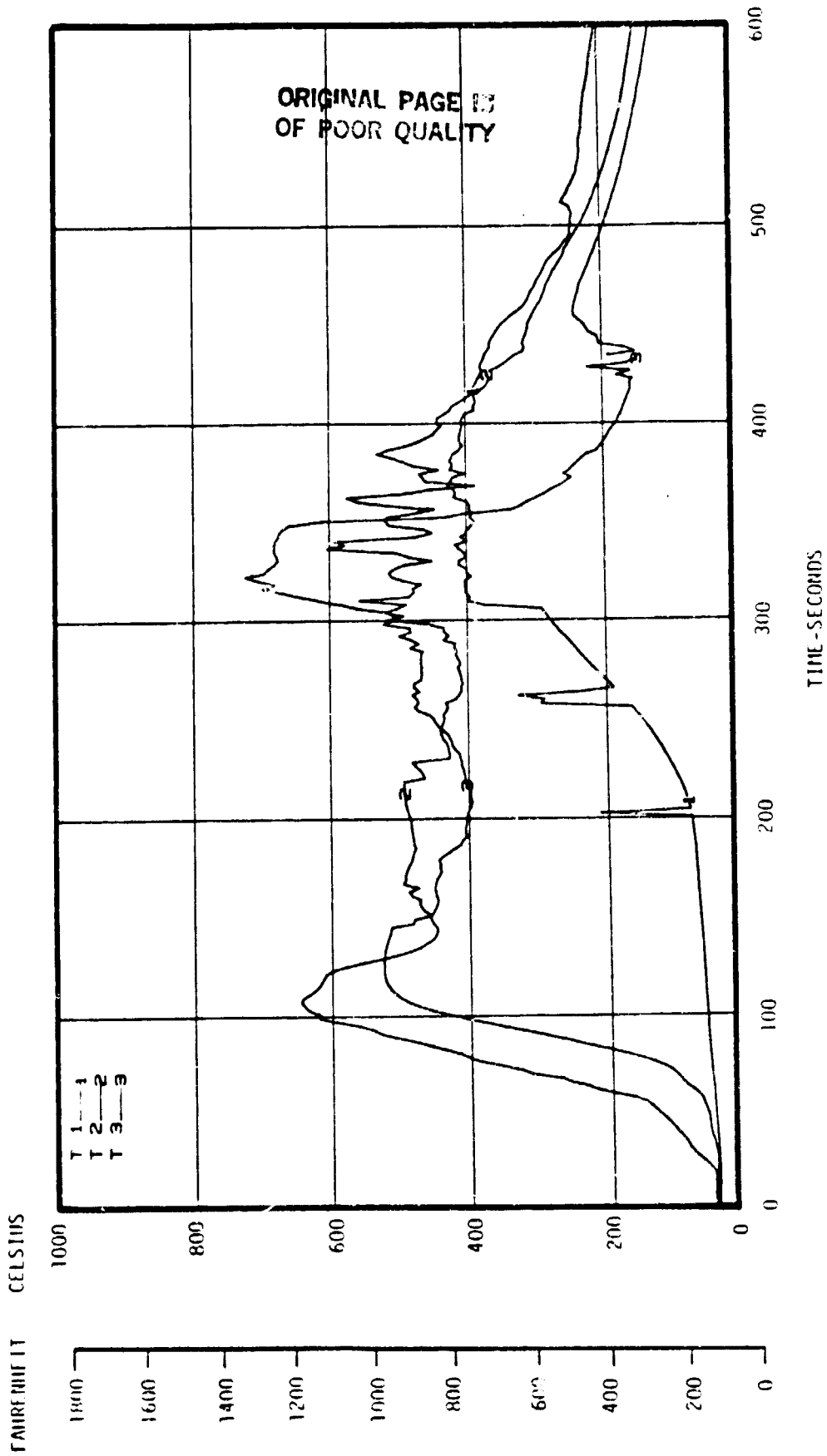


DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/1E/92 15.06

NASA-MIES FULL SCALE CUSHION BURN TEST NUMBER 18

CUSHION CONSTRUCTION NUMBER 8.0

SEAT CUSHION TEMPERATURES



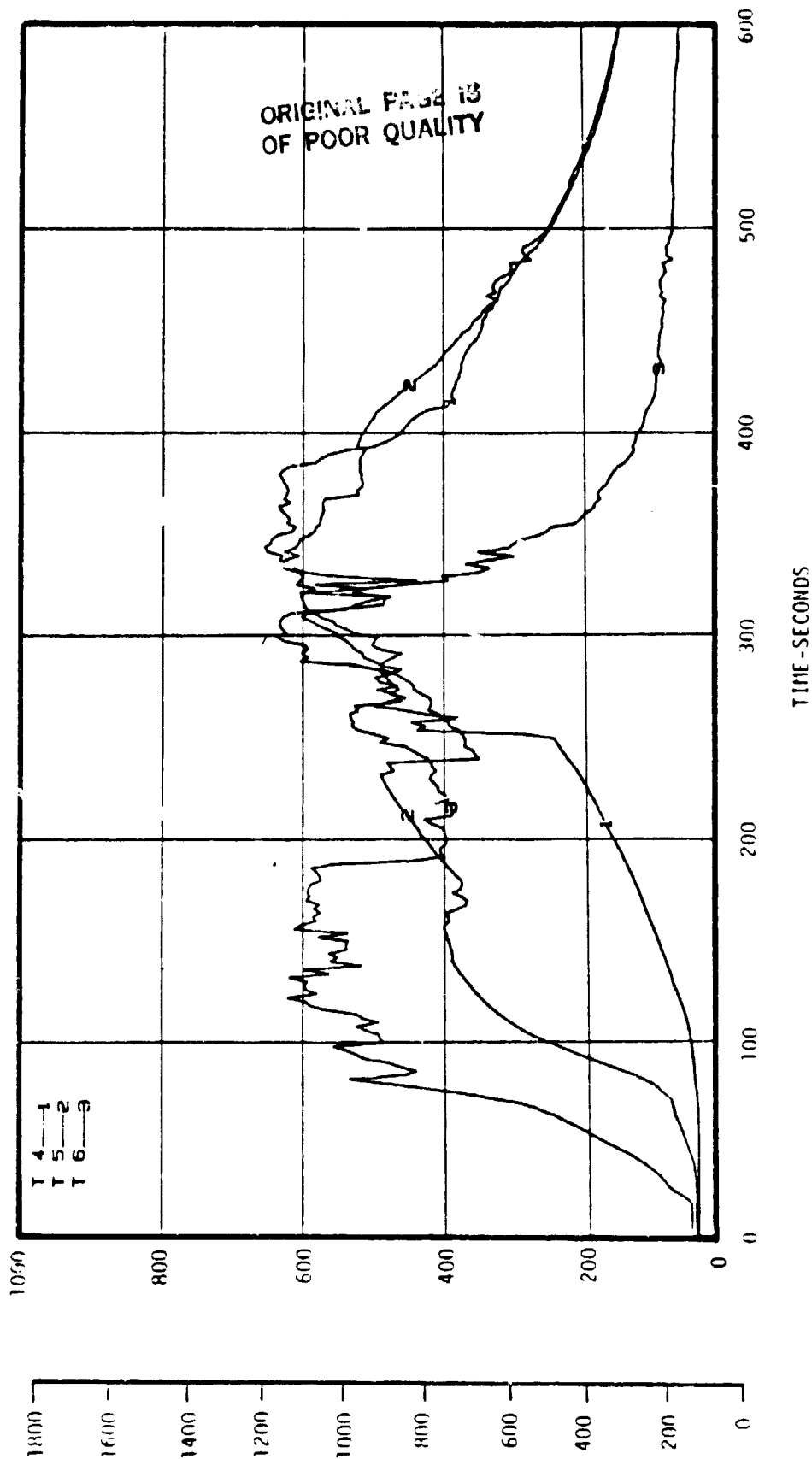
DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/18/82 15.06

NASA-AMES FULL SCALE CUSHION BURN TEST NUMBER 18

CUSHION CONSTRUCTION NUMBER 8.0

SEAT CUSHION TEMPERATURES

FAHRENHEIT CELSIUS

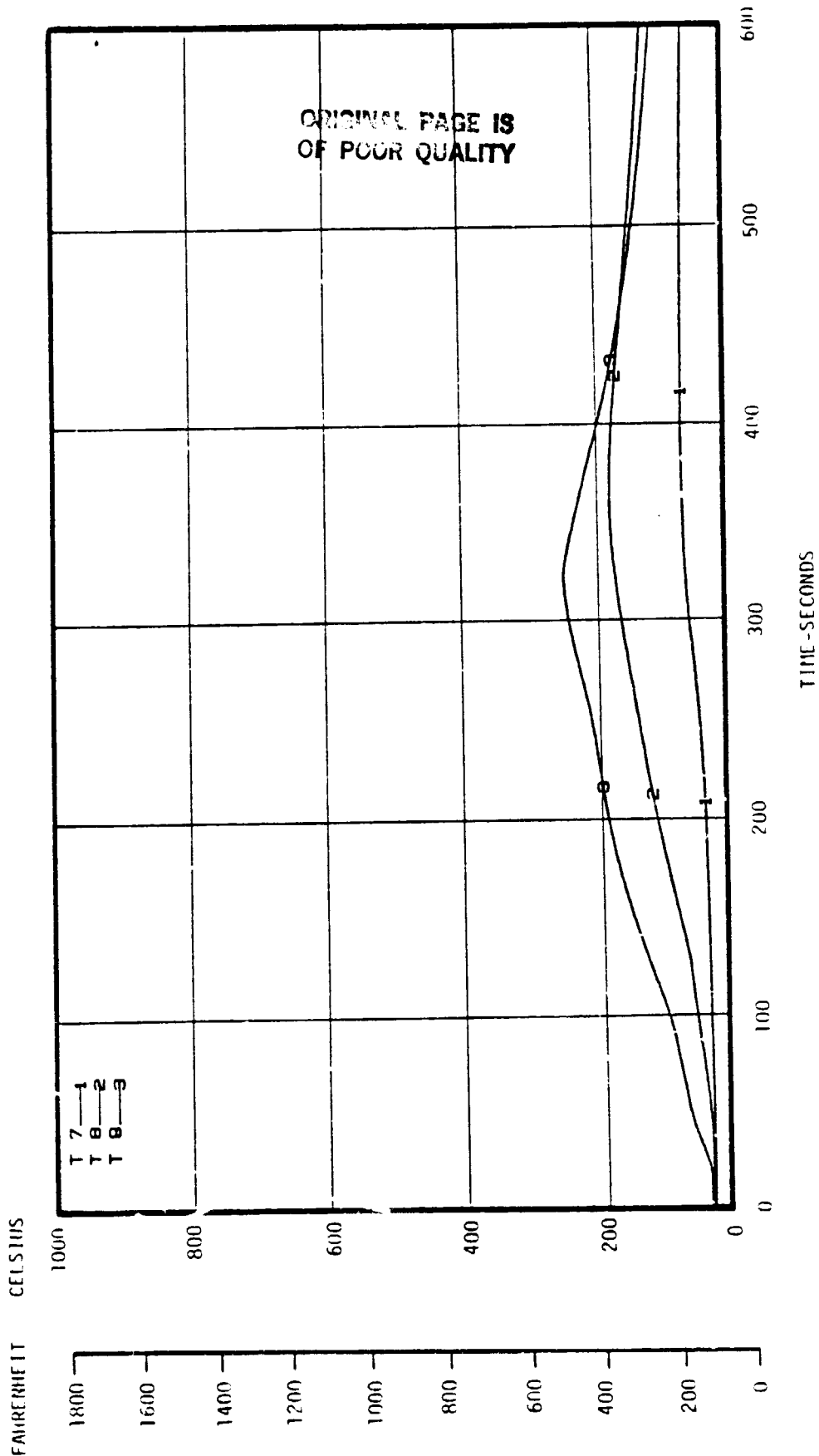


DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/18/82 15.00

NASA-AHS FULL SCALE CUSHION BURN TEST NUMBER 18

CUSHION CONSTRUCTION NUMBER 8.0

SEAT CUSHION TEMPERATURES

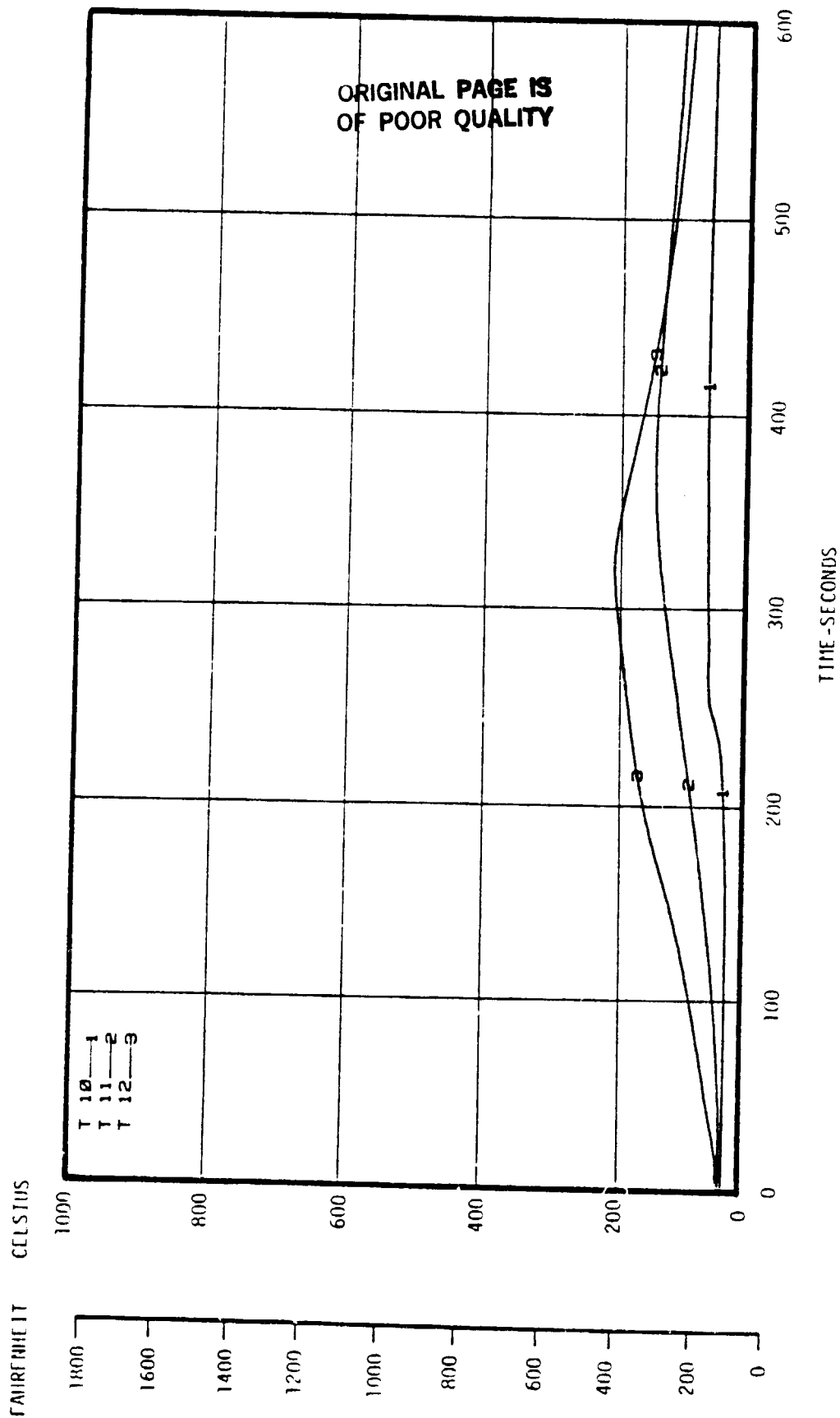


DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/18/82 15.06

NASA-MES FULL SCALE CUSHION BURN TEST NUMBER 18

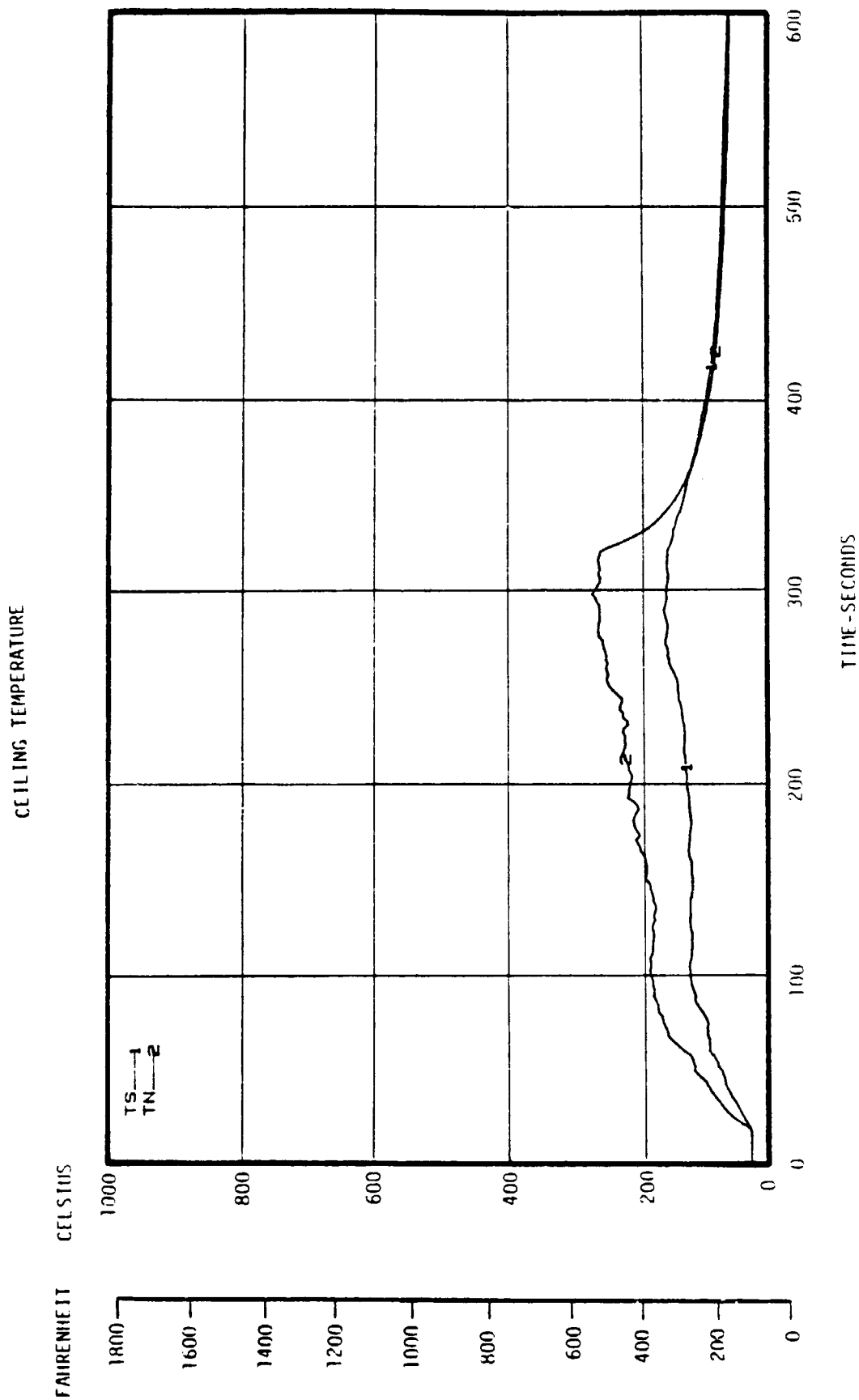
CUSHION CONSTRUCTION NUMBER 8.0

SEAT CUSHION TEMPERATURES



DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/18/82 15.06
 NASA-AMES FULL SCALE CUSHION BURN TEST NUMBER 18
 CUSHION CONSTRUCTION NUMBER 8.2

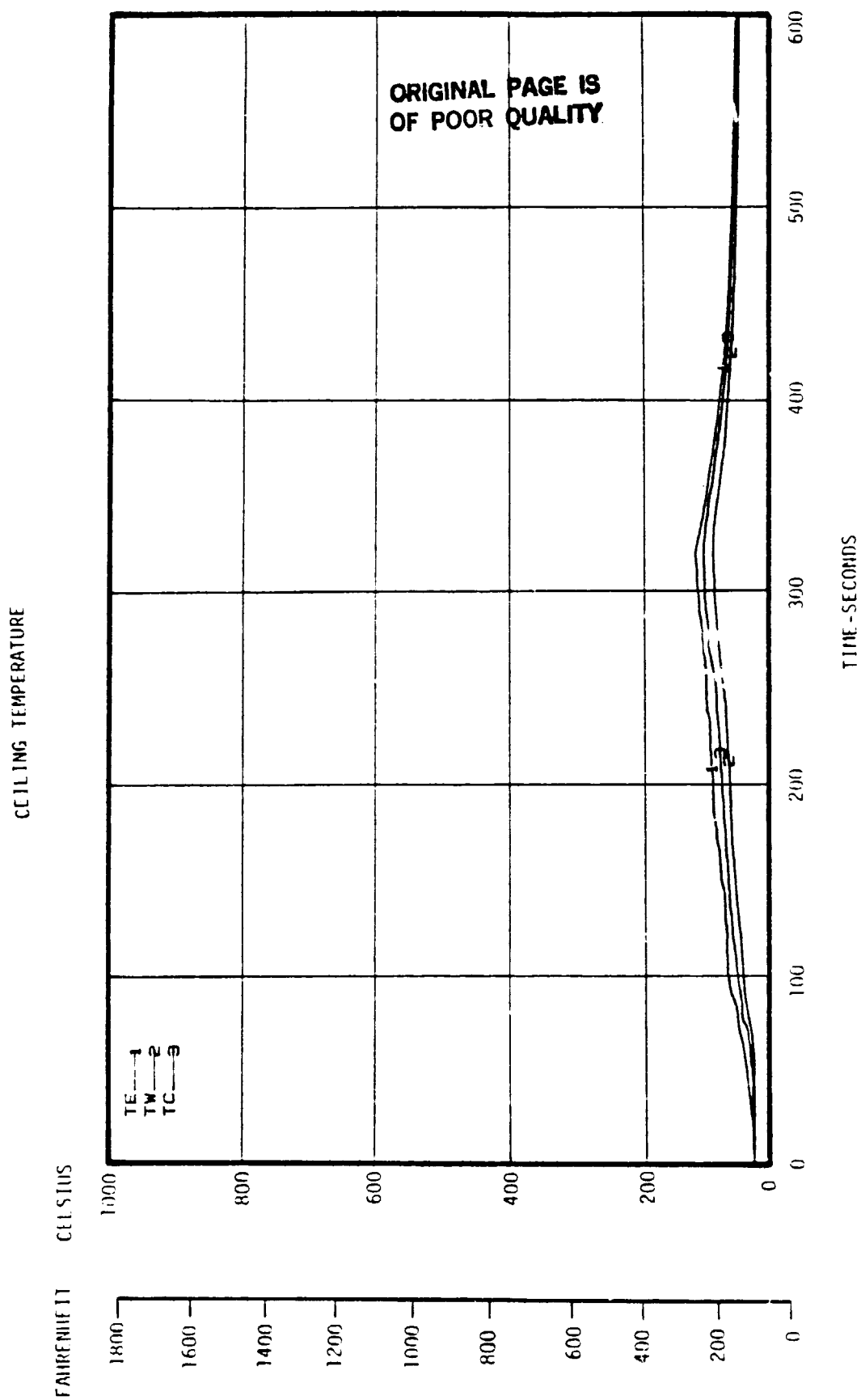
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DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/10/82 15.00

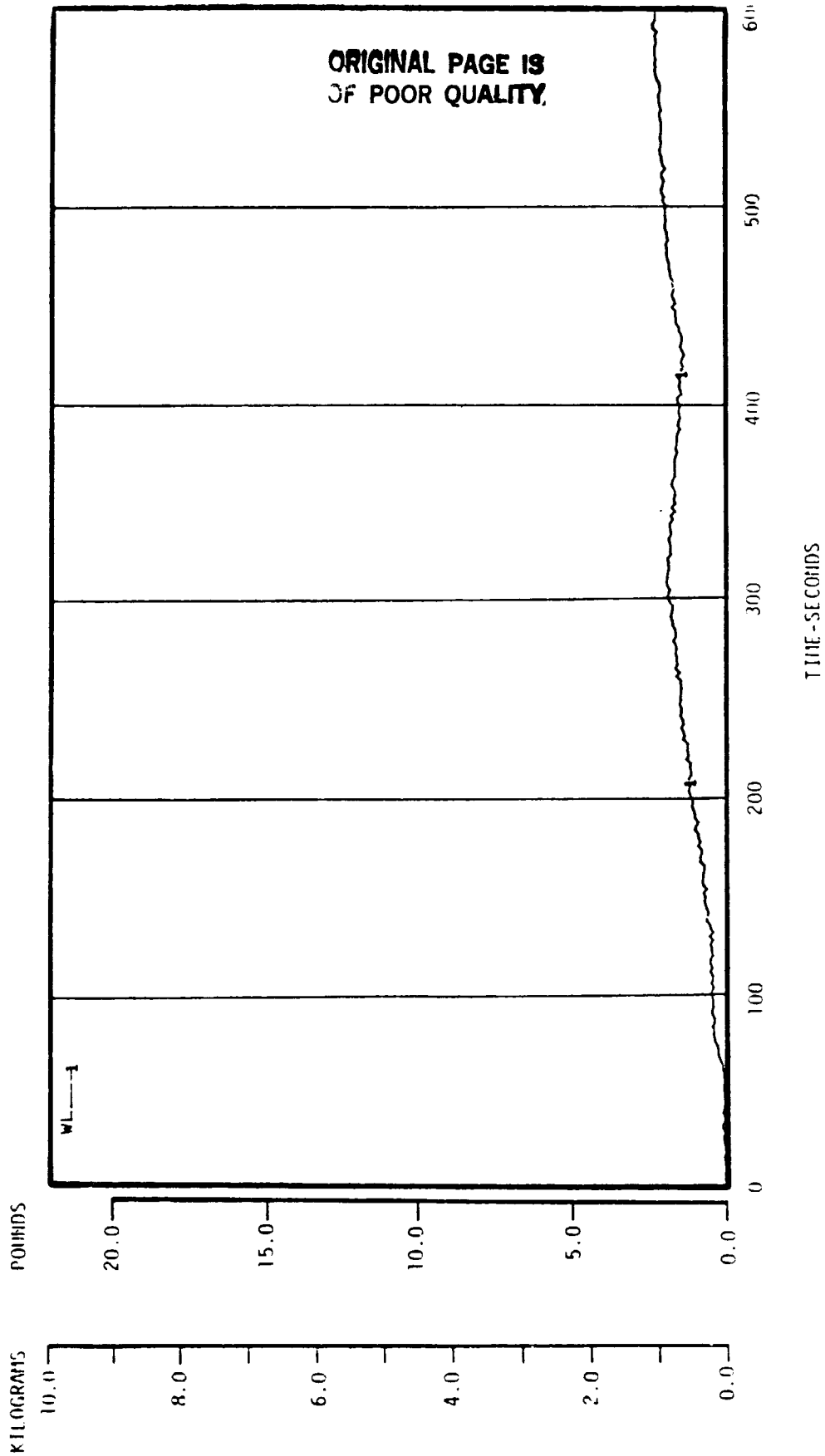
NASA-AMES FULL SCALE CUSHION BURST TEST NUMBER 10

CUSHION CONSTRUCTION NUMBER 0.0



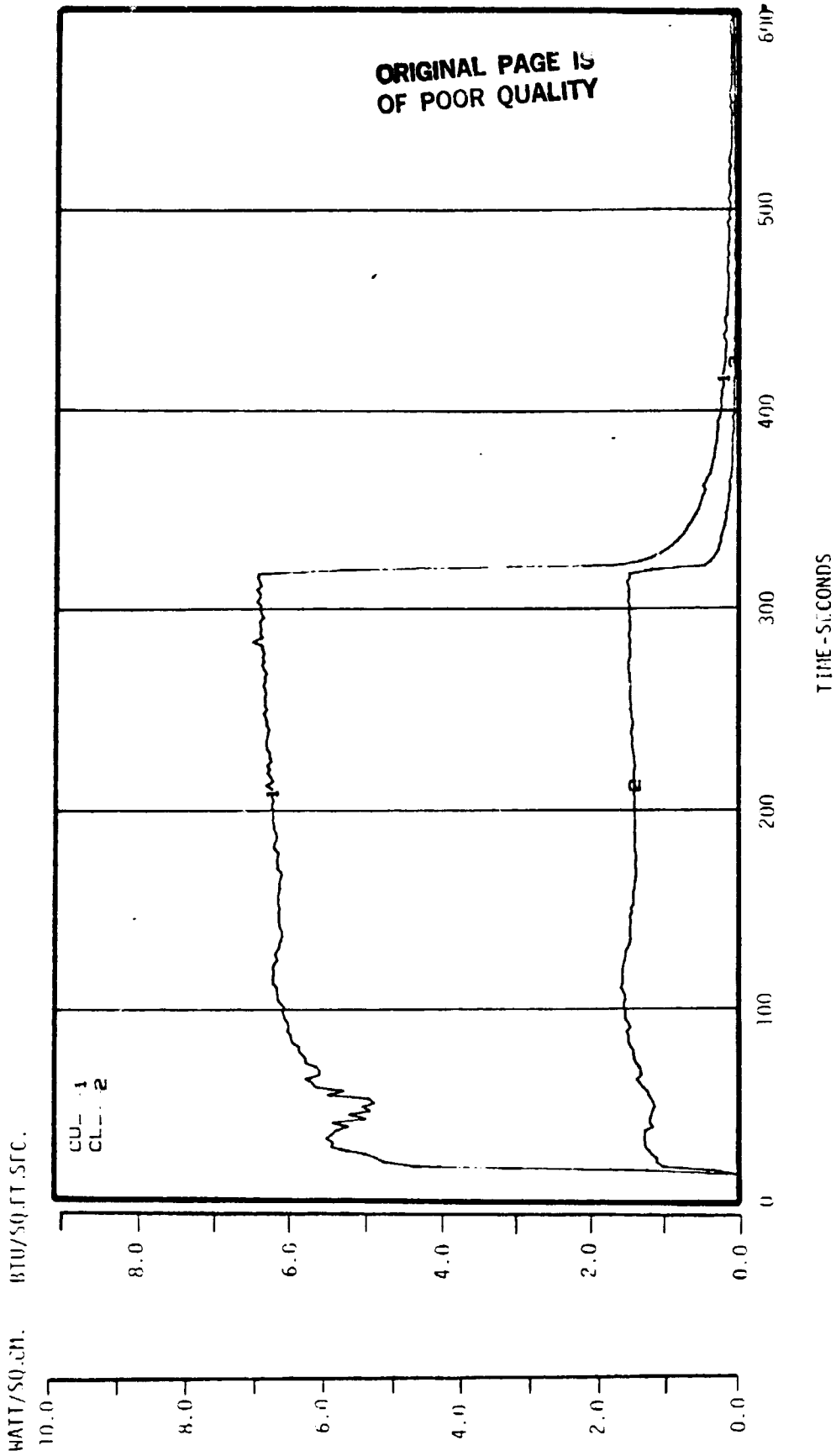
DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 09/16/82 15.08
 NASA-AMES FULL SCALE CUSHION BURST TEST NUMBER 18
 CUSHION CONSTRUCTION NUMBER 8.0

WEIGHT LOSS



DOUGLAS AIRCRAFT CABIN FIRE SIMULATION 03/18/82 15.08
 NASA-AMES FULL SCALE CUSHION BURN TEST NUMBER 18
 CUSHION CONSTRUCTION NUMBER 8.0

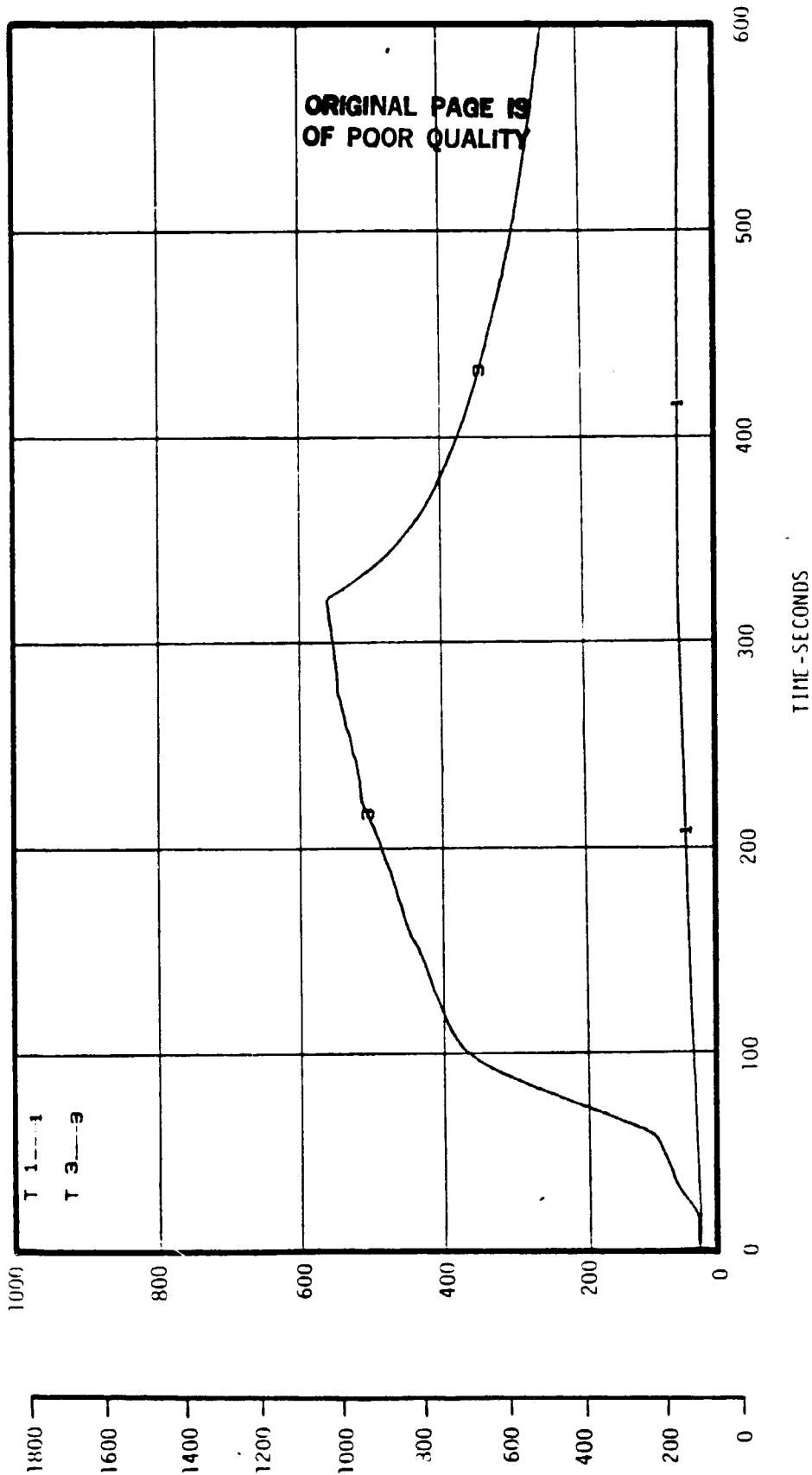
HEAT FLUX



DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/11/02 00.18
 NASA-AMES FULL SCALE CUSHION BURN TEST NUMBER 0
 CUSHION CONSTRUCTION NUMBER 0.0

SEAT CUSHION TEMPERATURES

FAHRENHEIT CELSIUS



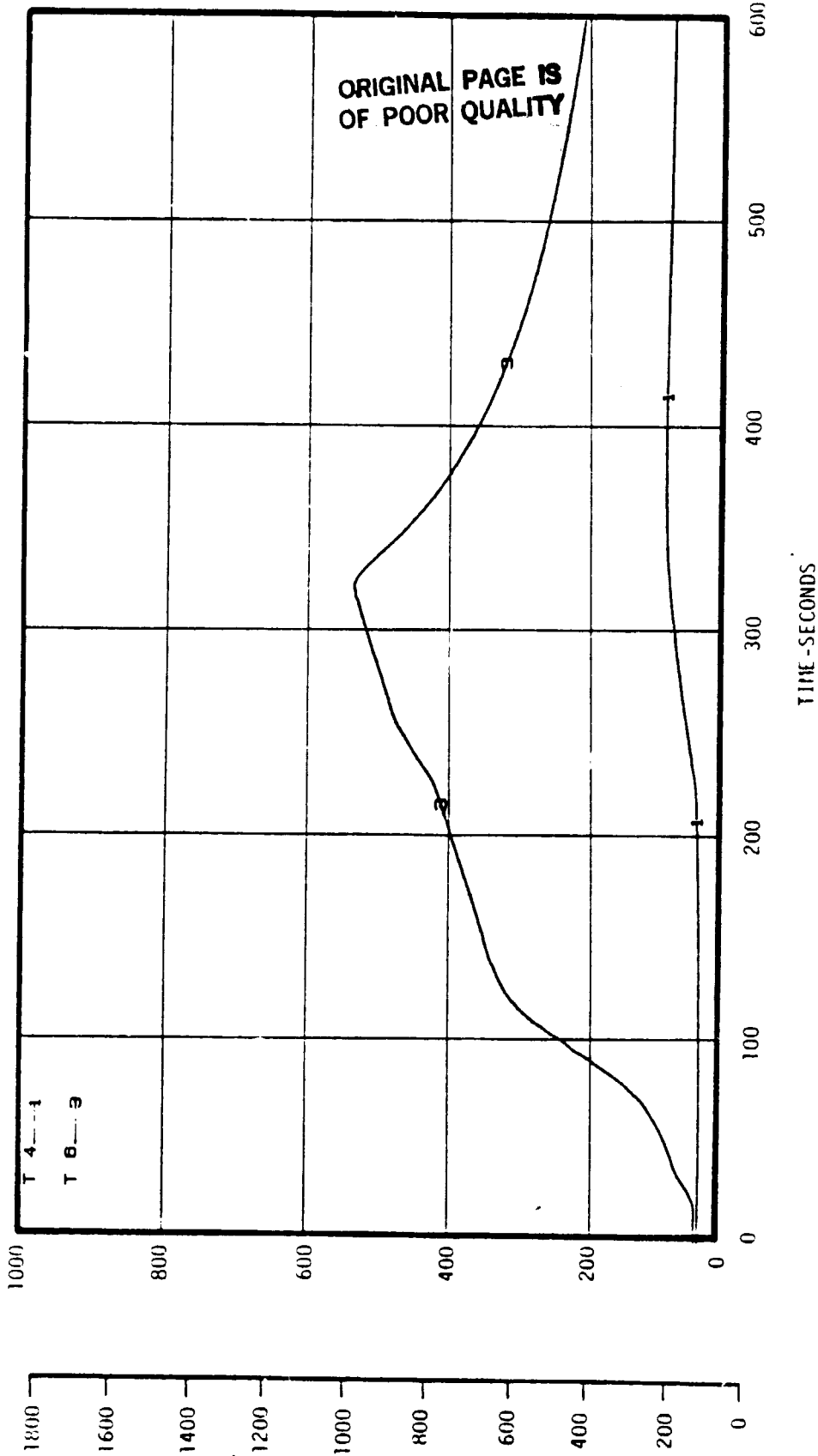
DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/11/82 00.40

NASA-AMES FULL SCALE CUSHION BURN TEST NUMBER 8.

CUSHION CONSTRUCTION NUMBER 9.0

SEAT CUSHION TEMPERATURES

FAHRENHEIT CELSIUS

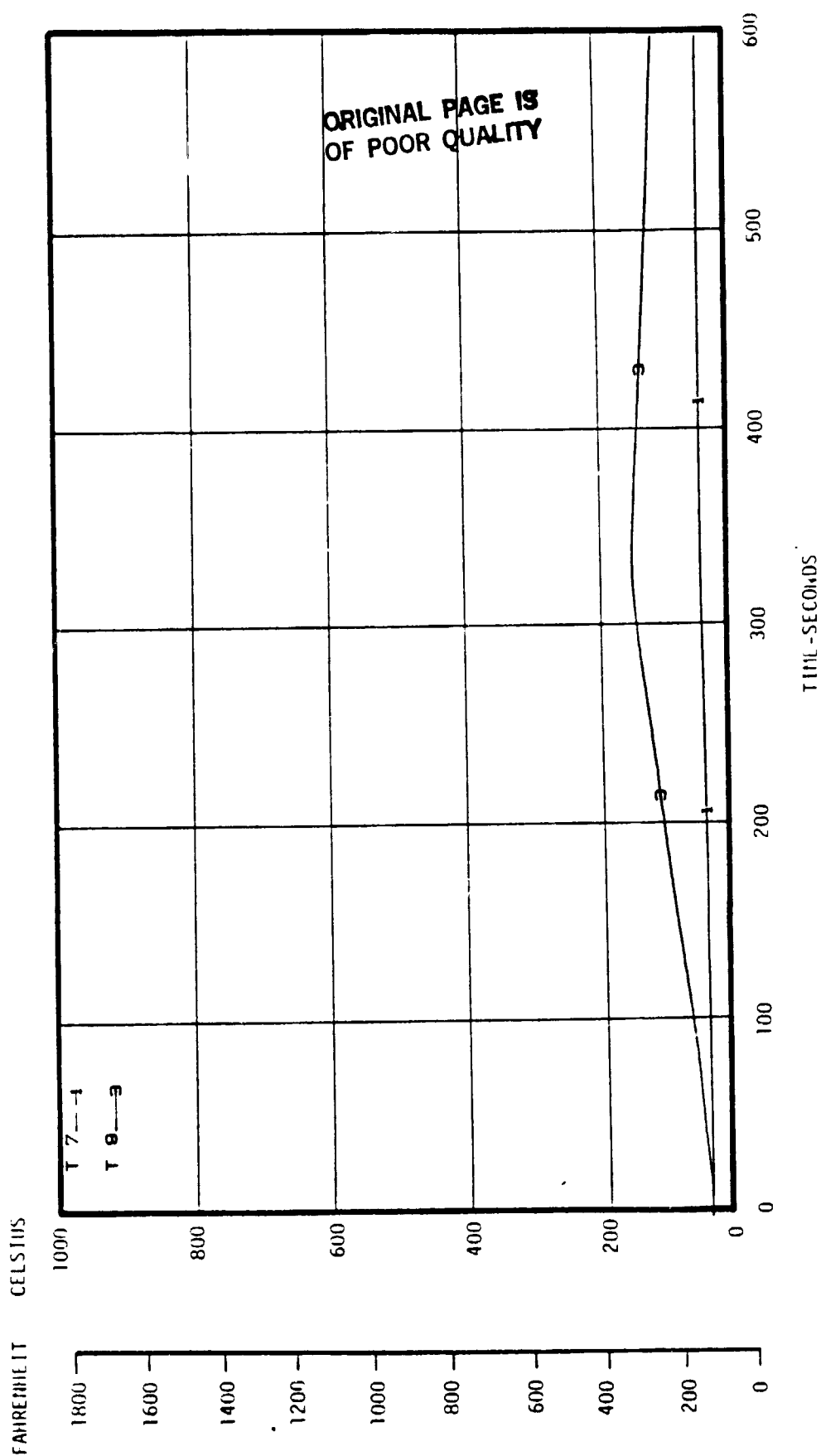


DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/11/62 00.40.

NASA-AMES FULL SCALE CUSHION BURN TEST NUMBER 8.

CUSHION CONSTRUCTION NUMBER 8-2

SEAT CUSHION TEMPERATURES



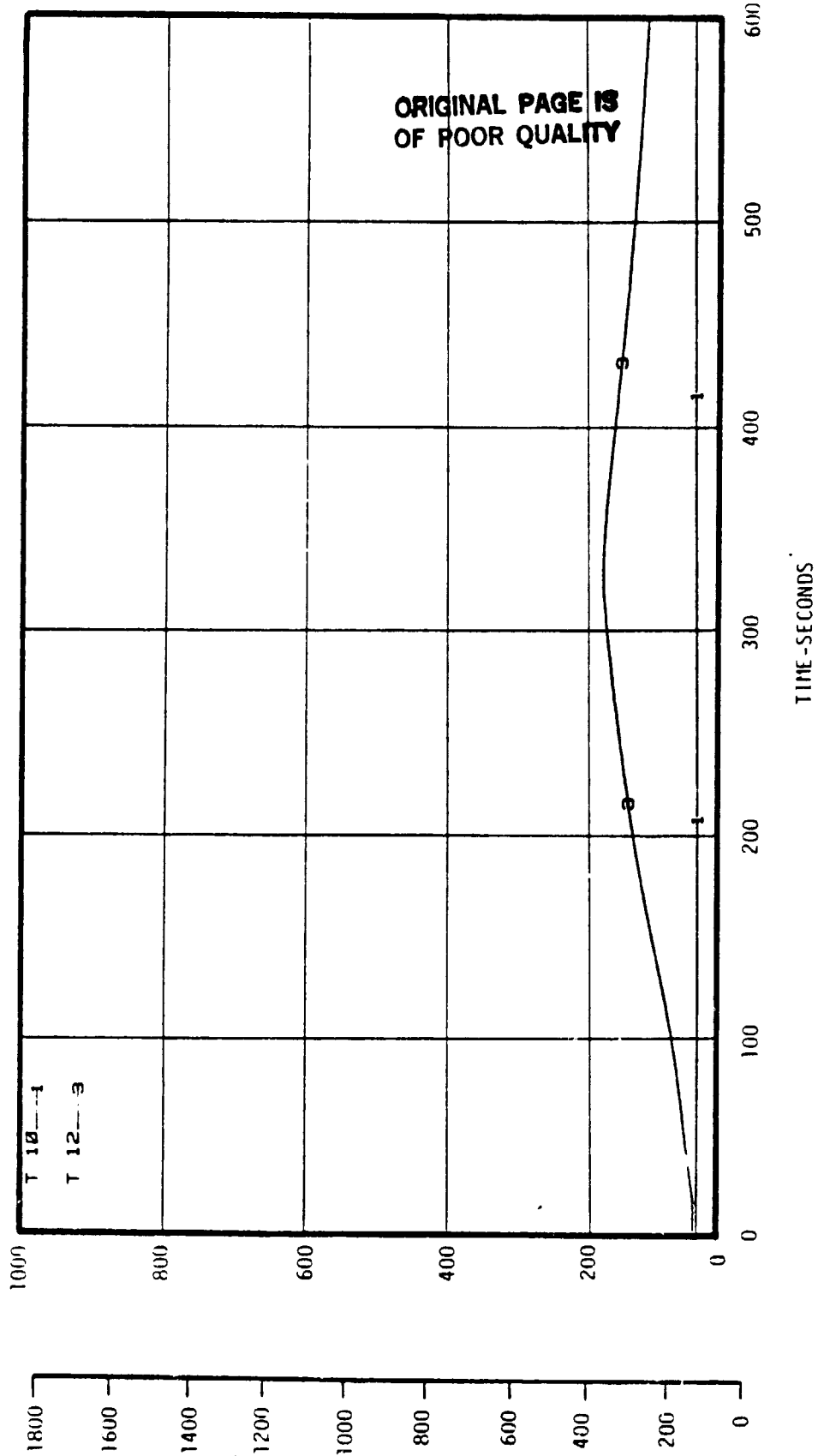
DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 002/11/82 009.40

NASA-AMES FULL SCALE CUSHION BURN TEST NUMBER 8

CUSHION CONSTRUCTION NUMBER 9-1

SEAT CUSHION TEMPERATURES

FAHRENHEIT CELSIUS



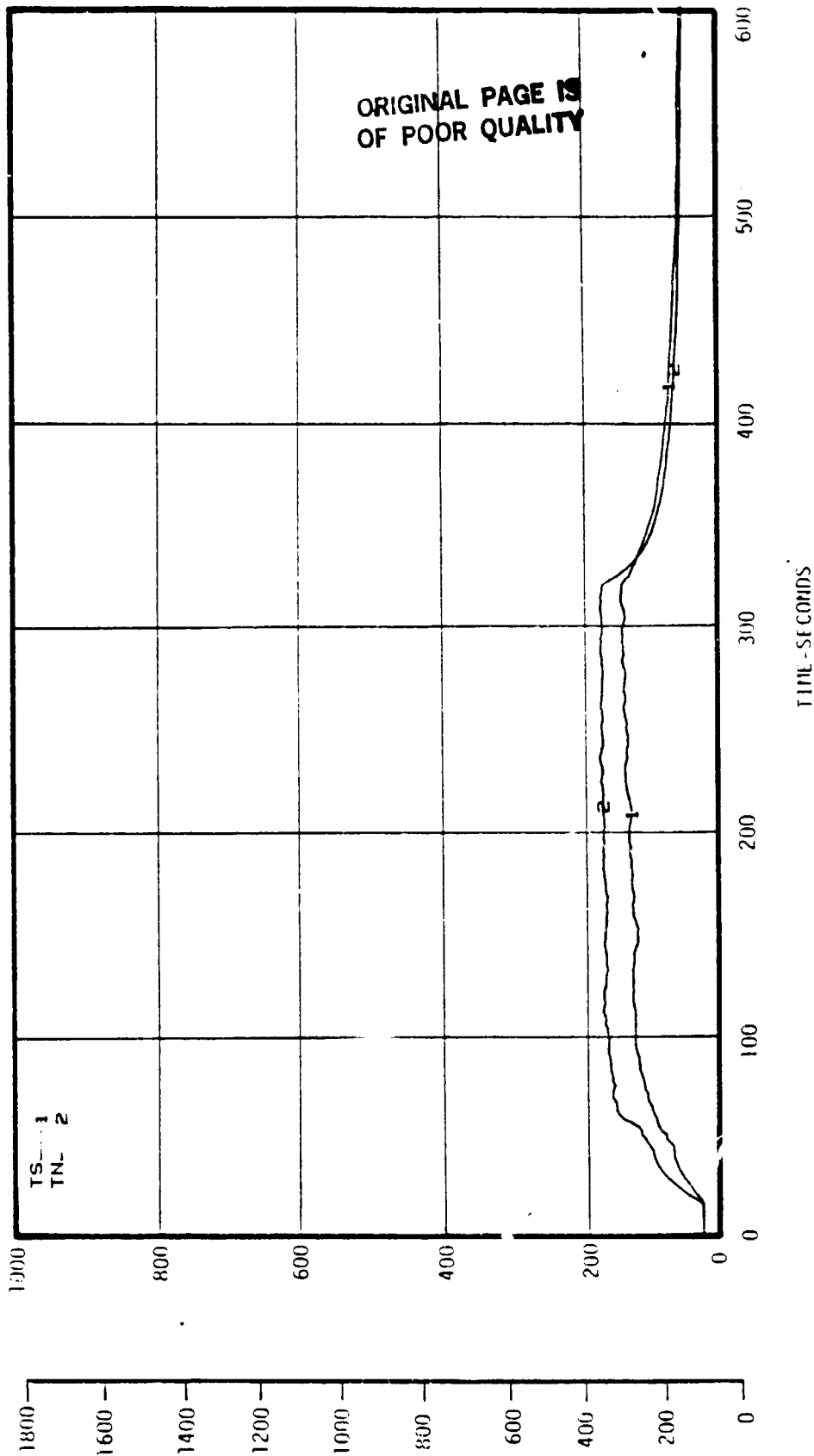
DOUGLAS AIRCRAFT CASE IN FIRE SIMULATOR 03/11/82--09-40

NASA-AMES FULL SCALE CUSHION BURN TEST NUMBER--8

CUSHION CONSTRUCTION NUMBER--9-2

CEILING TEMPERATURE

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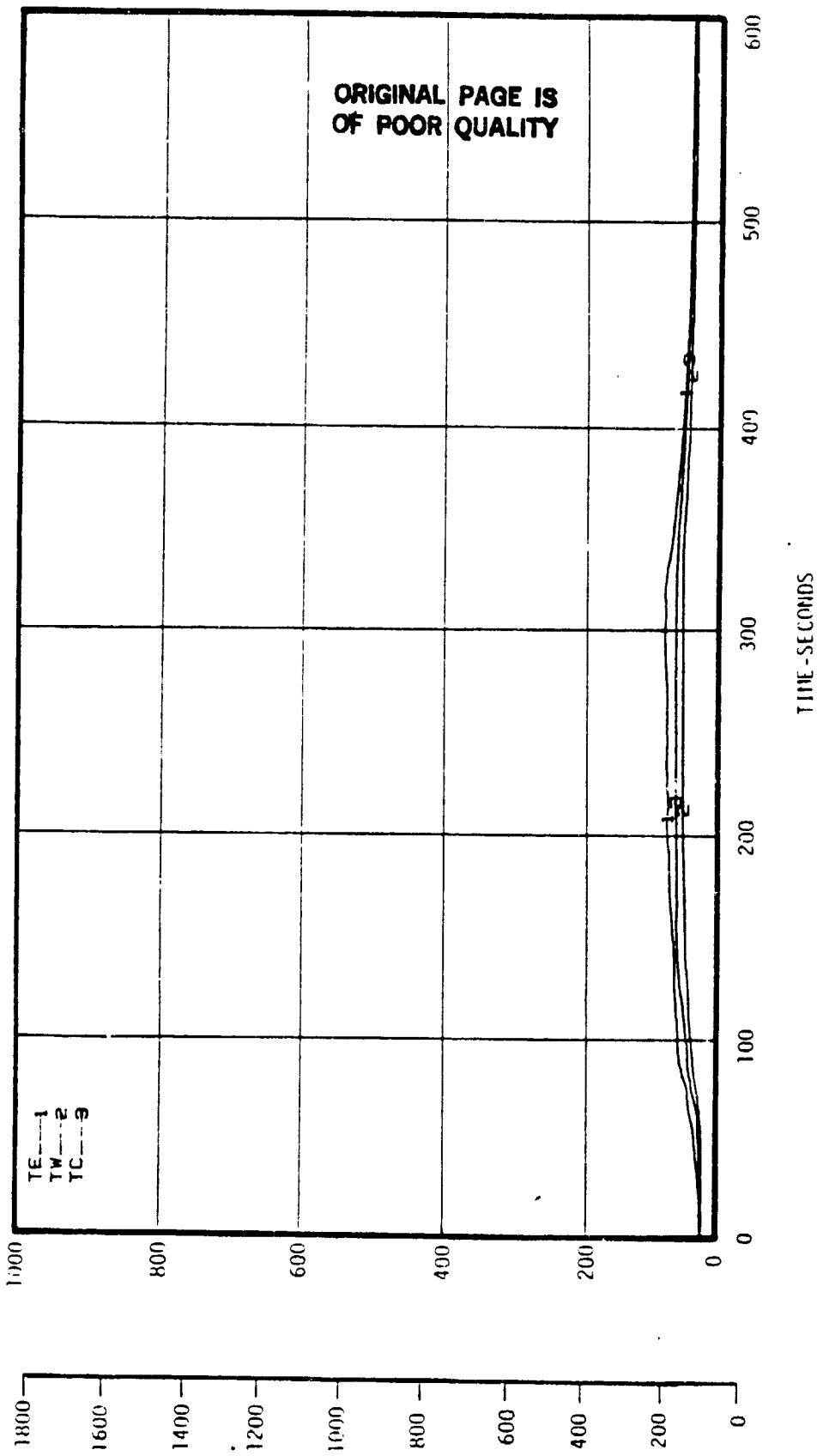
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CUSHION CONSTRUCTION NUMBER 9.0

CEILING TEMPERATURE

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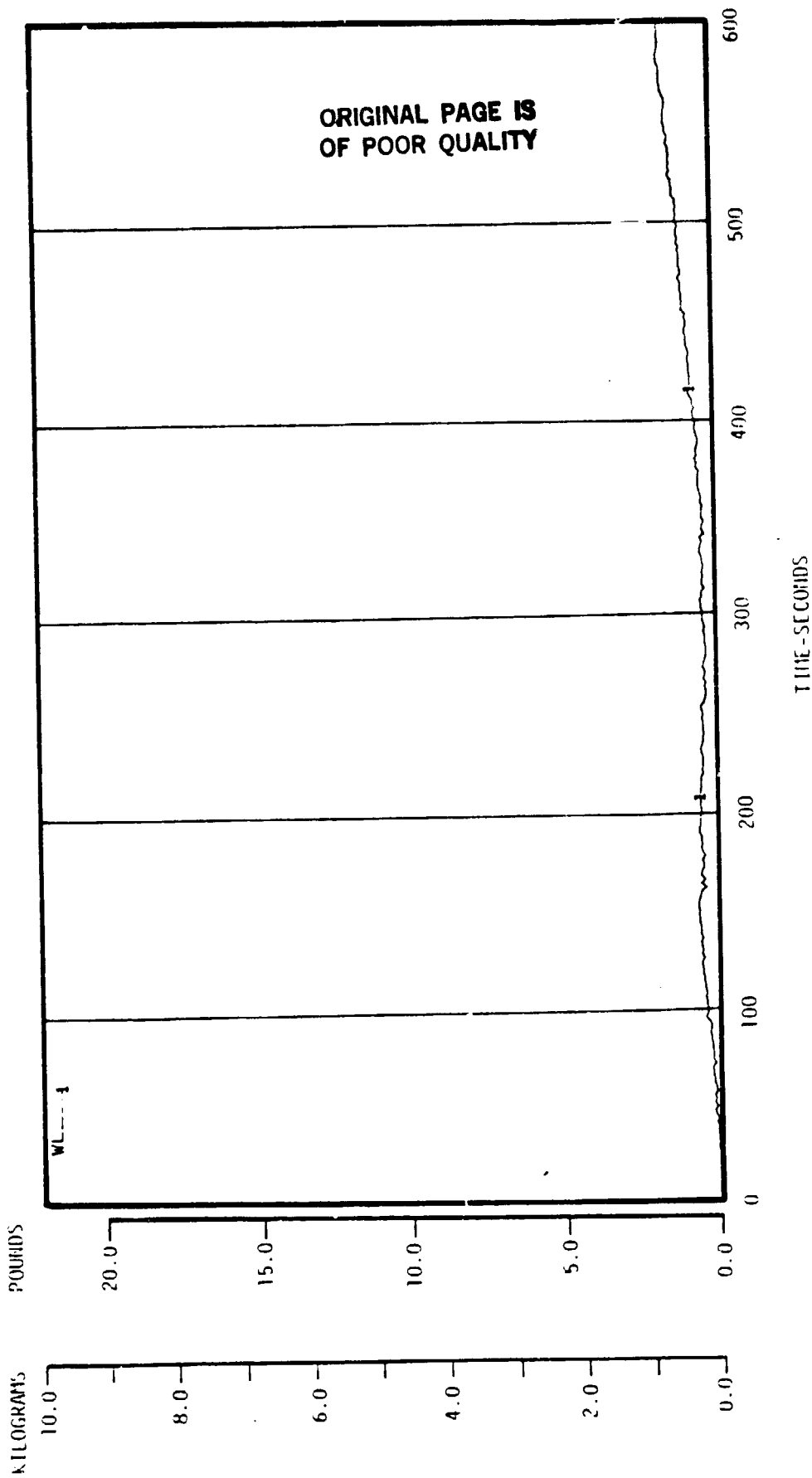


DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/11/82 09.40

NASA-AMES FULL SCALE CUSHION BURN TEST NUMBER 8

CUSHION CONSTRUCTION NUMBER 9.0

WEIGHT LOSS

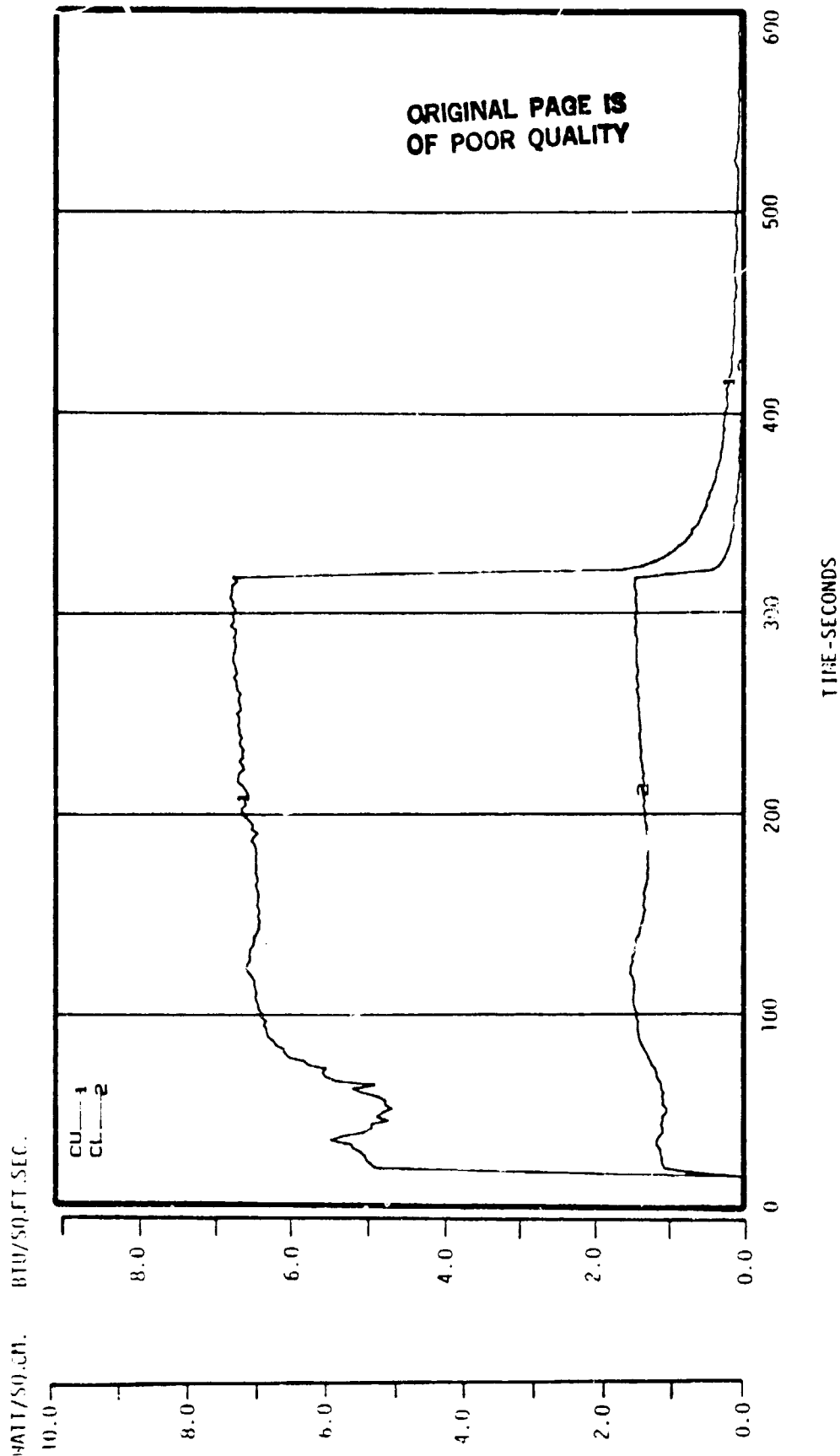


DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/11/82 09.40

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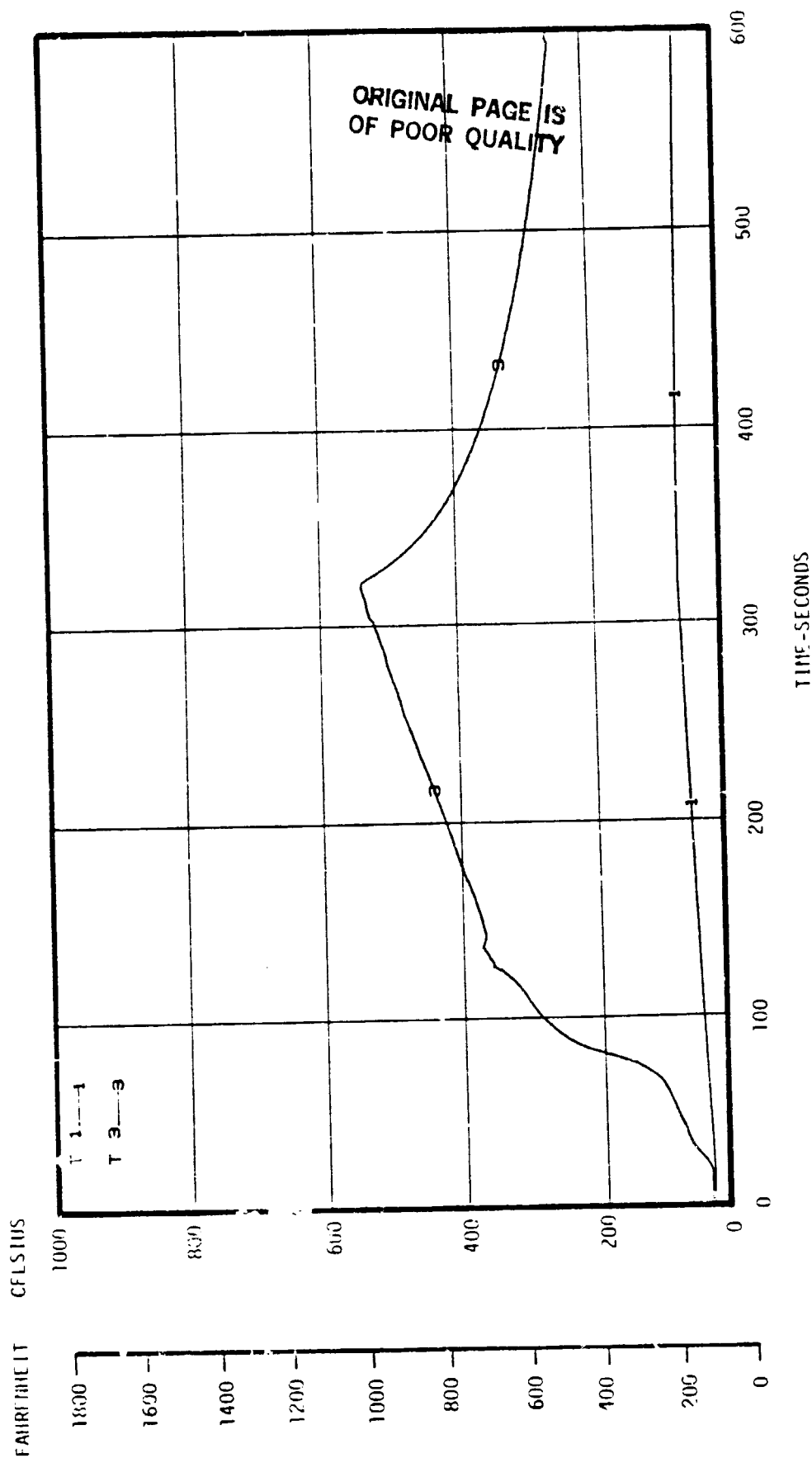
CUSHION CONSTRUCTION NUMBER 000

HEAT FLUX



DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 23/17/82 08.10
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 CUSHION CONSTRUCTION NUMBER 8.0

SEAT CUSHION TEMPERATURES

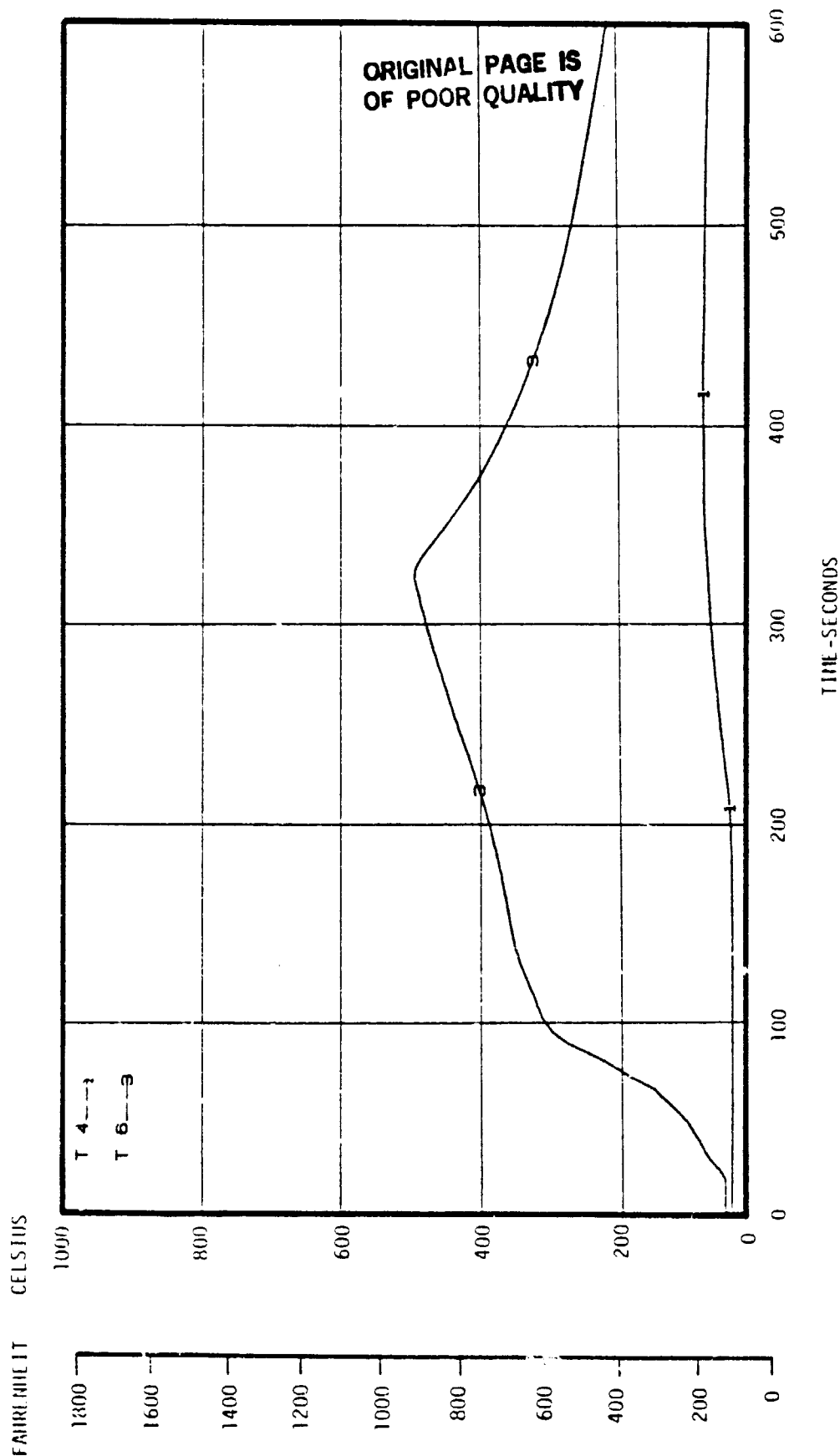


DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/17/82 08.10

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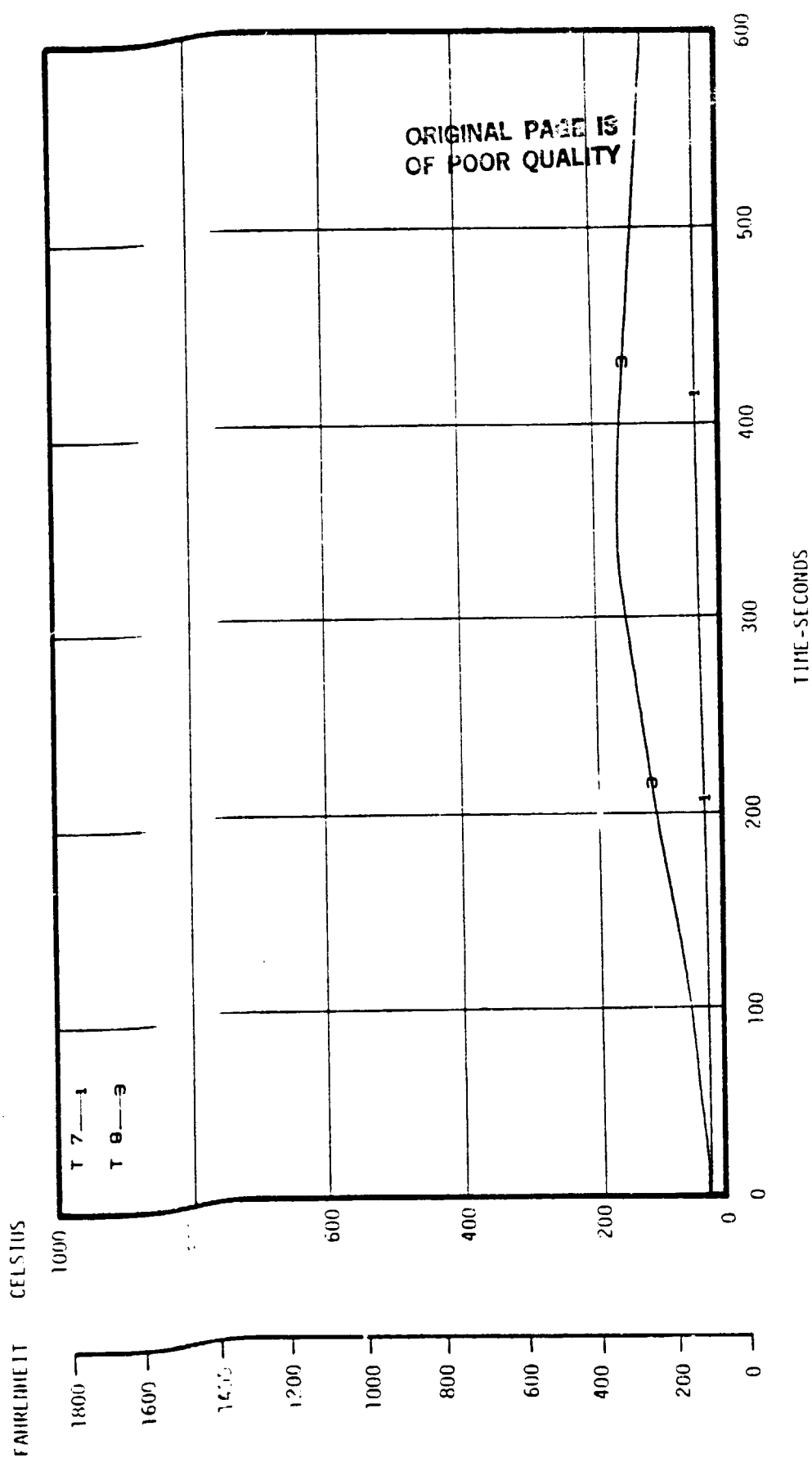
CUSHION CONSTRUCTION NUMBER 0.0

SEAT CUSHION TEMPERATURES



DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/17/82 08, 10
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SEAT CUSHION TEMPERATURES

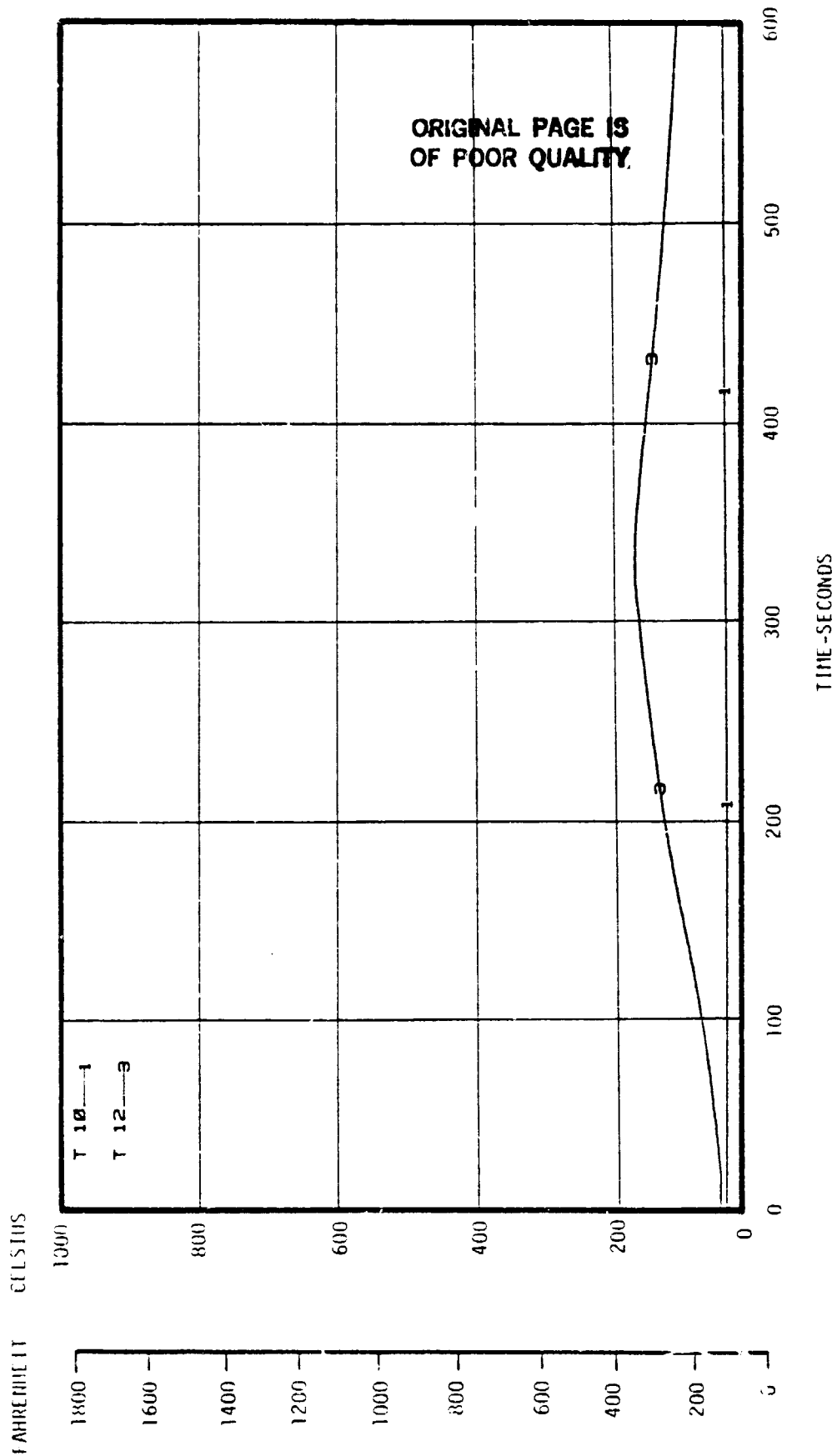


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NASA-AMES FULL SCALE CUSHION BURN TEST NUMBER 10

CUSHION CONSTRUCTION NUMBER 8.0

SEAT CUSHION TEMPERATURES



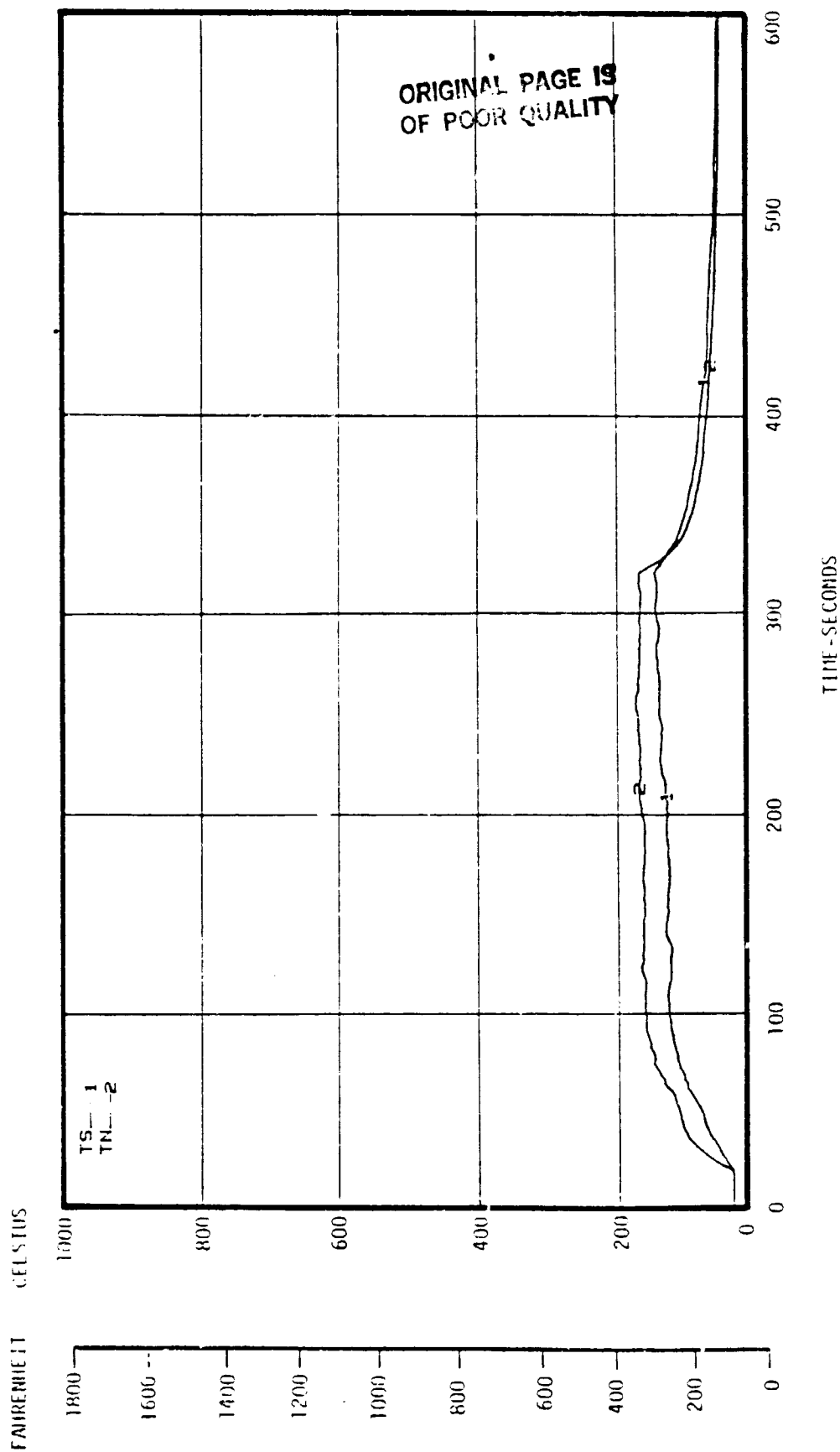
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DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/17/82 08.10

NASA-AMES FULL SCALE CUSHION BURN TEST NUMBER 18

CUSHION CONSTRUCTION NUMBER 8.0

CEILING TEMPERATURE



DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/17/82 00, 10

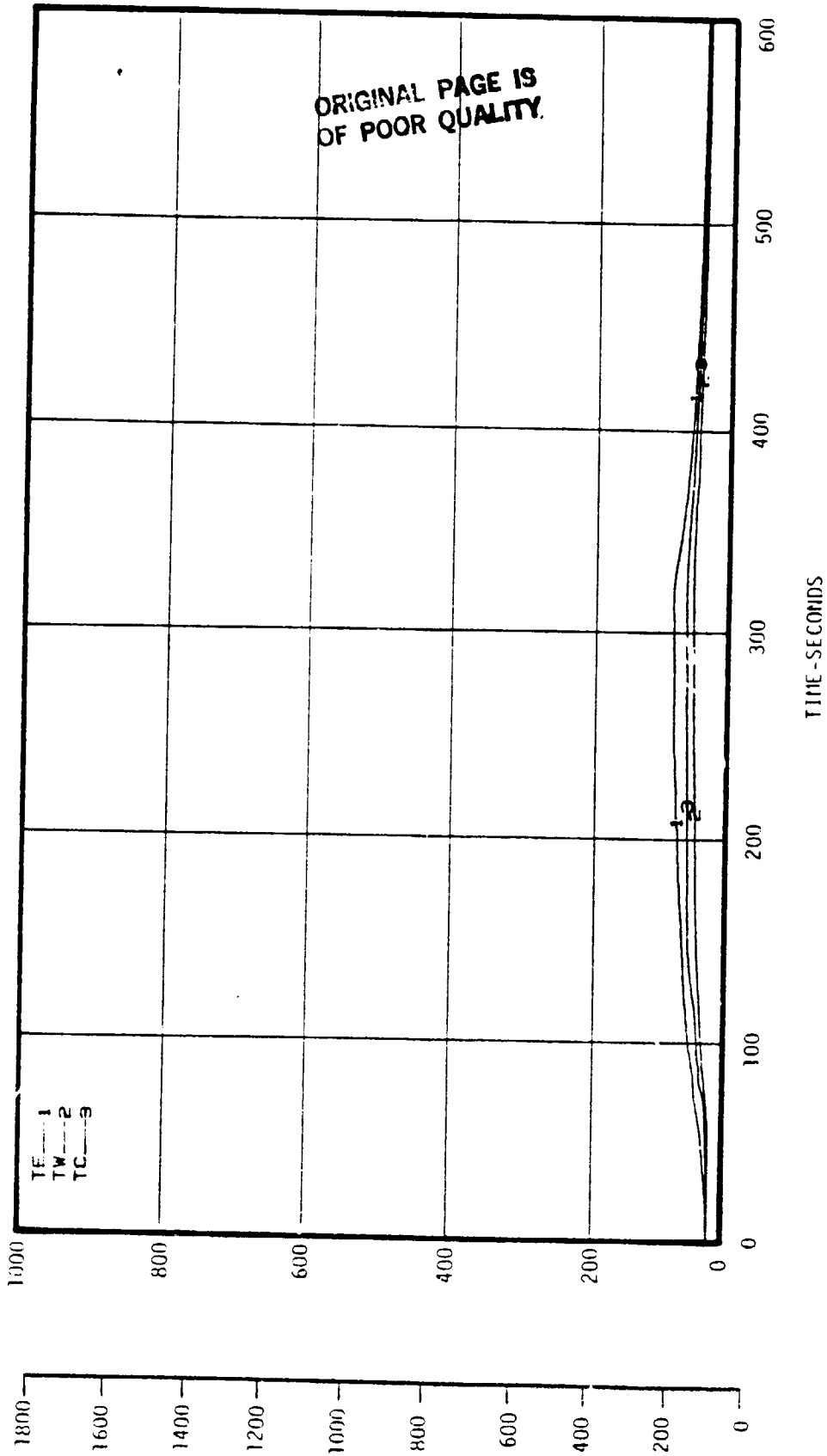
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CEILING TEMPERATURE

CELSIUS

FAHRENHEIT

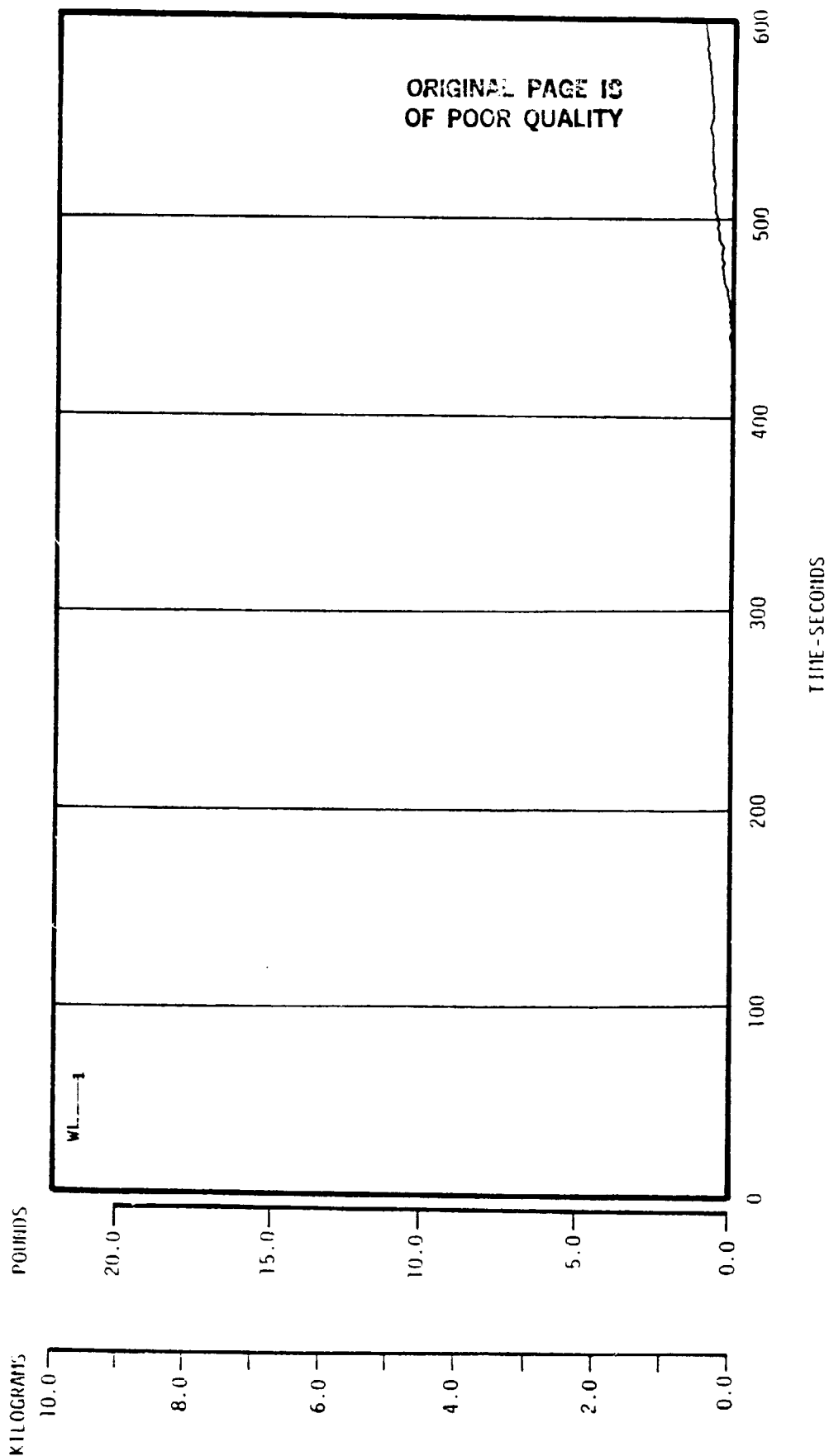


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CUSHION CONSTRUCTION NUMBER 0.0

WEIGHT LOSS

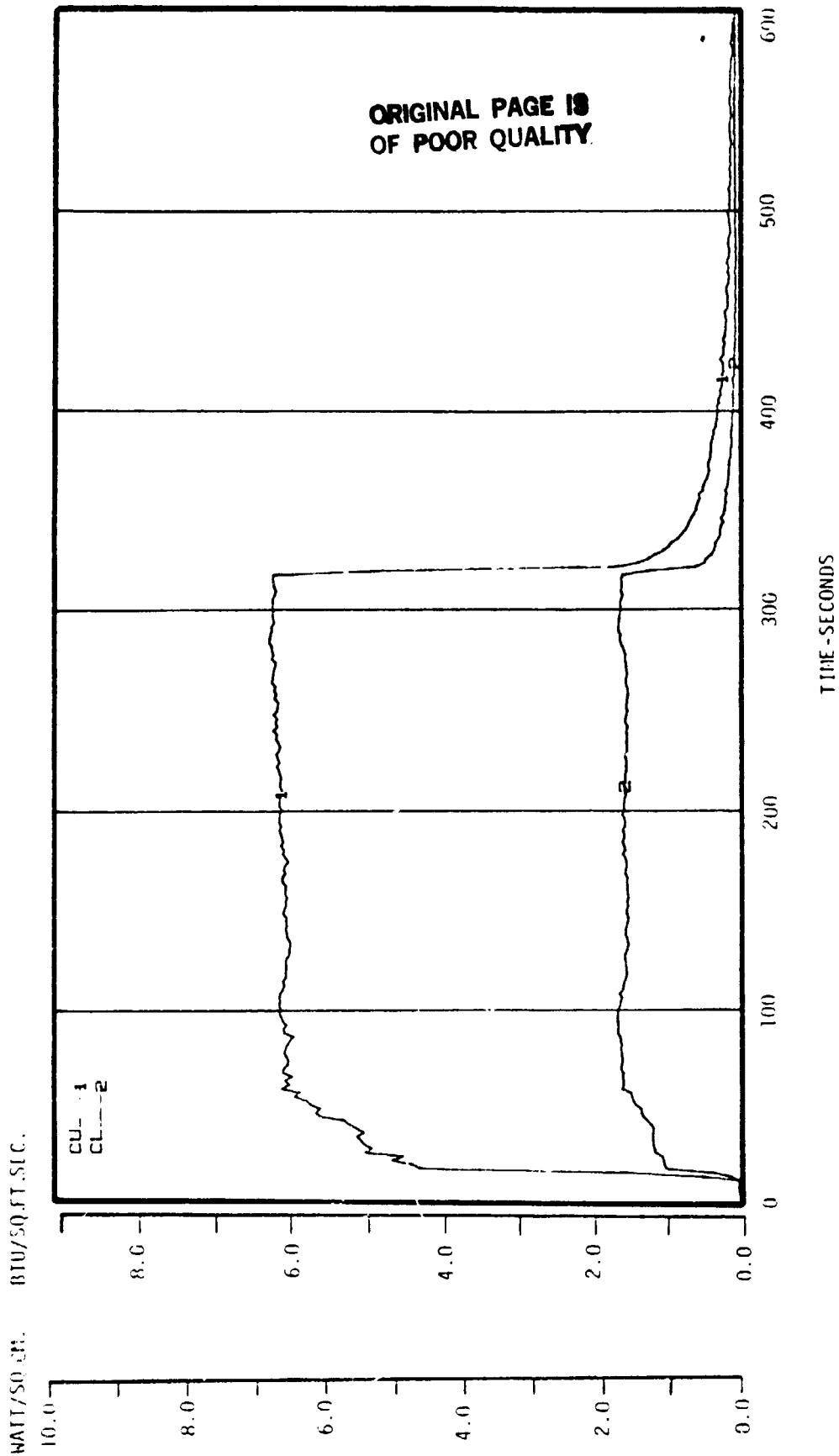


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NASA-AMES FULL SCALE CUSHION BURN TEST NUMBER 19

CUSHION CONSTRUCTION NUMBER 3.0

HEAT FLUX

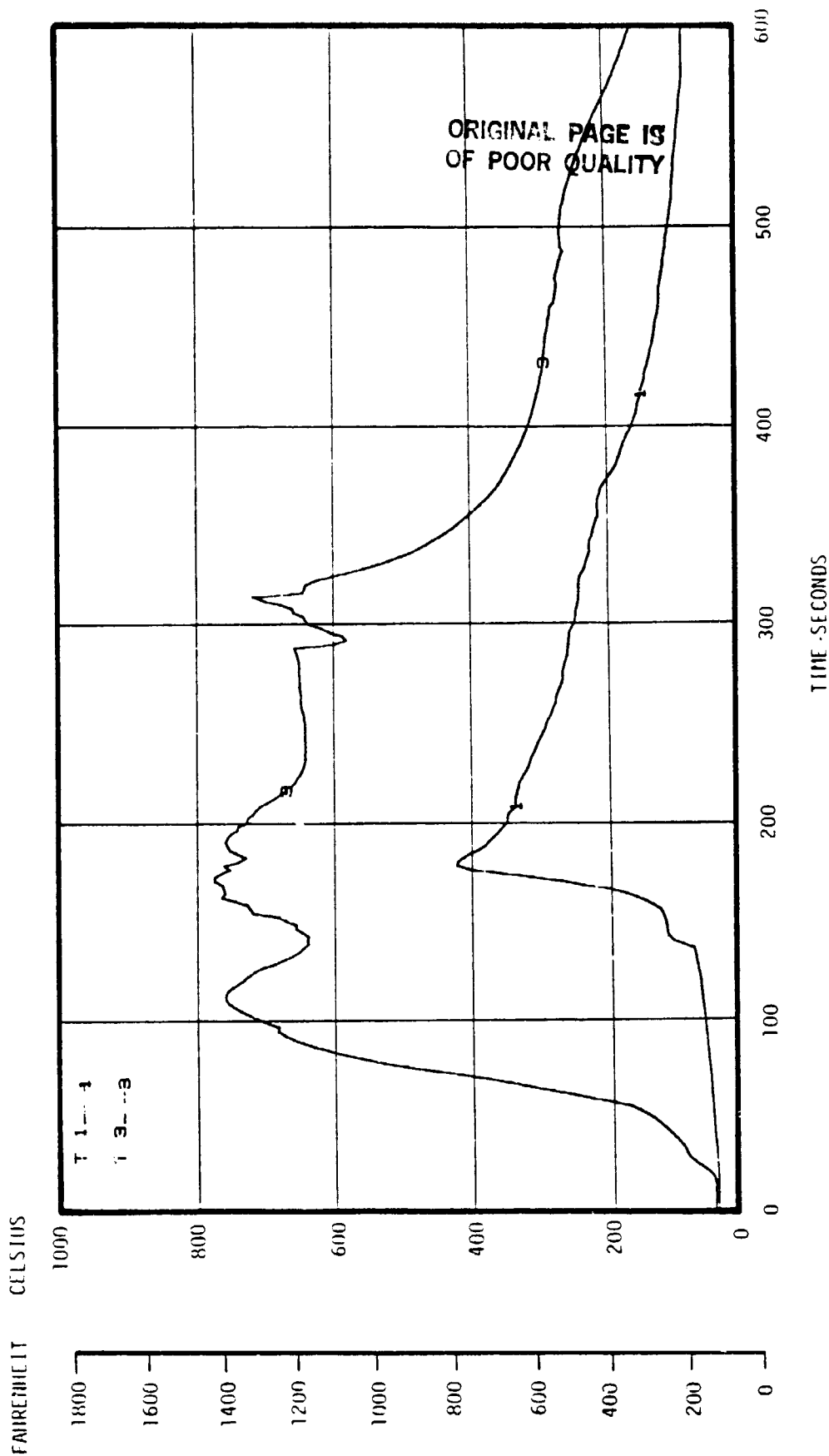


DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 09/11/82 14.91

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CUSHION CONSTRUCTION NUMBER 15.2

SEAT CUSHION TEMPERATURES

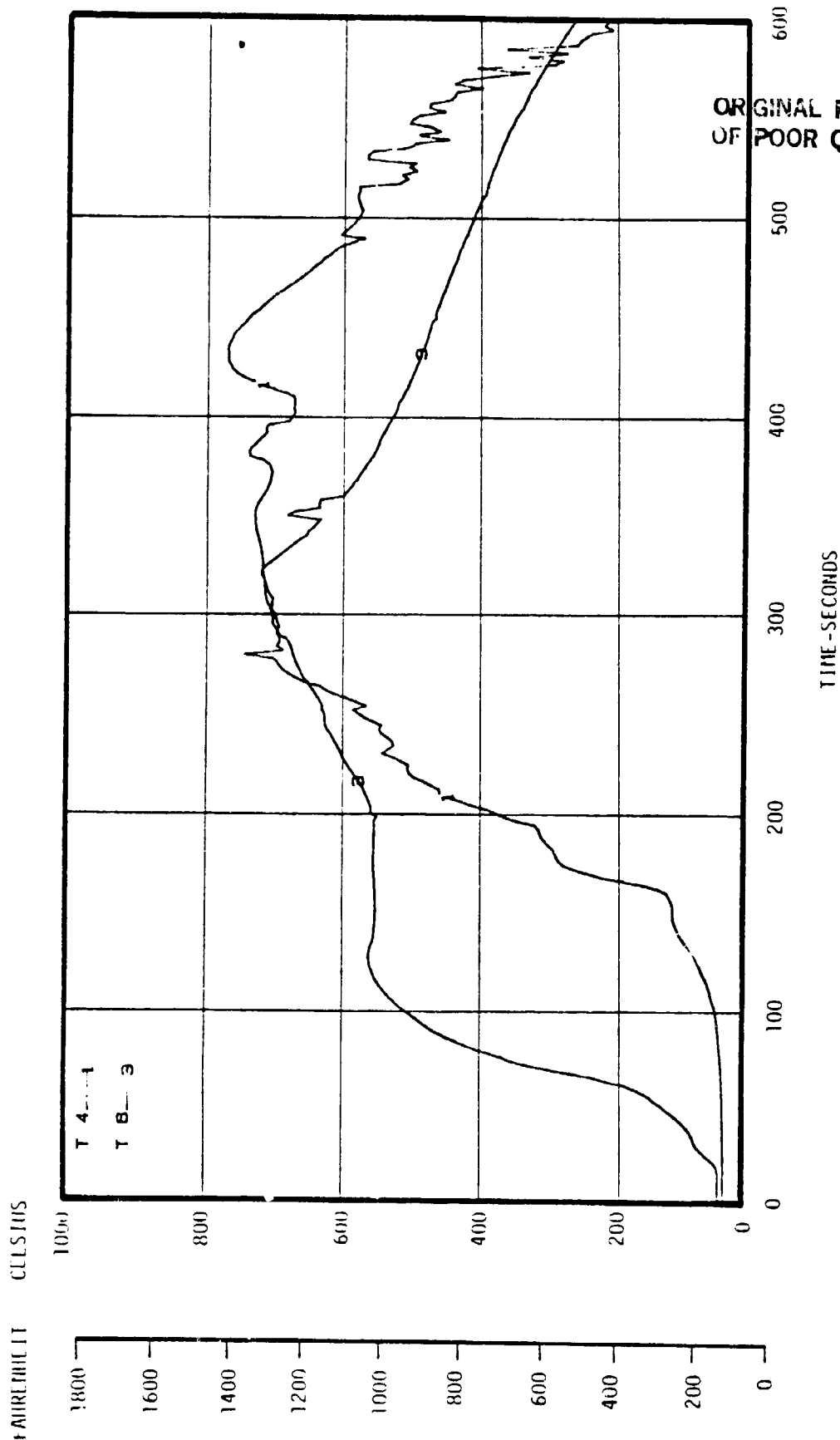


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CUSHION CONSTRUCTION NUMBER 10.0

SEAT CUSHION TEMPERATURES



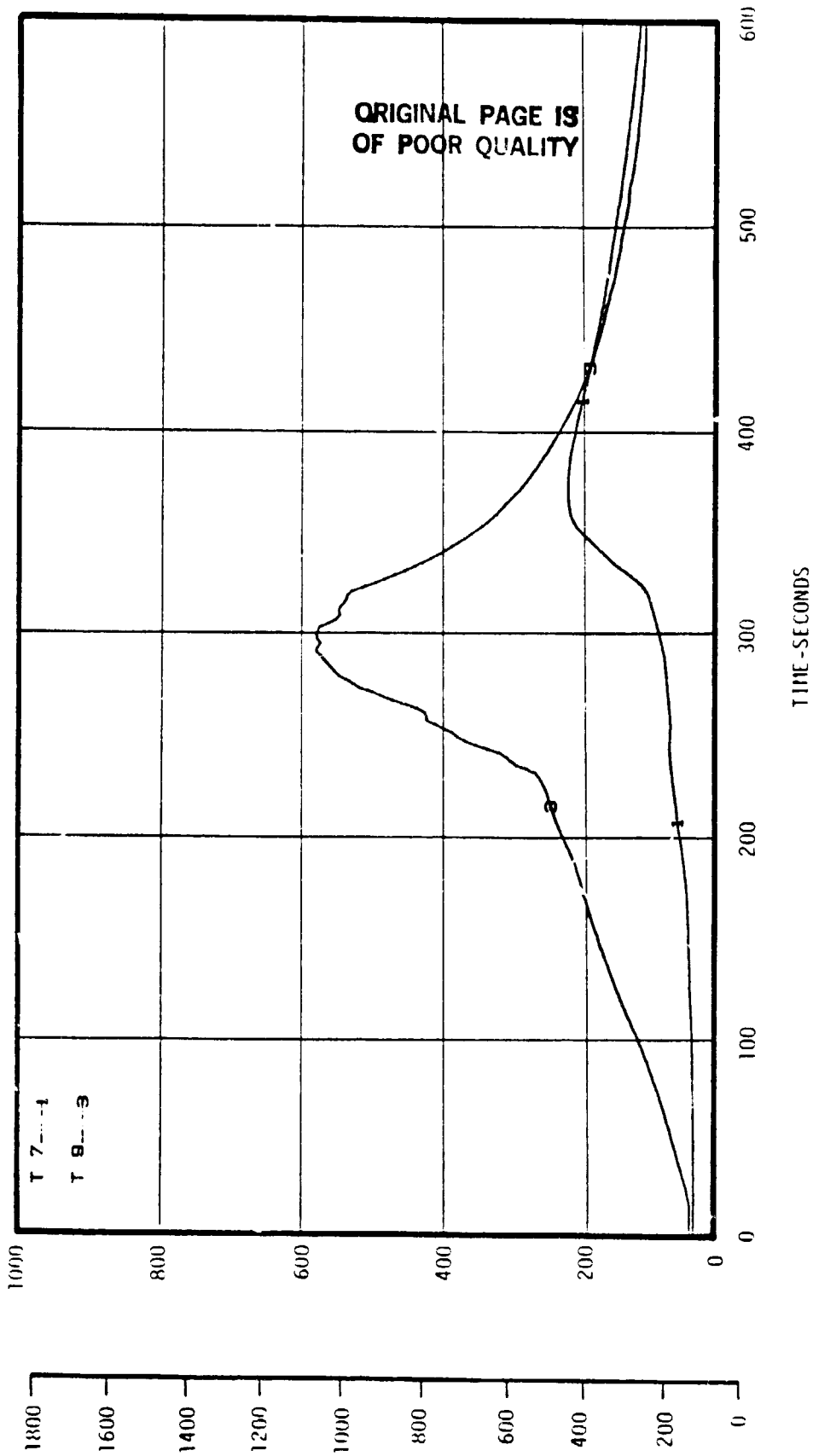
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CUSHION CONSTRUCTION NUMBER 10.0

SEAT CUSHION TEMPERATURES

FAHRENHEIT CELSIUS



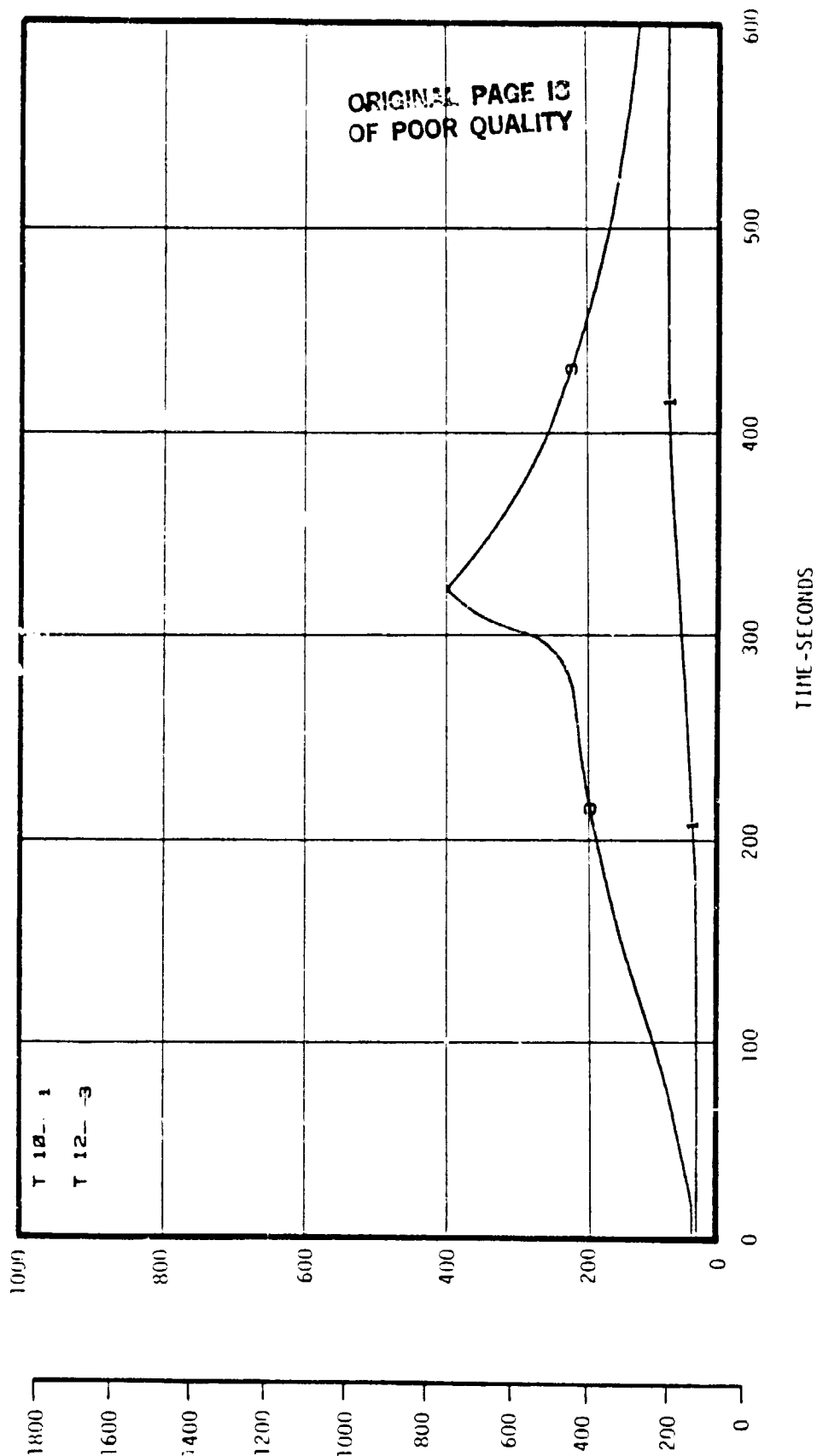
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CUSHION CONSTRUCTION NUMBER 10.0

SEAT CUSHION TEMPERATURES

FAHRENHEIT CELSIUS



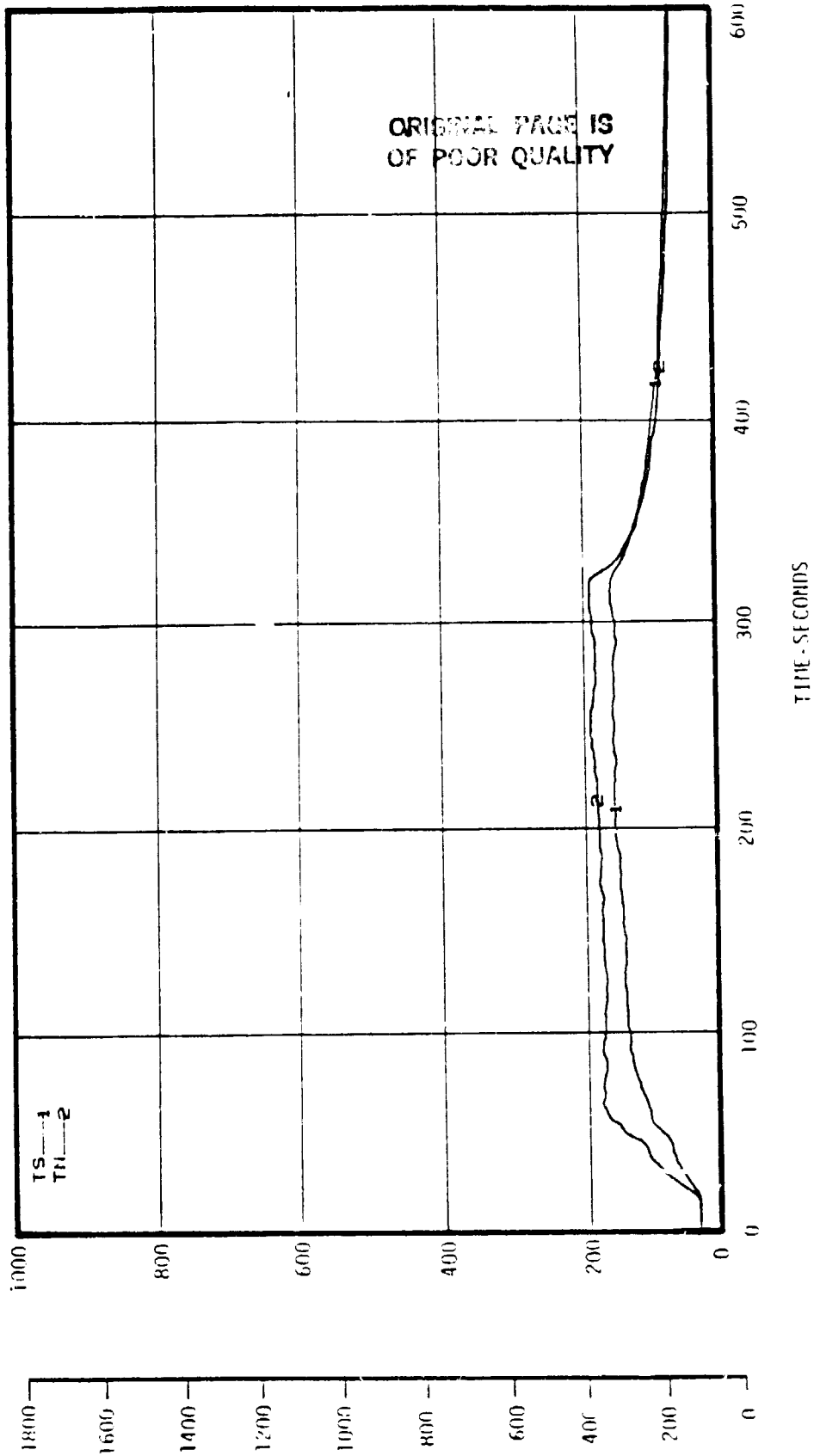
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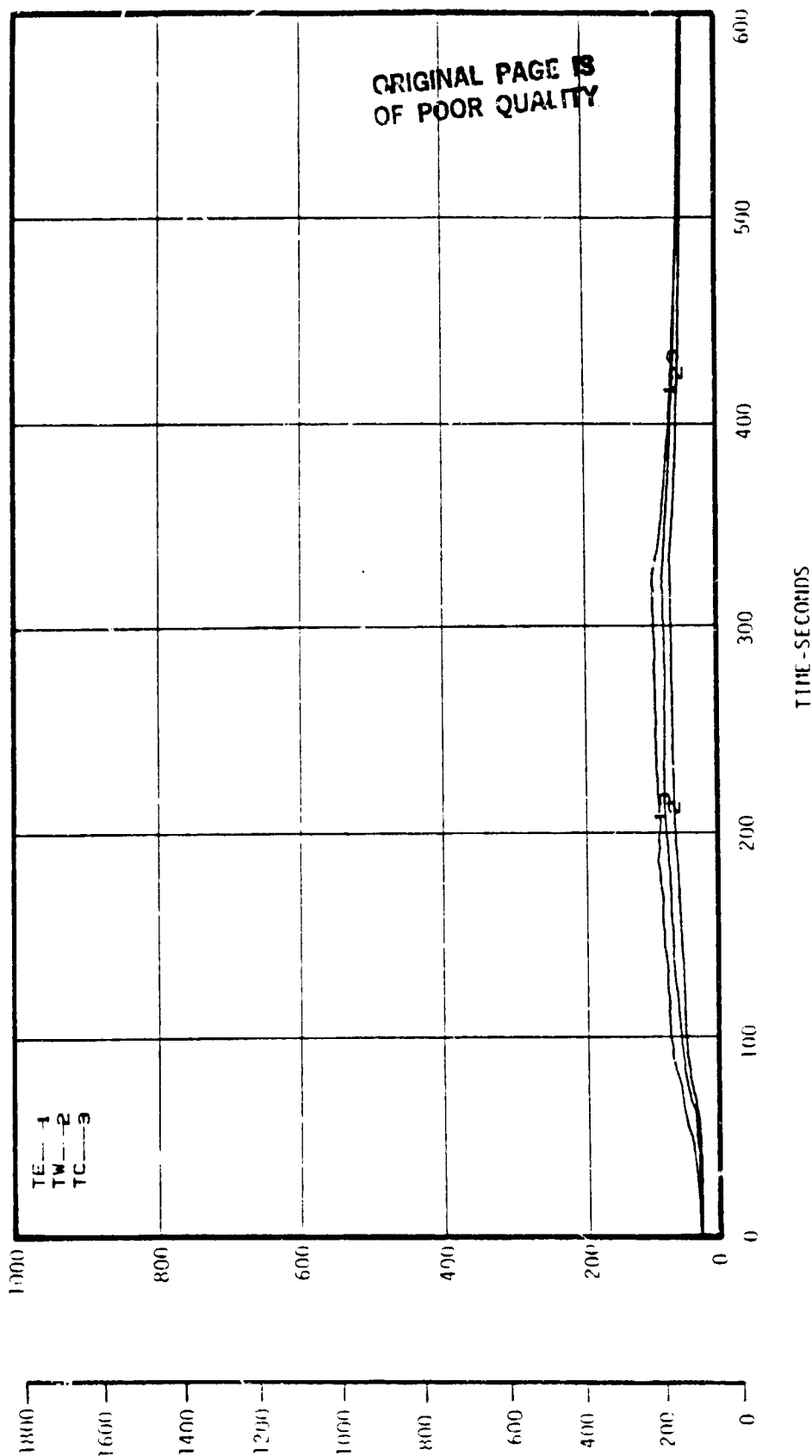
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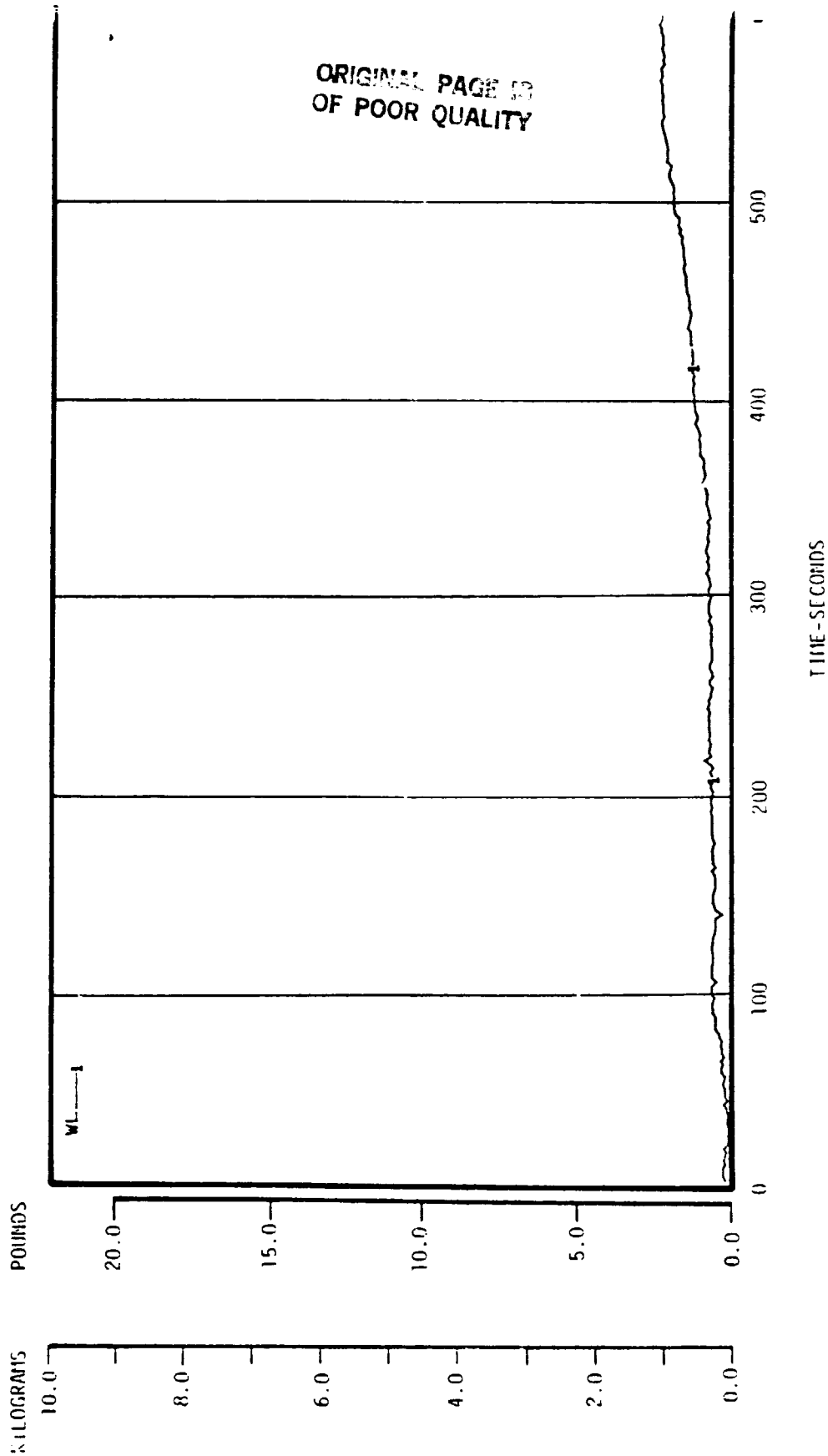


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CUSHION CONSTRUCTION NUMBER 10.0

WEIGHT LOSS



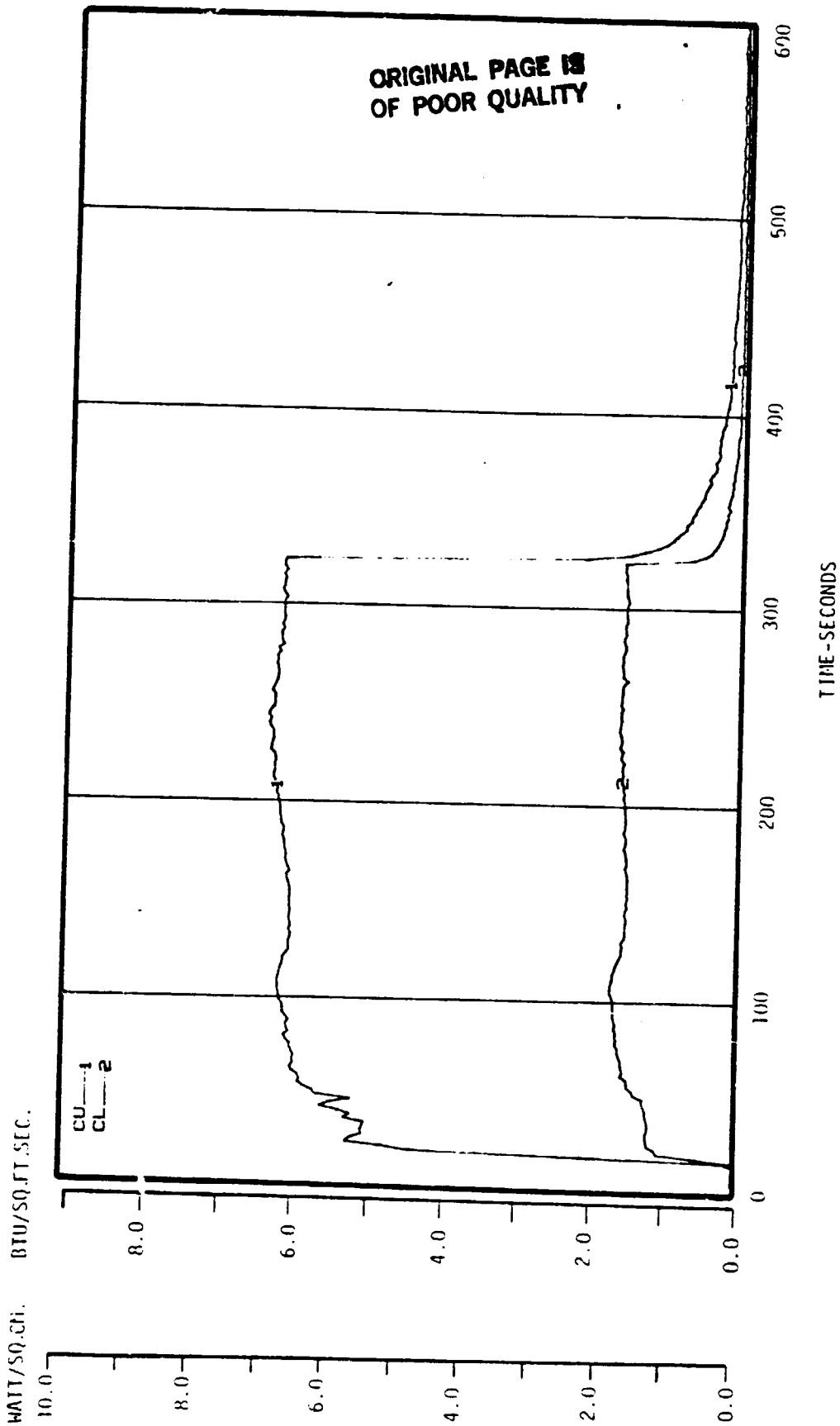
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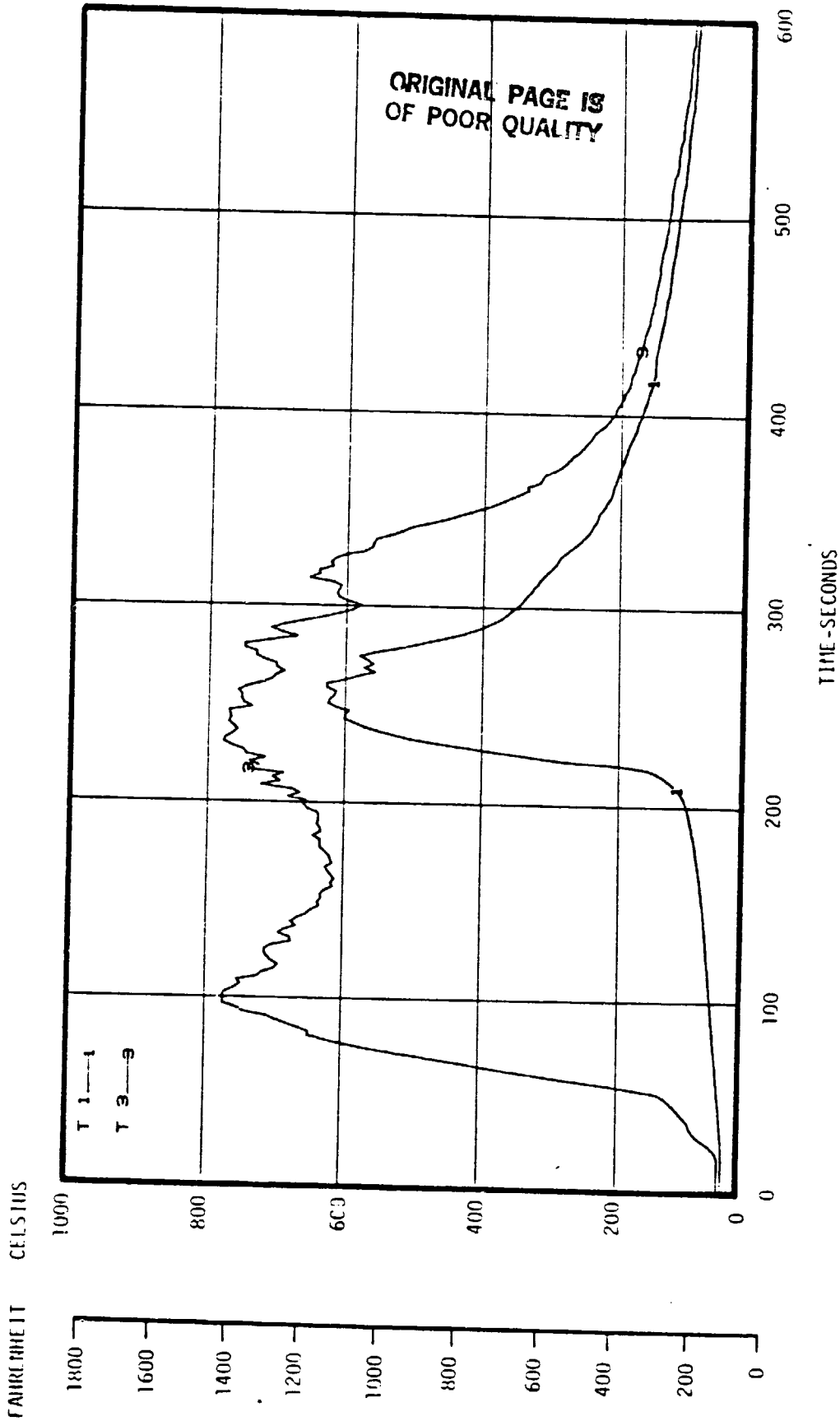
CUSHION CONSTRUCTION NUMBER 10.0

HEAT FLUX



DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/10/82 10.03
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 CUSHION CONSTRUCTION NUMBER 10.0

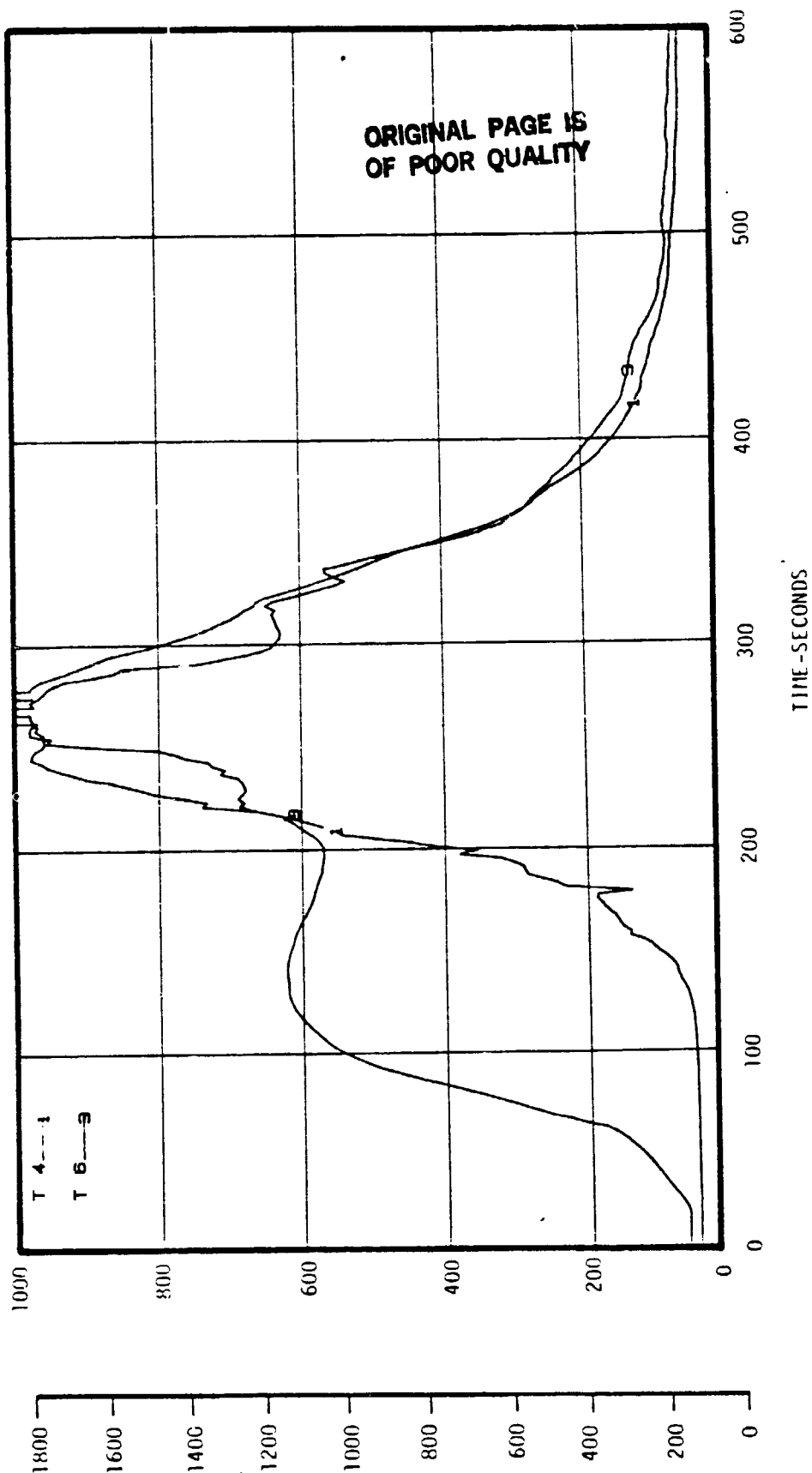
SEAT CUSHION TEMPERATURES



DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/10/82 10.03
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CUSHION CONSTRUCTION NUMBER 10.0

SEAT CUSHION TEMPERATURES

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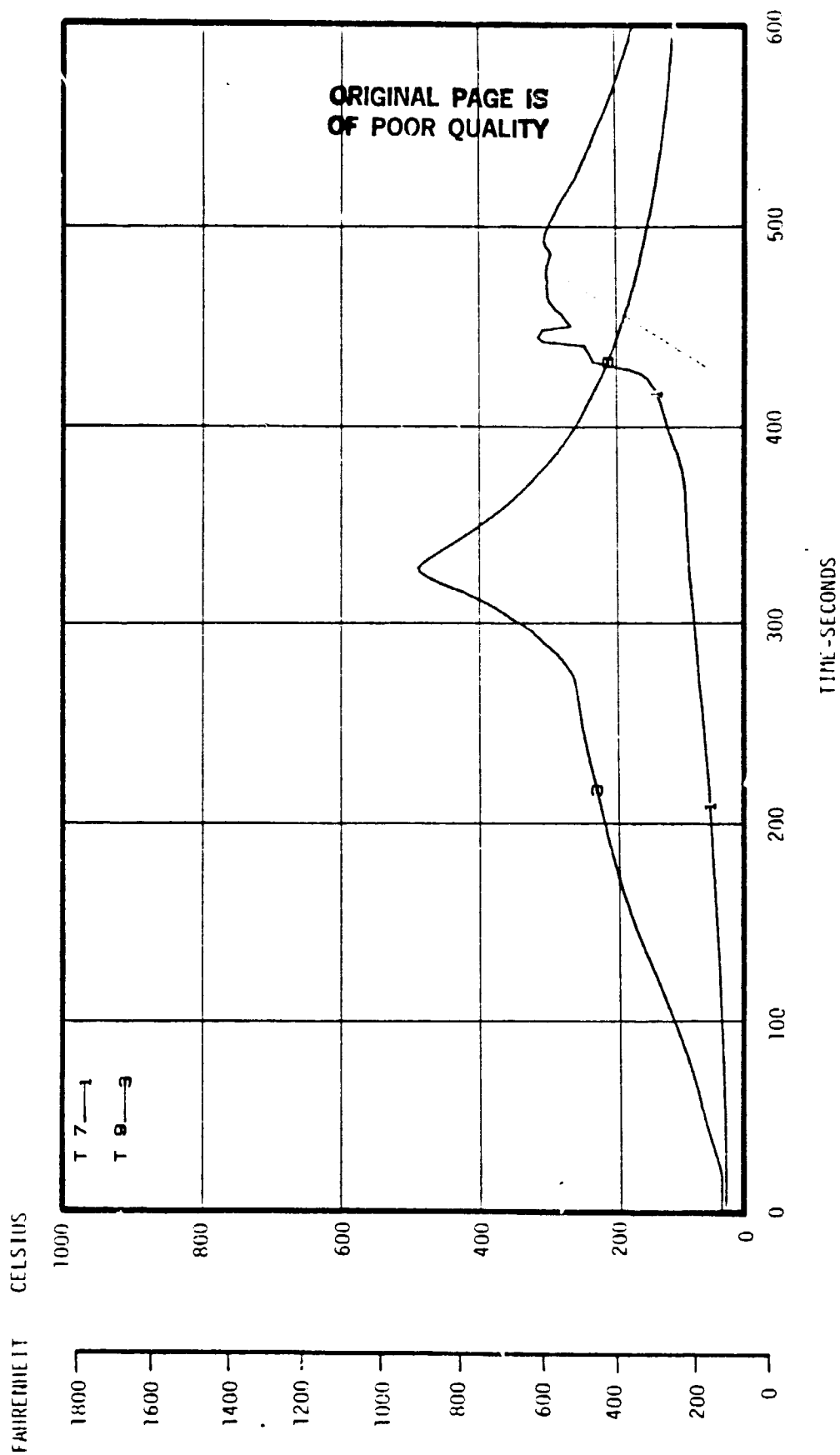


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CUSHION CONSTRUCTION NUMBER 10.0

SEAT CUSHION TEMPERATURES



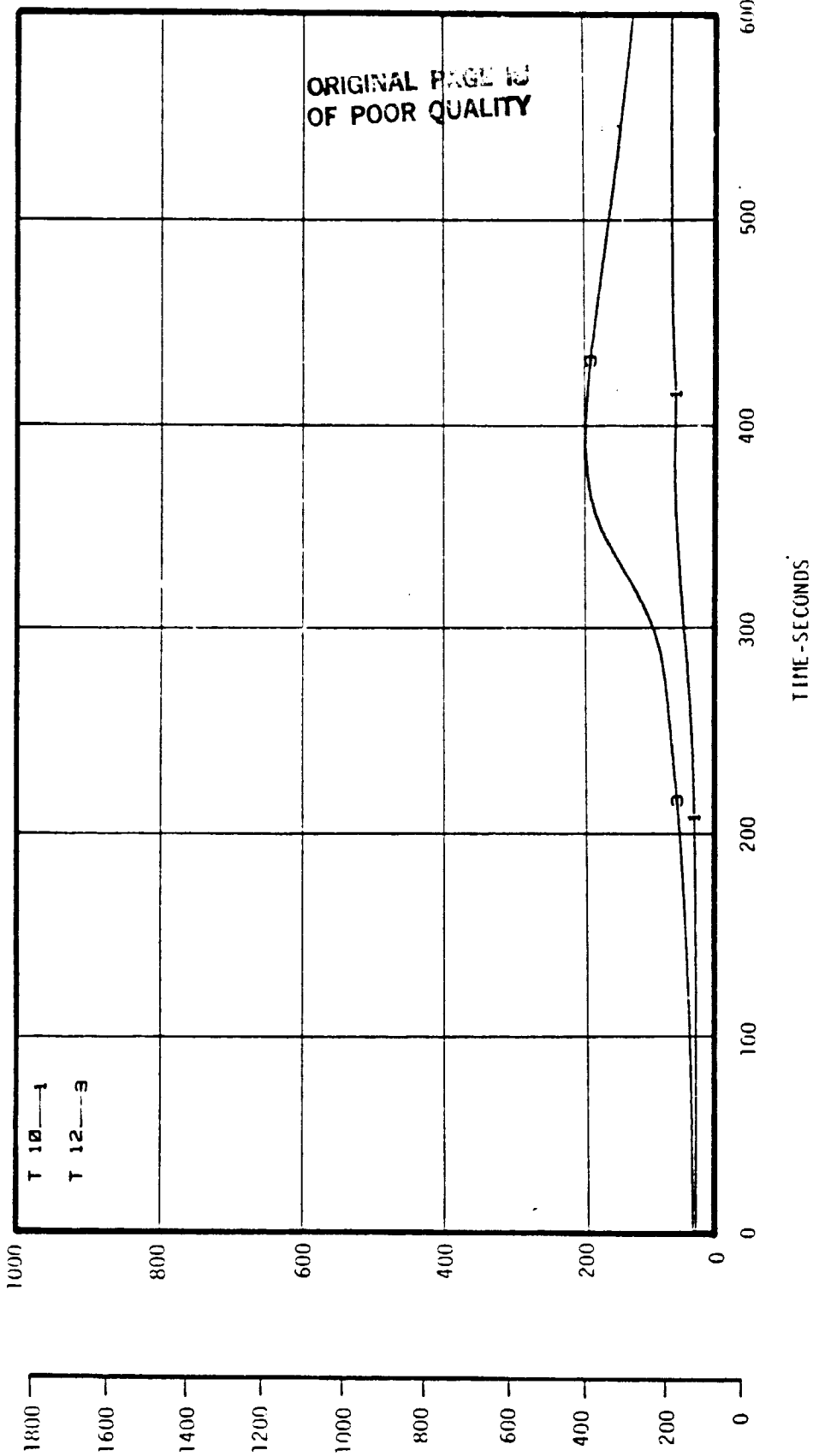
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NASA-M1ES FULL SCALE CUSHION BURN TEST NUMBER 8

CUSHION CONSTRUCTION NUMBER 10.0

SEAT CUSHION TEMPERATURES

FAHRENHEIT CELSIUS



DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/10/82 10.03

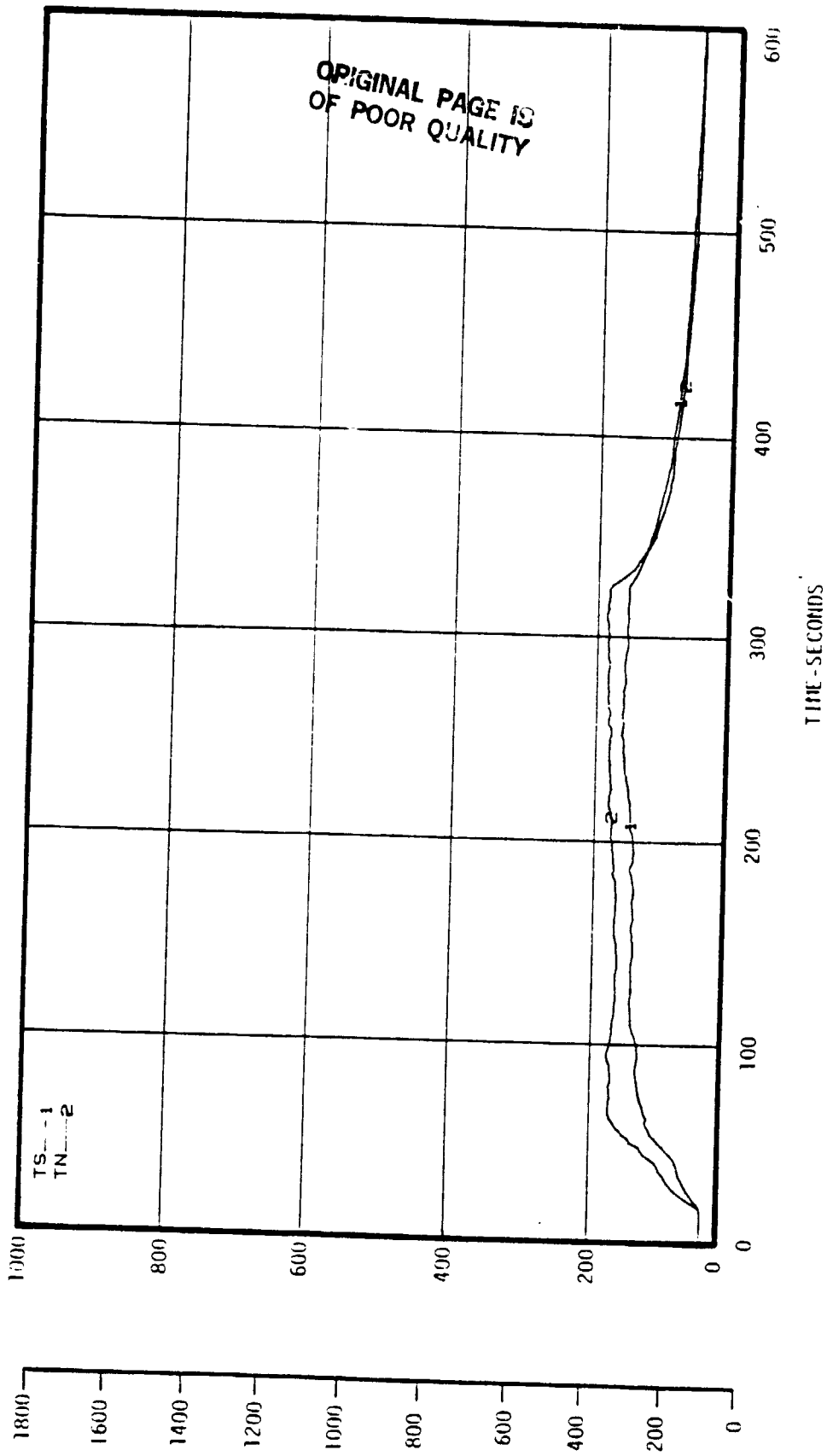
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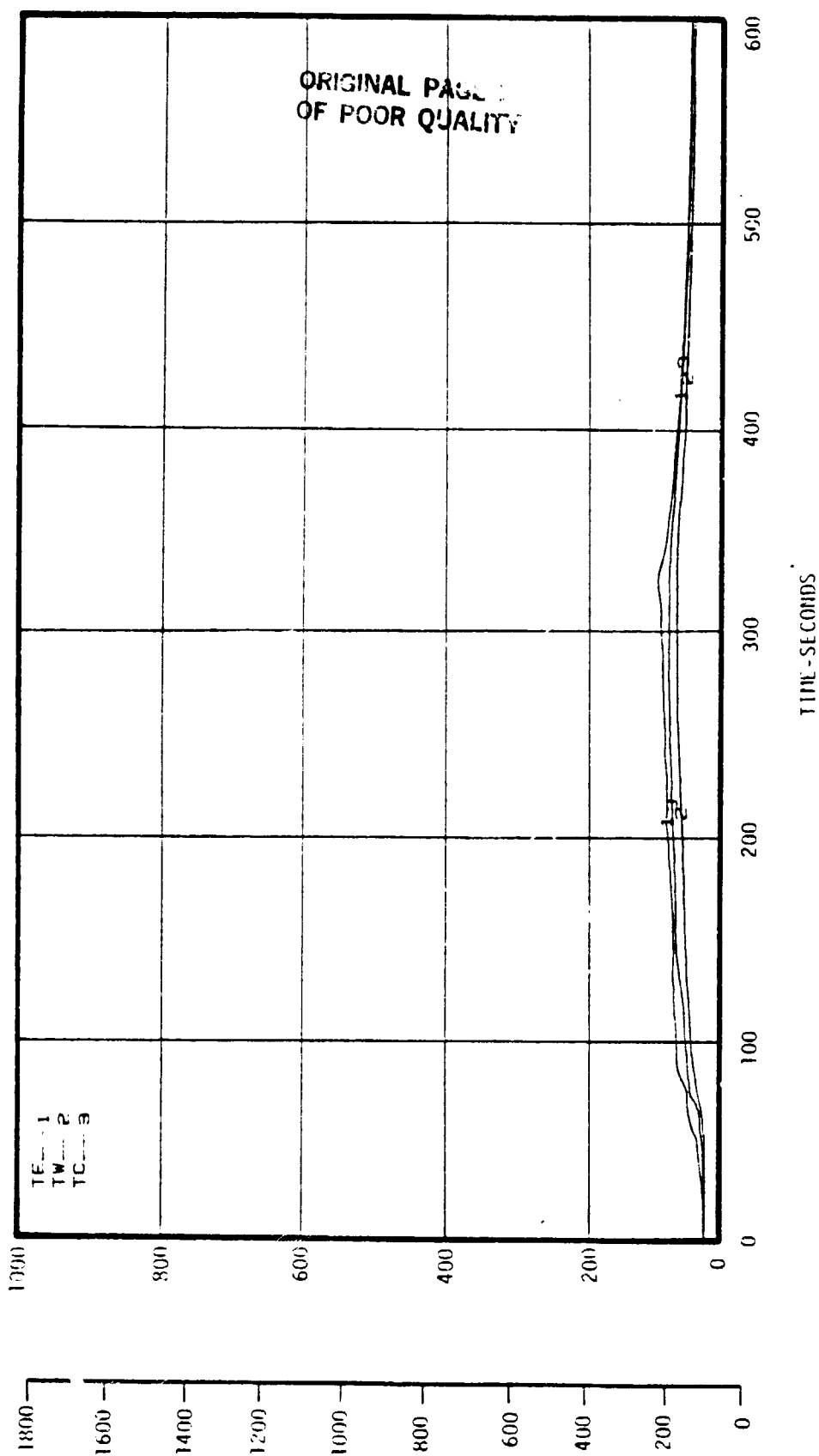
CELSIUS



DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/10/82 10.03
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CUSHION CONSTRUCTION NUMBER 10.0

CEILING TEMPERATURE

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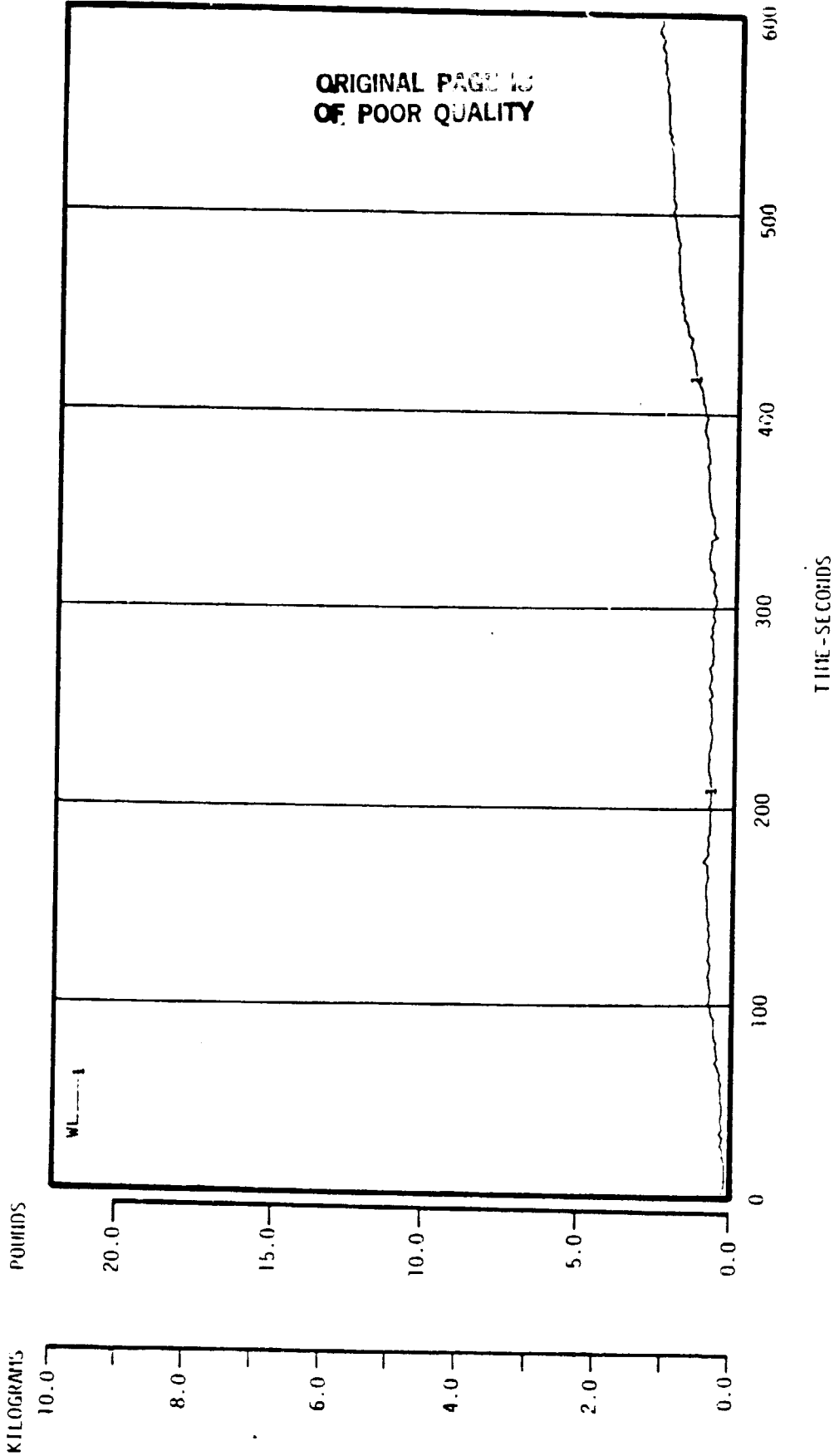


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NASA-MES FULL SCALE CUSHION BURR TEST NUMBER 6

CUSHION CONSTRUCTION NUMBER 10.0

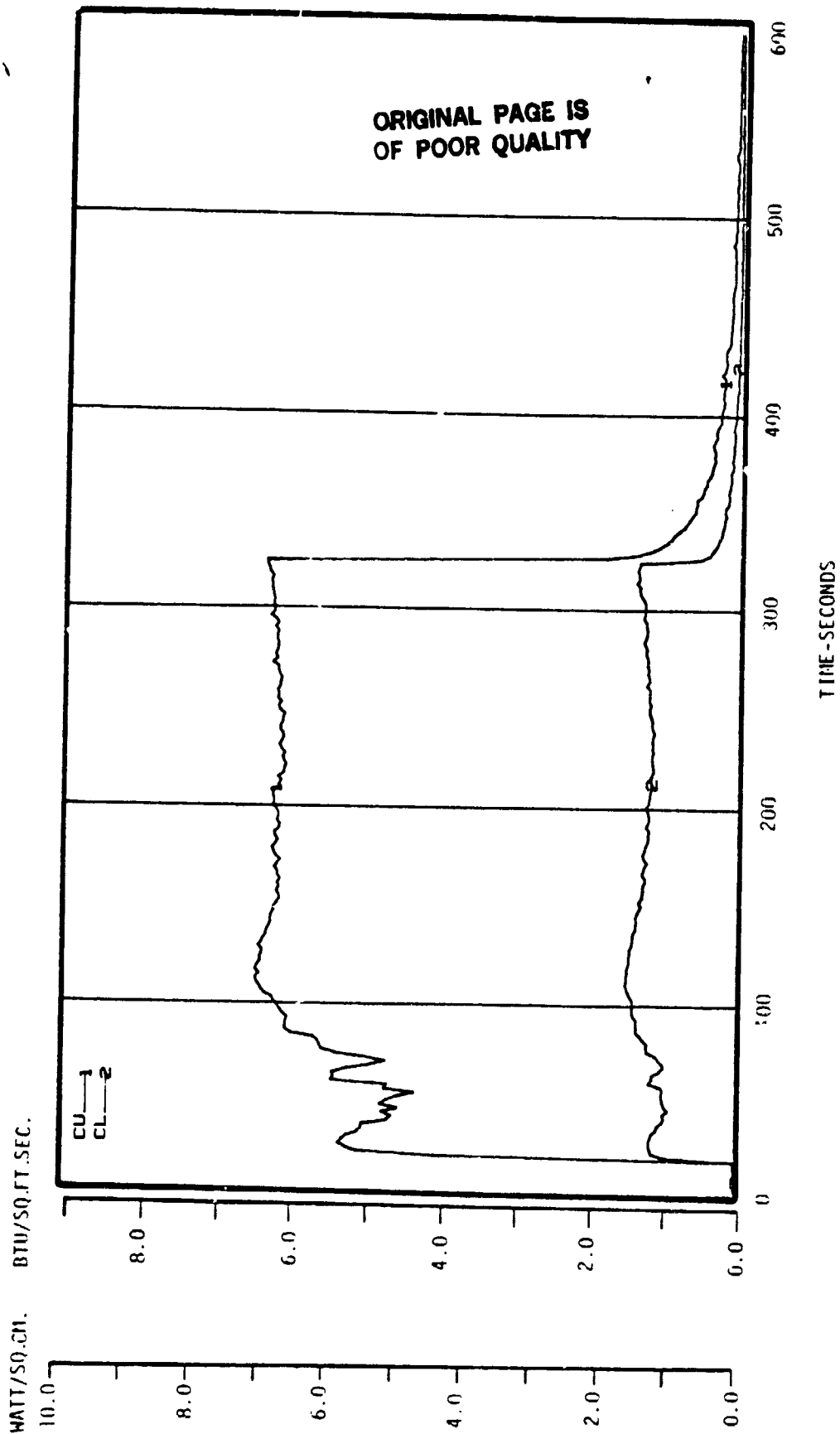
WEIGHT LOSS



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DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/10/82 10.03
NASA-AMES FULL SCALE CUSHION BURN TEST NUMBER 6
CUSHION CONSTRUCTION NUMBER 10.0

HEAT FLUX

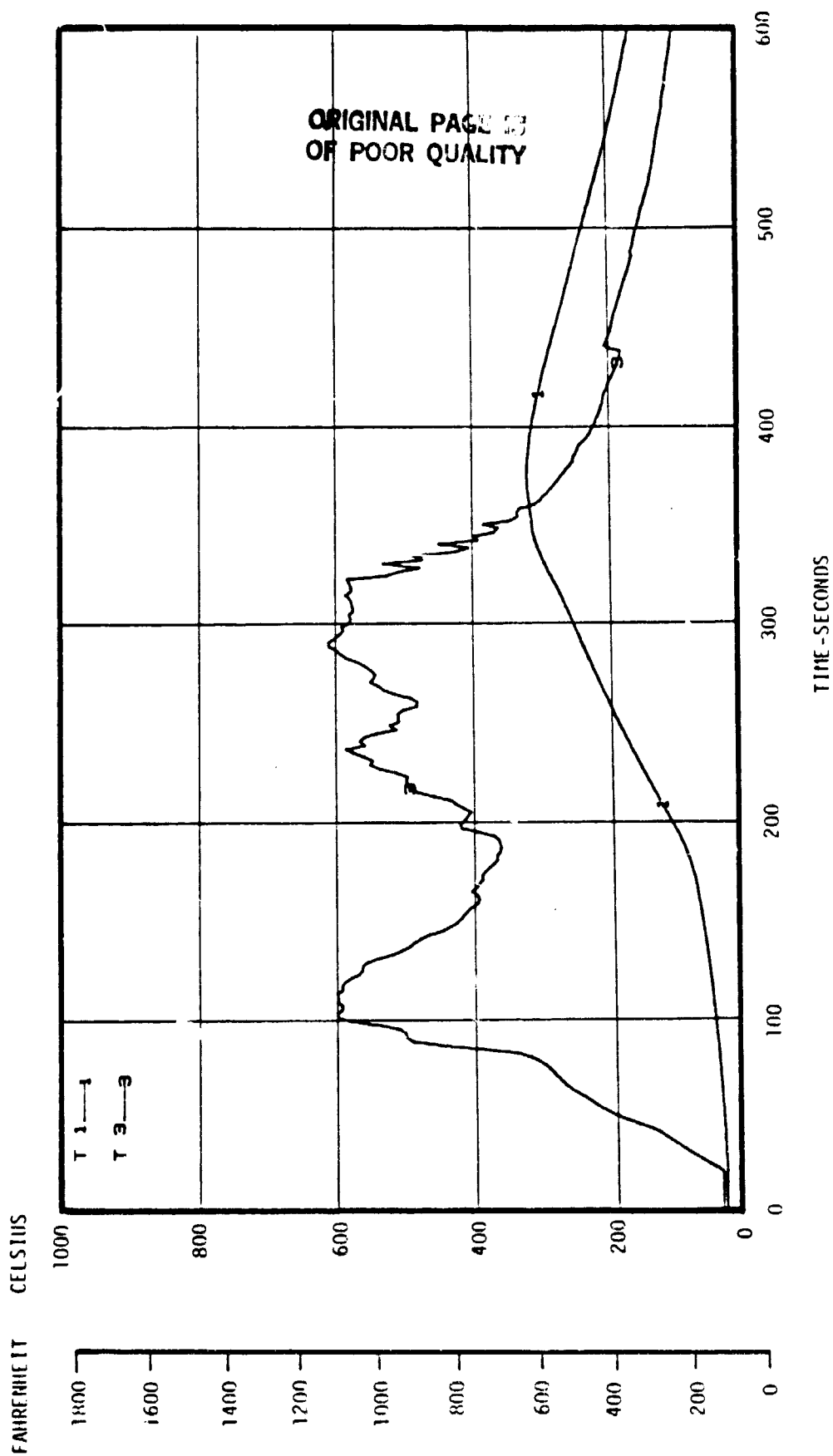


DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/18/82 09.04

NASA-AMES FULL SCALE CUSHION BURN TEST NUMBER 20

CUSHION CONSTRUCTION NUMBER 11.0

SEAT CUSHION TEMPERATURES

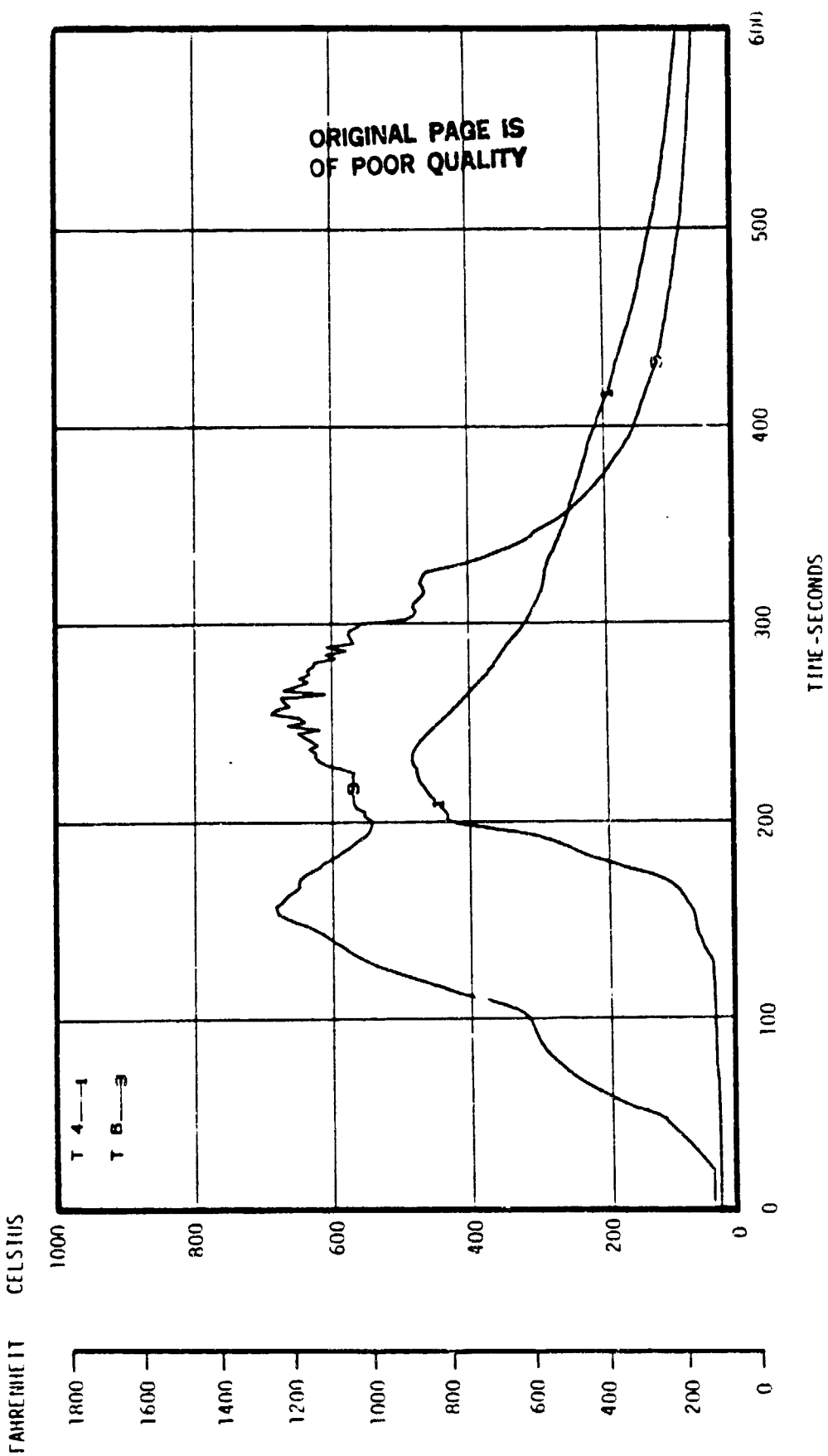


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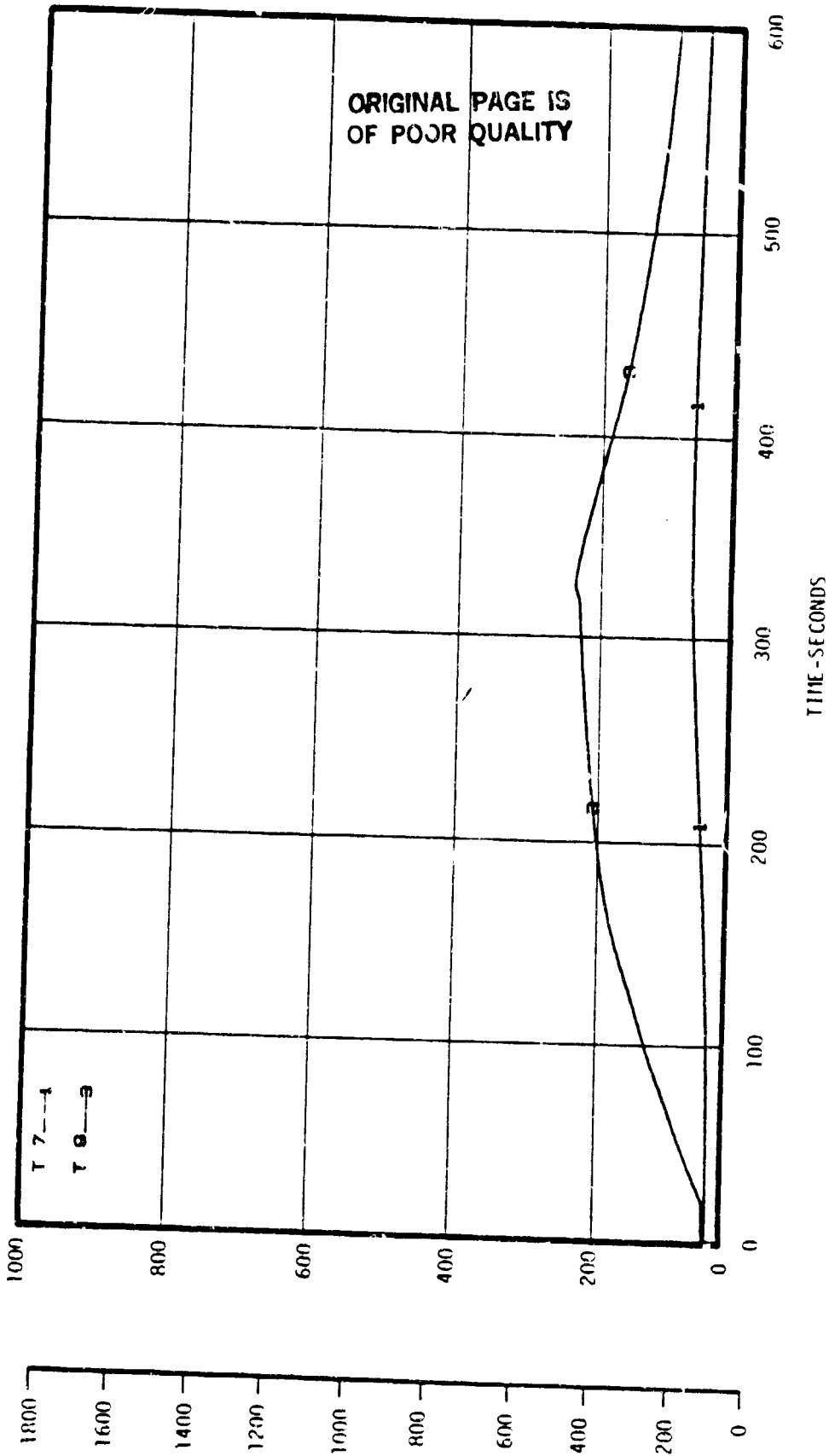
SEAT CUSHION TEMPERATURES



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 CUSHION CONSTRUCTION NUMBER 11.0

SEAT CUSHION TEMPERATURES

FAHRENHEIT CELSIUS

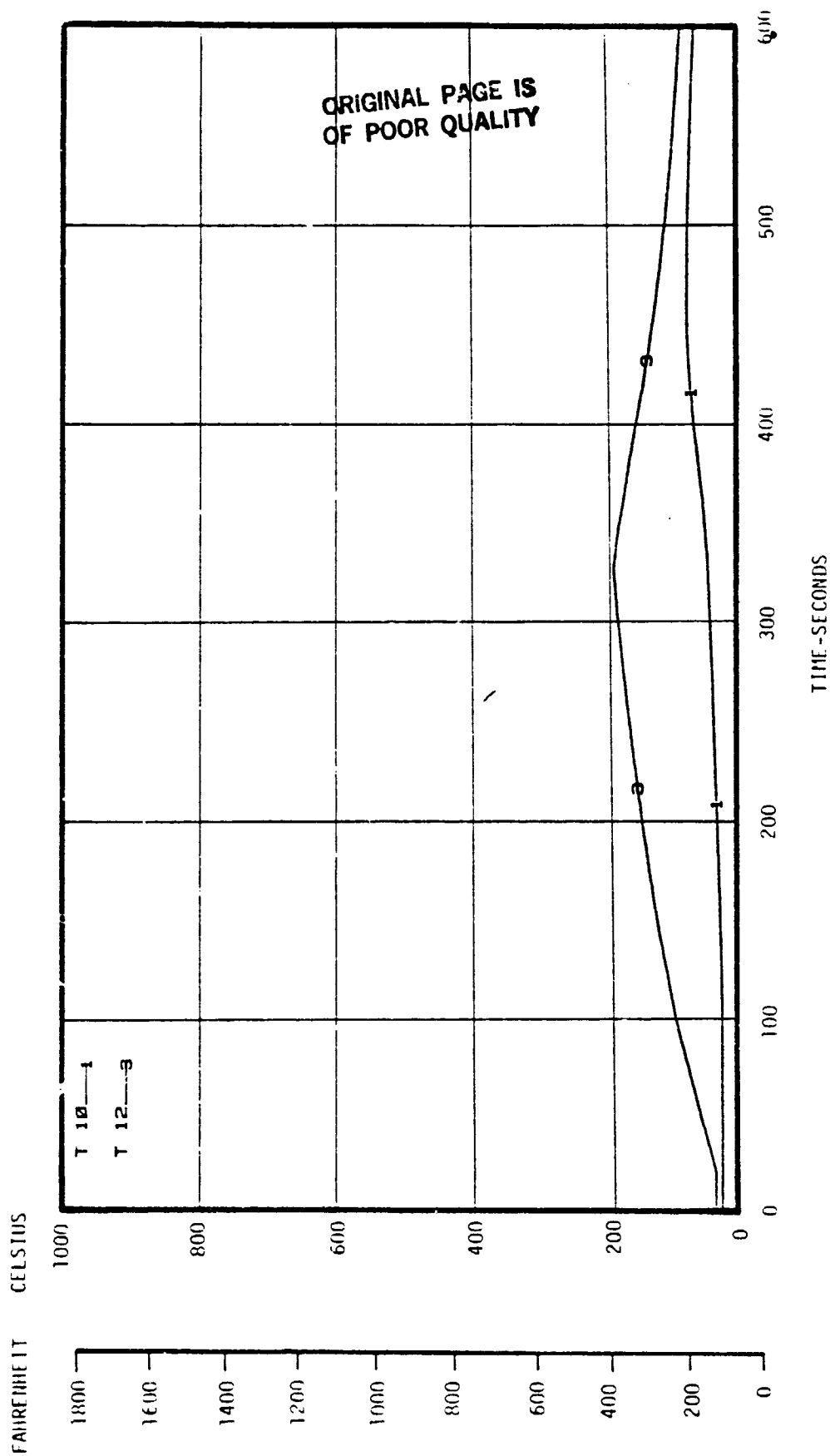


DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/18/82 PG. 04

NASA-AMES FULL SCALE CUSHION BURN TEST NUMBER 20

CUSHION CONSTRUCTION NUMBER 11.0

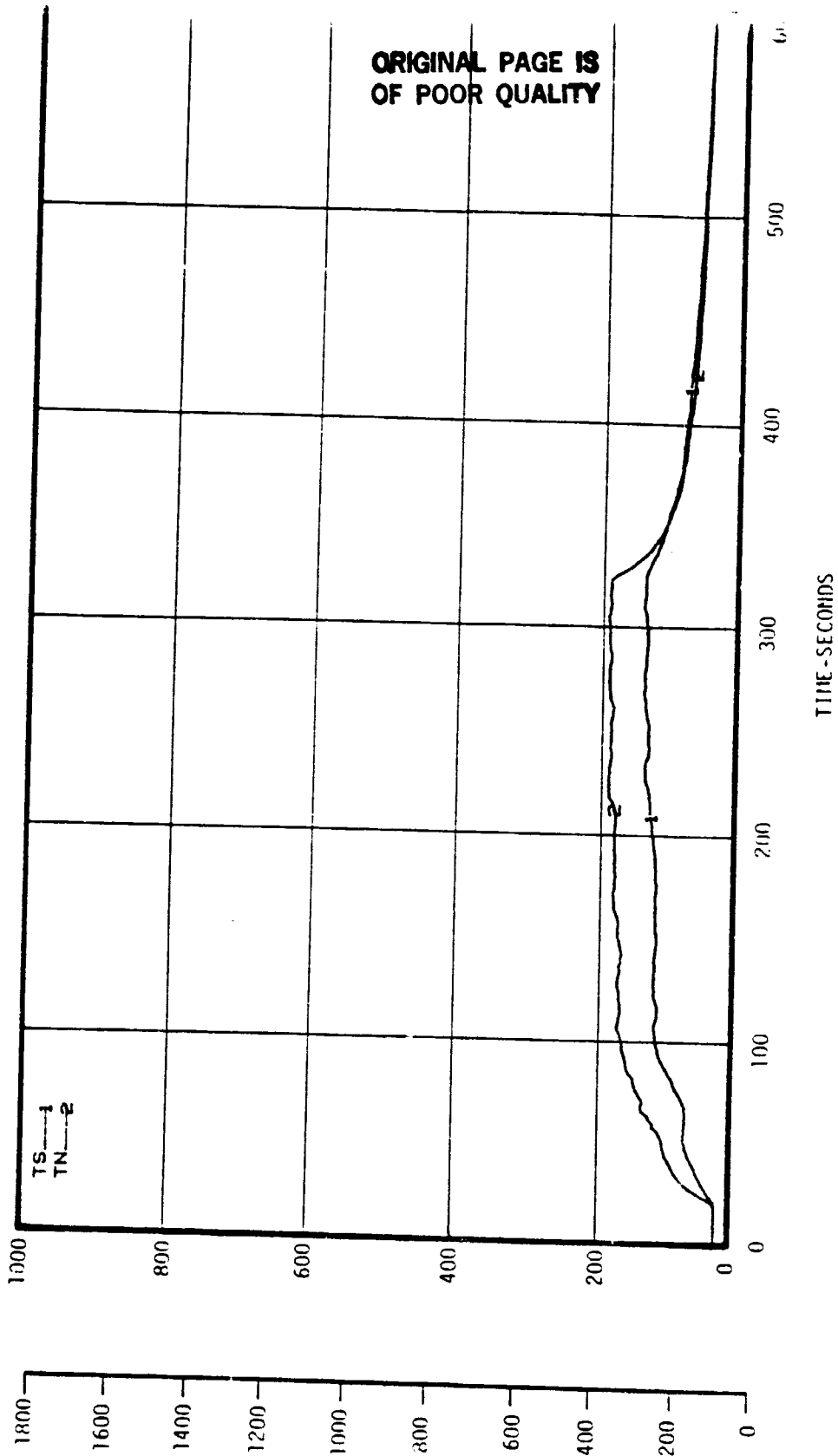
SEAT CUSHION TEMPERATURES



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CEILING TEMPERATURE

FAHRENHEIT CELSIUS

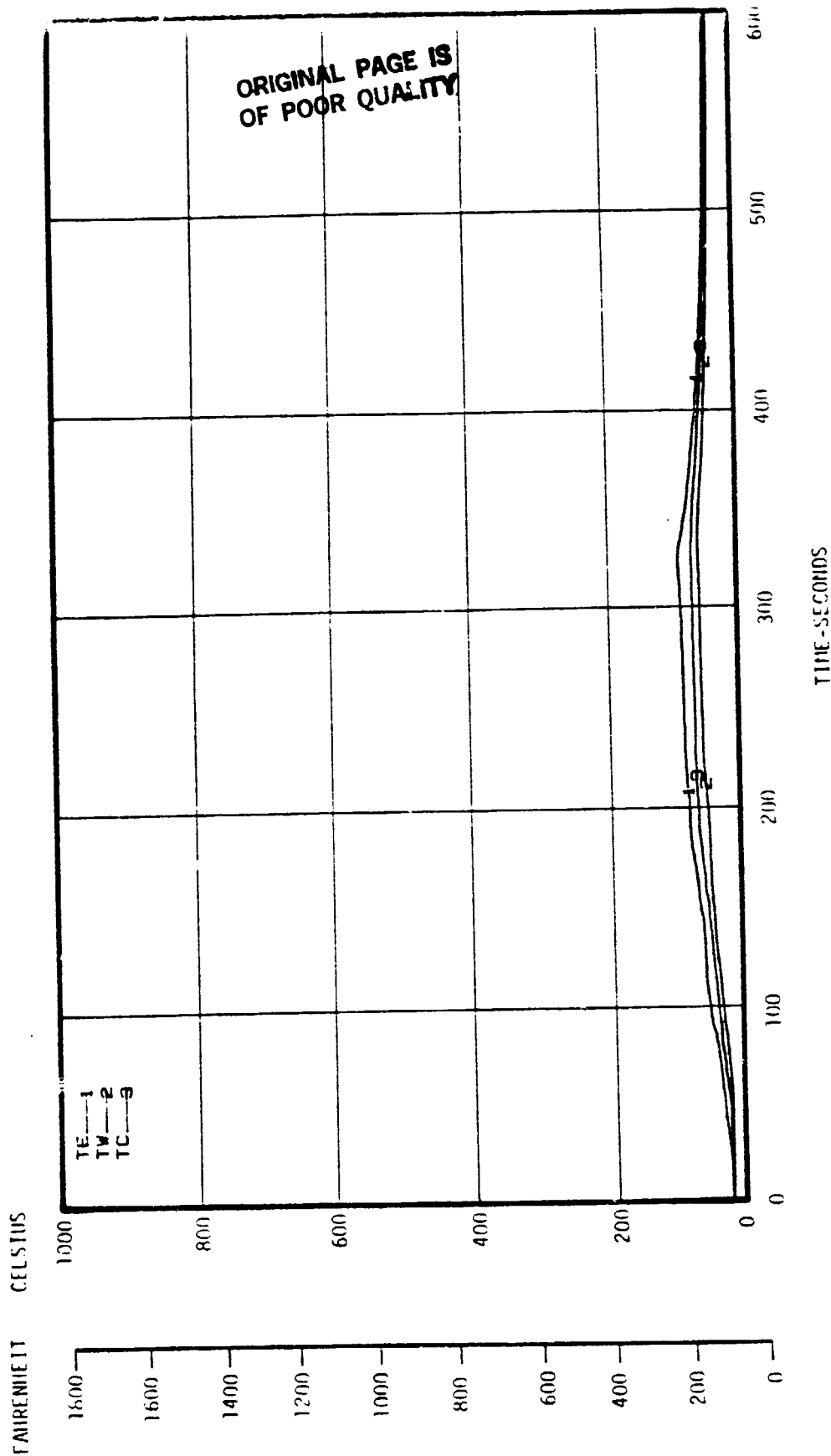


DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/10/82 00.04

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CUSHION CONSTRUCTION NUMBER 11.0

CEILING TEMPERATURE

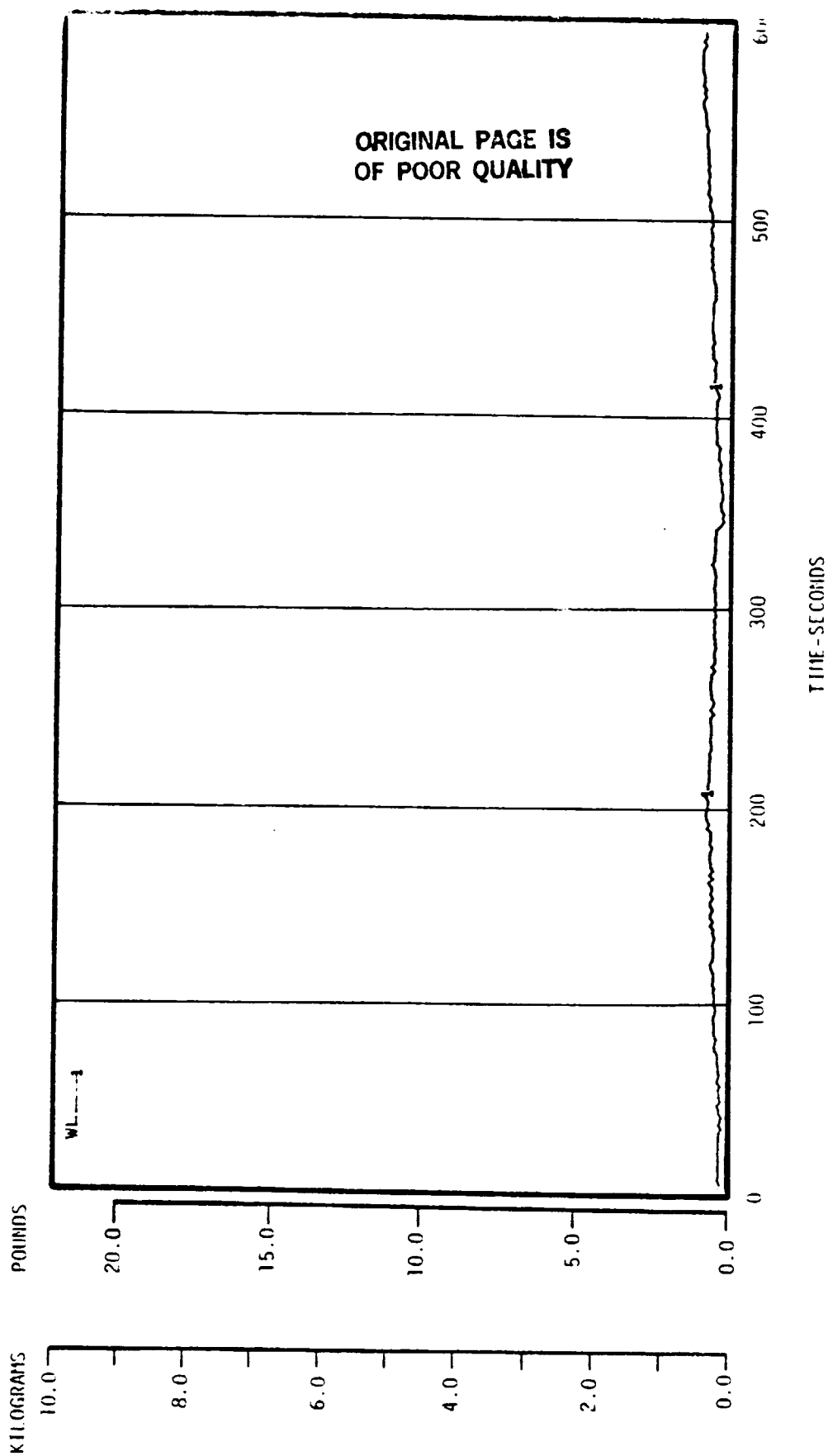


DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/18/82 00.04

NASA-MIS FULL SCALE CUSHION BURN TEST NUMBER 20

CUSHION CONSTRUCTION NUMBER 11.0

WEIGHT LOSS



JOHNSAS AIRCRAFT CABIN FIRE SIMULATOR 03/18/82 00.04

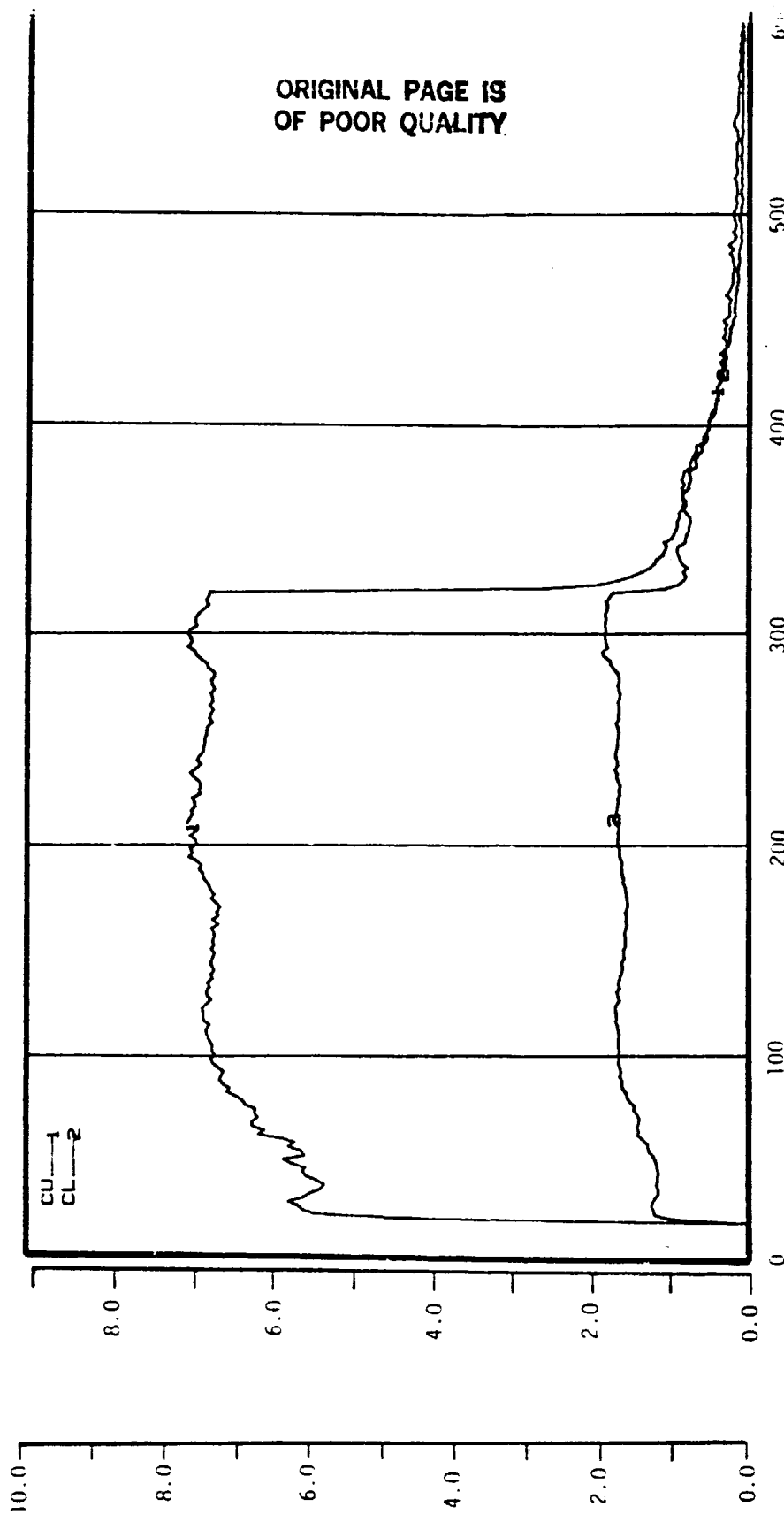
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CUSHION CONSTRUCTION NUMBER 11.0

HEAT FLUX

WATT/SQ.CM.

BTU/SQ.FT.SEC.



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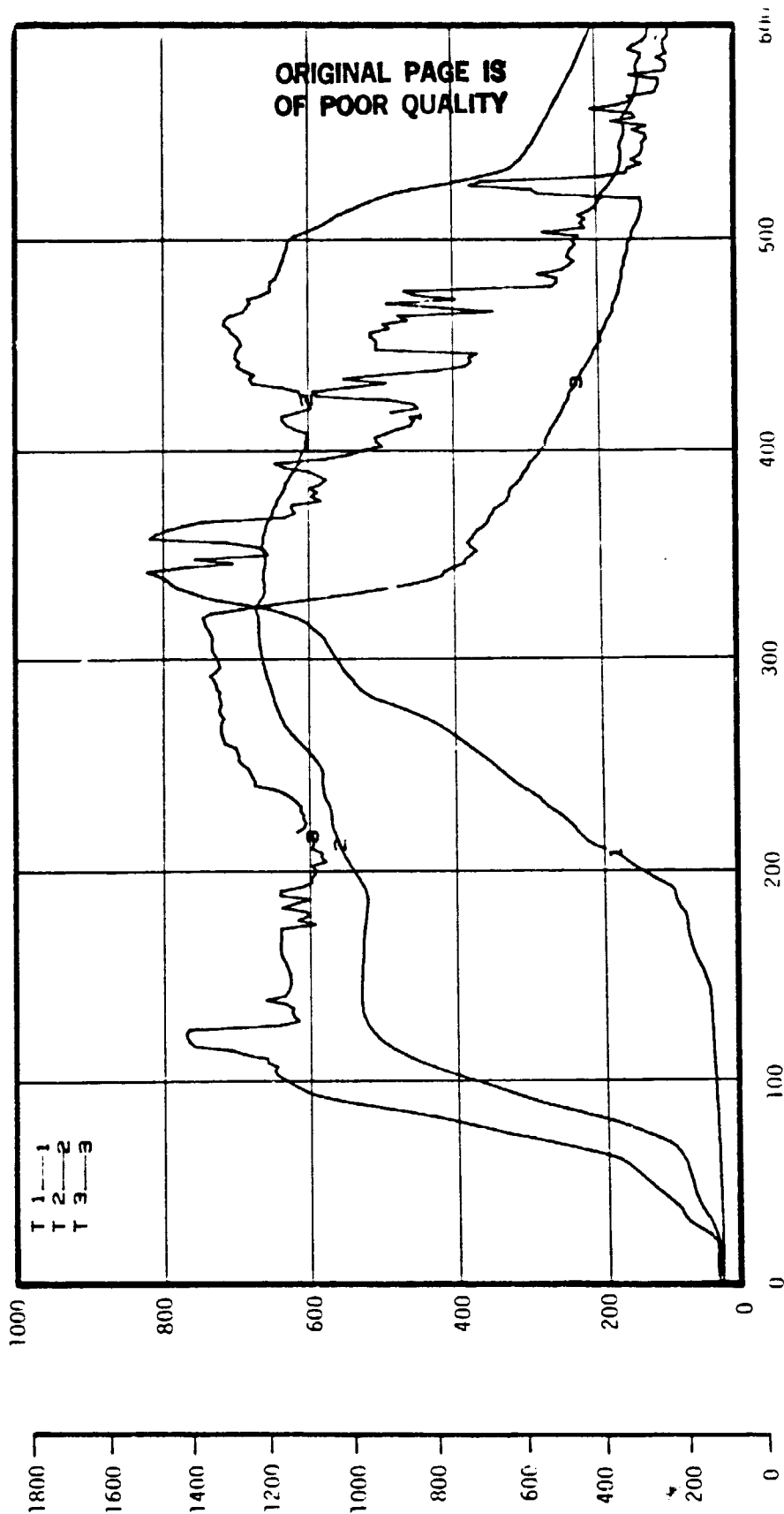
DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/10/82 11.00

NASA-WES FULL SCALE CUSHION BURN TEST NUMBER 21

CUSHION CONSTRUCTION NUMBER 12.0

SEAT CUSHION TEMPERATURES

FAHRENHEIT CELSIUS



DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/10/02 11.00

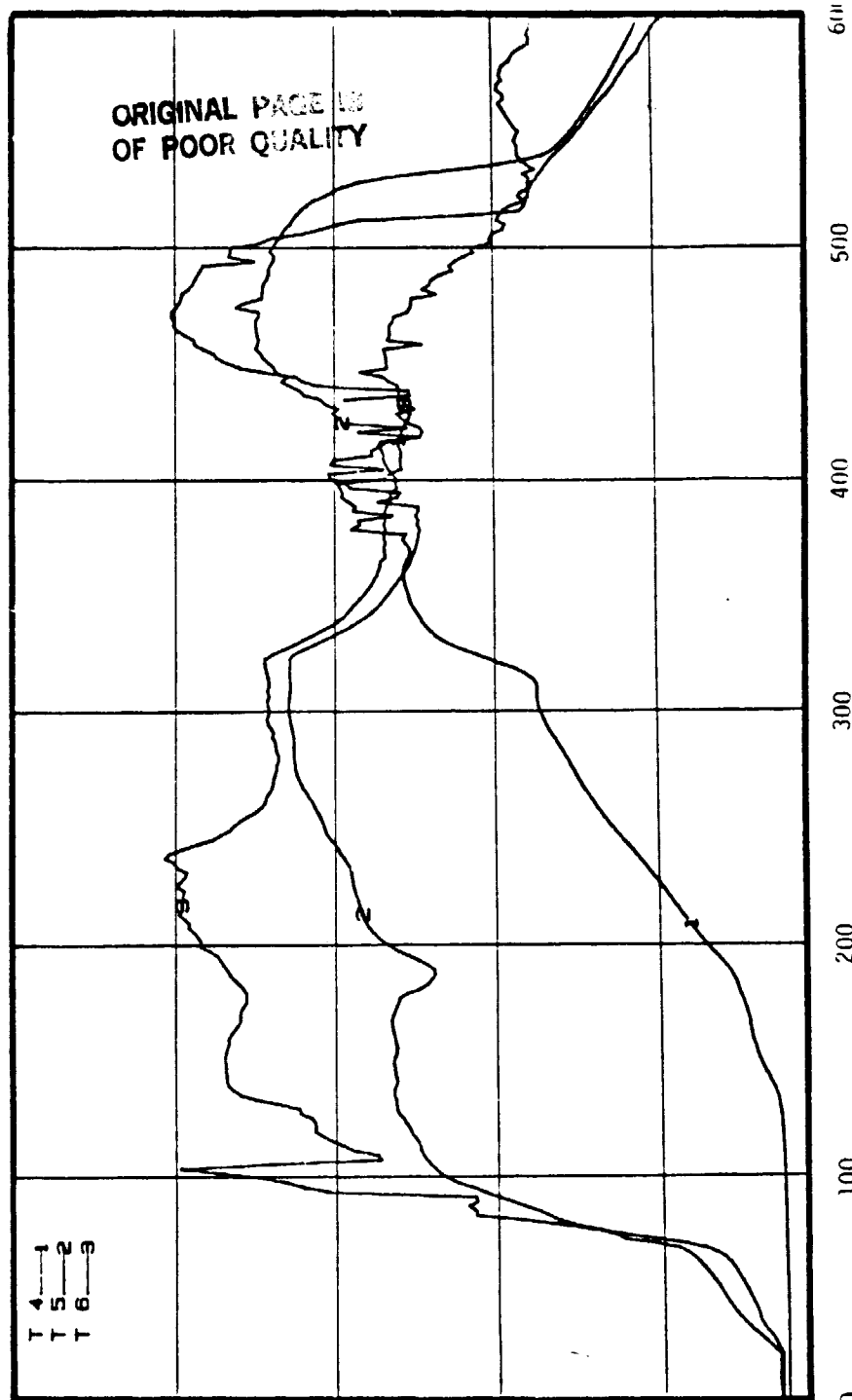
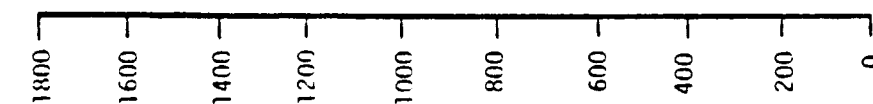
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CUSHION CONSTRUCTION NUMBER 12.0

SEAT CUSHION TEMPERATURES

FAHRENHEIT

CELSIUS



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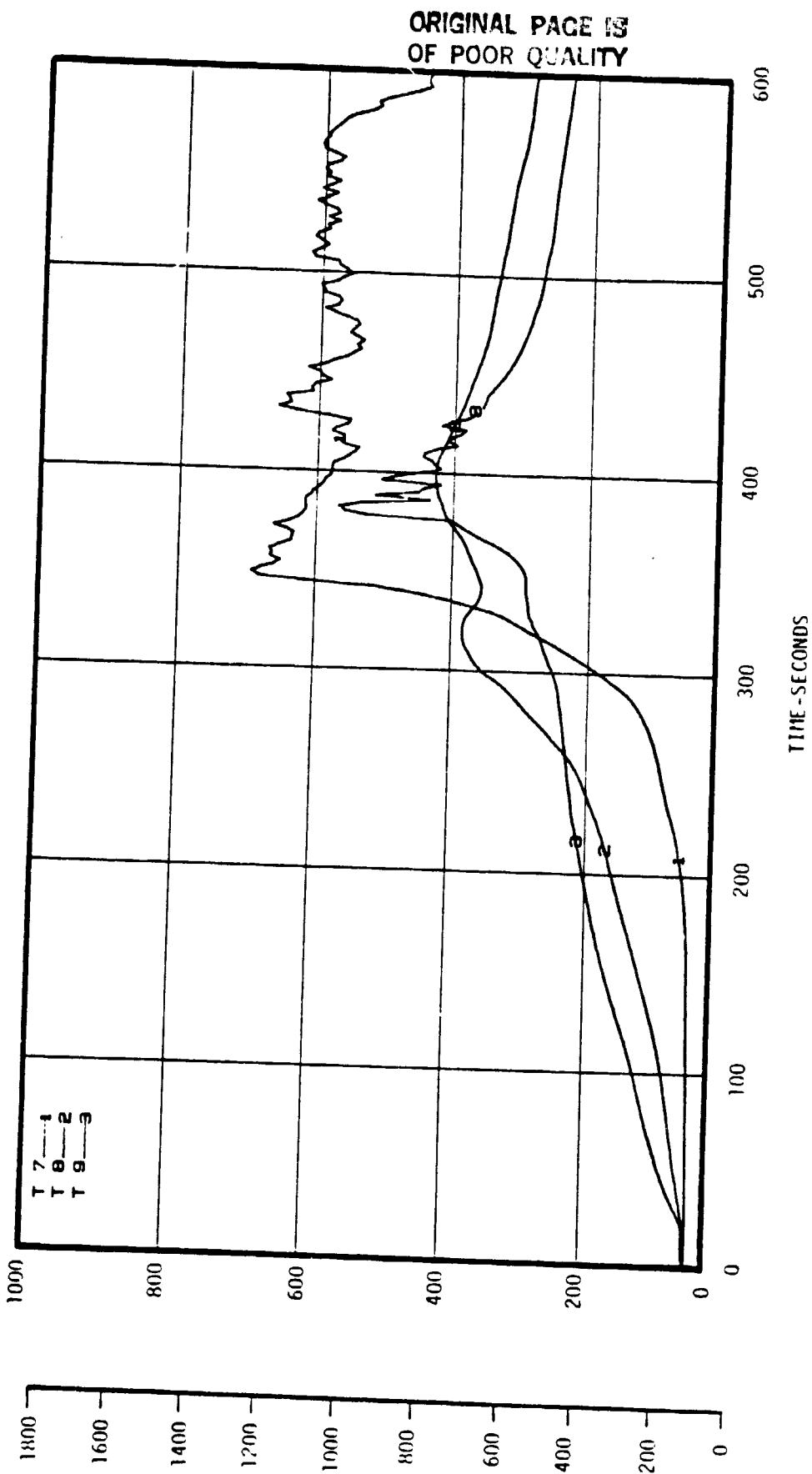
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CUSHION CONSTRUCTION NUMBER 12.0

SEAT CUSHION TEMPERATURES

FAHRENHEIT CELSIUS

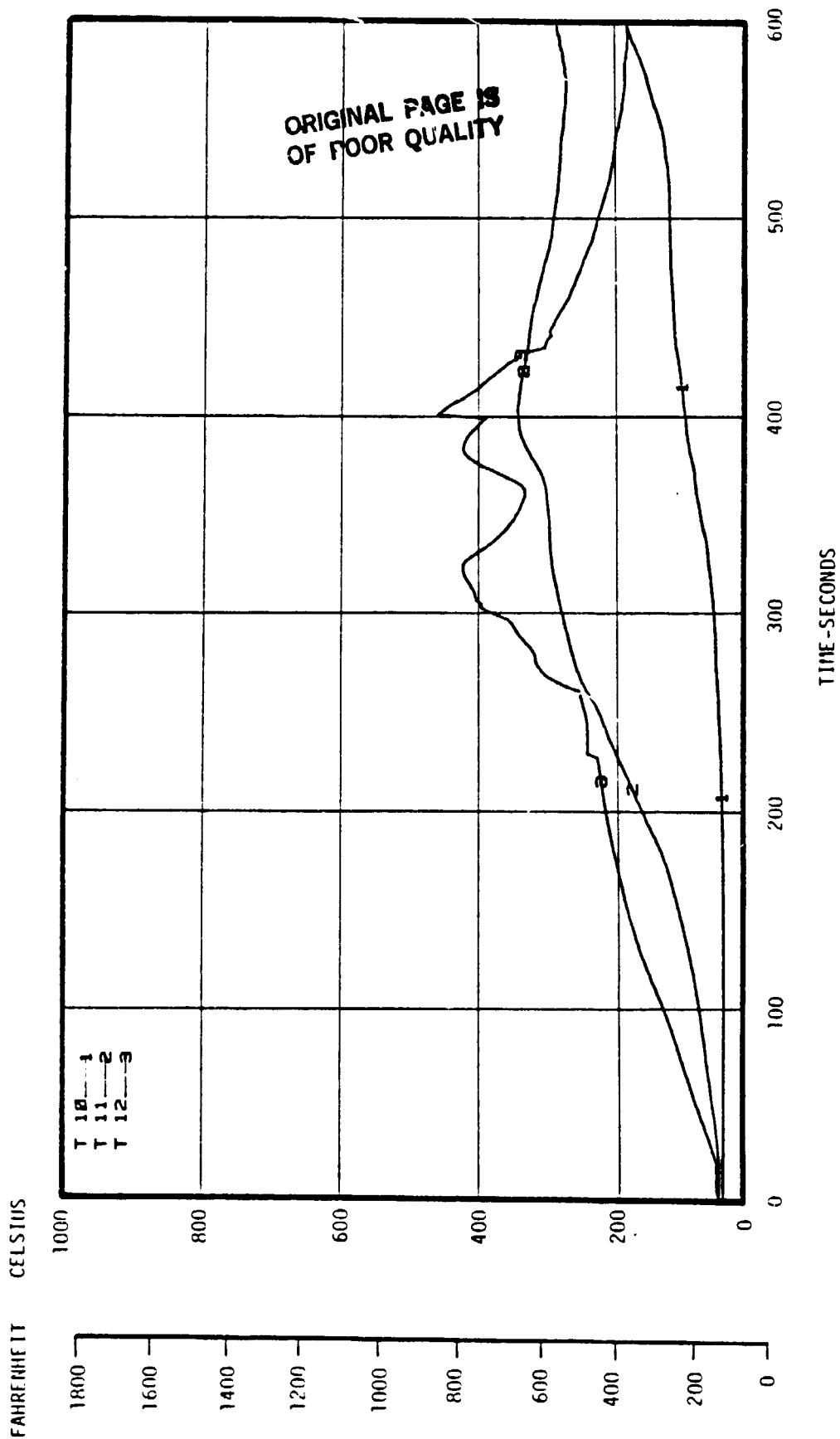


DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/10/02 11.00

NASA-AMES FULL SCALE CUSHION BURN TEST NUMBER 21

CUSHION CONSTRUCTION NUMBER 12.0

SEAT CUSHION TEMPERATURES

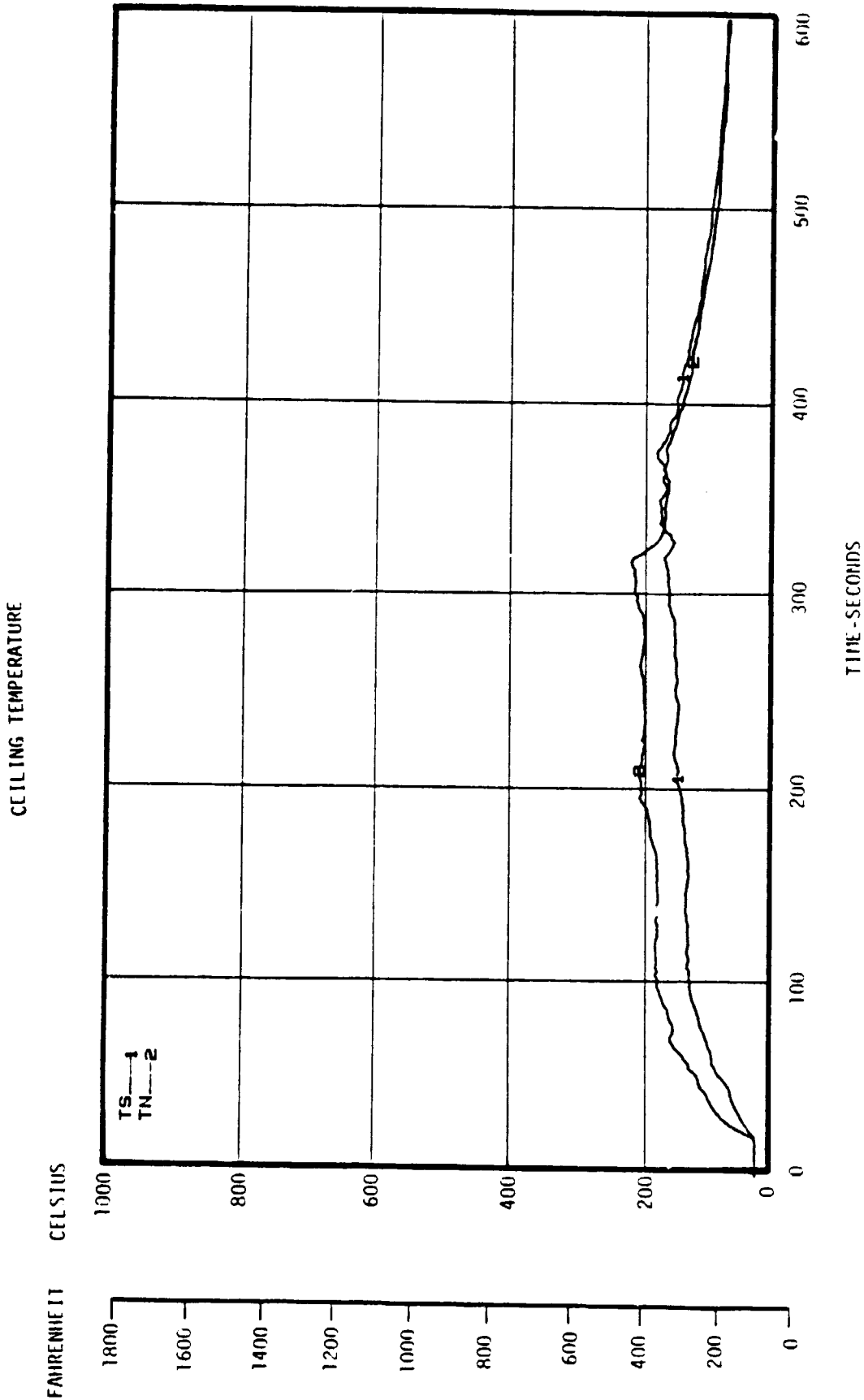


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CUSHION CONSTRUCTION NUMBER 12.0

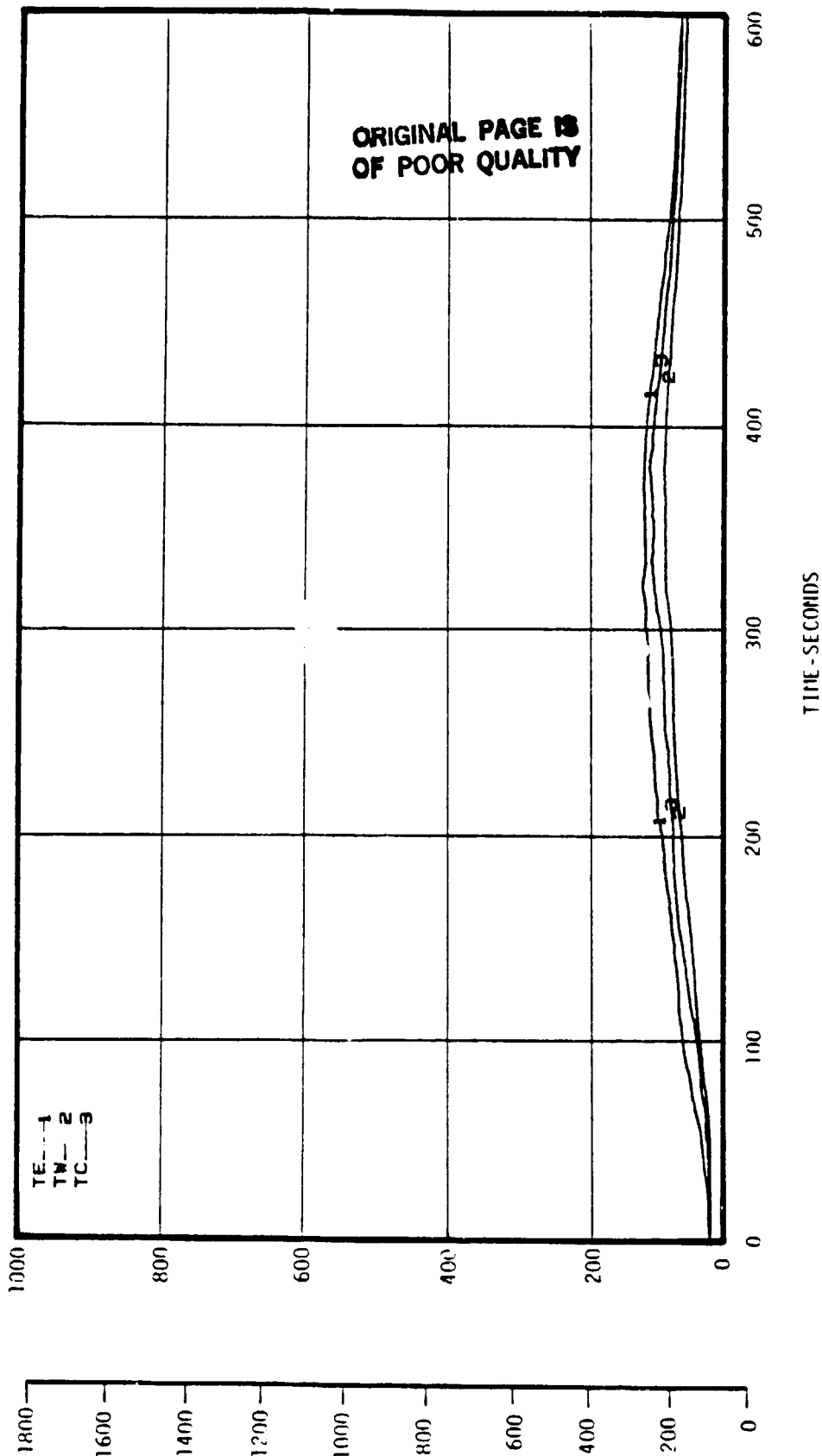
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DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/18/82 11.00
NASA-MFES FULL SCALE CUSHION BURN TEST NUMBER 21
CUSHION CONSTRUCTION NUMBER 12.0

CEILING TEMPERATURE

FAHRENHEIT CELSIUS

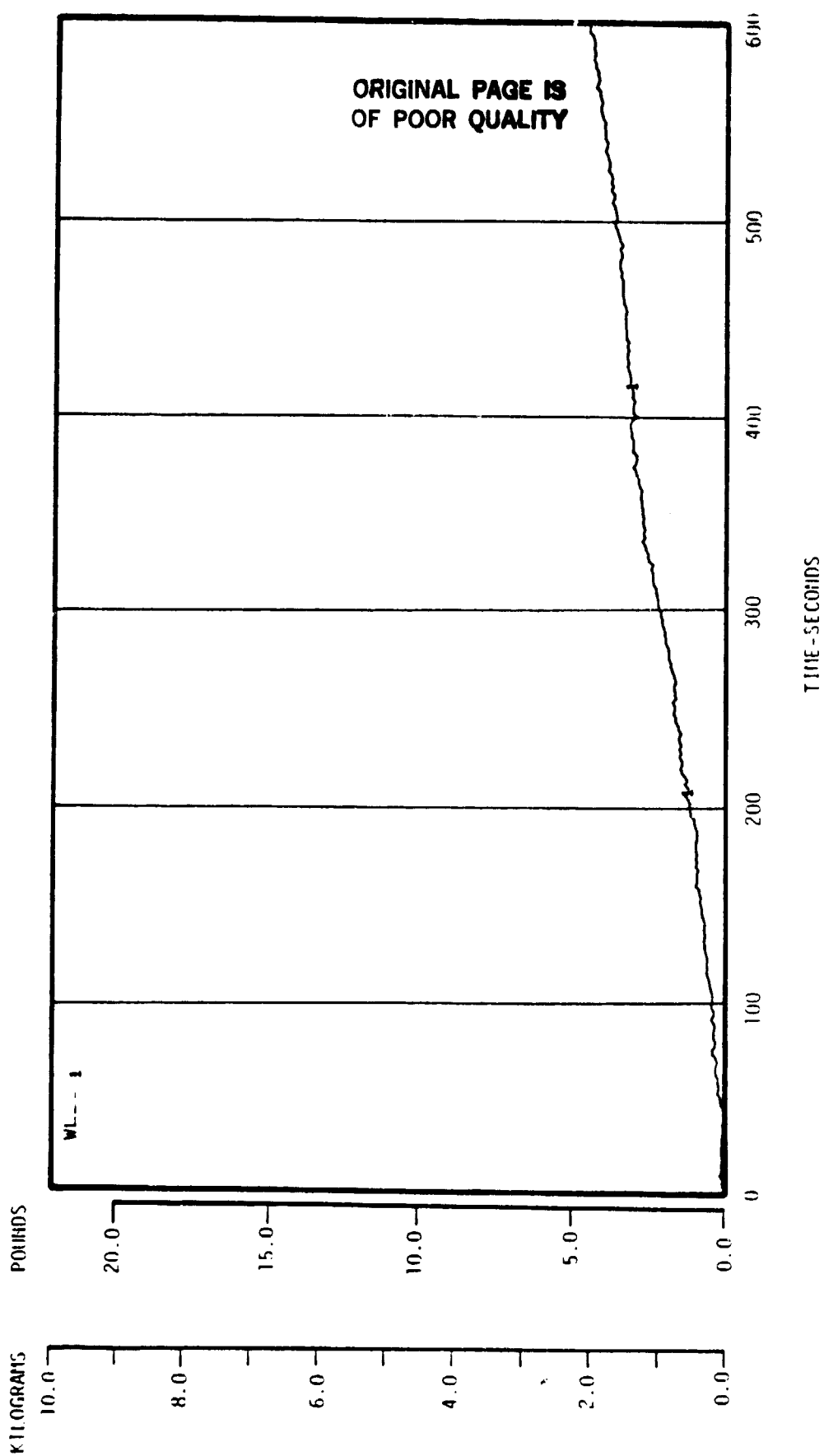


DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/10/82 11.00

NASA-AMES FULL SCALE CUSHION BURN TEST NUMBER 21

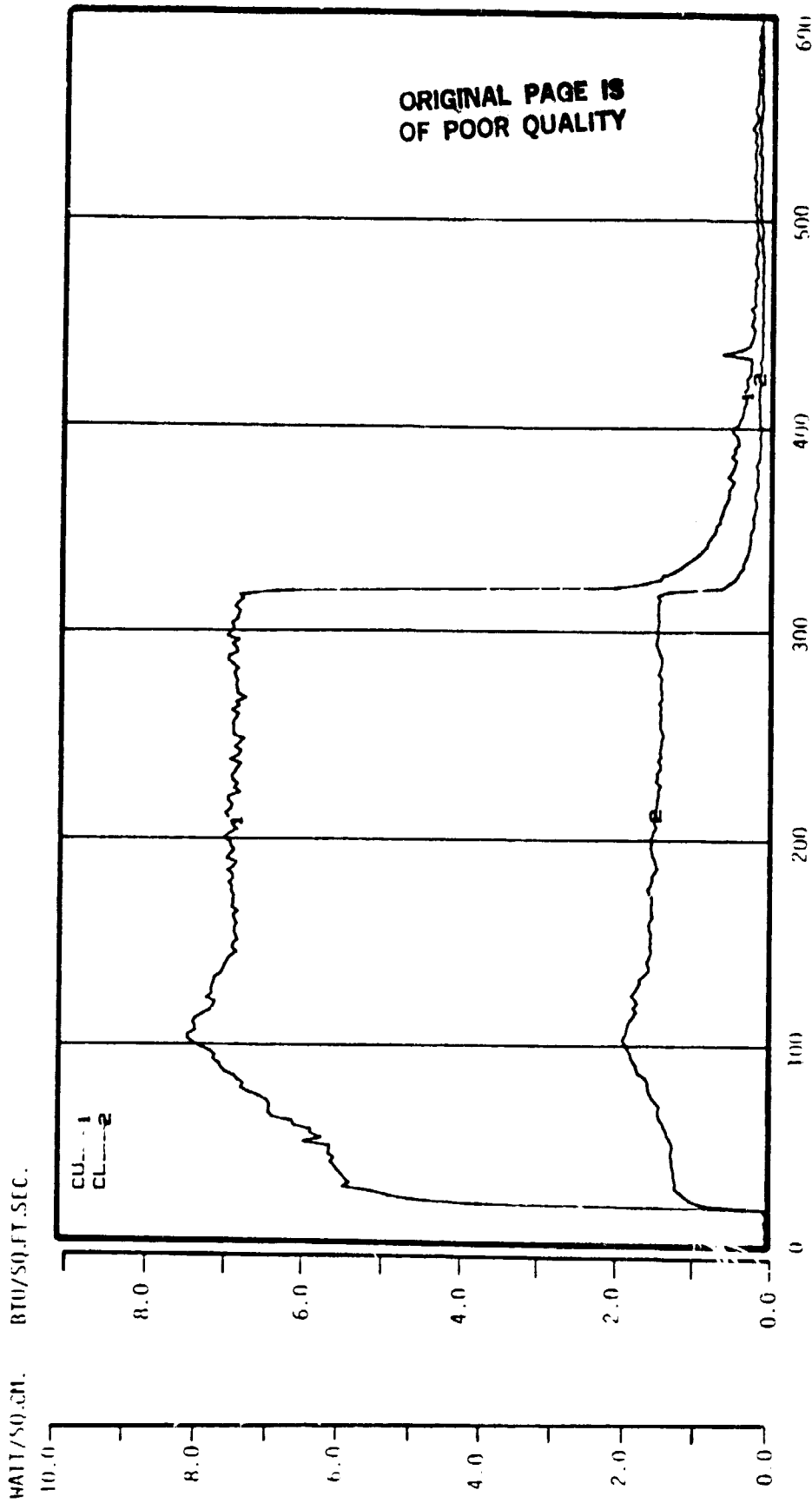
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WEIGHT LOSS



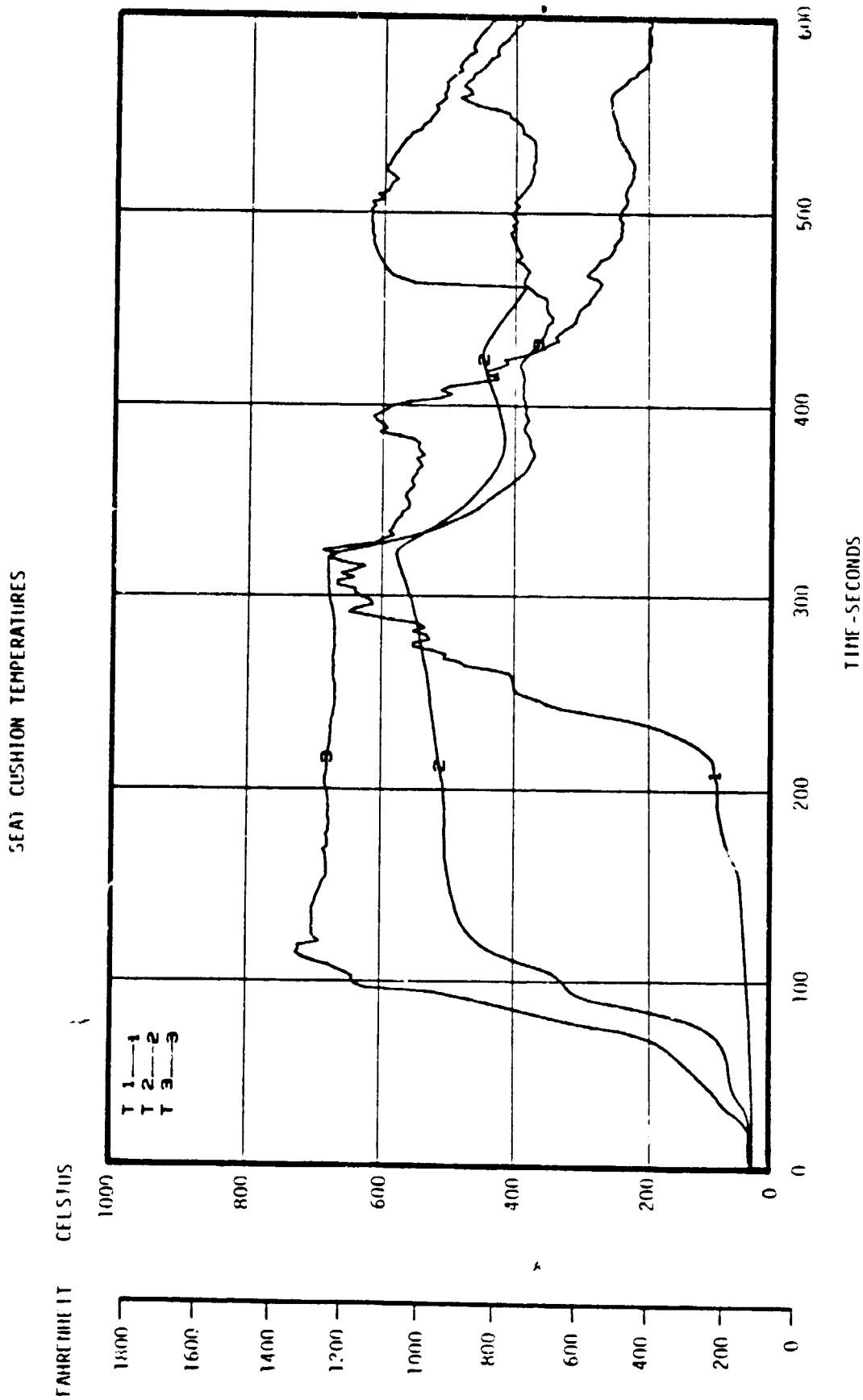
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 CUSHION CONSTRUCTION NUMBER 12.0

HEAT FLUX



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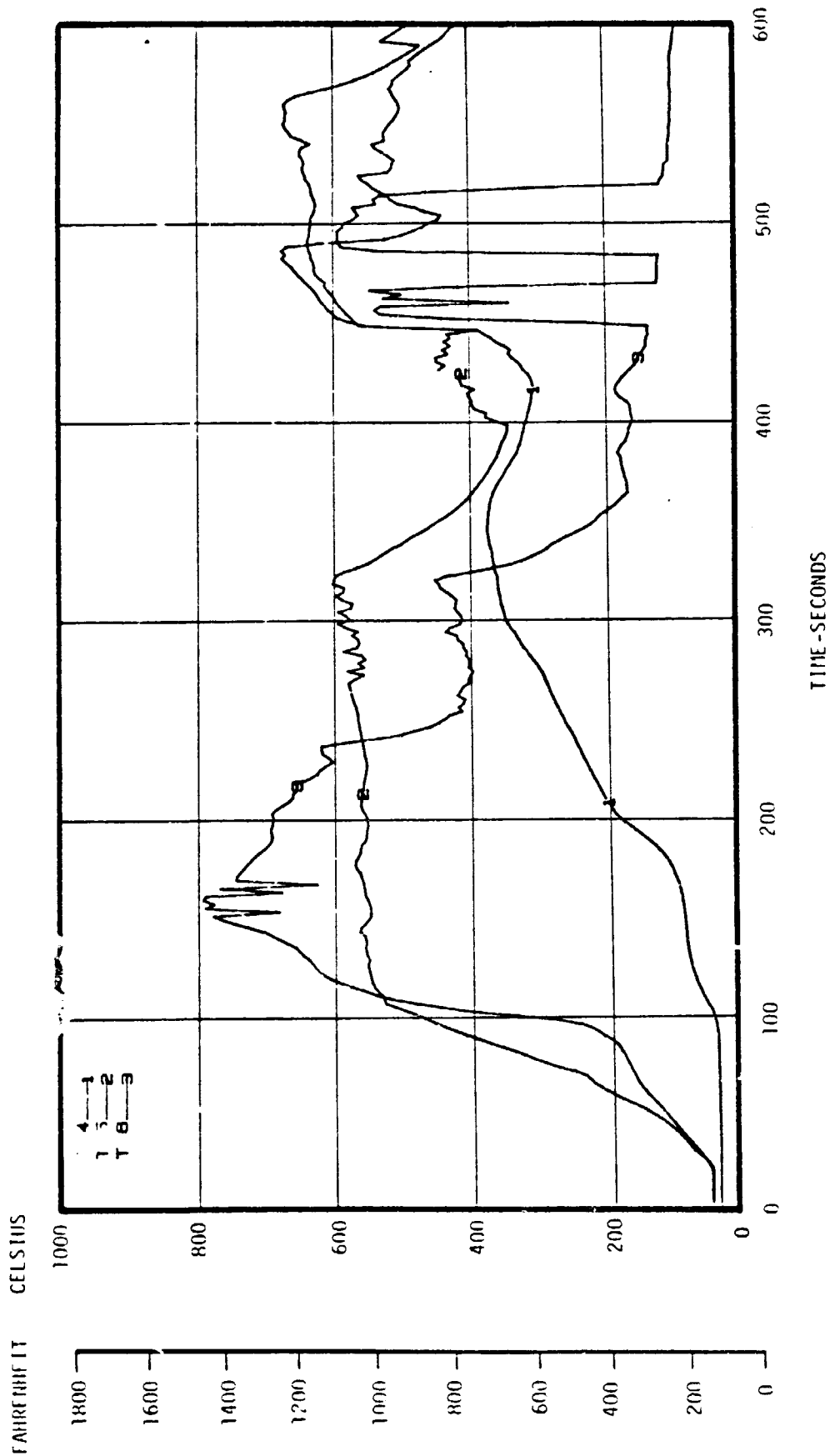
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NASA-NES FULL SCALE CUSHION BURN TEST NUMBER 22
CUSHION CONSTRUCTION NUMBER 13.0



DOUGLAS AIRCRAFT CABIN FIRE SIMULATOR 03/10/82 13.42
NASA-AFES FULL SCALE CUSHION BURN TEST NUMBER 22
CUSHION CONSTRUCTION NUMBER 13.0

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SEAT CUSHION TEMPERATURES



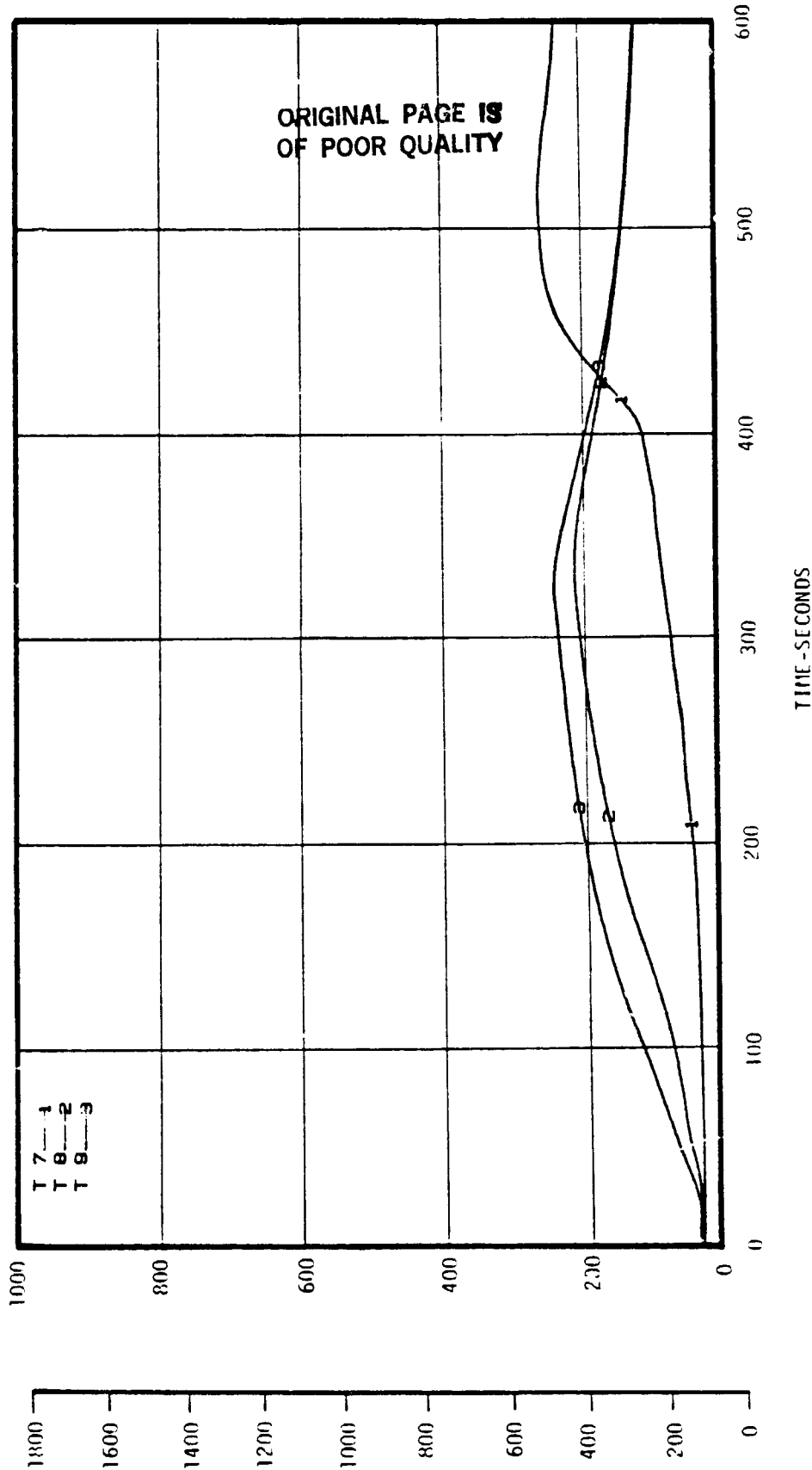
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CUSHION CONSTRUCTION NUMBER 13.0

SEAT CUSHION TEMPERATURES

FAHRENHEIT CELSIUS

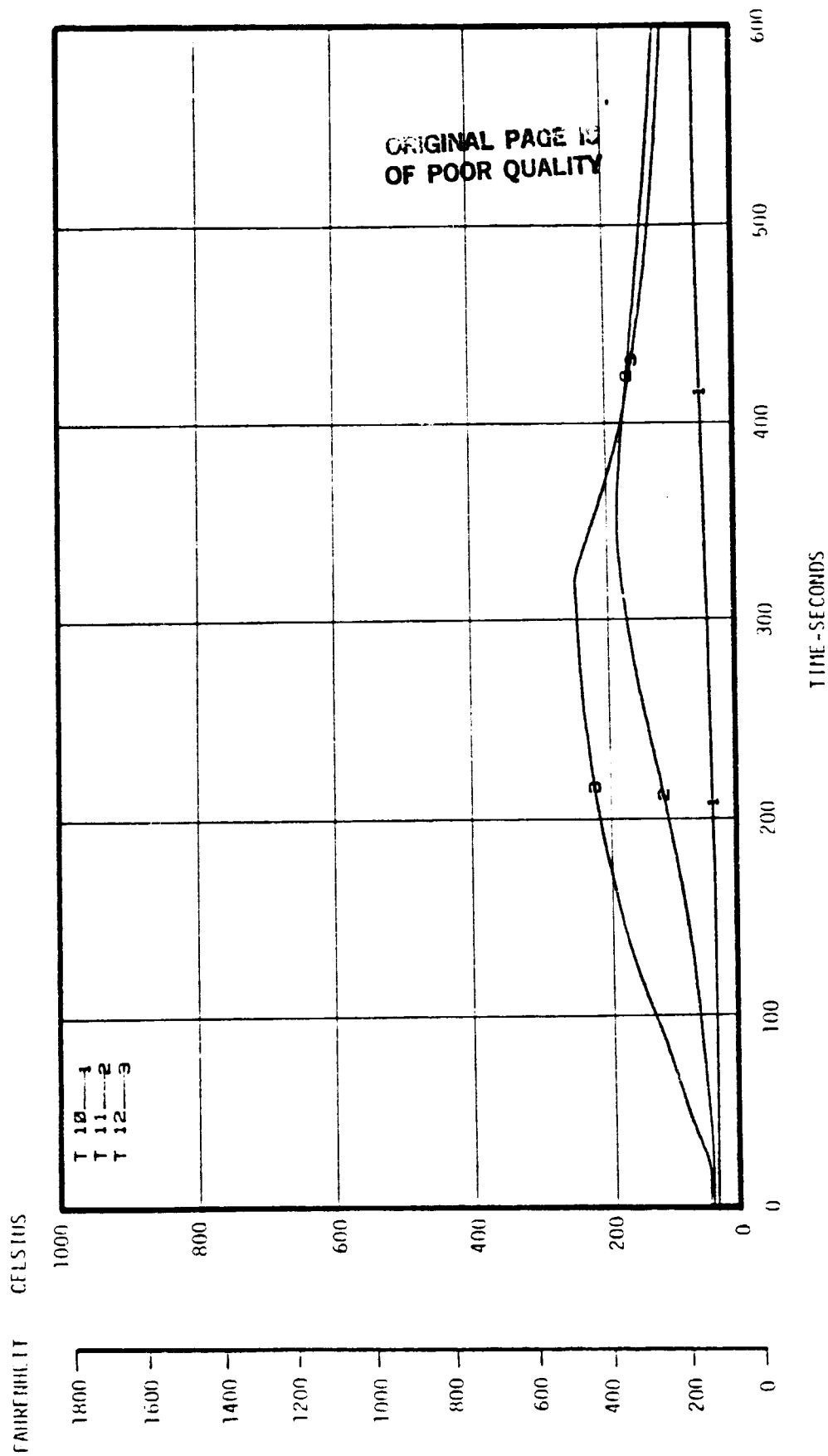


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CUSHION CONSTRUCTION NUMBER 13.0

SEAT CUSHION TEMPERATURES

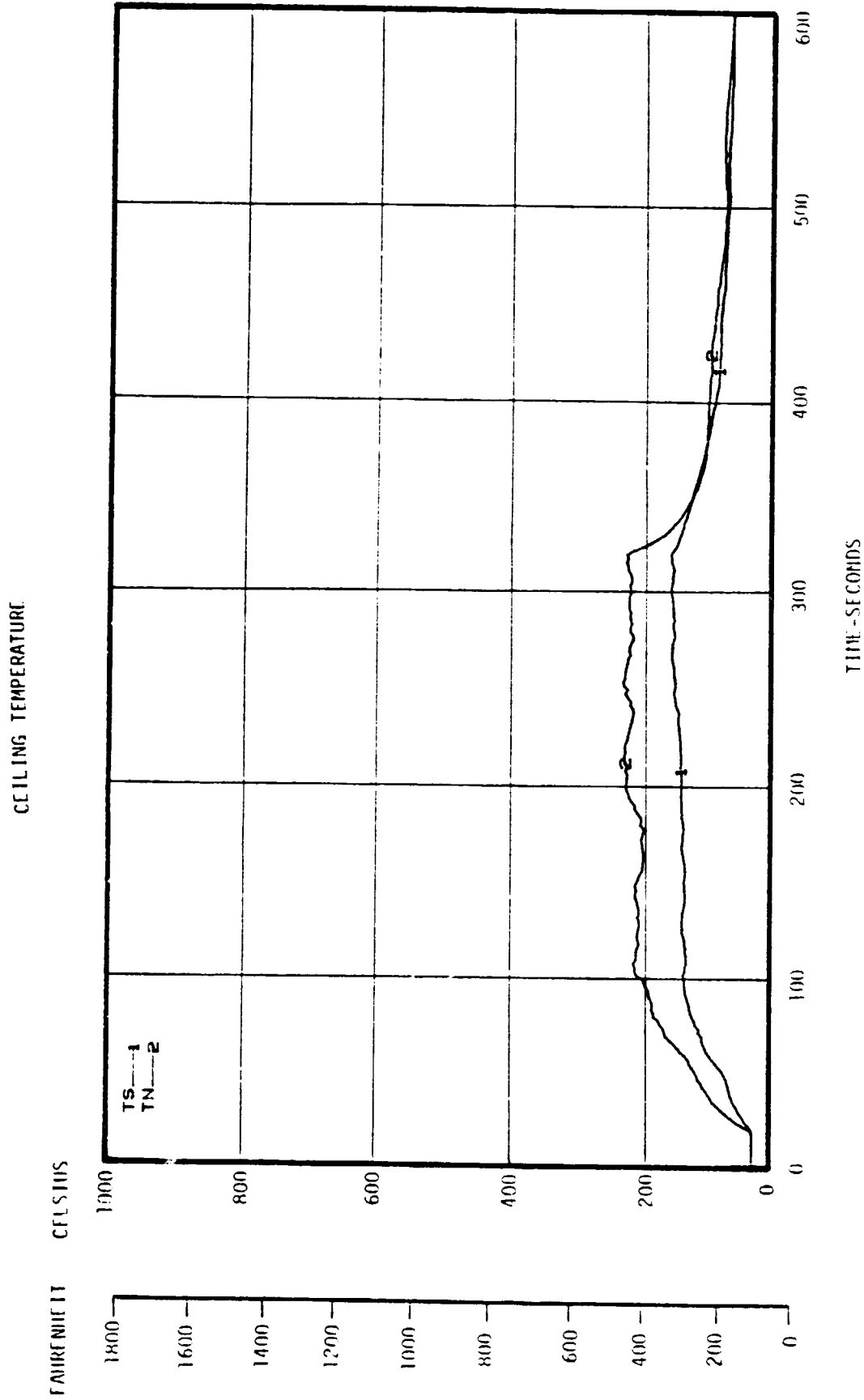


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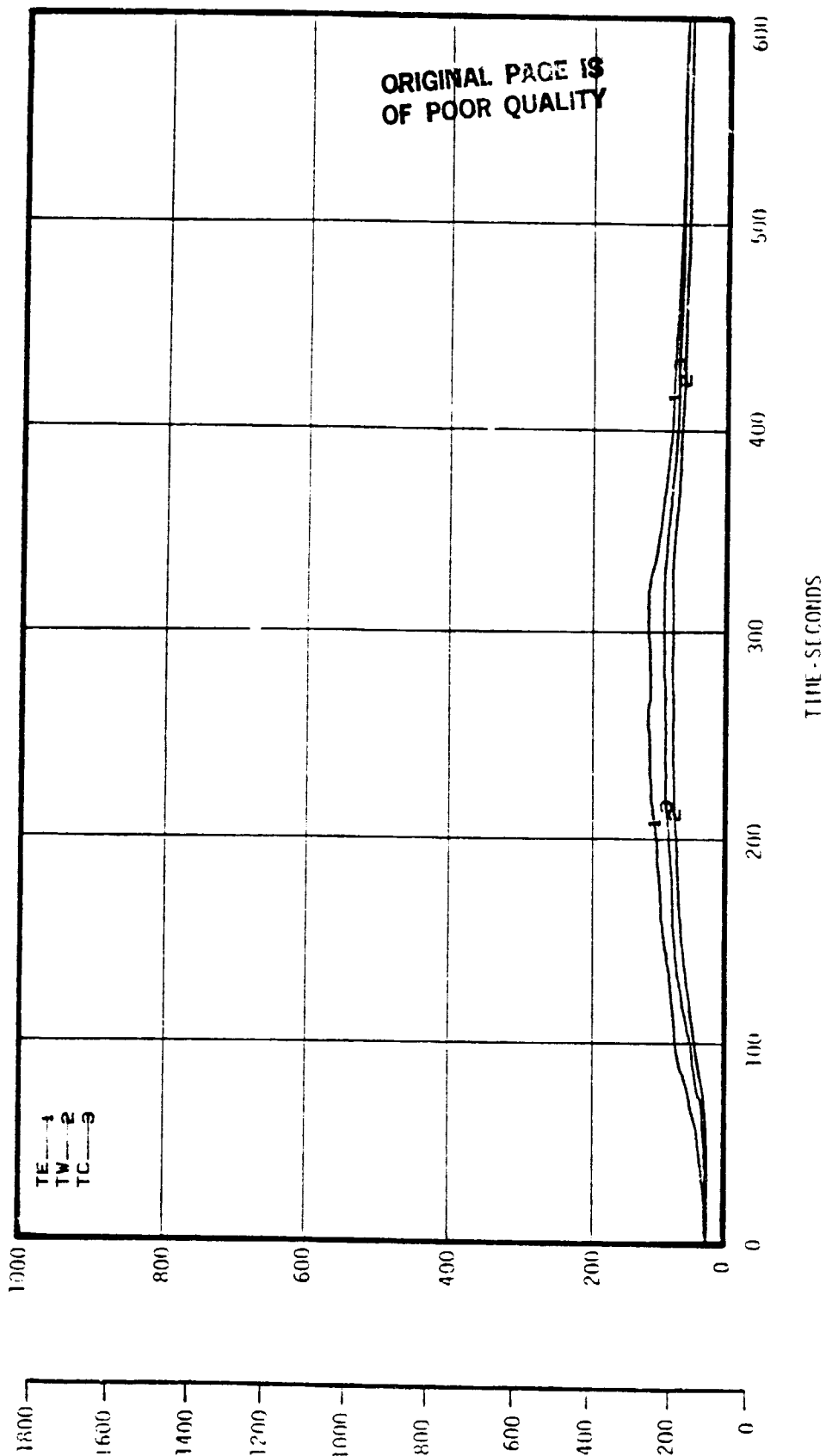
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NASA-AHS FULL SCALE CUSHION BURN TEST NUMBER 22

CUSHION CONSTRUCTION NUMBER 13.0

CEILING TEMPERATURE

FAHRENHEIT CELSIUS

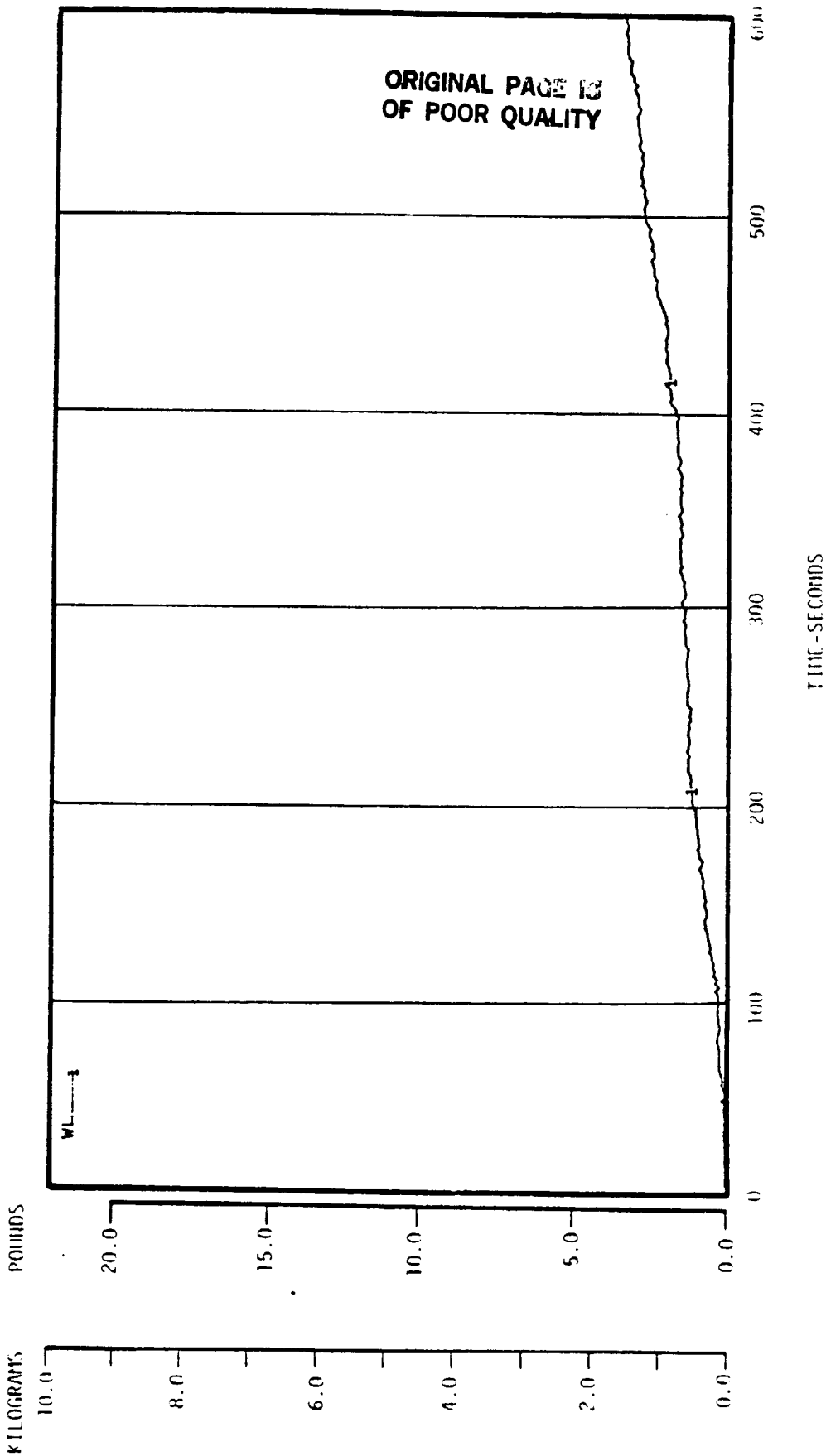


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NASA-MI'S FULL SCALE CUSHION BURN TEST NUMBER 22

CUSHION CONSTRUCTION NUMBER 13.0

WEIGHT LOSS



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NASA-MSC'S FUEL SCALING POSITION BURN TEST NUMBER 22

WEIGHT LOSS CONSTRUCTION NUMBER 13.0